List of Subjects in 14 CFR Part 39
Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment
Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

§ 39.13 [Amended]
■ 1. The authority citation for part 39 continues to read as follows:
Authority: 49 U.S.C. 106(g), 40113, 44701.
§ 39.13 [Amended]
■ 2. The FAA amends § 39.13 by adding the following new airworthiness directive:
(a) Effective Date
This airworthiness directive (AD) becomes effective November 18, 2020.
(b) Affected ADs
None.
(c) Applicability
This AD applies to all Airbus Helicopters Deutschland GmbH Model EC135P1, EC135P2, EC135P2+, EC135P3, EC135T1, EC135T2, EC135T2+, EC135T3, and EC635T2+ helicopters, certificated in any category.
(d) Subject
Joint Aircraft System Component (JASC) Codes 6500, Tail Rotor Drive.
(e) Reason
This AD was prompted by reports of improper heat treatment of titanium (Ti)-bolts installed on the forward and aft tail rotor drive shafts, resulting in a broken Ti-bolt. The FAA is issuing this AD to address improper heat treatment of Ti-bolts on the forward and aft tail rotor drive shafts, which could lead to rupture of a Ti-bolt installed in a critical location, possibly resulting in reduced control of the helicopter.
(f) Compliance
Comply with this AD within the compliance times specified, unless already done.
(g) Requirements
Except as specified in paragraph (h) of this AD: Comply with all required actions and compliance times specified in, and in accordance with, European Union Aviation Safety Agency (EASA) AD 2020–0099, dated May 5, 2020 (EASA AD 2020–0099).
(h) Exceptions to EASA AD 2020–0099
(1) Where EASA AD 2020–0099 refers to its effective date, this AD requires using the effective date of this AD.
(2) The “Remarks” section of EASA AD 2020–0099 does not apply to this AD.
(3) Although the service information referenced in EASA AD 2020–0099 specifies to discard certain parts, this AD does not include that requirement.
(i) Alternative Methods of Compliance (AMOCs)
The Manager, International Validation Branch, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the International Validation Branch, send it to the attention of the person identified in paragraph (j) of this AD. Information may be emailed to: 9-AVS-AIR-730-AMOC@faa.gov. Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/ certificate holding district office.
(j) Related Information
For more information about this AD, contact Kathleen Arrigotti, Aviation Safety Engineer, Large Aircraft Section, International Validation Branch, FAA, 2200 South 216th St., Des Moines, WA 98198; phone and fax: 206–231–3218; email: kathleen.arrigotti@faa.gov.
(k) Material Incorporated by Reference
(1) The Director of the Federal Register approved the incorporation by reference (IBR) of the service information listed in this paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.
(2) You must use this service information as applicable to do the actions required by this AD, unless this AD specifies otherwise.
(ii) [Reserved]
(3) For EASA AD 2020–0099, contact the EASA, Konrad-Adenauer-Ufer 3, 50668 Cologne, Germany; phone: +49 221 8999 1000; email: ADs@easa.europa.eu; internet: www.easa.europa.eu. You may find this EASA AD on the EASA website at https://ad.easa.europa.eu.
(4) You may view this material at the FAA, Office of the Regional Counsel, Southwest Region, 10101 Hillwood Pkwy., Room 6N–321, Fort Worth, TX. For information on the availability of this material at the FAA, call 206–231–3195. This material may be found in the AD docket on the internet at https://www.regulations.gov by searching for and locating Docket No. FAA–2020–0919.
(5) You may view this material that is incorporated by reference at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, email fedreg_legal@nara.gov, or go to: https://www.archives.gov/federal-register/cfr/ibr-locations.html.

Issued on October 13, 2020.
Lance T. Gant,
Director, Compliance & Airworthiness Division, Aircraft Certification Service.
[FR Doc. 2020–24263 Filed 11–2–20; 8:45 am]
BILLING CODE 4910–13–P

DEPARTMENT OF TRANSPORTATION
Federal Aviation Administration

14 CFR Part 39


RIN 2120–AA64
Airworthiness Directives; Polskie Zaklady Lotnicze Sp. z o.o Airplanes

AGENCY: Federal Aviation Administration (FAA), DOT.
ACTION: Final rule.

SUMMARY: The FAA is adopting a new airworthiness directive (AD) for certain Polskie Zaklady Lotnicze Sp. z o.o Model PZL M28 05 airplanes. This AD results from mandatory continuing airworthiness information (MCAI) originated by an aviation authority of another country to identify and correct an unsafe condition on an aviation product. The MCAI describes the unsafe condition as defective thermo-shrinkable tubes installed on the electrical harnesses located in the fuel tanks. This AD requires a one-time inspection of the electrical harnesses located in the fuel tanks and, depending on findings, replacement of the affected harness. The FAA is issuing this AD to address the unsafe condition on these products.

DATES: This AD is effective December 8, 2020.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in this AD as of December 8, 2020.

ADDRESSES: For service information identified in this final rule, contact Polskie Zaklady Lotnicze Sp. z o.o., Wojska Polskiego 3, 39–300 Mielec, Poland, telephone: +48 17 743 1901, email: pzl.lm@lmco.com, internet: http://www.pzlkielec.pl/. You may view this service information at the FAA, Airworthiness Products Section, Operational Safety Branch, 901 Locust, Kansas City, Missouri 64106. For information on the availability of this material at the FAA, call (816) 329–4148. It is also available on the internet at https://www.regulations.gov by searching for and locating Docket No. FAA–2020–0473.
Examining the AD Docket

You may examine the AD docket on the internet at https://www.regulations.gov by searching for and locating Docket No. FAA–2020–0473; or in person at Docket Operations between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this final rule, the MCAI, any comments received, and other information. The address for Docket Operations is U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue SE, Washington, DC 20590.

For Further Information Contact:
Doug Rudolph, Aerospace Engineer, FAA, General Aviation & Rotorcraft Section, International Validation Branch, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone: (816) 329–4059; fax: (816) 329–4090; email: doug.rudolph@faa.gov.

Supplementary Information:

Discussion

The FAA issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 by adding an AD that would apply to certain serial-numbered Polskie Zaklady Lotnicze Sp. z o.o. Model PZL M28 05 airplanes. The NPRM published in the Federal Register on May 14, 2020 (85 FR 28893). The NPRM proposed to correct an unsafe condition for the specified products and was based on MCAI originated by the European Union Aviation Safety Agency (EASA), which is the Technical Agent for the Member States of the European Union. EASA issued AD No. 2018–0242, dated October 8, 2018 (referred to after this as “the MCAI”), which states:

During accomplishment of maintenance on an M28 05 military version airplane, torn pieces of thermo-shrinkable tubes were found in the header section of the main fuel tank. These tubes are installed on electrical harnesses located in the fuel tanks and serve as marking and protection devices against mechanical damage during manufacturing and servicing. Pieces of these tubes may travel with the fuel flow and may block the jet pump or reduce its performance, particularly in the centre-wing fuel tank, in which the jet pump is the only way of further transfer of fuel to the engine. Subsequent investigation determined that degradation of the tube material was caused by a manufacturing deficiency, leading to insufficient material resistance against mechanical damage when a tube is located in a fuel.

This condition, if not detected and corrected, could lead to reduced fuel supply to the engines, inability to use all the fuel in fuel tanks and reduced available engine power, resulting in reduced aeroplane performance.

To address this potentially unsafe condition, PZL identified the batch of aeroplanes that are potentially equipped with thermo-shrinkable tubes having this manufacturing defect, and issued the [service bulletin] SB providing inspection and replacement instructions.

For the reasons described above, this [EASA] AD requires a one-time inspection of the electrical harnesses located in the fuel tanks and, depending on findings, replacement of the affected harness.

Polskie Zaklady Lotnicze Sp. z o.o. informed the FAA the potential for damage to the thermo-shrinkable tubes does not progress with time. Therefore, the FAA determined repetitive inspections are not required. You may examine the MCAI on the internet at https://www.regulations.gov by searching for and locating Docket No. FAA–2020–0473.

Comments

The FAA gave the public the opportunity to participate in developing this final rule. The following presents the comments received on the NPRM and the FAA’s response to each comment.

Support for the NPRM

An individual commenter supported the NPRM.

Request to Change the Cost of Compliance

Another individual commenter requested the FAA increase the estimated number of labor hours in the cost of compliance. The commenter stated the number of hours should be increased from 3 to at least 44 to 48, based on the time and personnel needed to open the center wing panels, properly shore the engines, open the tank covers, perform the inspection, and complete the close up. The commenter further stated that this would not include costs for any de-fueling, de-puddling, re-fueling, or leak checks that may need to be done.

The FAA disagrees. The cost analysis in AD rulemaking actions typically includes only the costs associated with complying with the AD. In the NPRM, the FAA estimated 3 work hours to perform the inspection and 60 work hours, if necessary, to replace the harness, based on information from the design approval holder. The compliance time for this AD allows the operator to do this inspection at the same time as other maintenance when the airplane has been prepared for other tasks. No changes were made to the proposed AD based on this comment.

Conclusion

The FAA reviewed the relevant data, considered the comments received, and determined that air safety and the public interest require adopting this final rule as proposed.

Related Service Information Under 1 CFR Part 51

The FAA reviewed Polskie Zaklady Lotnicze Sp. z o.o. Service Bulletin No. E/12.141/2018, dated May 15, 2018. The service information contains procedures for inspecting the thermo-shrinkable tubes on the electrical harnesses in the center and outer wing fuel tanks for damage and replacing any electrical harness with damaged thermo-shrinkable tubes. This service information is reasonably available because the interested parties have access to it through their normal course of business or by the means identified in the ADDRESSES section.

Costs of Compliance

The FAA estimates that this AD will affect 15 products of U.S. registry. The FAA also estimates that it will take about 3 work-hours per product to comply with the inspection requirement of this AD. The average labor rate is $85 per work-hour.

Based on these figures, the FAA estimates the cost of the AD on U.S. operators to be $3,825, or $255 per product.

In addition, the FAA estimates that any necessary follow-on replacement action will take about 60 work-hours and require parts costing $5,000, for a cost of $10,100 per electrical harness. The FAA has no way of determining the number of airplanes that may need these actions.

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA’s authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. Subtitle VII: Aviation Programs, describes in more detail the scope of the Agency’s authority.

The FAA is issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section 44701: General requirements. Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or
develop on products identified in this rulemaking action.

Regulatory Findings

This AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify this AD:

(1) Is not a “significant regulatory action” under Executive Order 12866,
(2) Will not affect intrastate aviation in Alaska, and
(3) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

§ 39.13 [Amended]

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. The FAA amends § 39.13 by adding the following new airworthiness directive (AD):


(a) Effective Date

This AD is effective December 8, 2020.

(b) Affected ADs

None.

(c) Applicability

This AD applies to Polskie Zaklady Lotnicze Sp. z o.o. Model PZL M28 05 airplanes, serial numbers AJE00301 through AJE00343, and AJE00345 through AJE00347, certificated in any category.

(d) Subject


(e) Reason

This AD was prompted by mandatory continuing airworthiness information (MCAI) originated by an aviation authority of another country to identify and correct an unsafe condition on an aviation product. The MCAI describes the unsafe condition as defective thermo-shrinkable tubes installed on the electrical harnesses located in the fuel tanks. The FAA is issuing this AD to prevent broken pieces of the thermo-shrinkable tubes from blocking the jet pump, reducing fuel supply to the engines, and resulting in the inability to use all the fuel in the fuel tanks. This condition could lead to reduced engine power and airplane performance.

(f) Actions and Compliance

Unless already done, do the following actions in paragraphs (f)(1) and (2) of this AD:

(1) Within the next 200 hours time-in-service (TIS) after the effective date of this AD or within the next 8 months after the effective date of this AD, whichever occurs first:
   (ii) If there is a tear or any cracking in or any seizing of an electrical wire harness thermo-shrinkable tube, before further flight, replace the harness in accordance with section II. a) Replacement of harness KL6 (KP), II. b) Replacement of Harness KL9 (KP9), or II. c) Replacement of harness KL10 (KP10), as applicable, of the Procedure for Bulletin Execution in Polskie Zaklady Lotnicze Sp. z o.o. Service Bulletin No. E/12.141/2018, dated May 15, 2018.
(2) As of the effective date of this AD, do not install any electrical wire harness part number 28.14.7205.073.000, 28.14.7205.074.000, 28.14.7205.075.000, 28.14.7205.076.000, 28.14.7205.077.000, or 28.14.7205.078.000, that has more than zero hours TIS on any airplane, unless it has passed the inspection required by paragraph (f)(1)(i) of this AD.

(g) Alternative Methods of Compliance (AMOCs)

The Manager, International Validation Branch, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. Send information to Doug Rudolph, Aerospace Engineer, FAA, General Aviation & Rotorcraft Section, International Validation Branch, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone: (816) 329–4148. For information on the availability of this material at the FAA, call (816) 329–4148.

(h) Related Information

Refer to European Union Aviation Safety Agency (EASA) AD No. 2018–0242, dated October 8, 2018, for more information. You may examine the EASA AD in the AD docket on the internet at https://www.regulations.gov.

(i) Material Incorporated by Reference

(1) The Director of the Federal Register approved the incorporation by reference (IBR) of the service information listed in this paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) You must use this service information as applicable to do the actions required by this AD, unless the AD specifies otherwise.


Issued on October 19, 2020.

Lance T. Gant,
Director, Compliance & Airworthiness Division, Aircraft Certification Service.