

(1) Is not a “significant regulatory action” under Executive Order 12866,  
(2) Will not affect intrastate aviation in Alaska, and

(3) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

#### List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

#### The Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

### PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

#### § 39.13 [Amended]

■ 2. The FAA amends § 39.13 by adding the following new airworthiness directive:

**2025–25–08 Airbus Helicopters:**  
Amendment 39–23216; Docket No. FAA–2025–2273; Project Identifier MCAI–2024–00689–R.

#### (a) Effective Date

This airworthiness directive (AD) is effective February 6, 2026.

#### (b) Affected ADs

None.

#### (c) Applicability

This AD applies to Airbus Helicopters Model H160–B helicopters, certificated in any category.

#### (d) Subject

Air Transport Association (ATA) of America Code 05, Time Limits/Maintenance Checks.

#### (e) Unsafe Condition

This AD was prompted by new or more restrictive airworthiness limitations. The FAA is issuing this AD to prevent failure of certain parts which, if not addressed, could result in loss of control of the helicopter.

#### (f) Compliance

Comply with this AD within the compliance times specified, unless already done.

#### (g) Required Actions

Except as specified in paragraphs (h) and (i) of this AD: Comply with all required actions and compliance times specified in, and in accordance with, European Union Aviation Safety Agency AD 2024–0223, dated November 26, 2024 (EASA AD 2024–0223).

#### (h) Exceptions to EASA AD 2024–0223

(1) Where EASA AD 2024–0223 refers to its effective date, this AD requires using the effective date of this AD.

(2) This AD does not adopt the requirements specified in paragraphs (1), (2), (4), and (5) of EASA AD 2024–0223.

(3) Where paragraph (3) of EASA AD 2024–0223 specifies “Within 12 months after the effective date of this AD, revise the approved AMP”, this AD requires replacing that text with “Within 30 days after the effective date of this AD, revise the airworthiness limitations section of the existing maintenance manual or instructions for continued airworthiness and the existing approved maintenance or inspection program, as applicable”.

(4) The initial compliance time for doing the tasks specified in paragraph (3) of EASA AD 2024–0223 is on or before the applicable “limitations” and “associated thresholds” as incorporated by the requirements of paragraph (3) of EASA AD 2024–0223 or within 30 days after the effective date of this AD, whichever occurs later.

(5) This AD does not adopt the “Remarks” section of EASA AD 2024–0223.

#### (i) Provisions for Alternative Actions and Intervals

No alternative actions and associated thresholds and intervals, including life limits, are allowed for compliance with paragraph (g) of this AD unless they are approved as specified in the provisions of the “Ref. Publications” section of EASA AD 2024–0223.

#### (j) Alternative Methods of Compliance (AMOCs)

(1) The Manager, International Validation Branch, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the manager of the International Validation Branch, send it to the attention of the person identified in paragraph (k) of this AD and email to: [AMOC@faa.gov](mailto:AMOC@faa.gov).

(2) Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office.

#### (k) Additional Information

For more information about this AD, contact Yves Petiotte, Aviation Safety Engineer, FAA, 1600 Stewart Avenue, Suite 410, Westbury, NY 11590; phone: (202) 975–4867; email: [yves.petiotte@faa.gov](mailto:yves.petiotte@faa.gov).

#### (l) Material Incorporated by Reference

(1) The Director of the Federal Register approved the incorporation by reference of the material listed in this paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) You must use this material as applicable to do the actions required by this AD, unless the AD specifies otherwise.

(i) European Union Aviation Safety Agency (EASA) AD 2024–0223, dated November 26, 2024.

(ii) [Reserved]

(3) For EASA material identified in this AD, contact EASA, Konrad-Adenauer-Ufer 3, 50668 Cologne, Germany; phone: +49 221 8999 000; email: [ADs@easa.europa.eu](mailto:ADs@easa.europa.eu); website: [easa.europa.eu](http://easa.europa.eu). You may find the EASA material on the EASA website at [ad.easa.europa.eu](http://ad.easa.europa.eu).

(4) You may view this material at the FAA, Office of the Regional Counsel, Southwest Region, 10101 Hillwood Parkway, Room 6N–321, Fort Worth, TX 76177. For information on the availability of this material at the FAA, call (817) 222–5110.

(5) You may view this material at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, visit [www.archives.gov/federal-register/cfr/ibr-locations](http://www.archives.gov/federal-register/cfr/ibr-locations) or email [fr-inspection@nara.gov](mailto:fr-inspection@nara.gov).

Issued on December 8, 2025.

**Steven W. Thompson,**

*Acting Deputy Director, Compliance & Airworthiness Division, Aircraft Certification Service.*

[FR Doc. 2025–24179 Filed 12–31–25; 8:45 am]

**BILLING CODE 4910–13–P**

## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 39

[Docket No. FAA–2025–2274; Project Identifier MCAI–2023–01244–R; Amendment 39–23219; AD 2025–25–11]

**RIN 2120–AA64**

### Airworthiness Directives; Airbus Helicopters Deutschland GmbH Helicopters

**AGENCY:** Federal Aviation Administration (FAA), DOT.

**ACTION:** Final rule.

**SUMMARY:** The FAA is adopting a new airworthiness directive (AD) for all Airbus Helicopters Deutschland GmbH Model MBB–BK 117 C–2, MBB–BK 117 D–2, and MBB–BK 117 D–3 helicopters. This AD was prompted by reports of damaged hoist hooks and hoist hook nuts. This AD requires performing an inspection of the affected hoist hook and affected hook nut and, depending on the results of the inspection, replacing the affected hoist hook and affected hook nut. This AD also prohibits installing an affected assembly (hoist hook attached to the hook damper) on any helicopter, and prohibits installing an affected hoist hook on any helicopter unless it is installed using updated procedures. The FAA is issuing this AD to address the unsafe condition on these products.

**DATES:** This AD is effective February 6, 2026.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in this AD as of February 6, 2026.

ADDRESSES:

**AD Docket:** You may examine the AD docket at *regulations.gov* under Docket No. FAA–2025–2274; or in person at Docket Operations between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this final rule, the mandatory continuing airworthiness information (MCAI), any comments received, and other information. The address for Docket Operations is U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue SE, Washington, DC 20590.

Material Incorporated by Reference:

- For European Union Aviation Safety Agency (EASA) material identified in this AD, contact EASA, Konrad-Adenauer-Ufer 3, 50668 Cologne, Germany; phone: +49 221 8999 000; email: *ADs@easa.europa.eu*; website: *easa.europa.eu*. You may find the EASA material on the EASA website at *ad.easa.europa.eu*.

- You may view this material at the FAA, Office of the Regional Counsel, Southwest Region, 10101 Hillwood Parkway, Room 6N–321, Fort Worth, TX 76177. For information on the availability of this material at the FAA, call (817) 222–5110. It is also available at *regulations.gov* under Docket No. FAA–2025–2274.

**FOR FURTHER INFORMATION CONTACT:** Ramon Walker Perez, Aviation Safety Engineer, FAA, 1600 Stewart Avenue, Suite 410, Westbury, NY 11590; phone: (847) 294–7337; email: *ramon.a.walker.perez@faa.gov*.

SUPPLEMENTARY INFORMATION:

Background

The FAA issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 by adding an AD that would apply to all Airbus Helicopters Deutschland GmbH Model MBB–BK 117

C–2, MBB–BK 117 D–2, and MBB–BK 117 D–3 helicopters. The NPRM was published in the **Federal Register** on September 5, 2025 (90 FR 42860). The NPRM was prompted by EASA AD 2023–0213, dated December 8, 2023 (EASA AD 2023–0213) (also referred to as the MCAI), issued by EASA, which is the Technical Agent for the Member States of the European Union. The MCAI states that occurrences of damage to the threaded part of the affected hoist hook and to the threads of the affected hook nut have been reported. Investigation identified galling as the root cause of the damage.

In the NPRM, the FAA proposed to require performing an inspection of the affected hoist hook and affected hook nut and, depending on the results of the inspection, replacing the affected hoist hook and affected hook nut, and installing a serviceable hoist hook in accordance with updated installation procedures. In the NPRM, the FAA also proposed to prohibit installing an affected assembly (hoist hook attached to the hook damper) on any helicopter, and to prohibit installing an affected hoist hook on any helicopter unless it is installed using updated procedures.

The FAA is issuing this AD to detect and correct damage to the hoist hook and hook nut, which if not corrected, could lead to failure of the affected assembly (hoist hook attached to the hook damper), which could result in loss of the hoist load and injury to persons.

You may examine the MCAI in the AD docket at *regulations.gov* under Docket No. FAA–2025–2274.

Discussion of Final Airworthiness Directive

Comments

The FAA received no comments on the NPRM or on the determination of the costs.

Conclusion

These products have been approved by the civil aviation authority of another

country and are approved for operation in the United States. Pursuant to the FAA’s bilateral agreement with this State of Design Authority, that authority has notified the FAA of the unsafe condition described in the MCAI referenced above. The FAA reviewed the relevant data, considered any comments received, and determined that air safety requires adopting this AD as proposed. Accordingly, the FAA is issuing this AD to address the unsafe condition on these products. Except for minor editorial changes, this AD is adopted as proposed in the NPRM. None of the changes will increase the economic burden on any operator.

Material Incorporated by Reference Under 1 CFR Part 51

The FAA reviewed EASA AD 2023–0213, which specifies procedures for inspecting the affected hoist hook and the affected hook and replacing both the hoist hook and hoist nut if any debris, dent, or crack is found in accordance with updated installation procedures. EASA AD 2023–0213 also prohibits installing an affected assembly (hoist hook attached to the hook damper) or any higher assembly having an affected assembly on a helicopter, and prohibits installing a hoist hook unless it is installed using the updated installation procedures. This material is reasonably available because the interested parties have access to it through their normal course of business or by the means identified in the **ADDRESSES** section.

Differences Between This AD and the MCAI

The MCAI applies to Airbus Helicopters Deutschland GmbH Model MBB–BK117 D–3m helicopters, whereas this AD does not because that model does not have an FAA type certificate.

Costs of Compliance

The FAA estimates that this AD affects 176 helicopters of U.S. registry. The FAA estimates the following costs to comply with this AD:

ESTIMATED COSTS				
Action	Labor cost	Parts cost	Cost per product	Cost on U.S. operators
Inspect hoist hook and hoist nut .....	2 work-hours × \$85 per hour = \$170 .....	\$0	\$170	\$29,920

The FAA estimates the following costs to do any replacements that would

be required based on the results of the inspection. The agency has no way of

determining the number of helicopters that might need this replacement.

## ON-CONDITION COSTS

Action	Labor cost	Parts cost	Cost per product
Install hoist hook .....	1 work-hour × \$85 per hour = \$85 .....	\$8,562	\$8,647

**Authority for This Rulemaking**

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. Subtitle VII: Aviation Programs, describes in more detail the scope of the Agency's authority.

The FAA is issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section 44701: General requirements. Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

**Regulatory Findings**

This AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that this AD:

- (1) Is not a "significant regulatory action" under Executive Order 12866,
- (2) Will not affect intrastate aviation in Alaska, and
- (3) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

**List of Subjects in 14 CFR Part 39**

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

**The Amendment**

Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

**PART 39—AIRWORTHINESS DIRECTIVES**

- 1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

**§ 39.13 [Amended]**

- 2. The FAA amends § 39.13 by adding the following new airworthiness directive:

**2025–25–11 Airbus Helicopters**

**Deutschland GmbH:** Amendment 39–23219; Docket No. FAA–2025–2274; Project Identifier MCAI–2023–01244–R.

**(a) Effective Date**

This airworthiness directive (AD) is effective February 6, 2026.

**(b) Affected ADs**

None.

**(c) Applicability**

This AD applies to Airbus Helicopters Deutschland GmbH Model MBB–BK 117 C–2, MBB–BK 117 D–2, and MBB–BK 117 D–3 helicopters, certificated in any category.

**(d) Subject**

Joint Aircraft System Component (JASC) Code 2500, Cabin Equipment/Furnishings.

**(e) Unsafe Condition**

This AD was prompted by reports of damage to the threaded part of the affected hoist hook and the threads of the affected hook nut due to galling. The FAA is issuing this AD to detect and correct damage to the hoist hook and hook nut, which if not corrected, could lead to failure of the affected assembly (hoist hook attached to the hook damper), which could result in loss of the hoist load and injury to persons.

**(f) Compliance**

Comply with this AD within the compliance times specified, unless already done.

**(g) Required Actions**

Except as specified in paragraphs (h) and (i) of this AD: Comply with all required actions and compliance times specified in, and in accordance with, European Union Aviation Safety Agency (EASA) AD 2023–0213, dated December 8, 2023 (EASA AD 2023–0213).

**(h) Exceptions to EASA AD 2023–0213**

(1) Where EASA AD 2023–0213 refers to its effective date, this AD requires using the effective date of this AD.

(2) This AD does not adopt the "Remarks" section of EASA AD 2023–0213.

**(i) No Reporting or Returning Parts Requirement**

Although the material referenced in EASA AD 2023–0213 specifies to submit information and return parts to the manufacturer, this AD does not require those actions.

**(j) Alternative Methods of Compliance (AMOCs)**

(1) The Manager, International Validation Branch, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the manager of the International Validation Branch, send it to the attention of the person identified in paragraph (k) of this AD and email to: [AMOC@faa.gov](mailto:AMOC@faa.gov).

(2) Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office.

**(k) Additional Information**

For more information about this AD, contact Ramon A. Walker Perez, Aviation Safety Engineer, FAA, 1600 Stewart Avenue, Suite 410, Westbury, NY 11590; phone: (847) 294–7337; email: [ramon.a.walker.perez@faa.gov](mailto:ramon.a.walker.perez@faa.gov).

**(l) Material Incorporated by Reference**

(1) The Director of the Federal Register approved the incorporation by reference of the material listed in this paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) You must use this material as applicable to do the actions required by this AD, unless the AD specifies otherwise.

(i) European Union Aviation Safety Agency (EASA) AD 2023–0213, dated December 8, 2023.

(ii) [Reserved]

(3) For EASA material identified in this AD, contact EASA, Konrad-Adenauer-Ufer 3, 50668 Cologne, Germany; phone: +49 221 8999 000; email: [ADs@easa.europa.eu](mailto:ADs@easa.europa.eu); website: [easa.europa.eu](http://easa.europa.eu). You may find the EASA material on the EASA website at [ad.easa.europa.eu](http://ad.easa.europa.eu).

(4) You may view this material at the FAA, Office of the Regional Counsel, Southwest Region, 10101 Hillwood Parkway, Room 6N–321, Fort Worth, TX 76177. For information on the availability of this material at the FAA, call (817) 222–5110.

(5) You may view this material at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, visit [www.archives.gov/federal-register/cfr/ibr-locations](http://www.archives.gov/federal-register/cfr/ibr-locations) or email [fr.inspection@nara.gov](mailto:fr.inspection@nara.gov).

Issued on December 10, 2025.

**Steven W. Thompson,**

*Acting Deputy Director, Compliance & Airworthiness Division, Aircraft Certification Service.*

[FR Doc. 2025–24181 Filed 12–31–25; 8:45 am]

BILLING CODE 4910–13–P

## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 39

[Docket No. FAA–2025–0612; Project Identifier MCAI–2023–00935–R; Amendment 39–23214; AD 2025–25–06]

RIN 2120–AA64

#### Airworthiness Directives; Airbus Helicopters

**AGENCY:** Federal Aviation Administration (FAA), DOT.

**ACTION:** Final rule.

**SUMMARY:** The FAA is adopting a new airworthiness directive (AD) for all Airbus Helicopters Model SA341G and SA342J helicopters. This AD was prompted by reports of disbonding of the stainless steel leading edge protection of certain part-numbered main rotor blades (MRB). This AD requires repetitively tap inspecting the MRB and, depending on the results, repairing or replacing the MRB. This AD also prohibits installing those MRB unless certain requirements are met. The FAA is issuing this AD to address the unsafe condition on these products.

**DATES:** This AD is effective February 6, 2026.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in this AD as of February 6, 2026.

#### ADDRESSES:

**AD Docket:** You may examine the AD docket at *regulations.gov* under Docket No. FAA–2025–0612; or in person at Docket Operations between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this final rule, the mandatory continuing airworthiness information (MCAI), any comments received, and other information. The address for Docket Operations is U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue SE, Washington, DC 20590.

#### Material Incorporated by Reference:

- For European Union Aviation Safety Agency (EASA) material identified in this AD, contact EASA, Konrad-Adenauer-Ufer 3, 50668

Cologne, Germany; phone: +49 221 8999 000; email: *ADs@easa.europa.eu*; website: *easa.europa.eu*. You may find this material on the EASA website at *ad.easa.europa.eu*.

- You may view this material at the FAA, Office of the Regional Counsel, Southwest Region, 10101 Hillwood Parkway, Room 6N–321, Fort Worth, TX 76177. For information on the availability of this material at the FAA, call (817) 222–5110. It is also available at *regulations.gov* under Docket No. FAA–2025–0612.

**FOR FURTHER INFORMATION CONTACT:** Zain Jamal, Aviation Safety Engineer, FAA, 1600 Stewart Avenue, Suite 410, Westbury, NY 11590; phone: (847) 294–7264; email: *zain.jamal@faa.gov*.

#### SUPPLEMENTARY INFORMATION:

##### Background

The FAA issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 by adding an AD that would apply to all Airbus Helicopters Model SA341G and SA342J helicopters. The NPRM was published in the **Federal Register** on April 7, 2025 (90 FR 14922). The NPRM was prompted by AD 2023–0155, dated July 31, 2023 (EASA AD 2023–0155) (also referred to as the MCAI), issued by EASA, which is the Technical Agent for the Member States of the European Union. The MCAI states that reports were received of the stainless steel leading edge protection disbonding on certain part-numbered MRBs. This condition, if not detected and addressed, could result in significant unbalance of the main rotor, a high level of vibration, failure of the main rotor and main gearbox, and consequent loss of control of the helicopter.

In the NPRM, the FAA proposed to require repetitively tap inspecting the MRB and, depending on the results, repairing or replacing the MRB. In the NPRM, the FAA also proposed to prohibit installing those MRB unless certain requirements are met. The FAA is issuing this AD to address the unsafe condition on these products.

You may examine the MCAI in the AD docket at *regulations.gov* under Docket No. FAA–2025–0612.

#### Discussion of Final Airworthiness Directive

##### Comments

The FAA received comments from an anonymous commenter who requested the FAA take into consideration the community, operator, and environmental impact when issuing an AD. The commenter made several other comments not specific to this AD, such

as systemic and social-justice recommendations, which were out of scope of this AD. The following presents the comments received on the NPRM and the FAA's response to the comments.

#### Request for Relief on Inspection Frequency and Cost Burden

The commenter requested the FAA consider using a tiered or risk-based inspection interval for low-utilization or community-serving operators. The commenter also requested that the FAA provide guidance for financial assistance to those disproportionately affected by ADs. The commenter stated that MRB inspections and MRB repair or replacement impose substantial labor and financial costs, especially on small operators and public sector agencies. The commenter requested the FAA consider cost offsets or grants for public-benefit entities.

The FAA disagrees with the commenter. The NPRM requires a tap inspection, which takes an hour to complete. The FAA is issuing this AD to address an unsafe condition on the Airbus Helicopters Model SA341G and SA342J helicopters. Changing the inspection interval for low-utilization operators or public sector agencies does not address this unsafe condition because the FAA would be allowing higher risk to the flying public by allowing low-utilization operators or public sector agencies to have different inspection intervals that could allow the unsafe condition to remain on the helicopter for an unacceptable amount of time. The NPRM originally included the cost to replace these blades at \$168,449 per blade. However, FAA revised the estimated cost of this final rule to reflect the repair cost, since it was determined that most operators have elected to repair their blades instead of replacing them with new blades. The repair cost is significantly less burdensome than the replacement cost and would not impose substantial labor and financial costs on small operators and public sector agencies.

Additionally, the FAA is unable to offer guidance on financial assistance for those impacted by this AD because that is outside the scope of an FAA AD response. No changes were made to this AD as a result of this comment.

#### Request for Worker Protection and Training

The commenter requested that all inspections and repairs be done by a unionized or properly certified mechanic. The commenter stated the FAA should require transparency in