

same document with the credit contract. Creditors are not required to give the consumer two separate copies of the document before consummation, one for the consumer to keep and a second copy for the consumer to execute. The disclosure requirement is satisfied if the creditor gives a copy of the document containing the unexecuted credit contract and disclosures to the consumer to read and sign; and the consumer receives a copy to keep at the time the consumer becomes obligated. It is not sufficient for the creditor merely to show the consumer the document containing the disclosures before the consumer signs and becomes obligated. The consumer must be free to take possession of and review the document in its entirety before signing.

i. *Example.* To illustrate:

A. A creditor gives a consumer a multiple-copy form containing a credit agreement and TILA disclosures. The consumer reviews and signs the form and returns it to the creditor, who separates the copies and gives one copy to the consumer to keep. The creditor has satisfied the disclosure requirement.

* * * * *

Subpart E—Special Rules for Certain Home Mortgage Transactions

* * * * *

§ 226.32—Requirements for Certain Closed-End Home Mortgages

* * * * *

Paragraph 32(c)(3) Regular payment; balloon payment.

1. *General.* The regular payment is the amount due from the borrower at regular intervals, such as monthly, bimonthly, quarterly, or annually. There must be at least two payments, and the payments must be in an amount and at such intervals that they fully amortize the amount owed. In disclosing the regular payment, creditors may rely on the rules set forth in § 226.18(g); however, the amounts for voluntary items, such as credit life insurance, may be included in the regular payment disclosure only if the consumer has previously agreed to the amounts.

i. If the loan has more than one payment level, the regular payment for each level must be disclosed. For example:

A. In a 30-year graduated payment mortgage where there will be payments of \$300 for the first 120 months, \$400 for the next 120 months, and \$500 for the last 120 months, each payment amount must be disclosed, along with the length of time that the payment will be in effect.

B. If interest and principal are paid at different times, the regular amount for each must be disclosed.

C. In discounted or premium variable-rate transactions where the creditor sets the initial interest rate and later rate adjustments are determined by an index or formula, the creditor must disclose both the initial payment based on the discount or premium and the payment that will be in effect thereafter. Additional explanatory material which does not detract from the required disclosures may accompany the disclosed amounts. For example, if a monthly payment

is \$250 for the first six months and then increases based on an index and margin, the creditor could use language such as the following: "Your regular monthly payment will be \$250 for six months. After six months your regular monthly payment will be based on an index and margin, which currently would make your payment \$350. Your actual payment at that time may be higher or lower."

* * * * *

By order of the Board of Governors of the Federal Reserve System, acting through the Director of the Division of Consumer and Community Affairs and the Secretary of the Board under delegated authority, April 2, 2002.

Jennifer J. Johnson,

Secretary of the Board.

[FR Doc. 02-8373 Filed 4-8-02; 8:45 am]

BILLING CODE 6210-01-P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 99-NM-86-AD; Amendment 39-12699; AD 2002-07-05]

RIN 2120-AA64

Airworthiness Directives; Airbus Model A300 B2, A300 B4, A300 B4-600, and A300 B4-600R Series Airplanes; and Model A300 F4-605R Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule.

SUMMARY: This amendment adopts a new airworthiness directive (AD), applicable to all Airbus Model A300 B2, A300 B4, A300 B4-600, and A300 B4-600R series airplanes, and Model A300 F4-605R airplanes. This AD requires repetitive inspections for cracking of certain fittings, corrective action if necessary, and, for certain airplanes, a modification. This AD also provides an optional terminating action for the repetitive inspections. The actions specified by this AD are intended to detect and correct propagation of cracks on the frame 40 aft fittings due to local stress concentrations at the upper flange runout of frame 40, which could result in reduced structural integrity of the airplane. This action is intended to address the identified unsafe condition.

DATES: Effective May 14, 2002.

The incorporation by reference of certain publications listed in the regulations is approved by the Director of the Federal Register as of May 14, 2002.

ADDRESSES: The service information referenced in this AD may be obtained

from Airbus Industrie, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France. This information may be examined at the Federal Aviation Administration (FAA), Transport Airplane Directorate, Rules Docket, 1601 Lind Avenue, SW., Renton, Washington; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

FOR FURTHER INFORMATION CONTACT: Dan Rodina, Aerospace Engineer, International Branch, ANM-116, FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 227-2125; fax (425) 227-1149.

SUPPLEMENTARY INFORMATION: A proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) to include an airworthiness directive (AD) that is applicable to all Airbus Model A300 B2, A300 B4, A300 B4-600, and A300 B4-600R series airplanes; and Model A300 F4-605R airplanes; was published as a supplemental notice of proposed rulemaking (NPRM) in the **Federal Register** on January 4, 2002 (67 FR 530). That action proposed to require repetitive inspections for cracking of certain fittings, corrective action if necessary, and, for certain airplanes, a modification; and would have provided for optional terminating action for the repetitive inspections.

Comments

Interested persons have been afforded an opportunity to participate in the making of this amendment. Due consideration has been given to the comments received from a single commenter in response to the second supplemental NPRM.

Provide Credit for Prior Inspection and Refer to Terminating Action

One commenter asks the FAA to revise the proposed rule to provide credit for an inspection already performed in accordance with the original issue of Airbus Service Bulletin A300-53-6048, dated January 16, 1996, provided that the inspection is accomplished in conjunction with Airbus Service Bulletin A300-57-6053. The commenter states that this would make the proposed rule consistent with the original issue of the corresponding French airworthiness directive, 98-481-270(B), dated December 2, 1998.

The same commenter also requests that we revise the proposed rule to provide for optional terminating action on Model A300 B4-600 and A300 B4-600R series airplanes and Model A300 F4-605R airplanes. The commenter states that inspection and rework per

Airbus Service Bulletins A300-57-6052 and A300-57-6053 constitutes terminating action for airplanes on which no cracks are found and no subsequent rework is required.

The FAA concurs with the intent of the commenter's request, but we have already accommodated the request previously. We added Note 2 to the first supplemental NPRM to provide credit for an inspection done in accordance with the original issue of Airbus Service Bulletin A300-53-6048. In addition, we revised paragraph (b)(8) and paragraph (e) in the first supplemental NPRM to clarify that modification per Airbus Service Bulletin A300-57-6053, Revision 1, dated October 31, 1995, or Revision 02, dated June 2, 1999, terminates the proposed requirements, regardless of the inspection results. No change to the final rule is necessary.

Conclusion

After careful review of the available data, including the comments noted above, the FAA has determined that air safety and the public interest require the adoption of the rule as proposed in the second supplemental NPRM.

Cost Impact

The FAA estimates that 70 Model A300 B2, A300 B4, A300 B4-600, and A300 B4-600R series airplanes; and Model A300 F4-605R airplanes; of U.S. registry will be affected by this AD.

For affected airplanes, it will take approximately 92 work hours per airplane to accomplish the required modification, at an average labor rate of \$60 per work hour. Required parts will cost as much as \$874 per airplane. Based on these figures, the cost impact of the required modification is estimated to be as much as \$6,394 per airplane.

It will take approximately 10 work hours per airplane to accomplish the required inspection, at an average labor rate of \$60 per work hour. Based on these figures, the cost impact of the required inspection on U.S. operators is estimated to be \$42,000, or \$600 per airplane, per inspection cycle.

The cost impact figures discussed above are based on assumptions that no operator has yet accomplished any of the requirements of this AD action, and that no operator would accomplish those actions in the future if this AD were not adopted. The cost impact figures discussed in AD rulemaking actions represent only the time necessary to perform the specific actions actually required by the AD. These figures typically do not include incidental costs, such as the time required to gain access and close up, planning time, or time necessitated by other administrative actions.

Regulatory Impact

The regulations adopted herein will not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, it is determined that this final rule does not have federalism implications under Executive Order 13132.

For the reasons discussed above, I certify that this action (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A final evaluation has been prepared for this action and it is contained in the Rules Docket. A copy of it may be obtained from the Rules Docket at the location provided under the caption **ADDRESSES**.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

Accordingly, pursuant to the authority delegated to me by the

Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:

2002-07-05 Airbus Industrie: Amendment 39-12699. Docket 99-NM-86-AD.

Applicability: All Model A300 B2, A300 B4, A300 B4-600, and A300 B4-600R series airplanes; and Model A300 F4-605R airplanes; certificated in any category.

Note 1: This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (g) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Required as indicated, unless accomplished previously.

To detect and correct propagation of cracks on the frame 40 aft fittings due to local stress concentrations at the upper flange runout of frame 40, which could result in reduced structural integrity of the airplane, accomplish the following:

Modification

(a) For airplanes on which Airbus Modification 10430 has not been done before the effective date of this AD: Concurrently with the inspection required by paragraph (b) of this AD, modify the profile of frame 40 aft fittings per the service information specified in Table 1, as follows:

TABLE 1.—SERVICE INFORMATION

For Model—	Do the actions in accordance with either—	Of Airbus Service Bulletin—	Dated—
(1) A300 B2 and A300 B4 series airplanes	(i) Revision 01 or	A300-53-0296	September 30, 1998.
	(ii) Revision 02	A300-53-0296	May 12, 1999.
(2) A300 B4-600 and A300 B4-600R series airplanes and Model A300 F4-605R airplanes.	(i) Revision 01 or	A300-53-6048	September 30, 1998.
	(ii) Revision 03	A300-53-6048	February 21, 2000.

Note 2: For Model A300 B4-600 and A300 B4-600R series airplanes and Model A300

F4-605R airplanes: Actions performed in accordance with Airbus Service Bulletin

A300-53-6048, dated January 16, 1996; or Revision 02, dated May 12, 1999; are

acceptable for compliance with the applicable requirements of this AD.

Note 3: Airbus Service Bulletin A300–53–6048 refers to Airbus Service Bulletin A300–

53–6063 as an additional source of service information for accomplishment of certain repairs.

Inspection

(b) For all airplanes, inspect the airplane per Table 2, as follows:

TABLE 2.—INSPECTION REQUIREMENTS

Requirements	Description
(1) Area to inspect	The frame 40 AFT fitting.
(2) Type of inspection	Nondestructive test (NDT).
(3) Compliance time	As specified by paragraph (c) of this AD.
(4) Discrepancies to detect	Cracking.
(5) Service information	Inspect in accordance with the applicable service bulletin listed in Table 1 of this AD.
(6) Follow-on actions if you find no cracking	Repeat the inspection thereafter at the applicable interval specified by Table 3 of this AD.
(7) Corrective actions if you find cracking	Do the actions specified by paragraph (d) of this AD.
(8) Terminating action	The modification specified by paragraph (e) of this AD terminates the requirements of this AD.

Note 4: An NDT per Non-destructive Testing Manual 53–15–30, Part 6, Procedure C, is also acceptable for compliance with the requirements of paragraph (b) of this AD.

(c) Perform the inspection required by paragraph (b) of this AD per the schedule in Table 3 of this AD. For airplanes on which this inspection has been accomplished before the effective date of this AD, the initial

compliance time may be extended by the repetitive interval following the date the inspection was accomplished. Table 3 follows:

TABLE 3.—COMPLIANCE TIMES FOR INSPECTION

For Model—	If the total flight cycles accumulated on the airplane as of the effective date of this AD is—	Then inspect—	And repeat the inspection at least every—
(1) A300 B4–600 and A300 B4–600R series airplanes and Model A300 F4–605R airplanes, pre-Modification 10430.	(i) Fewer than 6,200	Before the airplane accumulates 7,700 total flight cycles or 17,710 total flight hours, whichever occurs first.	7,500 flight cycles or 17,250 flight hours, whichever occurs first.
	(ii) At least 6,200 and fewer than 9,700.	Within 1,500 flight cycles or 3,450 flight hours after the effective date of this AD, whichever occurs first.	7,500 flight cycles or 17,250 flight hours, whichever occurs first.
	(iii) At least 9,700	Within 750 flight cycles or 1,725 flight hours after the effective date of this AD, whichever occurs first.	7,500 flight cycles or 17,250 flight hours, whichever occurs first.
(2) A300 B4–600 and A300 B4–600R series airplanes and Model A300 B–4605R airplanes, post-Modification 10430.	(i) Fewer than 19,600	Before the airplane accumulates 21,100 total flight cycles or 48,530 total flight hours, whichever occurs first.	7,500 flight cycles or 17,250 flight hours, whichever occurs first.
	(ii) At least 19,600 and fewer than 23,100.	Within 1,500 flight cycles or 3,450 flight hours after the effective date of this AD, whichever occurs first.	7,500 flight cycles or 17,250 flight hours, whichever occurs first.
	(iii) At least 23,100	Within 750 flight cycles or 1,725 flight hours after the effective date of this AD, whichever occurs first.	7,500 flight cycles or 17,250 flight hours, whichever occurs first.
(3) A300 B2 series airplanes	(i) Fewer than 12,00	Before the airplane accumulates 14,000 total flight cycles or 15,120 total flight hours, whichever occurs first.	5,500 flight cycles or 5,940 flight hours, whichever occurs first.
	(ii) At least 12,000 and fewer than 17,000.	Within 2,000 flight cycles or 2,160 flight hours after the effective date of this AD, whichever occurs first.	5,500 flight cycles or 5,940 flight hours, whichever occurs first.
	(iii) At least 17,000	Within 1,000 flight cycles or 1,080 flight hours after the effective date of this AD, whichever occurs first.	5,500 flight cycles or 5,940 flight hours, whichever occurs first.
(4) A300 B4–100 series airplanes.	(i) Fewer than 9,500	Before the airplane accumulates 11,500 total flight cycles or 15,295 total flight hours, whichever occurs first.	4,500 flight cycles or 5,985 flight hours, whichever occurs first.
	(ii) At least 9,500 and fewer than 14,500.	Within 2,000 flight cycles or 2,660 flight hours after the effective date of this AD, whichever occurs first.	4,500 flight cycles or 5,985 flight hours, whichever occurs first.
	(iii) At least 14,500	Within 1,000 flight cycles or 1,330 flight hours after the effective date of this AD, whichever occurs first.	4,500 flight cycles or 5,985 flight hours, whichever occurs first.

TABLE 3.—COMPLIANCE TIMES FOR INSPECTION—Continued

For Model—	If the total flight cycles accumulated on the airplane as of the effective date of this AD is—	Then inspect—	And repeat the inspection at least every—
(5) A300 B4–200 series airplanes.	(i) Fewer than 8,500	Before the airplane accumulates 10,500 total flight cycles or 21,840 total flight hours, whichever occurs first.	4,000 flight cycles or 8,320 flight hours, whichever occurs first.
	(ii) At least 8,500 and fewer than 13,500.	Within 2,000 flight cycles or 4,160 flight hours after the effective date of this AD, whichever occurs first.	4,000 flight cycles or 8,320 flight hours, whichever occurs first.
	(iii) At least 13,500	Within 1,000 flight cycles or 2,080 flight hours after the effective date of this AD, whichever occurs first.	4,000 flight cycles or 8,320 flight hours, whichever occurs first.

Note 5: An NDT inspection is also required by AD 98–25–07, amendment 39–10933, to be repetitively performed on Model A300 B4–600 and A300 B4–600R series airplanes and Model A300 F4–605R airplanes on which Airbus Modification 10453 has not been installed. For those airplanes, if the inspection is done within the applicable compliance time specified by paragraph (c) of this AD, the threshold for the initial inspection of paragraph (b) of this AD may be extended by 1,500 flight cycles.

Corrective Actions

(d) If any cracking is found during any inspection required by paragraph (b) of this AD: Except as required by paragraph (f) of this AD, prior to further flight, perform all applicable corrective actions in accordance with the applicable service bulletin identified in Table 1 of this AD.

Terminating Action

(e) Accomplishment of the applicable modification in accordance with the applicable service bulletin specified by

paragraph (e)(1) or (e)(2) of this AD terminates the requirements of this AD.

(1) For Model A300 B4–600 and A300 B4–600R series airplanes: In accordance with Airbus Service Bulletin A300–57–6053, Revision 1, dated October 31, 1995; or Revision 02, dated June 2, 1999.

(2) For Model A300 B2 and A300 B4 series airplanes: In accordance with Airbus Service Bulletin A300–53–0297, Revision 2, dated October 31, 1995.

Exception to Service Bulletin Instructions

(f) During any inspection required by this AD, if the service bulletin specifies to contact the manufacturer for an appropriate action: Prior to further flight, repair in accordance with a method approved by the Manager, International Branch, ANM–116, Transport Airplane Directorate, FAA; or the Direction Générale de l'Aviation Civile (DGAC) (or its delegated agent).

Alternative Methods of Compliance

(g) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager,

International Branch, ANM–116. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, International Branch, ANM–116.

Note 6: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the International Branch, ANM–116.

Special Flight Permits

(h) Special flight permits may be issued in accordance with §§ 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

Incorporation by Reference

(i) Except as required by paragraph (f) of this AD, the required actions shall be done in accordance with the applicable service documents identified in Table 4 and Table 5 of this AD, as follows:

TABLE 4.—REFERENCED SERVICE DOCUMENTS FOR REQUIRED ACTIONS

Service bulletin and date	Page numbers	Revision level shown on the page	Date shown on page
Airbus Service Bulletin A300–53–0296, Revision 01, September 30, 1998.	1–38	01	September 30, 1998.
Airbus Service Bulletin A300–53–0296, Revision 02, May 12, 1999.	1–3, 8, 15, 16, 18, 20, 22, 23, 24 ...	02	May 12, 1999.
	4–7, 9–14, 17, 19, 21, 25, 26, 27–38.	01	September 30, 1998.
Airbus Service Bulletin A300–53–6048, Revision 01, September 30, 1998.	1–31	01	September 30, 1998.
Airbus Service Bulletin A300–53–6048, Revision 03, February 21, 2000.	1–32	03	February 21, 2000.

TABLE 5.—REFERENCED SERVICE DOCUMENTS FOR OPTIONAL TERMINATING ACTION

Service bulletin and date	Page numbers	Revision level shown on the page	Date shown on page
Airbus Service Bulletin A300–57–6053, Revision 1, October 31, 1995.	1, 7–9, 11–15, 19–24, 35, 36, 41, 42, 45–47.	1	October 31, 1995.
	2–6, 10, 16–18, 25–34, 37–40, 43, 44 ..	Original	February 21, 1995.

TABLE 5.—REFERENCED SERVICE DOCUMENTS FOR OPTIONAL TERMINATING ACTION—Continued

Service bulletin and date	Page numbers	Revision level shown on the page	Date shown on page
Airbus Service Bulletin A300–57–6053, Revision 02, June 2, 1999.	1–6, 8, 23, 23a, 46, 47	02	June 2, 1999.
	7, 9, 11, 12, 13–15, 19–22, 35, 36, 41, 42, 45.	1	October 31, 1995.
	10, 16–18, 25, 26, 27–34, 37–40, 43, 44.	Original	February 21, 1995.
Airbus Service Bulletin A300–53–0297, Revision 2, October 31, 1995.	1–60	2	October 31, 1995.

This incorporation by reference was approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Copies may be obtained from Airbus Industrie, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France. Copies may be inspected at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

Note 7: The subject of this AD is addressed in French airworthiness directive 1998–481–270(B) R1, dated July 12, 2000.

Effective Date

(j) This amendment becomes effective on May 14, 2002.

Issued in Renton, Washington, on March 28, 2002.

Kalene C. Yanamura,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 02–8278 Filed 4–8–02; 8:45 am]

BILLING CODE 4910–13–P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2000–NM–324–AD; Amendment 39–12700; AD 2002–07–06]

RIN 2120–AA64

Airworthiness Directives; McDonnell Douglas Model DC–9–10, –20, –30, –40, and –50 Series Airplanes; and C–9 Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule.

SUMMARY: This amendment adopts a new airworthiness directive (AD), applicable to certain McDonnell Douglas Model DC–9–10, –20, –30, –40, and –50 series airplanes; and C–9 airplanes; that requires repetitive visual and x-ray inspections to detect cracks of the upper and lower corners and upper center of the door cutout of the aft

pressure bulkhead; corrective actions, if necessary; and follow-on actions. For certain airplanes, the amendment also requires modification of the ventral aft pressure bulkhead. The actions specified by this AD are intended to detect and correct fatigue cracks in the corners and upper center of the door cutout of the aft pressure bulkhead, which could result in rapid decompression of the fuselage and consequent reduced structural integrity of the airplane. This action is intended to address the identified unsafe condition.

DATES: Effective May 14, 2002.

The incorporation by reference of certain publications listed in the regulations is approved by the Director of the Federal Register as of May 14, 2002.

ADDRESSES: The service information referenced in this AD may be obtained from Boeing Commercial Aircraft Group, Long Beach Division, 3855 Lakewood Boulevard, Long Beach, California 90846, Attention: Data and Service Management, Dept. C1–L5A (D800–0024). This information may be examined at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the FAA, Los Angeles Aircraft Certification Office, 3960 Paramount Boulevard, Lakewood, California.

FOR FURTHER INFORMATION CONTACT: Wahib Mina, Aerospace Engineer, Airframe Branch, ANM–120L, FAA, Los Angeles Aircraft Certification Office, 3960 Paramount Boulevard, Lakewood, California 90712; telephone (562) 627–5324; fax (562) 627–5210.

SUPPLEMENTARY INFORMATION: A proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) to include an airworthiness directive (AD) that is applicable to certain McDonnell Douglas Model DC–9–10, –20, –30, –40, and –50 series airplanes; and C–9 airplanes; was published in the **Federal Register** on September 20, 2001 (66 FR 48384). That action proposed to require

repetitive general visual and x-ray inspections to detect cracks of the upper and lower corners and upper center of the door cutout of the aft pressure bulkhead; corrective actions, if necessary; and follow-on actions. For certain airplanes, the amendment also requires modification of the ventral aft pressure bulkhead.

Comments

Interested persons have been afforded an opportunity to participate in the making of this amendment. Due consideration has been given to the comments received.

Requests To Revise Certain Inspection Requirements

Three commenters request revision of the inspection requirements in paragraph (b) of the proposed rule. The rationales for these requests are as follows:

- One commenter suggests revising paragraph (b) of the proposed rule to specify the same inspections cited in McDonnell Douglas Service Bulletin DC9–53–137, Revision 07, dated February 6, 2001, which was cited as the appropriate source of service information for this AD. The commenter states that paragraph (b) of the proposed rule is misleading because it incorrectly implies that a repair will always be required or that a preventive modification is required. In addition, that paragraph does not allow for continuing visual and x-ray inspections as specified in the previously referenced service bulletin.

- One commenter requests clarification of the inspection procedures specified in the proposed rule. Paragraph (b) of the proposed rule specifies visual and eddy current inspections within 8,000 landings after accomplishment of the visual and x-ray inspections required by paragraph (a) of this AD. However, Service Bulletin DC9–53–137, Revision 07, specifies visual and eddy current inspections after a repair or preventive modification is installed. The proposed rule would