§ 39.13 [Amended]

2. The FAA amends § 39.13 by:

a. Removing Airworthiness Directive (AD) 2015–25–04, Amendment 39–18342 (80 FR 76381, December 9, 2015); and

b. Adding the following new AD:


(a) Applicability

This airworthiness directive (AD) applies to Leonardo S.p.a. (Type Certificate previously held by Agusta S.p.A.) Model A109A and A109A II helicopters, certificated in any category, with a slider assembly pitch control (slider) part number (P/N) 109–0130–11–7 installed, except those sliders marked with an “R” after the serial number.

(b) Unsafe Condition

This AD defines the unsafe condition as play on a slider. This condition could result in loss of tail rotor pitch control and consequently loss of helicopter control.

(c) Affected ADs


(d) Comments Due Date

The FAA must receive comments by April 22, 2021.

(e) Compliance

You are responsible for performing each action required by this AD within the specified compliance time unless it has already been accomplished prior to that time.

(f) Required Actions

(1) Within 25 hours time-in-service (TIS) after the effective date of this AD, and thereafter at intervals not to exceed 25 hours TIS, inspect the slider for play. If there is play greater than 2.3 millimeters (0.09 inch), before further flight, replace the slider with a slider P/N 109–0130–11–7 with suffix “R” marked after the serial number.

(2) Within 800 hours TIS after the effective date of this AD, if not previously required per paragraph (f)(1) of this AD, replace slider P/N 109–0130–11–7 with slider P/N 109–0130–11–7 with suffix “R” marked after the serial number.

(3) Installing slider P/N 109–0130–11–7 with suffix “R” marked after the serial number is a terminating action for the repetitive inspections required by paragraph (f)(1) of this AD.

(4) As of the effective date of this AD, do not install slider P/N 109–0130–11–7 on any helicopter unless the slider is marked with suffix “R” after the serial number.

(g) Alternative Methods of Compliance (AMOCs)

(1) The Manager, Strategic Policy Rotorcraft Unit, telephone 817–222–5110; email 9-ASW-FTW-AMOC-Requests@faa.gov.

(2) For operations conducted under a 14 CFR part 119 operating certificate or under 14 CFR part 91, subpart K, the FAA suggests that you notify your principal inspector, or lacking a principal inspector, the manager of the local flight standards district office or certificate holding district office before operating any aircraft complying with this AD through an AMOC.

(h) Additional Information


(i) Subject

Joint Aircraft Service Component (JASC) Code: 6720, Tail Rotor Control System.

Issued on February 4, 2021.

Gaetano A. Sciortino,
Deputy Director for Strategic Initiatives, Compliance & Airworthiness Division, Aircraft Certification Service.

[FR Doc. 2021–04232 Filed 3–5–21; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA–2021–0104; Project Identifier MCAI–2020–00477–R]

RIN 2120–AA64

Airworthiness Directives; Airbus Helicopters Deutschland GmbH Helicopters

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: The FAA proposes to adopt a new airworthiness directive (AD) for certain Airbus Helicopters Deutschland GmbH Model MBB–BK117 D–2 helicopters. This proposed AD was prompted by reports that collective lever switch units having certain part numbers did not have retaining rings installed in the cable cut switch guard. This proposed AD would require inspecting certain collective lever switch units for discrepancies (missing retaining rings, incorrectly installed retaining rings, and a missing axis in the cable cut switch guard), doing all applicable corrective actions, and marking affected parts, as specified in a European Union Aviation Safety Agency (EASA) AD, which is proposed for incorporation by reference (IBR). The FAA is proposing this AD to address the unsafe condition on these products.

DATES: The FAA must receive comments on this proposed AD by April 22, 2021.

ADDRESSES: You may send comments, using the procedures found in 14 CFR 11.35 and 11.45, by any of the following methods:

• Federal eRulemaking Portal: Go to https://www.regulations.gov. Follow the instructions for submitting comments.

• Fax: 202–493–2251.


Hand Delivery: Deliver to Mail address above between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

For material that is proposed for IBR in this AD, contact the EASA, Konrad-Adenauer-Ufer 3, 50668 Cologne, Germany; telephone +49 221 8999 000; email ADs@easa.europa.eu; internet www.easa.europa.eu. You may find this material on the EASA website at https://ad.easa.europa.eu. You may view this material at the FAA, Office of the Regional Counsel, Southwest Region, 10101 Hillwood Pkwy., Room 6N–321, Fort Worth, TX 76177. For information on the availability of this material at the FAA, call 817–222–5110. It is also available in the AD docket on the internet at https://www.regulations.gov by searching for and locating Docket No. FAA–2021–0104.

Examining the AD Docket

You may examine the AD docket on the internet at https://www.regulations.gov by searching for and locating Docket No. FAA–2021–0104; or in person at Docket Operations between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this NPRM, any comments received, and other information. The street address for Docket Operations is listed above. Comments will be available in the AD docket shortly after receipt.

FOR FURTHER INFORMATION CONTACT: Hal Jensen, Aerospace Engineer, Operational Safety Branch, FAA, 470 L’Enfant Plaza SW, Washington, DC 20024; telephone

Comments Invited

The FAA invites you to send any written relevant data, views, or arguments about this proposal. Send your comments to an address listed under ADDRESSES. Include “Docket No. FAA–2021–0104; Project Identifier MCAI–2020–00477–R” at the beginning of your comments. The most helpful comments reference a specific portion of the proposal, explain the reason for any recommended change, and include supporting data. The FAA will consider all comments received by the closing date and may amend this proposal because of those comments.

Except for Confidential Business Information (CBI) as described in the following paragraph, and other information as described in 14 CFR 11.35, the FAA will post all comments received, without change, to https://www.regulations.gov, including any personal information you provide. The agency will also post a report summarizing each substantive verbal contact received about this proposal.

Confidential Business Information

CBI is commercial or financial information that is both customarily and actually treated as private by its owner. Under the Freedom of Information Act (FOIA) (5 U.S.C. 552), CBI is exempt from public disclosure. If your comments responsive to this NPRM contain commercial or financial information that is customarily treated as private, that you actually treat as private, and that is relevant or responsive to this NPRM, it is important that you clearly designate the submitted comments as CBI. Please mark each page of your submission containing CBI as “PROPIN.” The FAA will treat such marked submissions as confidential under the FOIA, and they will not be placed in the public docket of this NPRM. Submissions containing CBI should be sent to Hal Jensen, Aerospace Engineer, Operational Safety Branch, FAA, 470 L’Enfant Plaza SW, Washington, DC 20024; telephone 202–267–9167; email hal.jensen@faa.gov.

Any commentary that the FAA receives that is not specifically designated as CBI will be placed in the public docket for this rulemaking.

Discussion

The EASA, which is the Technical Agent for the Member States of the European Union, has issued EASA AD 2020–0084, dated April 3, 2020 (EASA AD 2020–0084) (also referred to as the Mandatory Continuing Airworthiness Information, or the MCAI), to correct an unsafe condition for all Airbus Helicopters Deutschland GmbH Model MBB–BK117 D–2 helicopters. Although EASA AD 2020–0084 applies to all Model MBB–BK117 D–2 helicopters, this proposed AD would apply to helicopters with an affected part installed instead.

This proposed AD was prompted by reports that collective lever switch units having certain part numbers did not have retaining rings installed in the cable cut switch guard. The cable cut switch guard has an axis that holds, and allows the guard to turn over, the cable cut switch. This axis is secured with two retaining rings and if both retaining rings are missing, the axis can move out. The FAA is proposing this AD to address this condition, which could cause inadvertent activation of the rescue hoist cable cut function, resulting in personal injury. See the MCAI for additional background information.

Related Service Information Under 1 CFR Part 51

EASA AD 2020–0084 describes procedures for inspecting collective lever switch units having certain part numbers for discrepancies, applicable corrective actions, and marking affected parts. Discrepancies include missing retaining rings, incorrectly installed retaining rings, and a missing axis in the cable cut switch guard. Corrective actions include installing missing retaining rings, adjusting retaining rings that are installed incorrectly, and installing an axis in the cable cut switch guard. EASA AD 2020–0084 also specifies that an affected part can be installed on any helicopter, provided it has been marked. This material is reasonably available because the interested parties have access to it through their normal course of business or by the means identified in the ADDRESSES section.

FAA’s Determination and Requirements of This Proposed AD

This product has been approved by the aviation authority of another country, and is approved for operation in the United States. Pursuant to the bilateral agreement with the State of Design Authority, the FAA has been notified of the unsafe condition described in the MCAI referenced above. The FAA is proposing this AD after evaluating all the relevant information and determining the unsafe condition described previously is likely to exist or develop in other products of the same type design.

Proposed AD Requirements

This proposed AD would require accomplishing the actions specified in EASA AD 2020–0084, described previously, as incorporated by reference, except for any differences identified as exceptions in the regulatory text of this proposed AD.

Explanation of Required Compliance Information

In the FAA’s ongoing efforts to improve the efficiency of the AD process, the FAA initially worked with Airbus and EASA to develop a process to use certain EASA ADs as the primary source of information for compliance with requirements for corresponding FAA ADs. The FAA has since coordinated with other manufacturers and civil aviation authorities (CAAs) to use this process. As a result, EASA AD 2020–0084 will be incorporated by reference in the FAA final rule. This proposed AD would, therefore, require compliance with EASA AD 2020–0084 in its entirety, through that incorporation, except for any differences identified as exceptions in the regulatory text of this proposed AD.

Using common terms that are the same as the heading of a particular section in the EASA AD does not mean that operators need comply only with that section. For example, where the AD requirement refers to “all required actions and compliance times,” compliance with this AD requirement is not limited to the section titled “Required Action(s) and Compliance Time(s)” in the EASA AD. Service information specified in EASA AD 2020–0084 that is required for compliance with EASA AD 2020–0084 will be available on the internet at https://www.regulations.gov by searching for and locating Docket No. FAA–2021–0104 after the FAA final rule is published.

Costs of Compliance

The FAA estimates that this proposed AD affects 30 helicopters of U.S. registry. The FAA estimates the following costs to comply with this proposed AD:
The FAA estimates the following costs to do any necessary on-condition actions that would be required based on the results of any required actions. The FAA has no way of determining the number of helicopters that might need these on-condition actions:

### ESTIMATED COSTS FOR REQUIRED ACTIONS

<table>
<thead>
<tr>
<th>Labor cost</th>
<th>Parts cost</th>
<th>Cost per product</th>
<th>Cost on U.S. operators</th>
</tr>
</thead>
<tbody>
<tr>
<td>0.50 work-hour × $85 per hour = $42.50</td>
<td>$0</td>
<td>$42.50</td>
<td>$1,275</td>
</tr>
</tbody>
</table>

According to the manufacturer, some or all of the costs of this proposed AD may be covered under warranty, thereby reducing the cost impact on affected operators. The FAA does not control warranty coverage for affected operators. As a result, the FAA has included all known costs in the cost estimate.

### Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA’s authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. Subtitle VII: Aviation Programs, describes in more detail the scope of the Agency’s authority.

The FAA is issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section 44701: General requirements. Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

### Regulatory Findings

The FAA determined that this proposed AD would not have federalism implications under Executive Order 13132. This proposed AD would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify this proposed regulation:

1. Is not a “significant regulatory action” under Executive Order 12866,
2. Will not affect intrastate aviation in Alaska, and
3. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

### List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

### The Proposed Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA proposes to amend 14 CFR part 39 as follows:

### PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

   Authority: 49 U.S.C. 106(g), 40113, 44701.

### § 39.13 [Amended]

2. The FAA amends § 39.13 by adding the following new airworthiness directive:

   **Airbus Helicopters Deutschland GmbH:**

   **(a) Comments Due Date**
   - The FAA must receive comments by April 22, 2021.

   **(b) Affected Airworthiness Directives (ADs)**
   - None.

   **(c) Applicability**
   - This AD applies to Airbus Helicopters Deutschland GmbH Model MBB–BK117 D–2 helicopters, certificated in any category, all serial numbers, having an affected part defined in European Union Aviation Safety Agency (EASA) AD 2020–0084, dated April 3, 2020 (EASA AD 2020–0084).

   **(d) Subject**

   **(e) Reason**
   - This AD was prompted by reports that collective lever switch units having certain part numbers did not have retaining rings installed in the cable cut switch guard. The cable cut switch guard has an axis that holds, and allows the guard to turn over, the cable cut switch. This axis is secured with two retaining rings and if both retaining rings are missing, the axis can move out. The FAA is issuing this AD to address this condition, which could cause inadvertent activation of the rescue hoist cable cut function, resulting in personal injury.

   **(f) Compliance**
   - Comply with this AD within the compliance times specified, unless already done.

   **(g) Requirements**
   - Comply with all required actions and compliance times specified in, and in accordance with, EASA AD 2020–0084.

   **(h) Exceptions to EASA AD 2020–0084**
   1. Where EASA AD 2020–0084 refers to its effective date, this AD requires using the effective date of this AD.
   2. The “Remarks” section of EASA AD 2020–0084 does not apply to this AD.

   **(i) Special Flight Permit**
   - Special flight permits may be issued in accordance with 14 CFR 21.197 and 21.199 to operate the helicopter to a location where the helicopter can be modified (if the operator elects to do so), provided the helicopter is not used for hoist operations.

   **(j) Alternative Methods of Compliance (AMOCs)**
   1. The Manager, Strategic Policy Rotorcraft Section, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the manager of the Strategic Policy
This proposed AD would expand the applicability to include Model EC135P3 and EC135T3 helicopters. This proposed AD would also revise the life limits for certain parts and propose to remove each part that has reached or exceeded its life limit. The actions of this proposed AD are intended to address an unsafe condition on these products.

DATES: The FAA must receive comments on this proposed AD by April 22, 2021.

ADDRESSES: You may send comments by any of the following methods:

- Federal eRulemaking Docket: Go to https://www.regulations.gov. Follow the online instructions for sending your comments electronically.
- Hand Delivery: Deliver to the Dockets in Room W12–140, 1200 E. caramel. Street, Washington, DC 20590–0001. Comment pages are available for inspection only at that location.

The FAA invites you to send any written relevant data, views, or arguments about this proposal. Send your comments to an address listed under ADDRESSES. Include “Docket No. FAA–2019–0113; Product Identifier 2017–SW–140–AD” at the beginning of your comments. The most helpful comments reference a specific portion of the proposal, explain the reason for any recommended change, and include supporting data. The FAA will consider all comments received by the closing date and may amend this proposal because of those comments.

Confidential Business Information

CBI is commercial or financial information that is both customarily and actually treated as private by its owner. Under the Freedom of Information Act (FOIA) (5 U.S.C. 552), CBI is exempt from public disclosure. If your comments responsive to this NPRM contain commercial or financial information that is customarily treated as private, that you actually treat as private, and that is relevant or responsive to this NPRM, it is important that you clearly designate the submitted comments as CBI. Please mark each page of your submission containing CBI as ”PROPIN.” The FAA will treat such marked submissions as confidential under the FOIA, and they will not be placed in the public docket of this NPRM. Submissions containing CBI should be sent to Matt Fuller, AD Program Manager, Operational Safety Branch, Airworthiness Products Section, General Aviation & Rotorcraft Unit, 10101 Hillwood Pkwy., Fort Worth, TX 76177; telephone 817–222–5110; email matthew.fuller@faa.gov. Any commentary that the FAA receives which is not specifically designated as CBI will be placed in the public docket for this rulemaking.

Discussion

The FAA issued AD 2016–11–21, Amendment 2016–16, on June 6, 2016 (81 FR 36137, June 9, 2016) (AD 2016–11–21) for Airbus Helicopters Model EC135P1, EC135P2, EC135P2+, EC135T1,