

(g) Inspection and Corrective Actions

Within 200 hours time-in-service (TIS) or within 12 months, whichever occurs first after the effective date of this AD, prepare the airplane and gain access in accordance with steps 1 through 7 of the Accomplishment Instructions in ASI Aviation Service Bulletin No. F406-62, Revision 01, dated December 14, 2018 (SB F406-62R1), and inspect each avionics bus CB switch part number (P/N) CM3589-50 to identify the date code.

(1) If a CB switch does not have a date code, before further flight, remove the CB switch from service and install CB switch P/N 4061-2400-1 in accordance with steps 9 through 14 of the Accomplishment Instructions in SB F406-62R1.

(2) If a CB switch has a date code earlier than 0434, before the CB switch exceeds 1,000 hours TIS since first installation on an airplane, remove the CB switch from service and install CB switch P/N 4061-2400-1 in accordance with steps 9 through 14 of the Accomplishment Instructions in SB F406-62R1.

(3) If a CB switch has a date code 0434 or later, before the CB switch exceeds 6 years since first installation on an airplane or within 12 months after the effective date of this AD, whichever occurs later, remove the CB switch from service and install CB switch P/N 4061-2400-1 in accordance with steps 9 through 14 of the Accomplishment Instructions in SB F406-62R1.

(h) Replacements

Within 200 hours TIS or within 12 months, whichever occurs first after the effective date of this AD, remove each CB switch P/N CM3589-20 from service, re-identify the CB panel, and install CB switches with P/N 406E2450-00000-100 in accordance with Part 1, steps 1 through 13, of the Accomplishment Instructions in ASI Aviation Service Bulletin No. F406-90, dated December 14, 2018 (SB F406-90).

(i) Life Limit

Before exceeding 6 years since first installation on an airplane and thereafter at intervals not to exceed 6 years, remove each CB switch P/N 4061-2400-1 and P/N 406E2450-00000-100 from service and replace it in accordance with steps 9 through 14 of the Accomplishment Instructions in SB F406-62R1 or Part 1, steps 1 through 13, of the Accomplishment Instructions in SB F406-90, as applicable.

(j) Parts Installation Prohibition

As of the effective date of this AD, do not install a CB switch P/N CM3589-50 or P/N CM3589-20 on any airplane.

(k) Credit for Previous Actions

You may take credit for the actions required by paragraph (g) of this AD if you performed those actions before the effective date of this AD using Reims Aviation Industries Service Bulletin No. F406-62, dated March 8, 2006.

(l) Alternative Methods of Compliance (AMOCs)

(1) The Manager, International Validation Branch, has the authority to approve AMOCs

for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the manager of the certification office, send it to the attention of the person identified in paragraph (m)(1) of this AD or email: 9-AVS-AIR-730-AMOC@faa.gov.

(2) Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office.

(m) Related Information

(1) For more information about this AD, contact Gregory Johnson, Aviation Safety Engineer, International Validation Section, FAA, 901 Locust, Room 301, Kansas City, MO 64106-2641; phone: (720) 626-5462; email: gregory.johnson@faa.gov.

(2) Refer to European Aviation Safety Agency (EASA) AD 2019-0015, dated January 29, 2019, for more information. You may examine the EASA AD in the AD docket at <https://www.regulations.gov> by searching for and locating Docket No. FAA-2021-0714.

(3) Service information identified in this AD that is not incorporated by reference is available at the addresses specified in paragraphs (n)(3) and (4) of this AD.

(n) Material Incorporated by Reference

(1) The Director of the Federal Register approved the incorporation by reference of the service information listed in this paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) You must use this service information as applicable to do the actions required by this AD, unless the AD specifies otherwise.

(i) ASI Aviation Service Bulletin No. F406-62, Revision 01, dated December 14, 2018.

(ii) ASI Aviation Service Bulletin No. F406-90, dated December 14, 2018.

(3) For service information identified in this AD, contact ASI Aviation, A rodrome de Reims Prunay, 51360 Prunay, France; telephone: +33 3 26 48 46 84; fax: +33 3 26 49 18 57; email: contact@asi-aviation.fr; website: <https://asi-aviation.fr/page-Accueil.html>.

(4) You may view this service information at FAA, Airworthiness Products Section, Operational Safety Branch, 901 Locust, Kansas City, MO 64106. For information on the availability of this material at the FAA, call (816) 329-4148.

(5) You may view this service information that is incorporated by reference at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, email: fr.inspection@nara.gov, or go to: <https://www.archives.gov/federal-register/cfr/ibr-locations.html>.

Issued on October 22, 2021.

Lance T. Gant,

Director, Compliance & Airworthiness Division, Aircraft Certification Service.

[FR Doc. 2021-25688 Filed 11-24-21; 8:45 am]

BILLING CODE 4910-13-P

DEPARTMENT OF TRANSPORTATION**Federal Aviation Administration****14 CFR Part 39**

[Docket No. FAA-2021-0461; Project Identifier MCAI-2021-00156-R; Amendment 39-21775; AD 2021-22-02]

RIN 2120-AA64

Airworthiness Directives; Leonardo S.p.a. Helicopters

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final rule.

SUMMARY: The FAA is adopting a new airworthiness directive (AD) for all Leonardo S.p.a. Model AB139 and AW139 helicopters. This AD was prompted by a report of a short circuit caused by chafing of the electrical wiring in the overhead panel. This AD requires an initial detailed inspection inside the overhead panel for certain helicopters, repetitive detailed inspections inside the overhead panel for all helicopters, and corrective actions if necessary, as specified in a European Union Aviation Safety Agency (EASA) AD, which is incorporated by reference. The FAA is issuing this AD to address the unsafe condition on these products.

DATES: This AD is effective January 3, 2022.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in this AD as of January 3, 2022.

ADDRESSES: For material incorporated by reference (IBR) in this AD, contact the EASA, Konrad-Adenauer-Ufer 3, 50668 Cologne, Germany; phone: +49 221 8999 000; email: ADs@easa.europa.eu; internet: www.easa.europa.eu. You may find this material on the EASA website at <https://ad.easa.europa.eu>. You may view this material at the FAA, Office of the Regional Counsel, Southwest Region, 10101 Hillwood Pkwy., Room 6N-321, Fort Worth, TX 76177. For information on the availability of this material at the FAA, call 817-222-5110. It is also available in the AD docket on the internet at <https://www.regulations.gov> by searching for and locating Docket No. FAA-2021-0461.

Examining the AD Docket

You may examine the AD docket on the internet at <https://www.regulations.gov> by searching for and locating Docket No. FAA-2021-0461; or in person at Docket Operations between 9 a.m. and 5 p.m., Monday

through Friday, except Federal holidays. The AD docket contains this final rule, any comments received, and other information. The address for Docket Operations is U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE, Washington, DC 20590.

FOR FURTHER INFORMATION CONTACT: Jacob Fitch, Aerospace Engineer, COS Program Management Section, Operational Safety Branch, Compliance & Airworthiness Division, FAA, 10101 Hillwood Pkwy., Fort Worth, TX 76177; phone: (817) 222-4130; email: jacob.fitch@faa.gov.

SUPPLEMENTARY INFORMATION:

Background

The EASA, which is the Technical Agent for the Member States of the European Union, has issued EASA AD 2021-0044, dated February 5, 2021 (EASA AD 2021-0044) (also referred to as the Mandatory Continuing Airworthiness Information, or the MCAI), to correct an unsafe condition for all Leonardo S.p.a. Model AB139 and AW139 helicopters.

The FAA issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 by adding an AD that would apply to all Leonardo S.p.a. Model AB139 and AW139 helicopters. The NPRM published in the **Federal Register** on June 14, 2021 (86 FR 31451). The NPRM was prompted by a report of a short circuit caused by chafing of the electrical wiring in the overhead panel. The NPRM proposed to require an initial detailed inspection inside the overhead panel for certain helicopters, repetitive detailed inspections inside the overhead panel for all helicopters, and corrective actions if necessary, as specified in EASA AD 2021-0044.

The FAA is issuing this AD to address a short circuit caused by chafing of the electrical wiring in the overhead panel, which could cause damaged electrical

wiring, possible fire in the overhead panel, and loss of control of the helicopter. See the MCAI for additional background information.

Comments

The FAA gave the public the opportunity to participate in developing this final rule. The following presents the comment received on the NPRM and the FAA's response to each comment.

Request To Include Credit for Later Service Information

An anonymous commenter requested that the FAA revise the NPRM to include a paragraph that allows credit for the use of a later revision of the service information referenced in EASA AD 2021-0044.

The FAA disagrees with the request because, in this case, credit is unnecessary. The FAA is incorporating by reference EASA AD 2021-0044 as the method for accomplishing the actions required by this AD. EASA AD 2021-0044 includes the Ref. Publications section, which allows the use of later approved revisions of the service information referenced in EASA AD 2021-0044. Therefore, no change has been made to this AD.

Conclusion

The FAA reviewed the relevant data, considered the comment received, and determined that air safety and the public interest require adopting this final rule as proposed, except for minor editorial changes. The FAA has determined that these minor changes:

- Are consistent with the intent that was proposed in the NPRM for addressing the unsafe condition; and
- Do not add any additional burden upon the public than was already proposed in the NPRM.

Related Service Information Under 1 CFR Part 51

EASA AD 2021-0044 requires an initial detailed inspection (for certain

helicopters) inside the overhead panel for chafing of the cable harnesses and for correct clearance between the anchor nuts/screws and the cable harnesses, of the screws for correct length, and of the supports for sound bonding, and corrective actions if necessary; repetitive detailed inspections (for all helicopters) inside the overhead panel for the condition of the white protective tape on the anchor nuts, and for chafing of the cable harnesses and for correct clearance between the anchor nuts/screws and the cable harnesses, and corrective actions if necessary. Corrective actions include applying a white protective tape on the anchor nuts, replacement of incorrect length screws, replacement of damaged cables and fuses, rerouting of cable harnesses, replacement of supports, and removal and replacement of the white protective tape.

This material is reasonably available because the interested parties have access to it through their normal course of business or by the means identified in the **ADDRESSES** section.

Differences Between This AD and the MCAI

Although EASA AD 2021-0044 and the service information referenced in EASA AD 2021-0044 specify to submit certain information to the manufacturer, this AD does not include that requirement.

Interim Action

The FAA considers this AD interim action. If final action is later identified, the FAA might consider further rulemaking then.

Costs of Compliance

The FAA estimates that this AD affects 128 helicopters of U.S. Registry. The FAA estimates the following costs to comply with this AD:

ESTIMATED COSTS FOR REQUIRED ACTIONS

Action	Labor cost	Parts cost	Cost per product	Cost on U.S. operators
Inspection for chafing, clearance, screw length, and bonding.	1 work-hour × \$85 per hour = \$85	\$0	\$85	\$10,880.
Repetitive inspections for chafing, clearance, and tape condition.	1 work-hour × \$85 per hour = \$85 per inspection cycle.	\$0 per inspection cycle	\$85 per inspection cycle	\$10,880 per inspection cycle.

The FAA estimates the following costs to do any necessary on-condition actions that would be required based on

the results of any required actions. The FAA has no way of determining the

number of helicopters that might need these on-condition actions:

ESTIMATED COSTS OF ON-CONDITION ACTIONS

Action	Labor cost	Parts cost	Cost per product
Replace screws, cables, fuses, supports, and protective tape; reroute harnesses.	5 work-hours × \$85 per hour = \$425	\$600	\$1.025
Apply protective tape	1 work-hour × \$85 per hour = \$85	50	135
Replace cables, fuses and protective tape	1 work-hour × \$85 per hour = \$85	600	685

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. Subtitle VII: Aviation Programs, describes in more detail the scope of the Agency's authority.

The FAA is issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section 44701: General requirements. Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

This AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that this AD:

- (1) Is not a "significant regulatory action" under Executive Order 12866,
- (2) Will not affect intrastate aviation in Alaska, and
- (3) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

The Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

■ 2. The FAA amends § 39.13 by adding the following new airworthiness directive:

2021–22–02 Leonardo S.p.a.: Amendment 39–21775; Docket No. FAA–2021–0461; Project Identifier MCAI–2021–00156–R.

(a) Effective Date

This airworthiness directive (AD) is effective January 3, 2022.

(b) Affected ADs

None.

(c) Applicability

This AD applies to all Leonardo S.p.a. Model AB139 and AW139 helicopters, certificated in any category.

(d) Subject

Joint Aircraft System Component (JASC) Code 2400, Electrical Power System.

(e) Unsafe Condition

This AD was prompted by a report of a short circuit caused by chafing of the electrical wiring in the overhead panel. The FAA is issuing this AD to address a short circuit caused by chafing of the electrical wiring in the overhead panel, which could cause damaged electrical wiring, possible fire in the overhead panel, and loss of control of the helicopter.

(f) Compliance

Comply with this AD within the compliance times specified, unless already done.

(g) Requirements

Except as specified in paragraph (h) of this AD: Comply with all required actions and compliance times specified in, and in accordance with, European Union Aviation Safety Agency (EASA) AD 2021–0044, dated February 5, 2021 (EASA AD 2021–0044).

(h) Exceptions to EASA AD 2021–0044

(1) Where EASA AD 2021–0044 refers to its effective date, this AD requires using the effective date of this AD.

(2) The "Remarks" section of EASA AD 2021–0044 does not apply to this AD.

(3) Where EASA AD 2021–0044 refers to flight hours (FH), this AD requires using hours time-in-service.

(4) Where paragraphs (3) and (5) of EASA AD 2021–0044 refer to "any discrepancy," for this AD, discrepancies include chafing of the cable harnesses or incorrect clearance between the anchor nuts/screws and the cable harnesses, incorrect length of the screws, inadequately bonded supports, and poor condition of the white protective tape.

(i) No Reporting Requirement

Although EASA AD 2021–0044 and the service information referenced in EASA AD 2021–0044 specify to submit certain information to the manufacturer, this AD does not include that requirement.

(j) Special Flight Permit

Special flight permits may be issued in accordance with 14 CFR 21.197 and 21.199 to operate the helicopter to a location where the helicopter can be modified (if the operator elects to do so), provided the flight is straight, level, and avoids areas of known turbulence.

(k) Alternative Methods of Compliance (AMOCs)

(1) The Manager, International Validation Branch, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the manager of the International Validation Branch, send it to the attention of the person identified in paragraph (l) of this AD. Information may be emailed to: 9-AVS-AIR-730-AMOC@faa.gov.

(2) Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office.

(l) Related Information

For more information about this AD, contact Jacob Fitch, Aerospace Engineer, COS Program Management Section, Operational Safety Branch, Compliance & Airworthiness Division, FAA, 10101 Hillwood Pkwy., Fort Worth, TX 76177; phone: (817) 222–4130; email: jacob.fitch@faa.gov.

(m) Material Incorporated by Reference

(1) The Director of the Federal Register approved the incorporation by reference (IBR) of the service information listed in this paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) You must use this service information as applicable to do the actions required by this AD, unless this AD specifies otherwise.

(i) European Union Aviation Safety Agency (EASA) AD 2021–0044, dated February 5, 2021.

(ii) [Reserved]

(3) For EASA AD 2021–0044, contact the EASA, Konrad-Adenauer-Ufer 3, 50668 Cologne, Germany; phone: +49 221 8999 000; email: ADs@easa.europa.eu; Internet: www.easa.europa.eu. You may find this EASA AD on the EASA website at <https://ad.easa.europa.eu>.

(4) You may view this service information at the FAA, Office of the Regional Counsel, Southwest Region, 10101 Hillwood Pkwy., Room 6N–321, Fort Worth, TX 76177. For information on the availability of this material at the FAA, call 817–222–5110. This material may be found in the AD docket on the internet at <https://www.regulations.gov> by searching for and locating Docket No. FAA–2021–0461.

(5) You may view this material that is incorporated by reference at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, email fr.inspection@nara.gov, or go to <https://www.archives.gov/federal-register/cfr/ibr-locations.html>.

Issued on October 13, 2021.

Gaetano A. Sciortino,

Deputy Director for Strategic Initiatives, Compliance & Airworthiness Division, Aircraft Certification Service.

[FR Doc. 2021–25691 Filed 11–24–21; 8:45 am]

BILLING CODE 4910–13–P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA–2021–0693; Project Identifier MCAI–2020–01666–R; Amendment 39–21788; AD 2021–22–15]

RIN 2120–AA64

Airworthiness Directives; Airbus Helicopters

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final rule.

SUMMARY: The FAA is adopting a new airworthiness directive (AD) for certain Airbus Helicopters Model AS332L2 and EC225LP helicopters. This AD was prompted by a design deficiency. This AD requires modifying the hoist control power supply, as specified in a European Union Aviation Safety Agency (EASA) AD, which is incorporated by reference. The FAA is issuing this AD to address the unsafe condition on these products.

DATES: This AD is effective January 3, 2022.

The Director of the Federal Register approved the incorporation by reference

of a certain publication listed in this AD as of January 3, 2022.

ADDRESSES: For material incorporated by reference (IBR) in this AD, contact the EASA, Konrad-Adenauer-Ufer 3, 50668 Cologne, Germany; telephone +49 221 8999 000; email ADs@easa.europa.eu; internet www.easa.europa.eu. You may find this material on the EASA website at <https://ad.easa.europa.eu>. You may view this material at the FAA, Office of the Regional Counsel, Southwest Region, 10101 Hillwood Pkwy., Room 6N–321, Fort Worth, TX 76177. For information on the availability of this material at the FAA, call (817) 222–5110. It is also available in the AD docket at <https://www.regulations.gov> by searching for and locating Docket No. FAA–2021–0693.

Examining the AD Docket

You may examine the AD docket at <https://www.regulations.gov> by searching for and locating Docket No. FAA–2021–0693; or in person at Docket Operations between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this final rule, any comments received, and other information. The address for Docket Operations is U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue SE, Washington, DC 20590.

FOR FURTHER INFORMATION CONTACT:

Ronnea Derby, Aerospace Engineer, Denver ACO Branch, Compliance & Airworthiness Division, FAA, 26805 E 68th Ave., Mail Stop: Room 214, Denver, CO 80249; telephone (303) 342–1093; email Ronnea.L.Derby@faa.gov.

SUPPLEMENTARY INFORMATION:

Background

EASA, which is the Technical Agent for the Member States of the European Union, has issued EASA AD 2020–0281, dated December 16, 2020 (EASA AD 2020–0281), to correct an unsafe condition for certain serial-numbered Airbus Helicopters (AH), formerly Eurocopter, Eurocopter France, Aerospatiale Model AS 332 L2 and EC 225 LP helicopters.

The FAA issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 by adding an AD that would apply to certain serial-numbered Airbus Helicopters Model AS332L2 and EC225LP helicopters. The NPRM published in the **Federal Register** on August 25, 2021 (86 FR 47420). The NPRM was prompted by a design deficiency involving the incorrect wiring routing of the electrical hoist

installation. The affected wiring was not protected by the circuit breaker that was intended to provide electrical protection for that wiring. The NPRM proposed to require modifying the hoist control power supply, as specified in EASA AD 2020–0281.

The FAA is issuing this AD to correct the electrical hoist installation wiring routing. See EASA AD 2020–0281 for additional background information.

Discussion of Final Airworthiness Directive

Comments

The FAA gave the public the opportunity to participate in developing this final rule. The FAA received no comments on the NPRM or on the determination of the cost to the public.

Conclusion

The FAA reviewed the relevant data and determined that air safety and the public interest require adopting this final rule as proposed.

Related Service Information Under 1 CFR Part 51

EASA AD 2020–0281 requires modifying the hoist control power supply.

This material is reasonably available because the interested parties have access to it through their normal course of business or by the means identified in the **ADDRESSES** section.

Costs of Compliance

The FAA estimates that this AD affects 5 helicopters of U.S. Registry. Labor rates are estimated at \$85 per work-hour. Based on these numbers, the FAA estimates the following costs to comply with this AD.

Modifying the electrical hoist control power supply takes about 4 work-hours and parts cost about \$10, for an estimated cost of \$350 per helicopter and \$1,750 for the affected U.S. fleet.

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. Subtitle VII: Aviation Programs, describes in more detail the scope of the Agency's authority.

The FAA is issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section 44701: General requirements. Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds