(5) Deduct an amount equal to any PCC or NCA that the corporate credit union maintains at another corporate credit union;

(6) Deduct any amount of PCC received from federally insured credit unions that causes PCC minus retained earnings, all divided by moving daily average net assets, to exceed two percent when a corporate credit union’s retained earnings ratio is less than two and a half percent.

3. In Appendix B to part 704, in part I, revise paragraphs (b)(2) and (3) to read as follows:

Appendix B to Part 704—Expanded Authorities and Requirements

Part I

(b) * * * *

(2) 28 percent if the corporate credit union has a seven percent minimum leverage ratio and a two and a half percent retained earnings ratio, and is specifically approved by the NCUA; or

(3) 35 percent if the corporate credit union has an eight percent minimum leverage ratio and a three percent retained earnings ratio and is specifically approved by the NCUA.

* * * *

[FR Doc. 2017–25223 Filed 11–21–17; 8:45 am]

BILLING CODE 7535–01–P

BUREAU OF CONSUMER FINANCIAL PROTECTION

12 CFR Part 1040

[Docket No. CFPB–2016–0020]

RIN 3170–AA51

Arbitration Agreements

AGENCY: Bureau of Consumer Financial Protection.

ACTION: Final rule; CRA revocation.

SUMMARY: Under the Congressional Review Act, Congress has passed and the president has signed a joint resolution disapproving a final rule published by the Bureau of Consumer Financial Protection (Bureau) on July 19, 2017, to regulate arbitration agreements in contracts for specified consumer financial products and services. Under the joint resolution and by operation of the Congressional Review Act, the arbitration agreements rule has no force or effect. The Bureau is hereby removing it from the Code of Federal Regulations (CFR).

DATES: This action is effective November 22, 2017.
envisioned in the airworthiness standards for transport category airplanes. This design feature is an electronic flight-control system (EFCs) that provides control of the airplane through pilot inputs to the flight computer. The applicable airworthiness regulations do not contain adequate or appropriate safety standards for this design feature. These special conditions contain the additional safety standards that the Administrator considers necessary to establish a level of safety equivalent to that established by the existing airworthiness standards.

DATES: This action is effective on Mitsubishi on November 22, 2017. We must receive your comments January 8, 2018.

ADDRESSES: Send comments identified by docket number FAA–2017–0951 using any of the following methods:

- Federal eRegulations Portal: Go to http://www.regulations.gov/ and follow the online instructions for sending your comments electronically.
- Hand Delivery or Courier: Take comments to Docket Operations in Room W12–140 of the West Building Ground Floor at 1200 New Jersey Avenue SE., Washington, DC, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.
- Fax: Fax comments to Docket Operations at 202–493–2251.

Privacy: The FAA will post all comments it receives, without change, to http://www.regulations.gov/, including any personal information the commenter provides. Using the search function of the docket Web site, anyone can find and read the electronic form of all comments received into any FAA docket, including the name of the individual sending the comment (or signing the comment for an association, business, labor union, etc.). DOT’s complete Privacy Act Statement can be found in the Federal Register published on April 11, 2000 (65 FR 19477–19478).

Docket: Background documents or comments received may be read at http://www.regulations.gov/ at any time. Follow the online instructions for accessing the docket or go to Docket Operations in Room W12–140 of the West Building Ground Floor at 1200 New Jersey Avenue SE., Washington, DC, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.


SUPPLEMENTARY INFORMATION: The FAA has determined that notice of, and opportunity for prior public comment on, these special conditions is unnecessary because the substance of these special conditions has been subject to the public-comment process in several prior instances with no substantive comments received. The FAA finds good cause that prior notice and comment are unnecessary, and for the same reason finds that good cause exists for adopting these special conditions upon publication in the Federal Register.

Comments Invited
The FAA is requesting comments to allow interested persons to submit views that may not have been submitted in response to the prior opportunities for comment described above. We invite interested people to take part in this rulemaking by sending written comments, data, or views. The most helpful comments reference a specific portion of the special conditions, explain the reason for any recommended change, and include supporting data.

We will consider all comments we receive by the closing date for comments. We may change these special conditions based on the comments we receive.

Background
On August 19, 2009, Mitsubishi applied for a type certificate for their new Model MRJ–200 airplane. The Model MRJ–200 airplane is a low-wing, conventional-tail design with two wing-mounted turboprop engines. The airplane is equipped with an electronic flight-control system, has seating for 96 passengers and a maximum takeoff weight of 98,800 lbs.

Type Certification Basis
Under the provisions of title 14 Code of Federal Regulations (14 CFR) 21.17, Mitsubishi must show that the Model MRJ–200 airplane meets the applicable provisions of part 25, as amended by Amendments 25–1 through 25–141; part 36, as amended by Amendments 36–1 through 36–30; and part 34, as amended by Amendments 34–1 through the amendment effective at the time of design approval.

If the Administrator finds that the applicable airworthiness regulations (i.e., 14 CFR part 25) do not contain adequate or appropriate safety standards for the Model MRJ–200 airplane because of a novel or unusual design feature, special conditions are prescribed under the provisions of § 21.16.

Special conditions are initially applicable to the Model for which they are issued. Should the type certificate for that model be amended later to include any other model that incorporates the same novel or unusual design feature, these special conditions would also apply to the other model under § 21.101.

In addition to the applicable airworthiness regulations and special conditions, the model MRI–200 airplane must comply with the fuel-vent and exhaust-emission requirements of 14 CFR part 34, and the noise-certification requirements of 14 CFR part 36.

The FAA issues special conditions, as defined in 14 CFR 11.19, in accordance with § 11.38, and they become part of the type certification basis under § 21.17(a)(2).

Novel or Unusual Design Features
The Model MRI–200 airplane will incorporate the following novel or unusual design feature:

An electronic flight-control system that provides control of the airplane through pilot inputs to the flight computer. Current part 25 airworthiness regulations account for control laws where aileron deflection is proportional to control-stick deflection. They do not address any nonlinearities, i.e., situations where output does not change in the same proportion as input, or other effects on aileron actuation that may be caused by electronic flight controls.

Discussion
The flight-control system for the Model MRJ–200 airplane does not have a direct mechanical link, nor a linear gain, between the airplane flight-control surface and the pilot’s flight-deck control device, which is not accounted for in § 25.349(a). Instead, a flight-control computer commands the airplane flight-control surfaces, based on input received from the flight-deck control device. The flight-control computer modifies pilot input before the command is given to the flight-control surface.

These special conditions differ from current regulatory requirements in that they require that the roll maneuvers result from defined movements of the flight-deck roll control as opposed to defined aileron deflections. Also, these special conditions require an additional load condition at design maneuvering speed (V<sub>μ</sub>), in which the flight-deck roll
control is returned to neutral following the initial roll input. These special conditions differ from similar special conditions previously issued on this topic. These special conditions are limited to the roll axis only, whereas other special conditions also included pitch and yaw axes. Special conditions are no longer needed for the yaw axis because § 25.351 was revised at Amendment 25–91 to take into account effects of an electronic flight-control system. No special conditions are needed for the pitch axis because the method that Mitsubishi proposed for the pitch maneuver takes into account effects of an electronic flight-control system.

These special conditions contain the additional safety standards that the Administrator considers necessary to establish a level of safety equivalent to that established by the existing airworthiness standards.

**Applicability**

As discussed above, these special conditions are applicable to the Model MRJ–200 airplanes. Should Mitsubishi apply at a later date for a change to the type certificate to include another model incorporating the same novel or unusual design feature, these special conditions would apply to that model as well.

**Conclusion**

This action affects only certain novel or unusual design features on one model of airplanes. It is not a rule of general applicability.

**List of Subjects in 14 CFR Part 25**

Aircraft, Aviation safety, Reporting and recordkeeping requirements.

The authority citation for these special conditions is as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701, 44702, 44704.

**The Special Conditions**

Accordingly, pursuant to the authority delegated to me by the Administrator, the following special conditions are issued as part of the type certification basis for Mitsubishi Model MRJ–200 airplanes.

In lieu of compliance to 14 CFR 25.349(a), the following conditions, speeds, and flight-deck roll-control motions (except as the motions may be limited by pilot effort) must be considered in combination with an airplane load factor of zero, and of two-thirds of the positive maneuvering factor used in design. In determining the resulting control-surface deflections, the torsional flexibility of the wing must be considered in accordance with § 25.301(b).

(a) Conditions corresponding to steady rolling velocities must be investigated. In addition, conditions corresponding to maximum angular acceleration must be investigated for airplanes with engines or other weight concentrations outboard of the fuselage. For the angular acceleration conditions, zero rolling velocity may be assumed in the absence of a rational time-history investigation of the maneuver.

(b) At V_A, sudden movement of the flight-deck roll control up to the limit is assumed. The position of the flight-deck roll control must be maintained until a steady roll rate is achieved, and then must be returned suddenly to the neutral position.

(c) At V_C, the flight-deck roll control must be moved suddenly and maintained so as to achieve a roll rate not less than that obtained in special condition (b).

(d) At V_D, the flight-deck roll control must be moved suddenly and maintained so as to achieve a roll rate not less than one third of that obtained in special condition (b).

Vitor Wicklund,
Manager, Transport Standards Branch, Policy and Innovation Division, Aircraft Certification Service.

[FR Doc. 2017–25019 Filed 11–21–17; 8:45 am]

BILLING CODE 4910–13–P

**DEPARTMENT OF COMMERCE**

**National Oceanic and Atmospheric Administration**

**15 CFR Part 922**

[Docket No. 160413330–6330–01]

**RIN 0648–BF99**

Delay of Discharge Requirements for U.S. Coast Guard Activities in Greater Farallones and Cordell Bank National Marine Sanctuaries

**AGENCY:** Office of National Marine Sanctuaries (ONMS), National Ocean Service (NOS), National Oceanic and Atmospheric Administration (NOAA), Department of Commerce (DOC).

**ACTION:** Final rule; notification of delay of effectiveness for discharge requirements with regard to U.S. Coast Guard activities.

**SUMMARY:** The National Oceanic and Atmospheric Administration (NOAA) expanded the boundaries of Gulf of the Farallones National Marine Sanctuary (now renamed Greater Farallones National Marine Sanctuary or GFNMS) and Cordell Bank National Marine Sanctuary (CBNMS) to an area north and west of their previous boundaries with a final rule published on March 12, 2015. The final rule entered into effect on June 9, 2015. At that time, NOAA postponed, with regard to U.S. Coast Guard (USCG) activities, the effectiveness of the discharge requirements for six months in the regulations for both sanctuaries in the newly added areas. Since then, NOAA published four documents to extend the postponement of the discharge requirements to provide adequate time for completion of an environmental assessment, and subsequent rulemaking regarding USCG activities, as appropriate. The current extension would end on December 9, 2017. This document, published concurrently with a proposed rule to address discharges by the USCG and an environmental assessment, will extend the postponement of the discharge requirements for USCG activities in the expansion areas of GFNMS and CBNMS for one year beyond the end of the current extension to provide adequate time for completion, if appropriate, of a final environmental assessment and final rule. This extension will end on December 9, 2018, or 30 days after publication of a final rule, whichever comes first.

**DATES:** The effectiveness for the discharge requirements in both CBNMS and GFNMS expansion areas with regard to USCG activities is December 9, 2018.

**ADDRESSES:** Copies of documents relating to the expansion, including the Final Environmental Impact Statement (FEIS), final management plans, and the final rule published on March 12, 2015, can be viewed or downloaded at https://farallones.noaa.gov/manage/expansion_cbgf.html.

**FOR FURTHER INFORMATION CONTACT:** Maria Brown, Greater Farallones National Marine Sanctuary Superintendent, at Maria.Brown@noaa.gov or 415–561–6622.

**SUPPLEMENTARY INFORMATION:**

I. **Background**

On March 12, 2015, NOAA expanded the boundaries of Gulf of the Farallones National Marine Sanctuary (now renamed Greater Farallones National Marine Sanctuary or GFNMS) and Cordell Bank National Marine Sanctuary (CBNMS) to an area north and west of their previous boundaries with a final rule (80 FR 13078). The final rule entered into effect on June 9, 2015 (80 FR 34047). In the course of the rulemaking to expand GFNMS and