installation as follows, whichever occurs later:

(i) Before the timing chain exceeds 900 flight hours (FHs) since new, or;
(ii) Within 100 FHs after the windmilling restart, or;
(iii) Before further flight.

(2) For engines that have a windmill restart after the effective date of this AD, remove the timing chain before it exceeds 900 FHs since new or within 100 FHs after the windmilling restart, whichever occurs later, and replace with a new or repaired timing chain.

(3) Remove the fuel injectors and replace with parts eligible for installation before they exceed 900 FHs since new or before further flight after the effective date of this AD, whichever occurs later.

(i) Use the Accomplishment/Instructions, paragraph 2.1, of Austro Engine Mandatory Service Bulletin (MSB) No. MSB–E4–025, Rev. No. 3, dated January 8, 2019, to perform the required actions in paragraph (g)(3) of this AD.

(ii) [Reserved]

(iii) Thereafter, repeat the replacement of the fuel injectors required by paragraph (g)(3) of this AD at intervals not exceeding 900 FHs since new.

(b) Non-Required Actions

The tagging and returning of the removed fuel injectors to the manufacturer, referenced in the Accomplishment/Instructions, paragraph 2.1, of Austro Engine MSB No. MSB–E4–025, Rev. No. 3, dated January 8, 2019, are not required by this AD.

(i) Credit for Previous Actions

You may take credit for the replacement of the timing chain that is required by paragraph (g)(1) of this AD if you performed this replacement before the effective date of this AD using Austro Engine MSB No. MSB–E4–017/2, Revision 2, dated December 2, 2016.

(ii) Alternative Methods of Compliance (AMOCs)

(1) The Manager, ECO Branch, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the manager of the ECO Branch, send it to the attention of the person identified in paragraph (k)(1) of this AD. You may email your request to: ANE-AD-AMOC@faa.gov.

(2) Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office.

(k) Related Information

(1) For more information about this AD, contact Mehdi Lamnyi, Aerospace Engineer, ECO Branch, FAA, 1200 District Avenue, Burlington, MA 01803; phone: 781–238–7743; fax: 781–238–7199; email: Mehdi.Lamnyi@faa.gov.


(3) For Austro Engine GmbH service information identified in this AD, contact Austro Engine GmbH, Rudolf-Diesel-Straße 11, A–2700 Weiner Neustadt, Austria; phone: +43 2622 23000; fax: +43 2622 23000–2711; website: www.austroengine.at. You may view this referenced service information at the FAA, Engine and Propeller Standards Branch, 1200 District Avenue, Burlington, MA 01803. For information on the availability of this material at the FAA, call 781–238–7759.


Lance T. Gant,
Director, Compliance & Airworthiness Division, Aircraft Certification Service.

[FR Doc. 2020–05292 Filed 3–19–20; 8:45 am]
BILLING CODE 4910–13–P

DEPARTMENT OF TRANSPORTATION
Federal Aviation Administration

14 CFR Part 39


RIN 2120–AA64

Airworthiness Directives; Yábar Indústria Aeronáutica S.A. (Type Certificate Previously Held by Embraer S.A.) Airplanes

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: The FAA proposes to adopt a new airworthiness directive (AD) for all Yábar Indústria Aeronáutica S.A. (Type Certificate Previously Held by Embraer S.A.) Model ERJ 170 airplanes and Model ERJ 190–100 STD, –100 LR, –100 ECJ, –100 IGW, –200 STD, –200 LR, and –200 IGW airplanes. This proposed AD was prompted by reports of cracks discovered on the engine pylon inboard lower link lugs. This proposed AD would require repetitive detailed inspections of the engine inboard and outboard engine pylon lower link lugs for cracking, and repair if necessary, as specified in an Agência Nacional de Aviação Civil (ANAC) Brazilian AD, which will be incorporated by reference. The FAA is proposing this AD to address the unsafe condition on these products.

DATES: The FAA must receive comments on this proposed AD by May 4, 2020.

ADDRESSES: You may send comments, using the procedures found in 14 CFR 11.33 and 11.45, by any of the following methods:

• Federal eRulemaking Portal: Go to https://www.regulations.gov. Follow the instructions for submitting comments.
• Fax: 202–493–2251.

FOR FURTHER INFORMATION CONTACT:
Krista Greer, Aerospace Engineer, International Section, Transport Standards Branch, FAA, 2200 South 216th St., Des Moines, WA 98198; telephone and fax 206–231–3221; email krista.greer@faa.gov.

SUPPLEMENTARY INFORMATION:

Comments Invited

The FAA invites you to send any written relevant data, views, or arguments about this proposal. Send your comments to an address listed under the ADDRESSES section. Include “Docket No. FAA–2020–0202; Product
The FAA is proposing this AD to address cracking of the engine pylon lower link lugs, which could cause the loss of engine pylon integrity, and could result in engine separation from the wing, loss of airplane controllability, and possible injury to persons on the ground. See the MCAI for additional background information.

### Related IBR Material Under 1 CFR Part 51

ANAC Brazilian AD 2020–01–02 describes procedures for repetitive detailed inspections of LH and RH inboard and outboard engine pylon lower link lugs for cracking, and repair if necessary. This material is reasonably available because the interested parties have access to it through their normal course of business or by the means identified in the ADDRESSES section.

### FAA's Determination and Requirements of This Proposed AD

This product has been approved by the aviation authority of another country, and is approved for operation in the United States. Pursuant to the FAA's bilateral agreement with the State of the Design Authority, the FAA has been notified of the unsafe condition described in the MCAI referenced above. The FAA is proposing this AD because the FAA evaluated all the relevant information and determined the unsafe condition described previously is likely to exist or develop in other products of the same type design.

### Proposed AD Requirements

This proposed AD would require accomplishing the actions specified in Brazilian AD 2020–01–02 described previously, as incorporated by reference, except for any differences identified as exceptions in the regulatory text of this AD.

### Explanation of Required Compliance Information

In the FAA's ongoing efforts to improve the efficiency of the AD process, the FAA initially worked with Airbus and the European Union Aviation Safety Agency (EASA) to develop a process to use certain EASA ADs as the primary source of information for compliance with requirements for corresponding FAA ADs. The FAA has since coordinated with other manufacturers and civil aviation authorities (CAAs) to use this process. As a result, Brazilian AD 2020–01–02 will be incorporated by reference in the FAA final rule. This proposed AD would, therefore, require compliance with Brazilian AD 2020–01–02 in its entirety, through that incorporation, except for any differences identified as exceptions in the regulatory text of this proposed AD. Service information specified in Brazilian AD 2020–01–02 that is required for compliance with Brazilian AD 2020–01–02 will be available on the internet at [https://www.regulations.gov](https://www.regulations.gov) for searching and locating Docket No. FAA–2020–0202 after the FAA final rule is published.

### Costs of Compliance

The FAA estimates that this proposed AD affects 659 airplanes of U.S. registry. The FAA estimates the following costs to comply with this proposed AD:

<table>
<thead>
<tr>
<th>Labor cost</th>
<th>Parts cost</th>
<th>Cost per product</th>
<th>Cost on U.S. operators</th>
</tr>
</thead>
<tbody>
<tr>
<td>3 work-hours × $85 per hour = $255</td>
<td>$0</td>
<td>$255</td>
<td>$168,045</td>
</tr>
</tbody>
</table>

The FAA estimates that it would take about 1 work-hour per product to comply with the reporting requirement in this proposed AD. The average labor rate is $85 per hour. Based on these figures, the FAA estimates the cost on U.S. operators of reporting the inspection results to be $56,015, or $85 per product.

The FAA has received no definitive data that would enable the FAA to provide cost estimates for the on-condition actions specified in this proposed AD.

### Paperwork Reduction Act

A federal agency may not conduct or sponsor, and a person is not required to respond to, nor shall a person be subject to penalty for failure to comply with a collection of information subject to the requirements of the Paperwork Reduction Act unless that collection of information displays a current valid OMB control number. The control number for the collection of information required by this proposed AD is 2120–0056. The paperwork cost associated with this proposed AD has been detailed in the Costs of Compliance section of this document and includes time for reviewing instructions, as well as completing and reviewing the collection of information. Therefore, all reporting associated with this proposed AD is mandatory. Comments concerning the accuracy of this burden and suggestions for reducing the burden should be directed to Information Collection Clearance Officer, Federal Aviation Administration, 10101 Hillwood Parkway, Fort Worth, TX 76177–1524.

### Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA’s authority to issue
rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. Subtitle VII: Aviation Programs, describes in more detail the scope of the Agency’s authority.

The FAA is issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section 44701: “General requirements.” Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

The FAA determined that this proposed AD would not have federalism implications under Executive Order 13132. This proposed AD would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify this proposed regulation: (1) Is not a “significant regulatory action” under Executive Order 12866, (2) Will not affect intrastate aviation in Alaska, and (3) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

The Proposed Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA proposes to amend 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

§ 39.13 [Amended]

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. The FAA amends § 39.13 by adding the following new airworthiness directive (AD):


(a) Comments Due Date

The FAA must receive comments by May 4, 2020.

(b) Affected ADs

None.

(c) Applicability

This AD applies to all Yaborá Indústria Aeronáutica S.A. (Type certificate previously held by Embraer S.A.) airplanes specified in paragraphs (c)(1) and (2) of this AD, certified in any category, as identified in Brazilian AD 2020–01–02, effective January 28, 2020 (“Brazilian AD 2020–01–02”).


(d) Subject

Air Transport Association (ATA) of America Code 54, Nacelles/pylons.

(e) Reason

This AD was prompted by reports of cracking on the left hand (LH) and right hand (RH) sides of engine pylon inboard lower link lugs. The FAA is issuing this AD to address cracking of the engine pylon lower link lugs, which could cause the loss of engine pylon integrity, and could result in engine separation from the wing, loss of airplane controllability, and possible injury to persons on the ground.

(f) Compliance

Comply with this AD within the compliance times specified, unless already done.

(g) Requirements

Except as specified in paragraph (h) of this AD: Comply with all required actions and compliance times specified in, and in accordance with, Brazilian AD 2020–01–02.

(h) Exceptions to Brazilian AD 2020–01–02

(1) Where Brazilian AD 2020–01–02 refers to its effective date, this AD requires using the effective date of this AD.

(2) Where Brazilian AD 2020–01–02 requires contacting “the ANAC [Agência Nacional de Aviação Civil] and Embraer . . . to approve an adequate repair,” for this AD, obtain repair instructions using the procedures specified in paragraph (i)(2) of this AD and do the repair.

(3) The “Alternative methods of compliance (AMOCs)” section of Brazilian AD 2020–01–02 does not apply to this AD.

(4) Paragraph (e) of Brazilian AD 2020–01–02 specifies to report inspection results to ANAC and Yaborá Indústria Aeronáutica within a certain compliance time. For this AD, report inspection results at the applicable time specified in paragraph (h)(4)(i) or (ii) of this AD.

(i) If the inspection was done on or after the effective date of this AD: Submit the report within 30 days after the inspection.

(ii) If the inspection was done before the effective date of this AD: Submit the report within 30 days after the effective date of this AD.

(j) Other FAA AD Provisions

The following provisions also apply to this AD:

(1) Alternative Methods of Compliance (AMOCs): The Manager, International Section, Transport Standards Branch, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the International Section, send it to the attention of the person identified in paragraph (j)(2) of this AD. Information may be emailed to: 9-ANM-116-AMOC-REQUESTS@faa.gov. Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office.

(2) Contacting the Manufacturer: For any requirement in this AD, if obtained from a manufacturer, the instructions must be accomplished using a method approved by the Manager, International Section, Transport Standards Branch, FAA; or ANAC; or ANAC’s authorized Designee. If approved by the ANAC Designee, the approval must include the Designee’s authorized signature.

(3) Reporting Requirements: A federal agency may not conduct or sponsor, and a person is not required to respond to, nor shall a person be subject to a penalty for failure to comply with a collection of information subject to the requirements of the Paperwork Reduction Act unless that collection of information displays a current valid OMB Control Number. The OMB Control Number for this information collection is 2120–0056. Public reporting for this collection of information is estimated to be approximately 1 hour per response, including the time for reviewing instructions, searching existing data sources, gathering and maintaining the data needed, and completing and reviewing the collection of information. All responses to this collection of information are mandatory as required by this AD; the nature and extent of confidentiality to be provided, if any. Send comments regarding this burden estimate or any other aspect of this collection of information, including suggestions for reducing this burden to Information Collection Clearance Officer, Federal Aviation Administration, 10101 Hillwood Parkway, Fort Worth, TX 76177–1524.

(k) Related Information

(1) For information about Brazilian AD 2020–01–02, contact National Civil Aviation Agency, Aeronautical Technical Products Certification Branch (GCGP), Rua Laurent Martins, n°209, Jardim Espalanada, CEP 12242–431—São José dos Campos—SP, Brazil; telephone 55 (12) 3203–6600; email pac@anac.gov.br; internet www.anac.gov.br/en/. You may find this material on the ANAC website at https://sistemas.anac.gov.br/certificacao/DA/
DEPARTMENT OF TRANSPORTATION
Federal Aviation Administration

14 CFR Part 39
[Docket No. FAA–2018–0334; Product Identifier 2017–SW–133–AD]
RIN 2120–AA64
Airworthiness Directives; Bell Helicopter Textron Canada Limited Helicopters

AGENCY: Federal Aviation Administration (FAA), DOT.
ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: The FAA proposes to adopt a new airworthiness directive (AD) for Bell Helicopter Textron Canada Limited (BHTC) Model 429 helicopters. This proposed AD would require repetitive inspections of certain cyclic and collective assembly bearings. This proposed AD is prompted by reports that precipitation can lead to reduced effectiveness of the grease in the bearings. The actions of this proposed AD are intended to address an unsafe condition on these products.

DATES: The FAA must receive comments on this proposed AD by May 19, 2020.

ADDRESSES: You may send comments by any of the following methods:
- Federal eRulemaking Docket: Go to https://www.regulations.gov. Follow the online instructions for sending your comments electronically.
- Mail: Send comments to the U.S. Department of Transportation, Docket Operations, 400 Seventh Street SW, Washington, DC 20590–0001.

- Hand Delivery: Deliver to the “Mail” address between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

Examining the AD Docket

(2) For more information about this AD, contact Krista Greer, Aerospace Engineer, International Section, Transport Standards Branch, FAA, 2200 South 216th St., Des Moines, WA 98198; telephone and fax 206–231–3122; email krista.greer@faa.gov.

Issued on March 10, 2020.

Lance T. Gant,
Director, Compliance & Airworthiness Division, Aircraft Certification Service.

[FR Doc. 2020–05486 Filed 3–19–20; 8:45 am]
BILLING CODE 4910–13–P

Federal Register / Vol. 85, No. 55 / Friday, March 20, 2020 / Proposed Rules 16019

FOR FURTHER INFORMATION CONTACT:
David Hatfield, Aviation Safety Engineer, Safety Management Section, Rotorcraft Standards Branch, FAA, 10101 Hillwood Pkwy, Fort Worth, TX 76177; telephone 817–222–5110; email david.hatfield@faa.gov.

SUPPLEMENTARY INFORMATION:
Comments Invited
The FAA invites you to participate in this rulemaking by submitting written comments, data, or views. The FAA also invites comments relating to the economic, environmental, energy, or federalism impacts that might result from adopting the proposals in this document. The most helpful comments reference a specific portion of the proposal, explain the reason for any recommended change, and include supporting data. To ensure the docket does not contain duplicate comments, commenters should send only one copy of written comments, or if comments are filed electronically, commenters should submit only one time.

The FAA will file in the docket all comments received, as well as a report summarizing each substantive public contact with FAA personnel concerning this proposal. Before acting on this proposal, the FAA will consider all comments received on or before the closing date for comments. The FAA will consider comments filed after the comment period has closed if it is possible to do so without incurring expense or delay. The FAA may change this proposal in light of the comments received.

Discussion

Transport Canada advises that in-service reports show that bearings in the roof-mounted flight control bellcranks are adversely affected by precipitation. Pooling can occur at the forward portion of the roof, providing a source of contamination for bearings in the roof-mounted flight controls. Precipitation may reduce the effectiveness of the grease in the bearings, allowing corrosion to occur, and resulting in intermittent restrictions, such as binding and roughness in the flight controls. Transport Canada advises. Transport Canada also advises that an undetected corroded bearing could lead to restrictions in the collective, directional, or pitch control systems, resulting in difficulty controlling the helicopter.

Transport Canada consequently requires within 12 months after the helicopter was manufactured and thereafter at intervals not to exceed 6 months, inspecting the flight controls and replacing any discrepant bearings. If the helicopter’s age exceeds 12 months, Transport Canada requires the 12-month inspection within 30 days. Transport Canada also requires, within 30 days, performing a functional check and replacement, if applicable, of the bearings if the most recent functional check of the helicopter was performed with the alternate procedure of using a hydraulic test stand or if the inspection method is unknown.

FAA’s Determination
These helicopters have been approved by the aviation authority of Canada and are approved for operation in the United States. Pursuant the FAA’s bilateral agreement with Canada, Transport Canada, its technical representative, has notified the FAA about the unsafe condition described in its AD. The FAA is proposing this AD after evaluating all known relevant information and determining that an unsafe condition is