

credit.<sup>17</sup> Nevertheless, FHFA will consult with and refer matters to the Attorney General and coordinate with the Department of Justice on enforcement of fair lending matters as appropriate.

FHFA will consider whether the regulated entity has conducted any self-evaluations or undertaken any corrective actions when making supervisory and enforcement decisions. FHFA will view responsible business practices such as self-testing, implementation of management controls, and voluntary remedial action favorably when making fair lending supervisory and enforcement determinations. In particular, FHFA commits to taking into consideration an entity's cooperation and candor during examinations and monitoring. Regulated entities are not required to self-report potential violations of fair lending laws. However, self-reporting of violations of fair lending laws will be viewed favorably by FHFA as it exercises its discretion. FHFA also considers the number and duration of violations identified, the nature of the evidence of discrimination (*i.e.*, overt discrimination, disparate treatment, or disparate impact), the pervasiveness of the discrimination, the presence and effectiveness of any anti-discrimination policies, any history of discriminatory conduct, any corrective measures implemented or proposed by the regulated entity, and any other factors for determining the appropriateness of any potential action.

#### Consideration of Differences Between the Banks and the Enterprises

FHFA recognizes the important distinctions between the two types of regulated entities, the Enterprises and the Banks. In drafting this Policy Statement, FHFA has considered the differences between the Enterprises and the Banks with respect to the Banks' cooperative ownership structure, mission of providing liquidity to members, affordable housing and community development mission, capital structure, and joint and several liability, as well as other applicable differences.<sup>18</sup>

#### Sandra L. Thompson,

Acting Director, Federal Housing Finance Agency.

[FR Doc. 2021-14438 Filed 7-8-21; 8:45 am]

BILLING CODE 8070-01-P

## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 39

[Docket No. FAA-2021-0542; Project Identifier MCAI-2021-00117-R; Amendment 39-21641; AD 2021-14-14]

RIN 2120-AA64

#### Airworthiness Directives; Leonardo S.p.a. Helicopters

**AGENCY:** Federal Aviation Administration (FAA), DOT.

**ACTION:** Final rule; request for comments.

**SUMMARY:** The FAA is adopting a new airworthiness directive (AD) for certain Leonardo S.p.a. Model AW119 MKII helicopters. This AD was prompted by reports of detected smoke and burning smell during flight, caused by chafing of electrical wiring. This AD requires an inspection of the instrument panel electrical wiring, corrective actions if necessary, a modification of the wiring installation, and, for certain helicopters, an additional modification of the wiring installation, as specified in a European Union Aviation Safety Agency (EASA) AD, which is incorporated by reference. The FAA is issuing this AD to address the unsafe condition on these products.

**DATES:** This AD becomes effective July 26, 2021.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in this AD as of July 26, 2021.

The FAA must receive comments on this AD by August 23, 2021.

**ADDRESSES:** You may send comments, using the procedures found in 14 CFR 11.43 and 11.45, by any of the following methods:

- *Federal eRulemaking Portal:* Go to <https://www.regulations.gov>. Follow the instructions for submitting comments.

- *Fax:* 202-493-2251.

- *Mail:* U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE, Washington, DC 20590.

- *Hand Delivery:* Deliver to Mail address above between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

For material incorporated by reference (IBR) in this AD, contact the EASA, Konrad-Adenauer-Ufer 3, 50668 Cologne, Germany; phone: +49 221 8999 000; email: [ADs@easa.europa.eu](mailto:ADs@easa.europa.eu); internet: [www.easa.europa.eu](http://www.easa.europa.eu). You may find this material on the EASA website at <https://ad.easa.europa.eu>. You may

view this material at the FAA, Office of the Regional Counsel, Southwest Region, 10101 Hillwood Pkwy., Room 6N-321, Fort Worth, TX 76177. For information on the availability of this material at the FAA, call 817-222-5110. It is also available in the AD docket on the internet at <https://www.regulations.gov> by searching for and locating Docket No. FAA-2021-0542.

#### Examining the AD Docket

You may examine the AD docket at <https://www.regulations.gov> by searching for and locating Docket No. FAA-2021-0542; or in person at Docket Operations between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this AD, any comments received, and other information. The street address for Docket Operations is listed above.

**FOR FURTHER INFORMATION CONTACT:** Hal Jensen, Aerospace Engineer, Operational Safety Branch, Compliance & Airworthiness Division, FAA, 950 L'Enfant Plaza N SW, Washington, DC 20024; telephone (202) 267-9167; email [hal.jensen@faa.gov](mailto:hal.jensen@faa.gov).

#### SUPPLEMENTARY INFORMATION:

##### Background

The EASA, which is the Technical Agent for the Member States of the European Union, has issued EASA AD 2021-0040, dated January 27, 2021 (EASA AD 2021-0040) (also referred to as the Mandatory Continuing Airworthiness Information, or the MCAI), to correct an unsafe condition for certain Leonardo S.p.a. Model AW119 MKII helicopters.

This AD was prompted by reports of detected smoke and burning smell during flight, caused by chafing of electrical wiring. The FAA is issuing this AD to address detected smoke, burning smell during flight, and chafing of electrical wiring, which could lead to further occurrences of smoke in the cabin, or loss of function of avionics equipment, and possibly result in reduced control of the helicopter. See the MCAI for additional background information.

#### Related Service Information Under 14 CFR Part 51

EASA AD 2021-0040 specifies procedures for an inspection of the instrument panel electrical wiring for defects (including wire chafing; pinched, broken, or severely bent wires; deteriorated, cracked or missing wire shielding or insulation; and loose, corroded, or broken wire connectors), corrective actions (repair or replacement of the wiring and a pin to pin continuity

<sup>17</sup> 15 U.S.C. 1691e(g).

<sup>18</sup> 12 U.S.C. 4513(f).

check on the repaired wiring) if necessary, a modification of the wiring installation, and, for certain helicopters, an additional modification of the wiring installation. This material is reasonably available because the interested parties have access to it through their normal course of business or by the means identified in the **ADDRESSES** section.

#### **FAA's Determination**

This product has been approved by the aviation authority of another country, and is approved for operation in the United States. Pursuant to the FAA's bilateral agreement with the State of Design Authority, the FAA has been notified of the unsafe condition described in the MCAI referenced above. The FAA is issuing this AD after evaluating all pertinent information and determining that the unsafe condition exists and is likely to exist or develop on other products of the same type design.

#### **Requirements of This AD**

This AD requires accomplishing the actions specified in EASA AD 2021-0040, described previously, as incorporated by reference, except for any differences identified as exceptions in the regulatory text of this AD.

#### **Explanation of Required Compliance Information**

In the FAA's ongoing efforts to improve the efficiency of the AD process, the FAA initially worked with Airbus and EASA to develop a process to use certain EASA ADs as the primary source of information for compliance with requirements for corresponding FAA ADs. The FAA has since coordinated with other manufacturers and civil aviation authorities (CAAs) to use this process. As a result, EASA AD 2021-0040 is incorporated by reference in the FAA final rule. This AD will, therefore, require compliance with EASA AD 2021-0040 in its entirety, through that incorporation, except for any differences identified as exceptions in the regulatory text of this AD. Using common terms that are the same as the heading of a particular section in the EASA AD does not mean that operators need comply only with that section. For example, where the AD requirement refers to "all required actions and compliance times," compliance with this AD requirement is not limited to the section titled "Required Action(s) and Compliance Time(s)" in the EASA AD. Service information specified in EASA AD 2021-0040 that is required for

compliance with EASA AD 2021-0040 is available on the internet at <https://www.regulations.gov> by searching for and locating Docket No. FAA-2021-0542.

#### **FAA's Justification and Determination of the Effective Date**

Section 553(b)(3)(B) of the Administrative Procedure Act (APA) (5 U.S.C. 551 *et seq.*) authorizes agencies to dispense with notice and comment procedures for rules when the agency, for "good cause" finds that those procedures are "impracticable, unnecessary, or contrary to the public interest." Under this section, an agency, upon finding good cause, may issue a final rule without seeking comment prior to issuance. Further, section 553(d) of the APA authorizes agencies to make rules effective in less than thirty days, upon a finding of good cause.

An unsafe condition exists that requires the immediate adoption of this AD without providing an opportunity for public comments prior to adoption. The FAA has found that the risk to the flying public justifies foregoing notice and comment prior to adoption of this rule because of detected smoke, burning smell during flight, and chafing of electrical wiring, which could lead to further occurrences of smoke in the cabin, or loss of function of avionics equipment, and possibly result in reduced control of the helicopter. In addition, the compliance time for the inspection of the instrument panel electrical wiring is within 25 hours time-in-service or 3 months, whichever occurs first after the effective date of this AD, which is shorter than the time necessary for the public to comment and for publication of the final rule. Therefore, notice and opportunity for prior public comment are impracticable and contrary to public interest pursuant to 5 U.S.C. 553(b)(3)(B). In addition, the FAA finds that good cause exists pursuant to 5 U.S.C. 553(d) for making this amendment effective in less than 30 days, for the same reasons the FAA found good cause to forego notice and comment.

#### **Comments Invited**

The FAA invites you to send any written relevant data, views, or arguments about this AD. Send your comments to an address listed under **ADDRESSES**. Include "Docket No. FAA-2021-0542; Project Identifier MCAI-2021-00117-R" at the beginning of your comments. The most helpful comments reference a specific portion of the AD,

explain the reason for any recommended change, and include supporting data. The FAA will consider all comments received by the closing date and may amend this AD because of those comments.

Except for Confidential Business Information (CBI) as described in the following paragraph, and other information as described in 14 CFR 11.35, the FAA will post all comments received, without change, to <https://www.regulations.gov>, including any personal information you provide. The agency will also post a report summarizing each substantive verbal contact received about this AD.

#### **Confidential Business Information**

CBI is commercial or financial information that is both customarily and actually treated as private by its owner. Under the Freedom of Information Act (FOIA) (5 U.S.C. 552), CBI is exempt from public disclosure. If your comments responsive to this AD contain commercial or financial information that is customarily treated as private, that you actually treat as private, and that is relevant or responsive to this AD, it is important that you clearly designate the submitted comments as CBI. Please mark each page of your submission containing CBI as "PROPIN." The FAA will treat such marked submissions as confidential under the FOIA, and they will not be placed in the public docket of this AD. Submissions containing CBI should be sent to Hal Jensen, Aerospace Engineer, Operational Safety Branch, Compliance & Airworthiness Division, FAA, 950 L'Enfant Plaza N SW, Washington, DC 20024; telephone (202) 267-9167; email [hal.jensen@faa.gov](mailto:hal.jensen@faa.gov). Any commentary that the FAA receives that is not specifically designated as CBI will be placed in the public docket for this rulemaking.

#### **Regulatory Flexibility Act (RFA)**

The requirements of the RFA do not apply when an agency finds good cause pursuant to 5 U.S.C. 553 to adopt a rule without prior notice and comment. Because the FAA has determined that it has good cause to adopt this rule without notice and comment, RFA analysis is not required.

#### **Costs of Compliance**

The FAA estimates that this AD affects 10 helicopters of U.S. registry. The FAA estimates the following costs to comply with this AD:

ESTIMATED COSTS FOR REQUIRED ACTIONS

Labor cost	Parts cost	Cost per product	Cost on U.S. operators
Up to 11 work-hours × \$85 per hour = \$935 .....	\$73	Up to \$1,008 .....	Up to \$10,080.

The FAA has received no definitive data on which to base the cost estimates for the on-condition actions specified in this AD.

According to the manufacturer, some or all of the costs of this AD may be covered under warranty, thereby reducing the cost impact on affected individuals. The FAA does not control warranty coverage for affected individuals. As a result, the FAA has included all known costs in the cost estimate.

**Authority for This Rulemaking**

Title 49 of the United States Code specifies the FAA’s authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. Subtitle VII: Aviation Programs, describes in more detail the scope of the Agency’s authority.

The FAA is issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section 44701: General requirements. Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

**Regulatory Findings**

The FAA determined that this AD would not have federalism implications under Executive Order 13132. This AD would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify this regulation:

- (1) Is not a “significant regulatory action” under Executive Order 12866, and
- (2) Will not affect intrastate aviation in Alaska.

**List of Subjects in 14 CFR Part 39**

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

**The Amendment**

Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

**PART 39—AIRWORTHINESS DIRECTIVES**

- 1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

**§ 39.13 [Amended]**

- 2. The FAA amends § 39.13 by adding the following new airworthiness directive:

**2021–14–14 Leonardo S.p.a.:** Amendment 39–21641; Docket No. FAA–2021–0542; Project Identifier MCAI–2021–00117–R.

**(a) Effective Date**

This airworthiness directive (AD) becomes effective July 26, 2021.

**(b) Affected ADs**

None.

**(c) Applicability**

This AD applies to Leonardo S.p.a. Model AW119 MKII helicopters, certificated in any category, as identified in European Union Aviation Safety Agency (EASA) AD 2021–0040, dated January 27, 2021 (EASA AD 2021–0040).

**(d) Subject**

Joint Aircraft System Component (JASC) Code 3197, Instrument System Wiring.

**(e) Unsafe Condition**

This AD was prompted by reports of detected smoke and burning smell during flight, caused by chafing of electrical wiring. The FAA is issuing this AD to address detected smoke, burning smell during flight, and chafing of electrical wiring, which could lead to further occurrences of smoke in the cabin, or loss of function of avionics equipment, and possibly result in reduced control of the helicopter.

**(f) Compliance**

Comply with this AD within the compliance times specified, unless already done.

**(g) Requirements**

Except as specified in paragraph (h) of this AD: Comply with all required actions and compliance times specified in, and in accordance with, EASA AD 2021–0040.

**(h) Exceptions to EASA AD 2021–0040**

(1) Where EASA AD 2021–0040 refers to its effective date, this AD requires using the effective date of this AD.

(2) The “Remarks” section of EASA AD 2021–0040 does not apply to this AD.

(3) Where EASA AD 2021–0040 refers to flight hours (FH), this AD requires using hours time-in-service.

(4) Where paragraph (2) of EASA AD 2021–0040 specifies actions if “any defect is found,” for this AD a “defect” includes wire chafing; pinched, broken, or severely bent wires; deteriorated, cracked or missing wire shielding or insulation; and loose, corroded, or broken wire connectors.

(5) Where paragraph (1) of EASA AD 2021–0040 refers to “the instructions of Part I of the SB,” for this AD, use “the instructions of Part I, paragraph 3. of the Accomplishment Instructions of the SB.”

(6) Where paragraph (2) of EASA AD 2021–0040 refers to “the instructions of Part I of the SB,” for this AD, use “the instructions of Part I, paragraphs 4. and 5. of the Accomplishment Instructions of the SB.”

(7) Where paragraph (4) of EASA AD 2021–0040 refers to “the instructions of Part II of the SB,” for this AD, use “the instructions of Part II, paragraph 1. of the Accomplishment Instructions of the SB.”

(8) Where the service information referenced in EASA AD 2021–0040 specifies to contact Leonardo if the cargo hoist indicator cable is damaged, this AD requires repair or replacement using a method approved by the Manager, International Validation Branch, FAA. The Manager’s approval letter must specifically refer to this AD.

**(i) Special Flight Permit**

Special flight permits may be issued in accordance with 14 CFR 21.197 and 21.199 to operate the helicopter to a location where the helicopter can be modified (if the operator elects to do so), provided no passengers are onboard.

**(j) Alternative Methods of Compliance (AMOCs)**

(1) The Manager, International Validation Branch, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the manager of the International Validation Branch, send it to the attention of the person identified in paragraph (k) of this AD. Information may be emailed to: 9-AVS-AIR-730-AMOC@faa.gov.

(2) Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager

of the local flight standards district office/  
certificate holding district office.

#### (k) Related Information

For more information about this AD, contact Hal Jensen, Aerospace Engineer, Operational Safety Branch, Compliance & Airworthiness Division, FAA, 950 L'Enfant Plaza N SW, Washington, DC 20024; telephone (202) 267-9167; email [hal.jensen@faa.gov](mailto:hal.jensen@faa.gov).

#### (l) Material Incorporated by Reference

(1) The Director of the Federal Register approved the incorporation by reference (IBR) of the service information listed in this paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) You must use this service information as applicable to do the actions required by this AD, unless this AD specifies otherwise.

(i) European Union Aviation Safety Agency (EASA) AD 2021-0040, dated January 27, 2021.

(ii) [Reserved]

(3) For EASA AD 2021-0040, contact the EASA, Konrad-Adenauer-Ufer 3, 50668 Cologne, Germany; phone: +49 221 8999 000; email: [ADs@easa.europa.eu](mailto:ADs@easa.europa.eu); internet: [www.easa.europa.eu](http://www.easa.europa.eu). You may find this EASA AD on the EASA website at <https://ad.easa.europa.eu>.

(4) You may view this service information at the FAA, Office of the Regional Counsel, Southwest Region, 10101 Hillwood Pkwy., Room 6N-321, Fort Worth, TX 76177. For information on the availability of this material at the FAA, call 817-222-5110. This material may be found in the AD docket on the internet at <https://www.regulations.gov> by searching for and locating Docket No. FAA-2021-0542.

(5) You may view this material that is incorporated by reference at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, email [fedreg.legal@nara.gov](mailto:fedreg.legal@nara.gov), or go to <https://www.archives.gov/federal-register/cfr/ibr-locations.html>.

Issued on July 2, 2021.

#### Gaetano A. Sciortino,

Deputy Director for Strategic Initiatives,  
Compliance & Airworthiness Division,  
Airframe Certification Service.

[FR Doc. 2021-14690 Filed 7-7-21; 11:15 am]

BILLING CODE 4910-13-P

## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 39

[Docket No. FAA-2020-1180; Project Identifier MCAI-2020-00517-E; Amendment 39-21608; AD 2021-13-03]

RIN 2120-AA64

#### Airworthiness Directives; Safran Helicopter Engines, S.A. (Type Certificate Previously Held by Turbomeca, S.A.) Turboshaft Engines

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final rule.

**SUMMARY:** The FAA is adopting a new airworthiness directive (AD) for certain Safran Helicopter Engines, S.A. (Safran Helicopter Engines) Arriel 2B, 2B1, 2C, 2C1, 2C2, 2S1 and 2S2 model turboshift engines. This AD was prompted by reports of non-conforming fuel filter pre-blockage pressure switches. This AD requires repetitive visual inspections of the fuel filter by-pass indicator pop-up, a one-time operational test of the fuel filter pre-blockage pressure switch and, depending on the findings, replacement of the fuel filter pre-blockage pressure switch with a part eligible for installation. The FAA is issuing this AD to address the unsafe condition on these products.

**DATES:** This AD is effective August 13, 2021.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in this AD as of August 13, 2021.

**ADDRESSES:** For service information identified in this final rule, contact Safran Helicopter Engines, S.A., Avenue du 1er Mai, Tarnos, France; phone: +33 (0) 5 59 74 45 11. You may view this service information at the FAA, Airworthiness Products Section, Operational Safety Branch, 1200 District Avenue, Burlington, MA 01803. For information on the availability of this material at the FAA, call (781) 238-7759. It is also available at <https://www.regulations.gov> by searching for and locating Docket No. FAA-2020-1180.

#### Examining the AD Docket

You may examine the AD docket at <https://www.regulations.gov> by searching for and locating Docket No. FAA-2020-1180; or in person at Docket Operations between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this final rule, the mandatory continuing

airworthiness information (MCAI), any comments received, and other information. The address for Docket Operations is U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE, Washington, DC 20590.

#### FOR FURTHER INFORMATION CONTACT:

Wego Wang, Aviation Safety Engineer, ECO Branch, FAA, 1200 District Avenue, Burlington, MA 01803; phone: (781) 238-7134; fax: (781) 238-7199; email: [wego.wang@faa.gov](mailto:wego.wang@faa.gov).

#### SUPPLEMENTARY INFORMATION:

#### Background

The FAA issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 by adding an AD that would apply to certain Safran Helicopter Engines Arriel 2B, 2B1, 2C, 2C1, 2C2, 2S1 and 2S2 model turboshift engines. The NPRM published in the **Federal Register** on February 22, 2021 (86 FR 10501). The NPRM was prompted by reports of non-conforming fuel filter pre-blockage pressure switches. In the NPRM, the FAA proposed to require repetitive visual inspections of the fuel filter by-pass indicator pop-up, a one-time operational test of the fuel filter pre-blockage pressure switch and, depending on the findings, replacement of the fuel filter pre-blockage pressure switch with a part eligible for installation. The FAA is issuing this AD to address the unsafe condition on these products.

The European Union Aviation Safety Agency (EASA), which is the Technical Agent for the Member States of the European Community, has issued EASA AD 2019-0180, dated July 25, 2019 (referred to after this as “the MCAI”), to address the unsafe condition on these products. The MCAI states:

Occurrences have been reported of non-conforming fuel filter pre-blockage pressure switches, manufactured before December 2016. The non-conformity of the fuel filter pre-blockage pressure switch can cause its non-activation in case of fuel system contamination, with consequent opening of the by-pass without indication in the cockpit.

This condition, if not detected and corrected, and in case of fuel contamination, could lead to an uncommanded in-flight shut-down, possibly resulting in an emergency autorotation landing on a single engine helicopter, or to a double uncommanded in-flight shut-down on a twin engine helicopter.

To address this potential unsafe condition, SAFRAN issued the MSB, providing inspection instructions.

For the reasons described above, this [EASA AD] requires repetitive daily visual checks of the fuel filter by-pass indicator pop-up. This [EASA] AD also requires a one-