

The Proposed Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA proposes to amend 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

■ 2. The FAA amends § 39.13 by adding the following new airworthiness directive (AD):

Airbus SAS: Docket No. FAA–2019–0484; Product Identifier 2019–NM–065–AD.

(a) Comments Due Date

The FAA must receive comments by March 6, 2020.

(b) Affected ADs

None.

(c) Applicability

This AD applies to all Airbus SAS airplanes identified in paragraphs (c)(1) through (7) of this AD, certificated in any category.

- (1) Model A330–201, –202, –203, –223, and –243 airplanes.
- (2) Model A330–223F and –243F airplanes.
- (3) Model A330–301, –302, –303, –321, –322, –323, –341, –342, and –343 airplanes.
- (4) Model A340–211, –212, –213 airplanes.
- (5) Model A340–311, –312, and –313 airplanes.
- (6) Model A340–541 airplanes.
- (7) Model A340–642 airplanes.

(d) Subject

Air Transport Association (ATA) of America Code 32, Landing gear.

(e) Reason

This AD was prompted by a report that an airplane failed to extend its nose landing gear (NLG) using the free fall method, due to the loss of the green hydraulic system. The FAA is issuing this AD to address detached magnets on both electrical motors of the free fall actuators (FFAs), which could prevent landing gear extension by the free fall method, possibly resulting in loss of control of the airplane after landing.

(f) Compliance

Comply with this AD within the compliance times specified, unless already done.

(g) Requirements

Except as specified in paragraph (h) of this AD: Comply with all required actions and compliance times specified in, and in accordance with, European Union Aviation Safety Agency (EASA) AD 2019–0164, dated July 11, 2019 (“EASA AD 2019–0164”).

(h) Exceptions to EASA AD 2019–0164

(1) Where EASA AD 2019–0164 refers to its effective date or April 9, 2019 (the effective

date of EASA AD 2019–0063, dated March 26, 2019), this AD requires using the effective date of this AD.

(2) The “Remarks” section of EASA AD 2019–0164 does not apply to this AD.

(3) Where paragraph (3) of EASA AD 2019–0164 specifies credit for certain tasks “provided the continuity test specified in AMM task A330–32–33–00–710–809, or AMM task A340–32–33–00–710–806, as applicable, is accomplished concurrently,” this AD provides credit “provided the continuity test is accomplished concurrently in accordance with the instructions of an FAA-approved maintenance or inspection program.”

(i) No Reporting Requirement

Although the service information referenced in EASA AD 2019–0164 specifies to submit certain information to the manufacturer, this AD does not include that requirement.

(j) Other FAA AD Provisions

The following provisions also apply to this AD:

(1) *Alternative Methods of Compliance (AMOCs):* The Manager, International Section, Transport Standards Branch, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the International Section, send it to the attention of the person identified in paragraph (k)(2) of this AD. Information may be emailed to: 9-ANM-116-AMOC-REQUESTS@faa.gov. Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office.

(2) *Contacting the Manufacturer:* For any requirement in this AD to obtain instructions from a manufacturer, the instructions must be accomplished using a method approved by the Manager, International Section, Transport Standards Branch, FAA; or EASA; or Airbus SAS’s EASA Design Organization Approval (DOA). If approved by the DOA, the approval must include the DOA-authorized signature.

(3) *Required for Compliance (RC):* For any service information referenced in EASA AD 2019–0164 that contains RC procedures and tests: Except as required by paragraph (j)(2) of this AD, RC procedures and tests must be done to comply with this AD; any procedures or tests that are not identified as RC are recommended. Those procedures and tests that are not identified as RC may be deviated from using accepted methods in accordance with the operator’s maintenance or inspection program without obtaining approval of an AMOC, provided the procedures and tests identified as RC can be done and the airplane can be put back in an airworthy condition. Any substitutions or changes to procedures or tests identified as RC require approval of an AMOC.

(k) Related Information

(1) For information about EASA AD 2019–0164, contact the EASA, Konrad-Adenauer-Ufer 3, 50668 Cologne, Germany; phone: +49 221 89990 1000; email: ADs@easa.europa.eu; internet: www.easa.europa.eu. You may find this EASA AD on the EASA website at <https://ad.easa.europa.eu>. You may view this EASA AD at the FAA, Transport Standards Branch, 2200 South 216th St., Des Moines, WA. For information on the availability of this material at the FAA, call 206–231–3195. EASA AD 2019–0164 may be found in the AD docket on the internet at <https://www.regulations.gov> by searching for and locating Docket No. FAA–2019–0484.

(2) For more information about this AD, contact Vladimir Ulyanov, Aerospace Engineer, International Section, Transport Standards Branch, FAA, 2200 South 216th St., Des Moines, WA 98198; phone and fax: 206–231–3229.

Issued on January 3, 2020.

John Piccola, Jr.,

Acting Director, System Oversight Division, Aircraft Certification Service.

[FR Doc. 2020–00449 Filed 1–17–20; 8:45 am]

BILLING CODE 4910–13–P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA–2019–0800; Product Identifier 2005–NE–24–AD]

RIN 2120–AA64

Airworthiness Directives; General Electric Company Turbofan Engines

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: The FAA proposes to supersede Airworthiness Directive (AD) 2005–23–09, which applies to all General Electric Company (GE) CF6–80E1A1, –80E1A2, –80E1A3, –80E1A4, and –80E1A4/B model turbofan engines. AD 2005–23–09 requires initial and repetitive fluorescent-penetrant inspections (FPI) of certain areas of high-pressure compressor (HPC) cases, part number (P/N) 1509M97G07 and P/N 2083M69G03. Since the FAA issued AD 2005–23–09, GE performed an updated life analysis on the HPC case. As a result, GE found additional locations on the cases requiring FPI, revised the inspection interval for performing FPI of the existing location, and added an additional P/N HPC case that requires inspection. This proposed AD would require an update of the Airworthiness Limitations Section (ALS) of GE Engine Manual GEK99376

and the operator's existing continuous airworthiness maintenance program (CAMP). The FAA is proposing this AD to address the unsafe condition on these products.

DATES: The FAA must receive comments on this proposed AD by March 6, 2020.

ADDRESSES: You may send comments, using the procedures found in 14 CFR 11.43 and 11.45, by any of the following methods:

- *Federal eRulemaking Portal:* Go to <https://www.regulations.gov>. Follow the instructions for submitting comments.
- *Fax:* 202-493-2251.
- *Mail:* U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE, Washington, DC 20590.
- *Hand Delivery:* Deliver to Mail address above between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

For service information identified in this NPRM, contact General Electric Company, GE Aviation, Room 285, 1 Neumann Way, Cincinnati, OH 45215; phone: 513-552-3272; email: aviation.fleetsupport@ge.com. You may view this service information at the FAA, Engine and Propeller Standards Branch, 1200 District Avenue, Burlington, MA 01803. For information on the availability of this material at the FAA, call 781-238-7759.

Examining the AD Docket

You may examine the AD docket on the internet at <https://www.regulations.gov> by searching for and locating Docket No. FAA-2019-0800; or in person at Docket Operations between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this NPRM, the regulatory evaluation, any comments received, and other information. The street address for Docket Operations is listed above. Comments will be available in the AD docket shortly after receipt.

FOR FURTHER INFORMATION CONTACT: Scott Stevenson, Aerospace Engineer, ECO Branch, FAA, 1200 District Avenue, Burlington, MA 01803; phone: 781-238-7132; fax: 781-238-7199; email: scott.m.stevenson@faa.gov.

SUPPLEMENTARY INFORMATION:

Comments Invited

The FAA invites you to send any written relevant data, views, or arguments about this proposed AD. Send your comments to an address listed under the **ADDRESSES** section. Include "Docket No. FAA-2019-0800; Product Identifier 2005-NE-24-AD" at the beginning of your comments. The FAA specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of this NPRM. The FAA will consider all comments received by the closing date and may amend this NPRM because of those comments.

The FAA will post all comments received, without change, to <https://www.regulations.gov>, including any personal information you provide. The FAA will also post a report summarizing each substantive verbal contact received about this proposed AD.

Discussion

The FAA issued AD 2005-23-09, Amendment 39-14367 (70 FR 67901, November 9, 2005), ("AD 2005-23-09"), for all GE CF6-80E1A1, -80E1A2, -80E1A3, -80E1A4, and -80E1A4/B model turbofan engines. AD 2005-23-09 requires initial and repetitive FPI of certain areas of HPC cases, P/N 1509M97G07 and P/N 2083M69G03. AD 2005-23-09 resulted from the discovery that HPC cases, P/N 1509M97G07 and P/N 2083M69G03, were inadvertently left out of the ALS, Chapter 05-21-02, of GE Engine Manual, GEK 99376, Revision 17. The FAA issued AD 2005-23-09 to prevent failure of the HPC case aft mount flange, due to cracking.

Actions Since AD 2005-23-09 Was Issued

Since the FAA issued AD 2005-23-09, GE performed an updated lifing analysis on the HPC case. As a result, GE revised the inspection interval of the existing location for the FPI and found additional locations on the HPC case that require inspection. GE also found an additional HPC case, P/N 1509M97G05, that requires this inspection.

Related Service Information Under 1 CFR Part 51

The FAA reviewed TASK 05-21-02-200-001, dated September 15, 2015, from the ALS of the GE CF6-80E1 Engine Manual GEK99376, Revision 48, dated September 15, 2019. The service information describes procedures for performing FPIs of the HPC case. This service information is reasonably available because the interested parties have access to it through their normal course of business or by the means identified in the **ADDRESSES** section.

FAA's Determination

The FAA is proposing this AD because we evaluated all the relevant information and determined the unsafe condition described previously is likely to exist or develop in other products of the same type design.

Proposed AD Requirements

This proposed AD would retain certain requirements of AD 2005-23-09. This proposed AD would require an update of the ALS of GE Engine Manual GEK99376 and the operator's existing CAMP.

Costs of Compliance

The FAA estimates that this proposed AD affects 20 engines installed on airplanes of U.S. registry.

The FAA estimates the following costs to comply with this proposed AD:

ESTIMATED COSTS

Action	Labor cost	Parts cost	Cost per product	Cost on U.S. operators
Update ALS of engine manual	2 work-hours × \$85 per hour = \$170	\$0	\$170	\$3,400

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, Section 106, describes the authority of the FAA Administrator. Subtitle VII, Aviation Programs, describes in more

detail the scope of the Agency's authority.

The FAA is issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section 44701, "General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil

aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or

develop on products identified in this rulemaking action.

This AD is issued in accordance with authority delegated by the Executive Director, Aircraft Certification Service, as authorized by FAA Order 8000.51C. In accordance with that order, issuance of ADs is normally a function of the Compliance and Airworthiness Division, but during this transition period, the Executive Director has delegated the authority to issue ADs applicable to engines, propellers, and associated appliances to the Manager, Engine and Propeller Standards Branch, Policy and Innovation Division.

Regulatory Findings

The FAA has determined that this proposed AD would not have federalism implications under Executive Order 13132. This proposed AD would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that the proposed regulation:

- (1) Is not a "significant regulatory action" under Executive Order 12866,
- (2) Will not affect intrastate aviation in Alaska, and
- (3) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

The Proposed Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA proposes to amend 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

- 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

- 2. The FAA amends § 39.13 by removing Airworthiness Directive (AD) 2005–23–09, Amendment 39–14367 (70 FR 67901, November 9, 2005), and adding the following new AD:

General Electric Company: Docket No. FAA–2019–0800; Product Identifier 2005–NE–24–AD.

(a) Comments Due Date

The FAA must receive comments on this AD action by March 6, 2020.

(b) Affected ADs

This AD replaces AD 2005–23–09, Amendment 39–14367 (70 FR 67901, November 9, 2005).

(c) Applicability

This AD applies to General Electric Company (GE) CF6–80E1A1, –80E1A2, –80E1A3, –80E1A4, and –80E1A4/B model turbofan engines.

(d) Subject

Joint Aircraft System Component (JASC) Code 7230, Turbine Engine Compressor Section.

(e) Unsafe Condition

This AD was prompted by GE performing an updated lifing analysis on the high-pressure compressor (HPC) case. Based on this analysis, GE found new locations on the case that require fluorescent penetrant inspection (FPI), identified a new inspection interval for the existing FPI location, and added another part-numbered HPC case that requires inspection. The FAA is issuing this AD to prevent failure of the HPC case. The unsafe condition, if not addressed, could result in uncontained release of the HPC case, engine fire, and damage to the airplane.

(f) Compliance

Comply with this AD within the compliance times specified, unless already done.

(g) Required Actions

Within 180 days after the effective date of this AD, replace TASK 05–21–02–200–001 in GE CF6–80E1 Engine Manual GEK99376 and the operator's existing continuous airworthiness maintenance program with TASK 05–21–02–200–001, dated September 15, 2015, from the Airworthiness Limitations Section of GE CF6–80E1 Engine Manual GEK99376, Revision 48, dated September 15, 2019.

(h) Alternative Methods of Compliance (AMOCs)

(1) The Manager, ECO Branch, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the manager of the certification office, send it to the attention of the person identified in paragraph (i)(1) of this AD. You may email your request to: *ANE-AD-AMOC@faa.gov*.

(2) Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office.

(i) Related Information

(1) For more information about this AD, contact Scott Stevenson, Aerospace Engineer, ECO Branch, FAA, 1200 District Avenue, Burlington, MA 01803; phone: 781–238–

7132; fax: 781–238–7199; email: *scott.m.stevenson@faa.gov*.

(2) For service information identified in this AD, contact General Electric Company, GE Aviation, Room 285, 1 Neumann Way, Cincinnati, OH 45215; phone: 513–552–3272; email: *aviation.fleetsupport@ge.com*. You may view this referenced service information at the FAA, Engine and Propeller Standards Branch, 1200 District Avenue, Burlington, MA 01803. For information on the availability of this material at the FAA, call 781–238–7759.

Issued in Burlington, Massachusetts, on January 2, 2020.

Robert J. Ganley,

Manager, Engine & Propeller Standards Branch, Aircraft Certification Service.

[FR Doc. 2020–00010 Filed 1–17–20; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 71

[Docket No. FAA–2020–0008; Airspace Docket No. 19–ASO–10]

RIN 2120–AA66

Proposed Amendment of VOR Federal Airways V–7, V–52, and V–178 in the Vicinity of Central City, KY

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: This action proposes to amend three VHF Omnidirectional Range (VOR) Federal airways, V–7, V–52, and V–178, in the vicinity of Central City, KY. The modifications are necessary due to the planned decommissioning of the VOR portion of the Central City, KY, VOR/Tactical Air Navigation (VORTAC) navigation aid (NAVAID), which provides navigation guidance for portions of the affected air traffic service (ATS) routes. The Central City VOR is being decommissioned as part of the FAA's VOR Minimum Operational Network (MON) program.

DATES: Comments must be received on or before March 6, 2020.

ADDRESSES: Send comments on this proposal to the U.S. Department of Transportation, Docket Operations, 1200 New Jersey Avenue SE, West Building Ground Floor, Room W12–140, Washington, DC 20590; telephone: (800) 647–5527, or (202) 366–9826. You must identify FAA Docket No. FAA–2020–0008; Airspace Docket No. 19–ASO–10 at the beginning of your comments. You may also submit comments through the internet at <https://www.regulations.gov>.