

as hereby amended, have signed an amended marketing agreement; and

2. The issuance of this amendatory Order, further amending the aforesaid Order, is favored or approved by at least two-thirds of the producers who participated in a referendum on the question of approval and who, during the period of August 1, 2017, through July 31, 2018, were engaged within the production area in the production of such almonds. Such producers also produced for market at least two-thirds of the volume of such commodity represented in the referendum.

3. The issuance of this amendatory Order together with a signed marketing agreement advances the interests of growers of almonds in the production area pursuant to the declared policy of the Act.

Order Relative to Handling

It is therefore ordered, that on and after the effective date hereof, all handling of almonds grown in California shall be in conformity to, and in compliance with, the terms and conditions of the said Order as hereby proposed to be amended as follows:

The provisions amending the Order contained in the proposed rule issued by the Administrator on July 2, 2018, and published in the **Federal Register** on July 6, 2018, (83 FR 31473) will be and are the terms and provisions of this order amending the marketing order and are set forth in full herein.

List of Subjects in 7 CFR Part 981

Almonds, Marketing agreements, Nuts, Reporting and recordkeeping requirements.

Dated: September 18, 2019.

Bruce Summers,

Administrator, Agricultural Marketing Service.

For the reasons set out in the preamble, 7 CFR part 981 is amended as follows:

PART 981—ALMONDS GROWN IN CALIFORNIA

■ 1. The authority citation for 7 CFR part 981 continues to read as follows:

Authority: 7 U.S.C. 601–674.

■ 2. Amend § 981.32 by revising paragraph (a)(1) and adding paragraph (a)(3) to read as follows:

§ 981.32 Nominations.

(a) *Method.* (1) Each year the terms of office of three of the members elected pursuant to § 981.31(a) and (b) shall expire, except every third year when the term of office for two of those members shall expire. Nominees for each

respective member and alternate member shall be chosen by ballot delivered to the Board. Nominees chosen by the Board in this manner shall be submitted by the Board to the Secretary on or before June 1 of each year together with such information as the Secretary may require. If a nomination for any Board member or alternate is not received by the Secretary on or before June 1, the Secretary may select such member or alternate from persons belonging to the group to be represented without nomination. The Board shall mail to all handlers and growers, other than the cooperative(s) of record, the required ballots with all necessary voting information including the names of incumbents willing to accept renomination, and, to such growers, the name of any person proposed for nomination in a petition signed by at least 15 such growers and filed with the Board on or before April 1. Distribution of ballots shall be announced by press release, furnishing pertinent information on balloting, issued by the Board through newspapers and other publications having general circulation in the almond producing areas.

* * * * *

(3) The Board may recommend, subject to the approval of the Secretary, a change to the nomination method, should the Board determine that a revision is necessary.

* * * * *

■ 3. Amend § 981.33 by revising the first sentence of paragraphs (a) and (b) and the last sentence of paragraph (c) and adding paragraph (d) to read as follows:

§ 981.33 Selection and term of office.

(a) Members and their respective alternates for positions open on the Board shall be selected by the Secretary from persons nominated pursuant to § 981.32, or, at the discretion of the Secretary, from other qualified persons, for a term of office beginning August 1.
* * *

(b) The term of office of members of the Board shall be for a period of three years beginning on August 1 of the years selected except where otherwise provided. * * *

(c) * * * This limitation on tenure shall not apply to alternate members.

(d) The Board may recommend, subject to approval of the Secretary, revisions to the start date for the term of office of members of the Board.

[FR Doc. 2019–20533 Filed 9–25–19; 8:45 am]

BILLING CODE 3410–02–P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA–2019–0486; Product Identifier 2019–NM–061–AD; Amendment 39–19733; AD 2019–18–06]

RIN 2120–AA64

Airworthiness Directives; Airbus SAS Airplanes

AGENCY: Federal Aviation Administration (FAA), Department of Transportation (DOT).

ACTION: Final rule.

SUMMARY: The FAA is adopting a new airworthiness directive (AD) for certain Airbus SAS Model A318–112, –121, and –122; A319–111, –112, –115, –131, –132, and –133; A320–214, –216, –232, –233, –251N, and –271N; and A321–211, –212, –213, –231, –232, –251N, –253N, –271N, and –272N airplanes. This AD was prompted by reports of missing or loosened fasteners on connecting brackets of overhead stowage compartments (OHSC) and pivoting OHSC (POHSC). This AD requires modification of the OHSC and POHSC attachments, as specified in a European Union Aviation Safety Agency (EASA) AD, which is incorporated by reference. The FAA is issuing this AD to address the unsafe condition on these products.

DATES: This AD is effective October 31, 2019.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in this AD as of October 31, 2019.

ADDRESSES: For the material incorporated by reference (IBR) in this AD, contact the EASA, at Konrad-Adenauer-Ufer 3, 50668 Cologne, Germany; telephone +49 221 89990 1000; email ADs@easa.europa.eu; internet www.easa.europa.eu. You may find this IBR material on the EASA website at <https://ad.easa.europa.eu>. You may view this IBR material at the FAA, Transport Standards Branch, 2200 South 216th St., Des Moines, WA. For information on the availability of this material at the FAA, call 206–231–3195. It is also available in the AD docket on the internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA–2019–0486.

Examining the AD Docket

You may examine the AD docket on the internet at <http://www.regulations.gov> by searching for

and locating Docket No. FAA–2019–0486; or in person at Docket Operations between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this final rule, the regulatory evaluation, any comments received, and other information. The address for Docket Operations is U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue SE, Washington, DC 20590.

FOR FURTHER INFORMATION CONTACT:

Sanjay Ralhan, Aerospace Engineer, International Section, Transport Standards Branch, FAA, 2200 South 216th St., Des Moines, WA 98198; telephone and fax 206–231–3223.

SUPPLEMENTARY INFORMATION:

Discussion

The FAA issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 by adding an AD that would apply to certain Airbus SAS Model A318–112, –121, and –122; A319–111, –112, –115, –131, –132, and –133; A320–214, –216, –232, –233, –251N, and –271N; and A321–211, –212, –213, –231, –232, –251N, –253N, –271N, and –272N airplanes. The NPRM published in the *Federal Register* on June 27, 2019 (84 FR 30637). The NPRM was prompted by reports of missing or loosened fasteners on connecting brackets of OHSC and pivoting POHSC. The NPRM proposed to require modification of the OHSC and POHSC attachments.

The FAA is issuing this AD to address loosening of the OHSC or POHSC fasteners. This condition, if not corrected, could lead to detachment of OHSC or POHSC, possibly resulting in

injury to airplane occupants and/or impeding egress during an emergency evacuation.

The EASA, which is the Technical Agent for the Member States of the European Union, has issued EASA AD 2019–0069, dated March 28, 2019 (“EASA AD 2019–0069”) (also referred to as the Mandatory Continuing Airworthiness Information, or “the MCAI”), to correct an unsafe condition for certain Airbus SAS Model A318–112, –121, and –122; A319–111, –112, –115, –131, –132, and –133; A320–214, –216, –232, –233, –251N, and –271N; and A321–211, –212, –213, –231, –232, –251N, –253N, –271N, and –272N airplanes. The MCAI states:

During routine inspections, several screws were found missing or loose on the interconnecting brackets of certain overhead stowage compartments (OHSC) and pivoting OHSC (POHSC). Investigations and a sampling program have shown that loosening of fasteners can be generated by a relative movement of the OHSC/POHSC and vibrations inside the aeroplane, by elastic deformation of the aeroplane body and by take-off and landing manoeuvres.

This condition, if not corrected, could lead to detachment of an OHSC/POHSC, possibly resulting in injury to aeroplane occupants.

To address this potential unsafe condition, Airbus issued the original issue of the applicable SB [service bulletin], providing modification instructions to improve the robustness of the OHSC and POHSC. Prompted by new findings, the applicable SBs have been later issued, including additional work and associated instructions.

For the reasons described above, this [EASA] AD requires modification of the OHSC and POHSC attachments.

Comments

The FAA gave the public the opportunity to participate in developing

this final rule. The FAA has considered the comment received. Patrick Imperatrice indicated support for the NPRM.

Conclusion

The FAA reviewed the relevant data, considered the comment received, and determined that air safety and the public interest require adopting this final rule as proposed, except for minor editorial changes. The FAA has determined that these minor changes:

- Are consistent with the intent that was proposed in the NPRM for addressing the unsafe condition; and
- Do not add any additional burden upon the public than was already proposed in the NPRM.

Related IBR Material Under 1 CFR Part 51

EASA AD 2019–0069 describes procedures for modification of the OHSC and POHSC attachments. EASA AD 2019–0069 also describes an inspection for discrepancies (additional work) and corrective actions. The inspection includes checks of the dimensions of the threaded pins against tolerances and checks for damage. Corrective actions include replacing threaded pins and nuts and repairing damage. This material is reasonably available because the interested parties have access to it through their normal course of business or by the means identified in the **ADDRESSES** section.

Costs of Compliance

The FAA estimates that this AD affects 1,464 airplanes of U.S. registry. The FAA estimates the following costs to comply with this AD:

ESTIMATED COSTS FOR REQUIRED ACTIONS

Labor cost	Parts cost	Cost per product	Cost on U.S. operators
Up to 42 work-hours × \$85 per hour = Up to \$3,570	\$3,950	Up to \$7,520	Up to \$11,009,280

The FAA estimates the following costs to do any necessary on-condition actions that would be required based on

the results of any required or optional actions. The FAA has no way of

determining the number of aircraft that might need these on-condition actions:

ESTIMATED COSTS OF ON-CONDITION ACTIONS

Labor cost	Parts cost	Cost per product
Up to 19 work-hours × \$85 per hour = Up to \$1,615	[*]	Up to \$1,615*

* The FAA has received no definitive data for the on-condition parts costs.

According to the manufacturer, some or all of the costs of this AD may be covered under warranty, thereby

reducing the cost impact on affected individuals. The FAA does not control warranty coverage for affected

individuals. As a result, the FAA has included all known costs in our cost estimate.

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. Subtitle VII: Aviation Programs, describes in more detail the scope of the Agency's authority.

The FAA is issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section 44701: "General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

This AD is issued in accordance with authority delegated by the Executive Director, Aircraft Certification Service, as authorized by FAA Order 8000.51C. In accordance with that order, issuance of ADs is normally a function of the Compliance and Airworthiness Division, but during this transition period, the Executive Director has delegated the authority to issue ADs applicable to transport category airplanes and associated appliances to the Director of the System Oversight Division.

Regulatory Findings

This AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that this AD:

- (1) Is not a "significant regulatory action" under Executive Order 12866,
- (2) Will not affect intrastate aviation in Alaska, and
- (3) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

Accordingly, under the authority delegated to me by the Administrator,

the FAA amends 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

- 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

- 2. The FAA amends § 39.13 by adding the following new airworthiness directive (AD):

2019–18–06 Airbus SAS: Amendment 39–19733; Docket No. FAA–2019–0486; Product Identifier 2019–NM–061–AD.

(a) Effective Date

This AD is effective October 31, 2019.

(b) Affected ADs

None.

(c) Applicability

This AD applies to Airbus SAS Model A318–112, –121, and –122; A319–111, –112, –115, –131, –132, and –133; A320–214, –216, –232, –233, –251N, and –271N; and A321–211, –212, –213, –231, –232, –251N, –253N, –271N, and –272N airplanes, certificated in any category, as identified in European Union Aviation Safety Agency (EASA) AD 2019–0069, dated March 28, 2019 ("EASA AD 2019–0069").

(d) Subject

Air Transport Association (ATA) of America Code 25, Equipment/furnishings.

(e) Reason

This AD was prompted by reports of missing or loosened fasteners on connecting brackets of overhead stowage compartments (OHSC) and pivoting OHSC (POHSC). The FAA is issuing this AD to address loosening of the OHSC or POHSC fasteners. This condition, if not corrected, could lead to detachment of OHSC or POHSC, possibly resulting in injury to airplane occupants and/or impeding egress during an emergency evacuation.

(f) Compliance

Comply with this AD within the compliance times specified, unless already done.

(g) Requirements

Except as specified in paragraph (h) of this AD: Comply with all required actions and compliance times specified in, and in accordance with, EASA AD 2019–0069.

(h) Exceptions to EASA AD 2019–0069

(1) For purposes of determining compliance with the requirements of this AD: Where EASA AD 2019–0069 refers to its effective date, this AD requires using the effective date of this AD.

(2) For purposes of determining compliance with the requirements of this AD: Paragraph (1) of EASA AD 2019–0069 applies to all airplanes except for airplanes identified by paragraph (2) of EASA AD 2019–0069.

(3) The "Remarks" section of EASA AD 2019–0069 does not apply to this AD.

(i) Other FAA AD Provisions

The following provisions also apply to this AD:

(1) *Alternative Methods of Compliance (AMOCs):* The Manager, International Section, Transport Standards Branch, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the International Section, send it to the attention of the person identified in paragraph (j) of this AD. Information may be emailed to: 9-ANM-116-AMOC-REQUESTS@faa.gov. Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office.

(2) *Contacting the Manufacturer:* For any requirement in this AD to obtain instructions from a manufacturer, the instructions must be accomplished using a method approved by the Manager, International Section, Transport Standards Branch, FAA; or EASA; or Airbus SAS's EASA Design Organization Approval (DOA). If approved by the DOA, the approval must include the DOA-authorized signature.

(3) *Required for Compliance (RC):* For any service information referenced in EASA AD 2019–0069 that contains RC procedures and tests: Except as required by paragraph (i)(2) of this AD, RC procedures and tests must be done to comply with this AD; any procedures or tests that are not identified as RC are recommended. Those procedures and tests that are not identified as RC may be deviated from using accepted methods in accordance with the operator's maintenance or inspection program without obtaining approval of an AMOC, provided the procedures and tests identified as RC can be done and the airplane can be put back in an airworthy condition. Any substitutions or changes to procedures or tests identified as RC require approval of an AMOC.

(j) Related Information

For more information about this AD, contact Sanjay Ralhan, Aerospace Engineer, International Section, Transport Standards Branch, FAA, 2200 South 216th St., Des Moines, WA 98198; telephone and fax 206–231–3223.

(k) Material Incorporated by Reference

(1) The Director of the Federal Register approved the incorporation by reference (IBR) of the service information listed in this paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) You must use this service information as applicable to do the actions required by this AD, unless this AD specifies otherwise.

(i) European Union Aviation Safety Agency (EASA) AD 2019–0069, dated March 28, 2019.

(ii) [Reserved]

(3) For information about EASA AD 2019–0069, contact the EASA, at Konrad-Adenauer-Ufer 3, 50668 Cologne, Germany;

telephone +49 221 89990 6017; email ADs@easa.europa.eu; Internet www.easa.europa.eu. You may find this EASA AD on the EASA website at <https://ad.easa.europa.eu>.

(4) You may view this material at the FAA, Transport Standards Branch, 2200 South 216th St., Des Moines, WA. For information on the availability of this material at the FAA, call 206–231–3195. This material may be found in the AD docket on the internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA–2019–0486.

(5) You may view this material that is incorporated by reference at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, email fedreg.legal@nara.gov, or go to: <http://www.archives.gov/federal-register/cfr/ibr-locations.html>.

Issued in Des Moines, Washington, on September 6, 2019.

Michael Kaszycki,

Acting Director, System Oversight Division, Aircraft Certification Service.

[FR Doc. 2019–20898 Filed 9–25–19; 8:45 am]

BILLING CODE 4910–13–P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA–2019–0771; Product Identifier 2019–NE–27–AD; Amendment 39–19747; AD 2019–19–11]

RIN 2120–AA64

Airworthiness Directives; Pratt & Whitney Turbofan Engines

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final rule; request for comments.

SUMMARY: The FAA is adopting a new airworthiness directive (AD) for certain Pratt & Whitney (PW) PW1519G, PW1521G, PW1521GA, PW1524G, PW1525G, PW1521G–3, PW1524G–3, PW1525G–3, PW1919G, PW1921G, PW1922G, PW1923G, and PW1923G–A model turbofan engines. This AD requires initial and repetitive inspections of the low-pressure compressor (LPC) inlet guide vane (IGV) and the LPC rotor 1 (R1) and, depending on the results of the inspections, possible replacement of the LPC. This AD was prompted by two recent in-flight shutdowns (IFSDs) that occurred as the result of failures of the LPC R1. The FAA is issuing this AD to address the unsafe condition on these products. **DATES:** This AD is effective September 26, 2019.

The FAA must receive comments on this AD by November 12, 2019.

ADDRESSES: You may send comments, using the procedures found in 14 CFR 11.43 and 11.45, by any of the following methods:

- **Federal eRulemaking Portal:** Go to <http://www.regulations.gov>. Follow the instructions for submitting comments.

- **Fax:** 202–493–2251.

- **Mail:** U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue SE, Washington, DC 20590.

- **Hand Delivery:** U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue SE, Washington, DC 20590, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

For service information identified in this final rule, contact Pratt & Whitney, 400 Main Street, East Hartford, CT 06118; phone: 800–565–0140; fax: 860–565–5442; email: help24@pw.utc.com; internet: <http://fleetcare.pw.utc.com>. You may view this service information at the FAA, Engine and Propeller Standards Branch, 1200 District Avenue, Burlington, MA 01803. For information on the availability of this material at the FAA, call 781–238–7759. It is also available on the internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA–2019–0771.

Examining the AD Docket

You may examine the AD docket on the internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA–2019–0771; or in person at Docket Operations between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this final rule, the regulatory evaluation, any comments received, and other information. The street address for the Docket Operations is listed above. Comments will be available in the AD docket shortly after receipt.

FOR FURTHER INFORMATION CONTACT:

Kevin M. Clark, Aerospace Engineer, ECO Branch, FAA, 1200 District Avenue, Burlington, MA 01803; phone: 781–238–7088; fax: 781–238–7199; email: kevin.m.clark@faa.gov.

SUPPLEMENTARY INFORMATION:

Discussion

The FAA received reports of two recent IFSDs on PW PW1524G–3 model turbofan engines. The first IFSD occurred on July 25, 2019 and the second IFSD occurred on September 16, 2019. These IFSDs were due to failure of the LPC R1, which resulted in the

LPC R1 releasing from the LPC case and damaging the engine. LPC rotor failures historically have released high-energy debris that has resulted in damage to engines and airplanes (see Advisory Circular (AC) 39–8, “Continued Airworthiness Assessments of Powerplant and Auxiliary Power Unit Installations of Transport Category Airplanes,” dated September 8, 2003, available at rsl.faa.gov).

Although these IFSDs occurred on PW PW1524G–3 model turbofan engines, the FAA is including PW PW1900 engines in the applicability of the AD because similarities in type design make these engines susceptible to the same unsafe condition as PW PW1500 engines. This condition, if not addressed, could result in uncontained release of the LPC R1, in-flight shutdown, damage to the engine, damage to the airplane, and loss of control of the airplane. The FAA is issuing this AD to address the unsafe condition on these products.

Related Service Information

The FAA reviewed Pratt & Whitney Service Bulletin (SB) PW1000G–A–72–00–0125–00A–930A–D, Issue No. 001, dated September 23, 2019, and PW SB PW1000G–A–72–00–0075–00B–930A–D, Issue No. 001, dated September 23, 2019. The SBs contain procedures for performing borescope inspections of the LPC R1 and the LPC IGV actuation system on engines that have less than 300 flight cycles since new.

FAA’s Determination

The FAA is issuing this AD because it evaluated all the relevant information and determined the unsafe condition described previously is likely to exist or develop in other products of the same type design.

AD Requirements

This AD requires initial and repetitive borescope inspections of the LPC IGV and the LPC R1 and, depending on the results of the inspections, replacement of the LPC.

Interim Action

The FAA considers this AD interim action. The investigation into the two recent failures on the PW PW1524G–3 model turbofan engines is on-going and the FAA may pursue further rulemaking action at a later date.

Justification for Immediate Adoption and Determination of the Effective Date

Section 553(b)(3)(B) of the Administrative Procedure Act (APA) (5 U.S.C.) authorizes agencies to dispense with notice and comment procedures