

(f) Alternative Methods of Compliance (AMOCs)

(1) The Manager, Safety Management Section, Rotorcraft Standards Branch, FAA, may approve AMOCs for this AD. Send your proposal to: Matt Fuller, Senior Aviation Safety Engineer, Safety Management Section, Rotorcraft Standards Branch, FAA, 10101 Hillwood Pkwy., Fort Worth, TX 76177; telephone (817) 222-5110; email 9-ASW-FTW-AMOC-Requests@faa.gov.

(2) For operations conducted under a 14 CFR part 119 operating certificate or under 14 CFR part 91, subpart K, we suggest that you notify your principal inspector, or lacking a principal inspector, the manager of the local flight standards district office or certificate holding district office before operating any aircraft complying with this AD through an AMOC.

(g) Additional Information

The subject of this AD is addressed in European Aviation Safety Agency (EASA) AD No. 2017-0026, dated February 14, 2017. You may view the EASA AD on the internet at <http://www.regulations.gov> in the AD Docket.

(h) Subject

Joint Aircraft Service Component (JASC) Code: 2400, Electrical Power System.

Issued in Fort Worth, Texas, on May 31, 2018.

James A. Grigg,

Acting Director, Compliance & Airworthiness Division, Aircraft Certification Service.

[FR Doc. 2018-12227 Filed 6-6-18; 8:45 am]

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DEPARTMENT OF TRANSPORTATION**Federal Aviation Administration****14 CFR Part 39**

[Docket No. FAA-2018-0503; Product Identifier 2018-NM-048-AD]

RIN 2120-AA64

Airworthiness Directives; 328 Support Services GmbH (Type Certificate Previously Held by AvCraft Aerospace GmbH; Fairchild Dornier GmbH; Dornier Luftfahrt GmbH) Airplanes

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: We propose to adopt a new airworthiness directive (AD) for all 328 Support Services GmbH Model 328-100 and -300 airplanes. This proposed AD was prompted by reports indicating corrosion on the horizontal stabilizer bearing supports at the contact surface to the horizontal stabilizer rear spar. This proposed AD would require inspections for corrosion and any other damage (*i.e.*, cracking and chafing) of

the horizontal stabilizer rear bearing supports, replacement of the affected horizontal stabilizer rear bearing supports if necessary, and modification of the horizontal stabilizer rear spar. We are proposing this AD to address the unsafe condition on these products.

DATES: We must receive comments on this proposed AD by July 23, 2018.

ADDRESSES: You may send comments, using the procedures found in 14 CFR 11.43 and 11.45, by any of the following methods:

- *Federal eRulemaking Portal:* Go to <http://www.regulations.gov>. Follow the instructions for submitting comments.

- *Fax:* 202-493-2251.

- *Mail:* U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE, Washington, DC 20590.

- *Hand Delivery:* Deliver to Mail address above between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

For service information identified in this NPRM, contact 328 Support Services GmbH, Global Support Center, P.O. Box 1252, D-82231 Wessling, Federal Republic of Germany; telephone +49 8153 88111 6666; fax +49 8153 88111 6565; email gsc.op@328support.de; internet <http://www.328support.de>. You may view this service information at the FAA, Transport Standards Branch, 2200 South 216th St., Des Moines, WA. For information on the availability of this material at the FAA, call 206-231-3195.

Examining the AD Docket

You may examine the AD docket on the internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA-2018-0503; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this NPRM, the regulatory evaluation, any comments received, and other information. The street address for the Docket Operations office (telephone 800-647-5527) is in the **ADDRESSES** section. Comments will be available in the AD docket shortly after receipt.

FOR FURTHER INFORMATION CONTACT: Todd Thompson, Aerospace Engineer, International Section, Transport Standards Branch, FAA, 2200 South 216th St., Des Moines, WA 98198; telephone and fax 206-231-3228.

SUPPLEMENTARY INFORMATION:**Comments Invited**

We invite you to send any written relevant data, views, or arguments about

this proposal. Send your comments to an address listed under the **ADDRESSES** section. Include “Docket No. FAA-2018-0503; Product Identifier 2018-NM-048-AD” at the beginning of your comments. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of this NPRM. We will consider all comments received by the closing date and may amend this NPRM based on those comments.

We will post all comments we receive, without change, to <http://www.regulations.gov>, including any personal information you provide. We will also post a report summarizing each substantive verbal contact we receive about this NPRM.

Discussion

The European Aviation Safety Agency (EASA), which is the Technical Agent for the Member States of the European Union, has issued EASA Airworthiness Directive 2017-0239, dated November 30, 2017 (referred to after this as the Mandatory Continuing Airworthiness Information, or “the MCAI”), to correct an unsafe condition for all 328 Support Services GmbH Model 328-100 and -300 airplanes. The MCAI states:

Occurrences were reported on horizontal stabilizer bearing supports being found corroded at the contact surface to the horizontal stabilizer rear spar. The corroded area was at the lower flange position, which is connected to the stabilizer rear spar and not visible without detachment of the fitting. Investigation determined that the corrosion is triggered by galvanic effect, due to a direct contact between the horizontal stabilizer rear spar, made from CFRP (carbon fibre reinforced plastic), and the aluminium rear attachment fitting.

This condition, if not detected and corrected, could lead to failure of the fitting and loss of one load path of the horizontal stabilizer attachment, possibly resulting in reduced control of the aeroplane.

To address this potential unsafe condition, 328 Support Services GmbH (328 SSG) issued Service Bulletin (SB) SB-328-55-557 and SB-328J-55-324 to provide instructions for inspection of the affected area, replacement of the parts, and modification to improve corrosion behaviour by incorporating of glass fibre layer.

For the reasons described above, this [EASA] AD requires a one-time inspection [detailed visual inspection and an eddy current inspection for chafing and corrosion] of the horizontal stabilizer rear bearing supports, and, depending on findings, accomplishment of applicable corrective action(s) [replacement of the affected horizontal stabilizer rear bearing supports]. This [EASA] AD also requires a modification of the horizontal stabilizer rear spar, irrespective of findings.

You may examine the MCAI in the AD docket on the internet at <http://www.regulations.gov>

www.regulations.gov by searching for and locating Docket No. FAA-2018-0503.

Related Service Information Under 1 CFR Part 51

328 Support Services GmbH has issued Service Bulletin SB-328-55-557, Revision 1, dated February 1, 2018; and Service Bulletin SB-328J-55-324, Revision 1, dated February 1, 2018. This service information describes a detailed visual inspection and an eddy current inspection for corrosion and any other damage (i.e., cracking and chafing) of the horizontal stabilizer rear bearing supports, modification of the horizontal

stabilizer rear spar, and replacement of the affected horizontal stabilizer rear bearing supports if necessary. These documents are distinct since they apply to different airplane models. This service information is reasonably available because the interested parties have access to it through their normal course of business or by the means identified in the ADDRESSES section.

FAA’s Determination and Requirements of This Proposed AD

This product has been approved by the aviation authority of another country, and is approved for operation in the United States. Pursuant to our

bilateral agreement with the State of Design Authority, we have been notified of the unsafe condition described in the MCAI and service information referenced above. We are proposing this AD because we evaluated all pertinent information and determined an unsafe condition exists and is likely to exist or develop on other products of the same type design.

Costs of Compliance

We estimate that this proposed AD affects 27 airplanes of U.S. registry.

We estimate the following costs to comply with this proposed AD:

ESTIMATED COSTS

Action	Labor cost	Parts cost	Cost per product	Cost on U.S. operators
Detailed visual inspection and eddy current inspection.	4 work-hours × \$85 per hour = \$340	\$0	\$340	\$9,180
Modification	16 work-hours × \$85 per hour = \$1,360	0	1,360	36,720

We estimate the following costs to do any necessary replacement that would

be required based on the results of the proposed inspection. We have no way of

determining the number of aircraft that might need this replacement:

ON-CONDITION COSTS

Action	Labor cost	Parts cost	Cost per product
Replacement	24 work-hours × \$85 per hour = \$2,040	(*)	\$2,040

* We have received no definitive data that would enable us to provide part cost estimates for the on-condition action specified in this proposed AD.

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA’s authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. Subtitle VII: Aviation Programs, describes in more detail the scope of the Agency’s authority.

We are issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section 44701: “General requirements.” Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

This proposed AD is issued in accordance with authority delegated by the Executive Director, Aircraft Certification Service, as authorized by FAA Order 8000.51C. In accordance

with that order, issuance of ADs is normally a function of the Compliance and Airworthiness Division, but during this transition period, the Executive Director has delegated the authority to issue ADs applicable to transport category airplanes to the Director of the System Oversight Division.

Regulatory Findings

We determined that this proposed AD would not have federalism implications under Executive Order 13132. This proposed AD would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify this proposed regulation:

1. Is not a “significant regulatory action” under Executive Order 12866;
2. Is not a “significant rule” under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979);
3. Will not affect intrastate aviation in Alaska; and

4. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

The Proposed Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA proposes to amend 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

39.13 [Amended]

2. The FAA amends § 39.13 by adding the following new airworthiness directive (AD):

328 Support Services GmbH (Type Certificate Previously Held by AvCraft Aerospace GmbH; Fairchild Dornier GmbH; Dornier Luftfahrt GmbH): Docket No. FAA–2018–0503; Product Identifier 2018–NM–048–AD.

(a) Comments Due Date

We must receive comments by July 23, 2018.

(b) Affected ADs

None.

(c) Applicability

This AD applies to all 328 Support Services GmbH (Type Certificate previously held by AvCraft Aerospace GmbH; Fairchild Dornier GmbH; Dornier Luftfahrt GmbH) Model 328–100 and –300 airplanes, certificated in any category.

(d) Subject

Air Transport Association (ATA) of America Code 55, Stabilizers.

(e) Reason

This AD was prompted by reports indicating corrosion on the horizontal stabilizer bearing supports at the contact surface to the horizontal stabilizer rear spar. We are issuing this AD to address corrosion on the horizontal stabilizer bearing supports and rear spar, which could lead to failure of the fitting and loss of one load path of the horizontal stabilizer attachment, and possibly result in reduced controllability of the airplane.

(f) Compliance

Comply with this AD within the compliance times specified, unless already done.

(g) Inspection and Modification

(1) At the applicable time specified in paragraph (g)(3)(i) or (g)(3)(ii) of this AD, do a detailed visual inspection and an eddy current inspection for corrosion and any other damage (*i.e.*, cracking and chafing) of the horizontal stabilizer rear bearing supports in accordance with the Accomplishment Instructions of 328 Support Services GmbH Service Bulletin SB–328–55–557, Revision 1, dated February 1, 2018 (for Model 328–100 airplanes); or 328 Support Services GmbH Service Bulletin SB–328J–55–324, Revision 1, dated February 1, 2018 (for Model 328–300 airplanes); as applicable.

(2) At the applicable time specified in paragraph (g)(3)(i) or (g)(3)(ii) of this AD, modify the horizontal stabilizer rear spar in accordance with the Accomplishment Instructions of 328 Support Services GmbH Service Bulletin SB–328–55–557, Revision 1, dated February 1, 2018 (for Model 328–100 airplanes); or 328 Support Services GmbH Service Bulletin SB–328J–55–324, Revision 1, dated February 1, 2018 (for Model 328–300 airplanes); as applicable.

(3) Do the actions in paragraphs (g)(1) and (g)(2) at the applicable compliance time specified in paragraph (g)(3)(i) or (g)(3)(ii) of this AD.

(i) For Group 1 airplanes as identified in 328 Support Services GmbH Service Bulletin SB–328–55–557, Revision 1, dated February

1, 2018 (for Model 328–100 airplanes); or 328 Support Services GmbH Service Bulletin SB–328J–55–324, Revision 1, dated February 1, 2018 (for Model 328–300 airplanes); as applicable: Within 1,000 flight cycles or 8 months, whichever occurs first after the effective date of this AD.

(ii) For Group 2 airplanes as identified in 328 Support Services GmbH Service Bulletin SB–328–55–557, Revision 1, dated February 1, 2018 (for Model 328–100 airplanes); or 328 Support Services GmbH Service Bulletin SB–328J–55–324, Revision 1, dated February 1, 2018 (for Model 328–300 airplanes); as applicable: Within 5,000 flight hours or 30 months, whichever occurs first after the effective date of this AD.

(h) Corrective Action

If, during the inspections required by paragraph (g) of this AD, corrosion or any other damage (*i.e.*, cracking and chafing) is detected, before further flight, replace the affected horizontal stabilizer rear bearing supports in accordance with the Accomplishment Instructions of 328 Support Services GmbH Service Bulletin SB–328–55–557, Revision 1, dated February 1, 2018 (for Model 328–100 airplanes); or 328 Support Services GmbH Service Bulletin SB–328J–55–324, Revision 1, dated February 1, 2018 (for Model 328–300 airplanes); as applicable.

(i) Parts Installation Prohibition

As of the applicable time specified in paragraph (i)(1) or (i)(2) of this AD, no person may install a horizontal stabilizer rear bearing support, part number 001B551A1441000, on any airplane.

(1) For Group 1 airplanes as identified in 328 Support Services GmbH Service Bulletin SB–328–55–557, Revision 1, dated February 1, 2018 (for Model 328–100 airplanes); or 328 Support Services GmbH Service Bulletin SB–328J–55–324, Revision 1, dated February 1, 2018 (for Model 328–300 airplanes); as applicable: After replacement of the horizontal stabilizer rear bearing supports as required by paragraph (h) of this AD.

(2) For Group 2 airplanes as identified in 328 Support Services GmbH Service Bulletin SB–328–55–557, Revision 1, dated February 1, 2018 (for Model 328–100 airplanes); or 328 Support Services GmbH Service Bulletin SB–328J–55–324, Revision 1, dated February 1, 2018 (for Model 328–300 airplanes); as applicable: As of the effective date of this AD.

(j) Credit for Previous Actions

This paragraph provides credit for actions required by paragraph (g) of this AD, if those actions were performed before the effective date of this AD using 328 Support Services GmbH Service Bulletin SB–328–55–557, dated September 1, 2017 (for Model 328–100 airplanes); or 328 Support Services GmbH Service Bulletin SB–328J–55–324, dated September 1, 2017 (for Model 328–300 airplanes).

(k) Other FAA AD Provisions

The following provisions also apply to this AD:

(1) *Alternative Methods of Compliance (AMOCs):* The Manager, International Section, Transport Standards Branch, FAA,

has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the International Section, send it to the attention of the person identified in paragraph (l)(2) of this AD. Information may be emailed to: 9-ANM-116-AMOC-REQUESTS@faa.gov. Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office.

(2) *Contacting the Manufacturer:* For any requirement in this AD to obtain corrective actions from a manufacturer, the action must be accomplished using a method approved by the Manager, International Section, Transport Standards Branch, FAA; or the European Aviation Safety Agency (EASA); or 328 Support Services GmbH's EASA Design Organization Approval (DOA). If approved by the DOA, the approval must include the DOA-authorized signature.

(l) Related Information

(1) Refer to Mandatory Continuing Airworthiness Information (MCAI) EASA Airworthiness Directive 2017–0239, dated November 30, 2017, for related information. This MCAI may be found in the AD docket on the internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA–2018–0503.

(2) For more information about this AD, contact Todd Thompson, Aerospace Engineer, International Section, Transport Standards Branch, FAA, 2200 South 216th Street, Des Moines, WA 98198; telephone and fax 206–231–3228.

(3) For service information identified in this AD, contact 328 Support Services GmbH, Global Support Center, P.O. Box 1252, D–82231 Wessling, Federal Republic of Germany; telephone +49 8153 88111 6666; fax +49 8153 88111 6565; email gsc.op@328support.de; internet <http://www.328support.de>. You may view this service information at the FAA, Transport Standards Branch, 2200 South 216th St., Des Moines, WA. For information on the availability of this material at the FAA, call 206–231–3195.

Issued in Des Moines, Washington, on May 24, 2018.

James Cashdollar,

Acting Director, System Oversight Division, Aircraft Certification Service.

[FR Doc. 2018–12135 Filed 6–6–18; 8:45 am]

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