

**§ 39.13 [Amended]**

■ 2. The FAA amends § 39.13 by adding the following new airworthiness directive:

**The Boeing Company:** Docket No. FAA–2025–3997; Project Identifier AD–2025–01471–T.

**(a) Comments Due Date**

The FAA must receive comments on this airworthiness directive (AD) by January 2, 2026.

**(b) Affected ADs**

None.

**(c) Applicability**

This AD applies to all The Boeing Company Model 777–200, –200LR, –300, –300ER, and 777F series airplanes, certificated in any category.

**(d) Subject**

Air Transport Association (ATA) of America Code 29, Hydraulic Power.

**(e) Unsafe Condition**

This AD was prompted by a report of an overheated alternating current motor pump (ACMP) that caused a fire in the main landing gear (MLG) wheel well. A subsequent investigation found that the cause of the fire was an overheated ACMP. Further investigation found that a lock washer and nut was missing from a moving contact in the related electrical load control unit (ELCU). The FAA is issuing this AD to address a single failure of the ELCU in the ACMP. The unsafe condition, if not addressed, could result in the ACMP overheating and igniting a fire in the wheel well and consequent damage to the airplane, loss of continued safe flight and landing and/or personnel injury.

**(f) Compliance**

Comply with this AD within the compliance times specified, unless already done.

**(g) Required Actions**

Except as specified by paragraph (h) of this AD: At the applicable times specified in the “Compliance” paragraph of Boeing Alert Requirements Bulletin 777–29A0047 RB, dated September 11, 2025, do all applicable actions identified in, and in accordance with, the Accomplishment Instructions of Boeing Alert Requirements Bulletin 777–29A0047 RB, dated September 11, 2025.

**Note 1 to paragraph (g):** Guidance for accomplishing the actions required by this AD can be found in Boeing Alert Service Bulletin 777–29A0047, dated September 11, 2025, which is referred to in Boeing Alert Requirements Bulletin 777–29A0047 RB, dated September 11, 2025.

**(h) Exceptions to Requirements Bulletin Specifications**

(1) Where the Compliance Time column of the table in the “Compliance” paragraph of Boeing Alert Requirements Bulletin 777–29A0047 RB, dated September 11, 2025, refers to the original issue date of Requirements Bulletin 777–29A0047 RB, this AD requires using the effective date of this AD.

(2) Where Boeing Alert Requirements Bulletin 777–29A0047 RB, dated September 11, 2025, refers to part number 3033115–100 as a replacement part, for this AD, part numbers 66133–06 or 66068–08 are also acceptable replacement parts.

**(i) Parts Installation Prohibition**

As of the effective date of this AD, no person may install any ACMP, having part number 731966, on any airplane.

**(j) Alternative Methods of Compliance (AMOCs)**

(1) The Manager, AIR–520, Continued Operational Safety Branch, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or responsible Flight Standards Office, as appropriate. If sending information directly to the manager of the Continued Operational Safety Branch, send it to the attention of the person identified in paragraph (k)(1) of this AD. Information may be emailed to: *AMOC@faa.gov*. Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the responsible Flight Standards Office.

(2) An AMOC that provides an acceptable level of safety may be used for any repair, modification, or alteration required by this AD if it is approved by The Boeing Company Organization Designation Authorization (ODA) that has been authorized by the Manager, AIR–520, Continued Operational Safety Branch, FAA, to make those findings. To be approved, the repair method, modification deviation, or alteration deviation must meet the certification basis of the airplane, and the approval must specifically refer to this AD.

**(k) Additional Information**

(1) For more information about this AD, contact Michael Sheldon, Aviation Safety Engineer, FAA, 2200 South 216th St., Des Moines, WA 98198; phone: 651–955–7451; email: *michael.e.sheldon@faa.gov*.

(2) Material identified in this AD that is not incorporated by reference is available at the address specified in paragraph (l)(3) this AD.

**(l) Material Incorporated by Reference**

(1) The Director of the Federal Register approved the incorporation by reference of the material listed in this paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) You must use this material as applicable to do the actions required by this AD, unless the AD specifies otherwise.

(i) Boeing Alert Requirements Bulletin 777–29A0047 RB, dated September 11, 2025.

(ii) [Reserved]

(3) For Boeing material identified in this AD, contact Boeing Commercial Airplanes, Attention: Contractual & Data Services (C&DS), 2600 Westminister Blvd., MC 110–SK57, Seal Beach, CA 90740–5600; telephone 562–797–1717; website *myboeingfleet.com*.

(4) You may view this material at the FAA, Airworthiness Products Section, Operational Safety Branch, 2200 South 216th St., Des Moines, WA. For information on the availability of this material at the FAA, call 206–231–3195.

(5) You may view this material at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, visit *www.archives.gov/federal-register/cfr/ibr-locations* or email *fr.inspection@nara.gov*.

Issued on November 5, 2025.

**Peter A. White,**

*Deputy Director, Integrated Certificate Management Division, Aircraft Certification Service.*

[FR Doc. 2025–20013 Filed 11–14–25; 8:45 am]

**BILLING CODE 4910–13–P**

**DEPARTMENT OF TRANSPORTATION****Federal Aviation Administration****14 CFR Part 39**

**[Docket No. FAA–2025–3989; Project Identifier MCAI–2025–00160–T]**

**RIN 2120–AA64**

**Airworthiness Directives; Airbus SAS Airplanes**

**AGENCY:** Federal Aviation Administration (FAA), DOT.

**ACTION:** Notice of proposed rulemaking (NPRM).

**SUMMARY:** The FAA proposes to supersede Airworthiness Directive (AD) 2025–03–06 and AD 2025–17–07, which apply to certain Airbus SAS Model A318 and A320 series airplanes; Model A319–111, –112, –113, –114, –115, –131, –132, –133, –151N, –153N, and –171N airplanes; and Model A321–111, –112, –131, –211, –212, –213, –231, –232, –251N, –251NX, –252N, –252NX, –253N, –253NX, –271N, –271NX, –272N, and –272NX airplanes. AD 2025–17–07 also applies to Airbus SAS Model A321–253NY airplanes. AD 2025–03–06 and AD 2025–17–07 require revising the existing maintenance or inspection program, as applicable, to incorporate new or more restrictive airworthiness limitations. Since the FAA issued AD 2025–03–06 and AD 2025–17–07, the FAA has determined that additional new or more restrictive airworthiness limitations are necessary. This proposed AD would continue to require certain actions in AD 2025–03–06 and all actions in AD 2025–17–07. This proposed AD would also require revising the existing maintenance or inspection program, as applicable, to incorporate new or more restrictive airworthiness limitations and add new airplane models. The FAA is proposing this AD to address the unsafe condition on these products.

**DATES:** The FAA must receive comments on this proposed AD by January 2, 2026.

**ADDRESSES:** You may send comments, using the procedures found in 14 CFR 11.43 and 11.45, by any of the following methods:

- **Federal eRulemaking Portal:** Go to *regulations.gov*. Follow the instructions for submitting comments.

- **Fax:** 202–493–2251.

- **Mail:** U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue SE, Washington, DC 20590.

- **Hand Delivery:** Deliver to Mail address above between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

**AD Docket:** You may examine the AD docket at *regulations.gov* under Docket No. FAA–2025–3989; or in person at Docket Operations between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this NPRM, the mandatory continuing airworthiness information (MCAI), any comments received, and other information. The street address for Docket Operations is listed above.

**Material Incorporated by Reference:**

- For European Union Aviation Safety Agency (EASA) material identified in this proposed AD, contact EASA, Konrad-Adenauer-Ufer 3, 50668 Cologne, Germany; telephone +49 221 8999 000; email *ADs@easa.europa.eu*. You may find this material on the EASA website at *ad.easa.europa.eu*. It is also available at *regulations.gov* under Docket No. FAA–2025–3989.

- You may view this material at the FAA, Airworthiness Products Section, Operational Safety Branch, 2200 South 216th St., Des Moines, WA. For information on the availability of this material at the FAA, call 206–231–3195.

#### FOR FURTHER INFORMATION CONTACT:

Aaron Nguyen, Aviation Safety Engineer, FAA, 2200 South 216th St., Des Moines, WA 98198; phone: 817–222–5134; email: *aaron.t.nguyen@faa.gov*.

#### SUPPLEMENTARY INFORMATION:

##### Comments Invited

The FAA invites you to send any written relevant data, views, or arguments about this proposal. Send your comments using a method listed under the **ADDRESSES** section. Include “Docket No. FAA–2025–3989; Project Identifier MCAI–2025–00160–T” at the beginning of your comments. The most helpful comments reference a specific portion of the proposal, explain the reason for any recommended change, and include supporting data. The FAA will consider all comments received by the closing date and may amend this proposal because of those comments.

Except for Confidential Business Information (CBI) as described in the following paragraph, and other information as described in 14 CFR 11.35, the FAA will post all comments received, without change, to *regulations.gov*, including any personal information you provide. The agency will also post a report summarizing each substantive verbal contact received about this NPRM.

#### Confidential Business Information

CBI is commercial or financial information that is both customarily and actually treated as private by its owner. Under the Freedom of Information Act (FOIA) (5 U.S.C. 552), CBI is exempt from public disclosure. If your comments responsive to this NPRM contain commercial or financial information that is customarily treated as private, that you actually treat as private, and that is relevant or responsive to this NPRM, it is important that you clearly designate the submitted comments as CBI. Please mark each page of your submission containing CBI as “PROPIN.” The FAA will treat such marked submissions as confidential under the FOIA, and they will not be placed in the public docket of this NPRM. Submissions containing CBI should be sent to Aaron Nguyen, Aviation Safety Engineer, FAA, 2200 South 216th St., Des Moines, WA 98198; phone: 817–222–5134; email: *aaron.t.nguyen@faa.gov*. Any commentary that the FAA receives that is not specifically designated as CBI will be placed in the public docket for this rulemaking.

#### Background

The FAA issued AD 2025–03–06, Amendment 39–22954 (90 FR 9595, February 14, 2025) (AD 2025–03–06), for certain Airbus SAS Model A318 and A320 series airplanes; Model A319–111, –112, –113, –114, –115, –131, –132, –133, –151N, –153N, and –171N airplanes; and Model A321–111, –112, –131, –211, –212, –213, –231, –232, –251N, –251NX, –252N, –252NX, –253N, –253NX, –271N, –271NX, –272N, and –272NX airplanes. AD 2025–03–06 was prompted by an MCAI originated by EASA, which is the Technical Agent for the Member States of the European Union. EASA issued AD 2024–0031, dated January 31, 2024; corrected February 1, 2024 (EASA AD 2024–0031), to correct an unsafe condition.

AD 2025–03–06 requires revising the existing maintenance or inspection program, as applicable, to incorporate new or more restrictive airworthiness limitations (specified in Airbus A318/

A319/A320/A321 Airworthiness Limitations Section (ALS) Part 2, Damage Tolerant Airworthiness Limitation Items (DT—ALI), Revision 10, Issue 02, dated November 30, 2023). The FAA issued AD 2025–03–06 to address fatigue cracking, accidental damage, or corrosion in principal structural elements, which could result in reduced structural integrity of the airplane.

The FAA issued AD 2025–17–07, Amendment 39–23117 (90 FR 41771, August 27, 2025) (AD 2025–17–07), for certain Airbus SAS Model A318, A320, and A321 series airplanes; and Model A319–111, –112, –113, –114, –115, –131, –132, –133, –151N, –153N, and –171N airplanes. AD 2025–17–07 was prompted by EASA AD 2024–0208, dated October 25, 2024 (EASA AD 2024–0208), to correct an unsafe condition.

AD 2025–17–07 requires revising the existing maintenance or inspection program, as applicable, to incorporate new airworthiness limitations (specified in Airbus A318/A319/A320/A321 ALS Part 2, Damage Tolerant Airworthiness Limitation Items (DT—ALI), Variation 10.3, dated August 7, 2024). The FAA issued AD 2025–17–07 to address fatigue cracking, accidental damage, or corrosion in principal structural elements, which could result in reduced structural integrity of the airplane.

#### Actions Since EASA AD 2024–0031 and AD 2024–0208 Were Issued

EASA issued AD 2025–0030, dated February 10, 2025 (EASA AD 2025–0030) (also referred to as the MCAI) to supersede AD 2024–0031 and AD 2024–0208 (corresponding to FAA AD 2025–03–06 and AD 2025–17–07, respectively), for certain Airbus SAS Model A318, A319, and A321 series airplanes; and Model A320–211, –212, –214, –215, –216, –231, –232, –233, –251N, –252N, –253N, –271N, –272N, and –273N airplanes. Model A320–215 airplanes are not certificated by the FAA and are not included on the U.S. type certificate data sheet; this proposed AD therefore does not include those airplanes in the applicability. The MCAI states that new or more restrictive airworthiness limitations have been developed, as specified in Airbus A318/A319/A320/A321 ALS Part 2, Damage Tolerant Airworthiness Limitation Items (DT—ALI), Revision 11, dated November 4, 2024.

The FAA is proposing this AD to address the unsafe condition on these products. The FAA is also proposing to add Model A319–173N and A321–253NY airplanes to the applicability. You may examine the MCAI in the AD

docket at *regulations.gov* under Docket No. FAA–2025–3989.

### Material Incorporated by Reference Under 1 CFR Part 51

The FAA reviewed EASA AD 2025–0030. This material specifies new or more restrictive airworthiness limitations for airplane structures and safe life limits.

This proposed AD would also require EASA AD 2024–0208, which the Director of the Federal Register approved for incorporation by reference as of October 1, 2025 (90 FR 41771, August 27, 2025).

This proposed AD would also require EASA AD 2024–0031, which the Director of the Federal Register approved for incorporation by reference as of March 21, 2025 (90 FR 9595, February 14, 2025).

This material is reasonably available because the interested parties have access to it through their normal course of business or by the means identified in the **ADDRESSES** section.

### FAA's Determination

These products have been approved by the civil aviation authority of another country and are approved for operation in the United States. Pursuant to the FAA's bilateral agreement with this State of Design Authority, that authority has notified the FAA of the unsafe condition described in the MCAI referenced above. The FAA is issuing this NPRM after determining that the unsafe condition described previously is likely to exist or develop in other products of the same type design.

### Proposed AD Requirements in This NPRM

This proposed AD would retain certain requirements of AD 2025–03–06 and all requirements of AD 2025–17–07. This proposed AD would also require revising the existing maintenance or inspection program, as applicable, to incorporate additional new or more restrictive airworthiness limitations, which are specified in EASA AD 2025–0030 already described, as proposed for incorporation by reference. Any differences with EASA AD 2025–0030 are identified as exceptions in the regulatory text of this proposed AD.

This proposed AD would require revisions to certain operator maintenance documents to include new actions (e.g., inspections) and Critical Design Configuration Control Limitations (CDCCLs). Compliance with these actions and CDCCLs is required by 14 CFR 91.403(c). For airplanes that have been previously modified, altered, or repaired in the areas addressed by

this proposed AD, the operator may not be able to accomplish the actions described in the revisions. In this situation, to comply with 14 CFR 91.403(c), the operator must request approval for an alternative method of compliance (AMOC) according to paragraph (p)(1) of this proposed AD.

### Explanation of Required Compliance Information

In the FAA's ongoing efforts to improve the efficiency of the AD process, the FAA developed a process to use some civil aviation authority (CAA) ADs as the primary source of information for compliance with requirements for corresponding FAA ADs. The FAA has been coordinating this process with manufacturers and CAAs. As a result, the FAA proposes to retain the IBRs of EASA AD 2024–0031 and EASA AD 2024–0208 and incorporate EASA AD 2025–0030 by reference in the FAA final rule. This proposed AD would, therefore, require compliance with EASA AD 2024–0031, EASA AD 2024–0208, and EASA AD 2025–0030 through that incorporation, except for any differences identified as exceptions in the regulatory text of this proposed AD. Using common terms that are the same as the heading of a particular section in EASA AD 2024–0031, EASA AD 2024–0208, or EASA AD 2025–0030 does not mean that operators need comply only with that section. For example, where the AD requirement refers to "all required actions and compliance times," compliance with this proposed AD requirement is not limited to the section titled "Required Action(s) and Compliance Time(s)" in EASA AD 2024–0031, EASA AD 2024–0208, or EASA AD 2025–0030. Material required by EASA AD 2024–0031, EASA AD 2024–0208, and EASA AD 2025–0030 for compliance will be available at *regulations.gov* by searching for and locating Docket No. FAA–2025–3989 after the FAA final rule is published.

### Airworthiness Limitation ADs Using the New Process

The FAA's process of incorporating by reference MCAI ADs as the primary source of information for compliance with corresponding FAA ADs has been limited to certain MCAI ADs (primarily those with service bulletins as the primary source of information for accomplishing the actions required by the FAA AD). However, the FAA is now expanding the process to include MCAI ADs that require a change to airworthiness limitation documents, such as airworthiness limitation sections.

For these ADs that incorporate by reference an MCAI AD that changes airworthiness limitations, the FAA requirements are unchanged. Operators must revise the existing maintenance or inspection program, as applicable, to incorporate the information specified in the new airworthiness limitation document. The airworthiness limitations must be followed according to 14 CFR 91.403(c) and 91.409(e).

The previous format of the airworthiness limitation ADs included a paragraph that specified that no alternative actions (e.g., inspections), intervals, or CDCCLs may be used unless the actions, intervals, and CDCCLs are approved as an AMOC in accordance with the procedures specified in the AMOCs paragraph under "Additional AD Provisions." This new format includes a "New Provisions for Alternative Actions, Intervals, and CDCCLs" paragraph that does not specifically refer to AMOCs, but operators may still request an AMOC to use an alternative action, interval, or CDCCL.

### Costs of Compliance

The FAA estimates that this AD, if adopted as proposed, would affect 1,900 airplanes of U.S. registry. The FAA estimates the following costs to comply with this proposed AD:

The FAA estimates the total cost per operator for the retained actions from AD 2025–03–06 to be \$7,650 (90 work-hours × \$85 per work-hour).

The FAA estimates the total cost per operator for the retained actions from AD 2025–17–07 to be \$7,650 (90 work-hours × \$85 per work-hour).

The FAA has determined that revising the existing maintenance or inspection program takes an average of 90 work-hours per operator, although the agency recognizes that this number may vary from operator to operator. Since operators incorporate maintenance or inspection program changes for their affected fleet(s), the FAA has determined that a per-operator estimate is more accurate than a per-airplane estimate.

The FAA estimates the total cost per operator for the new proposed actions to be \$7,650 (90 work-hours × \$85 per work-hour).

### Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. Subtitle VII: Aviation Programs, describes in more detail the scope of the Agency's authority.

The FAA is issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section 44701: General requirements. Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

### Regulatory Findings

The FAA determined that this proposed AD would not have federalism implications under Executive Order 13132. This proposed AD would not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify this proposed regulation:

- (1) Is not a “significant regulatory action” under Executive Order 12866,
- (2) Would not affect intrastate aviation in Alaska, and
- (3) Would not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

### List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

### The Proposed Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA proposes to amend 14 CFR part 39 as follows:

### PART 39—AIRWORTHINESS DIRECTIVES

- 1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

#### § 39.13 [Amended]

- 2. The FAA amends § 39.13 by:
- a. Removing Airworthiness Directive (AD) 2025–03–06, Amendment 39–22954 (90 FR 9595, February 14, 2025); and AD 2025–17–07, Amendment 39–23117 (90 FR 41771, August 27, 2025) and
  - b. Adding the following new AD:
- Airbus SAS:** Docket No. FAA–2025–3989; Project Identifier MCAI–2025–00160–T.

#### (a) Comments Due Date

The FAA must receive comments on this airworthiness directive (AD) by January 2, 2026.

#### (b) Affected ADs

(1) This AD replaces AD 2025–03–06, Amendment 39–22954 (90 FR 9595, February 14, 2025) (AD 2025–03–06).

(2) This AD replaces AD 2025–17–07, Amendment 39–23117 (90 FR 41771, August 27, 2025).

#### (c) Applicability

This AD applies to Airbus SAS Model airplanes identified in paragraphs (c)(1) through (4) of this AD, certificated in any category, with an original airworthiness certificate or original export certificate of airworthiness issued on or before November 4, 2024.

(1) Model A318–111, –112, –121, and –122 airplanes.

(2) Model A319–111, –112, –113, –114, –115, –131, –132, –133, –151N, –153N, –171N, and –173N airplanes.

(3) Model A320–211, –212, –214, –216, –231, –232, –233, –251N, –252N, –253N, –271N, –272N, and –273N airplanes.

(4) Model A321–111, –112, –131, –211, –212, –213, –231, –232, –251N, –251NX, –252N, –252NX, –253N, –253NX, –253NY, –271N, –271NX, –272N, and –272NX airplanes.

#### (d) Subject

Air Transport Association (ATA) of America Code 05, Time Limits/Maintenance Checks.

#### (e) Unsafe Condition

This AD was prompted by a determination that new or more restrictive airworthiness limitations are necessary. The FAA is issuing this AD to address fatigue cracking, accidental damage, or corrosion in principal structural elements. The unsafe condition, if not addressed, could result in reduced structural integrity of the airplane.

#### (f) Compliance

Comply with this AD within the compliance times specified, unless already done.

#### (g) Retained Revision of the Existing Maintenance or Inspection Program From AD 2025–03–06, With a New Terminating Action

This paragraph restates the requirements of paragraph (n) of AD 2025–03–06, with a new terminating action. For airplanes with an original airworthiness certificate or original export certificate of airworthiness issued on or before December 19, 2023: Except as specified in paragraph (h) of this AD, comply with all required actions and compliance times specified in, and in accordance with, European Union Aviation Safety Agency (EASA) AD 2024–0031, dated January 31, 2024; corrected February 1, 2024 (EASA AD 2024–0031). Accomplishing the revision of the existing maintenance or inspection program required by paragraph (m) of this AD terminates the requirements of this paragraph.

#### (h) Retained Exceptions to EASA AD 2024–0031, With No Changes

This paragraph restates the exceptions specified in paragraph (o) of AD 2025–03–06, with no changes.

(1) This AD does not adopt the requirements specified in paragraphs (1) and (2) of EASA AD 2024–0031.

(2) Paragraph (3) of EASA AD 2024–0031 specifies revising “the approved AMP” within 12 months after its effective date, but this AD requires revising the existing maintenance or inspection program, as applicable, within 90 days after March 21, 2025 (the effective date of AD 2025–03–06).

(3) The initial compliance time for doing the tasks specified in paragraph (3) of EASA AD 2024–0031 is at the applicable “associated thresholds” as incorporated by the requirements of paragraph (3) of EASA AD 2024–0031, or within 90 days after March 21, 2025 (the effective date of AD 2025–03–06), whichever occurs later.

(4) This AD does not adopt the provisions specified in paragraphs (4), (5), and (6) of EASA AD 2024–0031.

(5) This AD does not require incorporating Section 4, “Damage Tolerant—Airworthiness Limitations Items—tasks beyond MPPT,” of “the ALS” specified in EASA AD 2024–0031.

(6) This AD does not adopt the “Remarks” section of EASA AD 2024–0031.

#### (i) Retained Restrictions on Alternative Actions, Intervals, and Critical Design Configuration Control Limitations (CDCCLs) From AD 2025–03–06, With a New Exception

This paragraph restates the requirements of paragraph (p) of AD 2025–03–06, with a new exception. Except as required by paragraphs (j) and (m) of this AD, after the existing maintenance or inspection program has been revised as required by paragraph (g) of this AD, no alternative actions (e.g., inspections), intervals, and CDCCLs are allowed unless they are approved as specified in the provisions of the “Ref. Publications” section of EASA AD 2024–0031.

#### (j) Retained Revision of the Existing Maintenance or Inspection Program From AD 2025–17–07, With a New Terminating Action

This paragraph restates the requirements of paragraph (g) of AD 2025–17–07, with a new terminating action. For airplanes with an original airworthiness certificate or original export certificate of airworthiness issued on or before August 7, 2024: Except as specified in paragraph (k) of this AD, comply with all required actions and compliance times specified in, and in accordance with, EASA AD 2024–0208, dated October 25, 2024 (EASA AD 2024–0208). Accomplishing the revision of the existing maintenance or inspection program required by paragraph (m) of this AD terminates the requirements of this paragraph.

#### (k) Retained Exceptions to EASA AD 2024–0208, With No Changes

This paragraph restates the exceptions specified in paragraph (h) of AD 2025–17–07, with no changes.

(1) This AD does not adopt the requirements specified in paragraphs (1) and (2) of EASA AD 2024–0208.

(2) Paragraph (3) of EASA AD 2024–0208 specifies revising “the approved AMP” within 12 months after its effective date, but this AD requires revising the existing maintenance or inspection program, as applicable, within 90 days after October 1, 2025 (the effective date of AD 2025–17–07).

(3) The initial compliance time for doing the tasks specified in paragraph (3) of EASA AD 2024–0208 is at the applicable “associated thresholds” as incorporated by the requirements of paragraph (3) of EASA AD 2024–0208, or within 90 days after October 1, 2025 (the effective date of AD 2025–17–07), whichever occurs later.

(4) This AD does not adopt the provisions specified in paragraph (4) of EASA AD 2024–0208.

(5) This AD does not adopt the “Remarks” section of EASA AD 2024–0208.

#### **(l) Retained Restrictions on Alternative Actions and Intervals From AD 2025–17–07, With a New Exception**

This paragraph restates the requirements of paragraph (i) of AD 2025–17–07, with a new exception. Except as required by paragraph (m) of this AD, after the existing maintenance or inspection program has been revised as required by paragraph (j) of this AD, no alternative actions (e.g., inspections) and intervals are allowed unless they are approved as specified in the provisions of the “Ref. Publications” section of EASA AD 2024–0208.

#### **(m) New Revision of the Existing Maintenance or Inspection Program**

Except as specified in paragraph (n) of this AD: Comply with all required actions and compliance times specified in, and in accordance with, EASA AD 2025–0030, dated February 10, 2025. Accomplishing the revision of the existing maintenance or inspection program required by this paragraph terminates the requirements of paragraphs (g) and (j) of this AD.

#### **(n) Exceptions to EASA AD 2025–0030**

(1) This AD does not adopt the requirements specified in paragraphs (1) and (2) of EASA AD 2025–0030.

(2) Paragraph (3) of EASA AD 2025–0030 specifies revising “the approved AMP” within 12 months after its effective date, but this AD requires revising the existing maintenance or inspection program, as applicable, within 90 days after the effective date of this AD.

(3) The initial compliance time for doing the tasks specified in paragraph (3) of EASA AD 2025–0030 is at the applicable “associated thresholds” as incorporated by the requirements of paragraph (3) of EASA AD 2025–0030, or within 90 days after the effective date of this AD, whichever occurs later.

(4) This AD does not adopt the provisions specified in paragraphs (4) and (5) of EASA AD 2025–0030.

(5) This AD does not adopt the “Remarks” section of EASA AD 2025–0030.

(6) This AD does not require incorporating Section 4, “Damage Tolerant—Airworthiness Limitations Items—tasks beyond MPPT,” of “the ALS” specified in EASA AD 2025–0030.

#### **(o) New Provisions for Alternative Actions, Intervals, and CDCCLs**

After the existing maintenance or inspection program has been revised as required by paragraph (m) of this AD, no alternative actions (e.g., inspections), intervals, and CDCCLs are allowed unless they are approved as specified in the provisions of the “Ref. Publications” section of EASA AD 2025–0030.

#### **(p) Additional AD Provisions**

The following provisions also apply to this AD:

(1) *Alternative Methods of Compliance (AMOCs)*: The Manager, AIR–520, Continued Operational Safety Branch, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or responsible Flight Standards Office, as appropriate. If sending information directly to the manager of the Continued Operational Safety Branch, send it to the attention of the person identified in paragraph (q) of this AD and email to: [AMOC@faa.gov](mailto:AMOC@faa.gov). Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the responsible Flight Standards Office.

(2) *Contacting the Manufacturer*: For any requirement in this AD to obtain instructions from a manufacturer, the instructions must be accomplished using a method approved by the Manager, Continued Operational Safety Branch, FAA; or EASA; or Airbus SAS’s EASA Design Organization Approval (DOA). If approved by the DOA, the approval must include the DOA-authorized signature.

#### **(q) Additional Information**

For more information about this AD, contact Aaron Nguyen, Aviation Safety Engineer, FAA, 2200 South 216th St., Des Moines, WA 98198; phone: 817–222–5134; email: [aaron.t.nguyen@faa.gov](mailto:aaron.t.nguyen@faa.gov).

#### **(r) Material Incorporated by Reference**

(1) The Director of the Federal Register approved the incorporation by reference (IBR) of the material listed in this paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) You must use this material as applicable to do the actions required by this AD, unless this AD specifies otherwise.

(3) The following material was approved for IBR on [DATE 35 DAYS AFTER PUBLICATION OF THE FINAL RULE].

(i) European Union Aviation Safety Agency (EASA) AD 2025–0030, dated February 10, 2025.

(ii) [Reserved]

(4) The following material was approved for IBR on October 1, 2025 (90 FR 41771, August 27, 2025).

(i) EASA AD 2024–0208, dated October 25, 2024.

(ii) [Reserved]

(5) The following material was approved for IBR on March 21, 2025 (90 FR 9595, February 14, 2025).

(i) EASA AD 2024–0031, dated January 31, 2024; corrected February 1, 2024.

(ii) [Reserved]

(6) For EASA material identified in this AD, contact EASA, Konrad-Adenauer-Ufer 3,

50668 Cologne, Germany; telephone +49 221 8999 000; email [ADs@easa.europa.eu](mailto:ADs@easa.europa.eu). You may find this material on the EASA website at [ad.easa.europa.eu](http://ad.easa.europa.eu).

(7) You may view this material at the FAA, Airworthiness Products Section, Operational Safety Branch, 2200 South 216th St., Des Moines, WA. For information on the availability of this material at the FAA, call 206–231–3195.

(8) You may view this material at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, visit [www.archives.gov/federal-register/cfr/ibr-locations](http://www.archives.gov/federal-register/cfr/ibr-locations) or email [fr.inspection@nara.gov](mailto:fr.inspection@nara.gov).

Issued on October 22, 2025.

**Lona C. Saccomando,**

*Acting Deputy Director, Integrated Certificate Management Division, Aircraft Certification Service.*

[FR Doc. 2025–20012 Filed 11–14–25; 8:45 am]

**BILLING CODE 4910–13–P**

## **DEPARTMENT OF TRANSPORTATION**

### **Federal Aviation Administration**

#### **14 CFR Part 39**

**[Docket No. FAA–2025–3988; Project Identifier MCAI–2025–00443–T]**

**RIN 2120–AA64**

#### **Airworthiness Directives; Airbus SAS Airplanes**

**AGENCY:** Federal Aviation Administration (FAA), DOT.

**ACTION:** Notice of proposed rulemaking (NPRM).

**SUMMARY:** The FAA proposes to adopt a new airworthiness directive (AD) for certain Airbus SAS Model A330–200, –200 Freighter, –300, –800, and –900 series airplanes. This proposed AD was prompted by the identification of an incorrect shot peening application implemented in production. This proposed AD would require repetitive special detailed inspections (SDIs) of affected central windshield frames and applicable corrective actions. The FAA is proposing this AD to address the unsafe condition on these products.

**DATES:** The FAA must receive comments on this proposed AD by January 2, 2026.

**ADDRESSES:** You may send comments, using the procedures found in 14 CFR 11.43 and 11.45, by any of the following methods:

- *Federal eRulemaking Portal:* Go to [regulations.gov](http://regulations.gov). Follow the instructions for submitting comments.

- *Fax:* 202–493–2251.

- *Mail:* U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room