paragraph (h) of this AD have been accomplished on the pre-Airbus Modification 204615 part.

(l) Credit for Previous Actions

This paragraph provides credit for actions required by paragraph (h) of this AD, if those actions were performed before the effective date of this AD using Airbus Service Bulletin A330–71–3025, dated January 10, 2011, which was incorporated by reference in AD 2012–22–12; or Airbus Service Bulletin A330–71–3025, Revision 01, dated October 24, 2012, which is not incorporated by reference in this AD; provided that, within 1,050 flight cycles after the effective date of this AD, a special detailed inspection for pulled, loose, and missing attachment rivets of the IBAs and OBAs of the forward bulkhead is accomplished; and all applicable corrective actions are done; in accordance with the Accomplishment Instructions of Airbus Service Bulletin A330–71–3033, dated December 14, 2015. Do all applicable corrective actions before further flight. Accomplishment of corrective actions does not constitute terminating action for the repetitive inspections required by paragraph (h) of this AD.

(m) Other FAA AD Provisions

The following provisions also apply to this AD:

(1) Alternative Methods of Compliance (AMOCs): The Manager, International Branch, ANM–116, Transport Airplane Directorate, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the International Branch, send it to ATTN: Vladimir Ulyanov, Aerospace Engineer, International Branch, ANM–116, Transport Airplane Directorate, FAA, 1601 Lind Avenue SW., Renton, WA 98057–3556; telephone 425–227–1138; fax 425–227–1149. Information may be emailed to: 9-ANM-116-AMOC-REQUESTS@faa.gov.

(i) Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/ certification holding district office.

(ii) AMOCs approved previously for AD 2012–22–12 are not approved as AMOCs with this AD.

(2) Contacting the Manufacturer: As of the effective date of this AD, for any requirement in this AD to obtain corrective actions from a manufacturer, the action must be accomplished using a method approved by the Manager, International Branch, ANM–116, Transport Airplane Directorate, FAA; or the EASA; or Airbus’s EASA DOA. If approved by the DOA, the approval must include the DOA authorized signature.

(3) Required for Compliance (RC): If any service information contains procedures or tests that are identified as RC, those procedures and tests must be done to comply with this AD; any procedures or tests that are not identified as RC are recommended. Those procedures and tests that are not identified as RC may be deviated from using accepted methods in accordance with the operator’s maintenance or inspection program without obtaining approval of an AMOC, provided the procedures and tests identified as RC can be done and the airplane can be put back in an airworthy condition. Any substitutions or changes to procedures or tests identified as RC require approval of an AMOC.

(n) Related Information

(1) Refer to Mandatory Continuing Airworthiness Information (MCAI) EASA AD 2016–0086R1, dated May 13, 2016, for related information. This MCAI may be found in the AD docket on the Internet at http://www.regulations.gov by searching for and locating Docket No. FAA–2016–9305.

(2) Service information identified in this AD that is not incorporated by reference is available at the addresses specified in paragraphs (o)(3) and (o)(5) of this AD.

(o) Material Incorporated by Reference

(1) The Director of the Federal Register approved the incorporation by reference (IBR) of the service information listed in this paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) You must use this service information as applicable to do the actions required by this AD, unless this AD specifies otherwise.


(3) For Airbus service information identified in this AD, contact Airbus SAS—Airworthiness Office—EAL, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France; telephone +33 5 61 93 45 80; email airworthiness.A330-A340@airbus.com; Internet http://www.airbus.com.

(4) For Rolls-Royce service information identified in this AD, contact Rolls-Royce Plc, Technical Publications, P.O. Box 31, Derby, DE24 8BJ, United Kingdom; telephone 44 (0) 1332 245882; fax 44 (0) 1332 249936; Internet http://www.Rolls-Royce.com.

(5) You may view this service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, WA. For information on the availability of this material at the FAA, call 425–227–1221.

(6) You may view this service information that is incorporated by reference at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call 202–741–6036, or go to: http://www.archives.gov/federal-register/cfr/ibr-locations.html.

Issued in Renton, Washington, on February 3, 2017.

Michael Kazyscki,
Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 2017–03260 Filed 2–21–17; 8:45 am]

BILLING CODE 4910–13–P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39


RIN 2120–AA64

Airworthiness Directives; Textron Aviation Inc. (Type Certificate Previously Held by Cessna Aircraft Company) Airplanes

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final rule; request for comments.

SUMMARY: We are adopting a new airworthiness directive (AD) for certain Textron Aviation Inc. Model 560XL airplanes. This AD requires inspections of the fuel tube and right alternating current (AC) generator wires. This AD was prompted by reports of inadequate separation between the electrical wire bundle and fuel tube. We are issuing this AD to address the unsafe condition on these products.

DATES: This AD is effective March 9, 2017.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in this AD as of March 9, 2017.

We must receive comments on this AD by April 10, 2017.

ADDRESSES: You may send comments, using the procedures found in 14 CFR 11.43 and 11.45, by any of the following methods:

• Federal eRulemaking Portal: Go to http://www.regulations.gov. Follow the instructions for submitting comments.

• Fax: 202–493–2251.


• Hand Delivery: Deliver to Mail address above between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

For service information identified in this final rule, contact Textron Aviation Inc., P.O. Box 7706, Wichita, KS 67277; telephone 316–517–6215; fax 316–517–5802; email citationpubs@ttxav.com; Internet https://support.cessna.com/custsupport/cssupport/newlogin.jsp. You may view this referenced service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, WA. For information on the availability of this material at the

Examining the AD Docket
You may examine the AD docket on the Internet at http://www.regulations.gov by searching for and locating Docket No. FAA–2017–0122; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this AD, the regulatory evaluation, any comments received, and other information. The street address for the Docket Office (phone: 800–647–5527) is in the ADDRESSES section. Comments will be available in the AD docket shortly after receipt.

FOR FURTHER INFORMATION CONTACT:
Craig Henrichsen, Aerospace Engineer, Electrical Systems and Avionics, ACE–119W, FAA, Wichita Aircraft Certification Office (ACO), 1801 Airport Road, Room 100, Dwight D. Eisenhower Airport, Wichita, KS 67209; phone: 316–946–4110; fax: 316–946–4107; email: Wichita-COS@faa.gov.

SUPPLEMENTARY INFORMATION:
Discussion
We have been notified by Textron Aviation Inc. of a windshield heat failure reported by a flight crew. During follow-on troubleshooting activities, the maintenance crew discovered an open circuit breaker in the tail cone area after finding the AC generator electrical wires chafed against a motive flow fuel tube that was leaking. Additional inspections by the operator found two other airplanes with chafing damage to the wires and fuel tube. Textron Aviation Inc. also found evidence of damage to the electrical wire bundle on two other airplanes. Further inspections have found inadequate separation between the electrical wire bundle and fuel tube on 19 airplanes in service. The airplane quality inspection failed to identify the wire separation problem prior to airplane delivery. Inadequate separation between the fuel tube and electrical wire bundle, if not corrected, could result in chafing, a possible fuel leak, and electrical arcing, which could lead to a fuel ignition source and possible uncontrolled fire in the tail cone of the airplane.

Related Service Information Under 1 CFR Part 51
We reviewed Textron Aviation Service Letter SL560XL–24–07, including Attachment, dated January 13, 2017. The service information describes procedures for inspections of the fuel tube and right AC generator wires, applicable adjustments, and corrective actions. This service information is reasonably available because the interested parties have access to it through their normal course of business or by the means identified in the ADDRESSES section.

FAA’s Determination
We are issuing this AD because we evaluated all the relevant information and determined the unsafe condition described previously is likely to exist or develop in other products of the same type design.

AD Requirements
This AD requires accomplishing the actions specified in the service information described previously. This AD also requires sending the inspection reports to Textron Aviation Inc. For information on the procedures and compliance times, see this service information at http://www.regulations.gov by searching for and locating Docket No. FAA–2017–0122.

The phrase “related investigative actions” is used in this AD. Related investigative actions are follow-on actions that (1) are related to the primary action, and (2) further investigate the nature of any condition found.

The phrase “corrective actions” is used in this AD. Corrective actions correct or address any condition found.

Interim Action
This AD is considered to be interim action. The inspection reports that are required by this AD will enable the manufacturer to obtain better insight into the nature, cause, and extent of the damage, and eventually to develop final action to address the unsafe condition. Once final action has been identified, we might consider further rulemaking.

FAA’s Justification and Determination of the Effective Date
An unsafe condition exists that requires the immediate adoption of this AD. The FAA has found that the risk to the flying public justifies waiving notice and comment prior to adoption of this rule because inadequate separation between the electrical wire bundle and fuel tube could result in chafing, electrical arcing, and a possible fuel leak, leading to a fuel ignition source and possible uncontrolled fire in the tail cone of the airplane. Therefore, we find that notice and opportunity for prior public comment are impracticable and that good cause exists for making this amendment effective in less than 30 days.

Comments Invited
This AD is a final rule that involves requirements affecting flight safety and was not preceded by notice and an opportunity for public comment. However, we invite you to send any written data, views, or arguments about this AD. Send your comments to an address listed under the ADDRESSES section. Include the docket number FAA–2017–0122 and Directorate Identifier 2017–NM–010–AD at the beginning of your comments. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of this AD. We will consider all comments received by the closing date and may amend this AD because of those comments.

We will post all comments we receive, without change, to http://www.regulations.gov, including any personal information you provide. We will also post a report summarizing each substantive verbal contact we receive about this AD.

Costs of Compliance
We estimate that this AD affects 639 airplanes of U.S. registry. We estimate the following costs to comply with this AD:

<table>
<thead>
<tr>
<th>Action</th>
<th>Labor cost</th>
<th>Parts cost</th>
<th>Cost per product</th>
<th>Cost on U.S. operators</th>
</tr>
</thead>
<tbody>
<tr>
<td>Inspections</td>
<td>1 work-hour × $85</td>
<td>$0</td>
<td>$85</td>
<td>$54,315</td>
</tr>
<tr>
<td>Reporting</td>
<td>$85</td>
<td>85</td>
<td>$54,315</td>
<td></td>
</tr>
</tbody>
</table>
We estimate the following costs to do any necessary repairs or replacements that would be required based on the results of the inspection. We have no way of determining the number of aircraft that might need these repairs or replacements:

### ON-CONDITION COSTS

<table>
<thead>
<tr>
<th>Action</th>
<th>Labor cost</th>
<th>Parts cost</th>
<th>Cost per product</th>
</tr>
</thead>
<tbody>
<tr>
<td>Fuel tube repair or replacement</td>
<td>7 work-hours × $85 per hour = $595</td>
<td>$321</td>
<td>$916</td>
</tr>
<tr>
<td>Electrical wiring replacement</td>
<td>8 work-hours × $85 per hour = $680</td>
<td>1/4</td>
<td>1684</td>
</tr>
</tbody>
</table>

1These figures are based on 1 foot of wire at $4 per foot.

**Paperwork Reduction Act**

A federal agency may not conduct or sponsor, and a person is not required to respond to, nor shall a person be subject to penalty for failure to comply with a collection of information subject to the requirements of the Paperwork Reduction Act unless that collection of information displays a current valid OMB control number. The control number for the collection of information required by this AD is 2120–0056. The paperwork cost associated with this AD has been detailed in the Costs of Compliance section of this document and includes time for reviewing instructions, as well as completing and reviewing the collection of information. Therefore, all reporting associated with this AD is mandatory. Comments concerning the accuracy of this burden and suggestions for reducing the burden should be directed to the FAA at 800 Independence Ave. SW., Washington, DC 20591. ATTN: Information Collection Clearance Officer, AES–200.

**Authority for This Rulemaking**

Title 49 of the United States Code specifies the FAA’s authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. “Subtitle VII: Aviation Programs” describes in more detail the scope of the Agency’s authority.

We are issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section 44701: “General requirements.” Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

**Regulatory Findings**

This AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that this AD:

1. Is not a “significant regulatory action” under Executive Order 12866,
2. Is not a “significant rule” under DOT Regulatory Policies and Procedures (49 FR 11384, February 26, 1979),
3. Will not affect intrastate aviation in Alaska, and
4. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

**List of Subjects in 14 CFR Part 39**

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

**Adoption of the Amendment**

Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

**PART 39—AIRWORTHINESS DIRECTIVES**

§ 39.13 [Amended]

1. The authority citation for part 39 continues to read as follows:

   Authority: 49 U.S.C. 106(g), 40113, 44701.

2. The FAA amends § 39.13 by adding the following new airworthiness directive (AD):


   **2017–04–14**

   **(a) Effective Date**

   This AD is effective March 9, 2017.

   **(b) Affected ADs**

   None.

   **(c) Applicability**

   This AD applies to Textron Aviation Inc. (Type Certificate previously held by Cessna Aircraft Company) Model 560XL airplanes, certificated in any category, as identified in Textron Aviation Service Letter SL560XL–24–07, dated January 13, 2017.

   **(d) Subject**

   Air Transport Association (ATA) of America Code 24, Electrical power.

   **(e) Unsafe Condition**

   This AD was prompted by reports of inadequate separation between the electrical wire bundle and fuel tube. We are issuing this AD to detect and correct inadequate separation and consequent chafing, which could result in electrical arcing and a fuel leak, leading to a fuel ignition source and possible uncontrolled fire in the tail cone of the airplane.

   **(f) Compliance**

   Comply with this AD within the compliance times specified, unless already done.

   **(g) Inspections, Adjustments, and Corrective Actions**

   Within 25 flight hours or 180 days after the effective date of this AD, whichever occurs first, do general visual inspections of the fuel tube and right alternating current (AC) generator wires for evidence of damage, do all applicable adjustments, and do all applicable corrective actions, in accordance with the Accomplishment Instructions of Textron Aviation Service Letter SL560XL–24–07, including Attachment, dated January 13, 2017. All applicable adjustments and corrective actions must be done before further flight.

   **(b) Reporting**

   Submit a report of the findings (both positive and negative) of the inspections required by paragraph (g) of this AD to Textron Aviation Inc. as specified in Textron Aviation Service Letter SL560XL–24–07, including Attachment, dated January 13, 2017, at the applicable time specified in paragraph (h)(1) or (h)(2) of this AD.

   1. If the inspection was done on or after the effective date of this AD: Submit the report within 10 days after the inspection.
   2. If the inspection was done before the effective date of this AD: Submit the report within 10 days after the effective date of this AD.
SUMMARY: We are superseding Airworthiness Directive (AD) 2012–16–08 for certain BAE Systems (Operations) Limited Model BAe 146 and Avro 146–R series airplanes. AD 2012–16–08 required repetitive inspections for bulging, surface anomalies, and cracking of the fuselage skin adjacent to the discharge valves, and repair and application of additional sealant in the affected area if necessary. This new AD retains the requirements of AD 2012–16–08, expands the applicability, and requires an additional one-time inspection for the presence of water traps/air driers to determine which airplanes must be inspected. This AD was prompted by reports of cracking and surface anomalies of the fuselage skin at the water trap/air drier unit of the forward discharge valve due to corrosion, and the determination that airplanes on which a certain auto-pressurization modification was incorporated during production were also affected by this condition. We are issuing this AD to address the unsafe condition on these products.

DATES: This AD is effective March 29, 2017.

The Director of the Federal Register approved the incorporation by reference of certain publications listed in this AD as of March 29, 2017.

The Director of the Federal Register approved the incorporation by reference of certain other publications listed in this AD as of September 18, 2012 (77 FR 48420, August 14, 2012).

ADDRESSES: For service information identified in this final rule, contact BAE Systems (Operations) Limited, Customer Information Department, Prestwick International Airport, Ayrshire, KA9 1RW, Scotland, United Kingdom; telephone +44 1292 675207; fax +44 1292 675704; email BAepublications@baesystems.com; Internet http://www.baesystems.com/Businesses/RegionalAircraft/index.htm. You may view this referenced service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, WA. For information on the availability of this material at the FAA, call 425–227–1221. It is also available on the Internet at http://www.regulations.gov by searching for and locating Docket No. FAA–2016–9186.

Examining the AD Docket
You may examine the AD docket on the Internet at http://www.regulations.gov by searching for and locating Docket No. FAA–2016–9186; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this AD, the regulatory evaluation, any comments received, and other information. The address for the Docket Office (telephone 800–647–5527) is Docket Management Facility, U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue SE., Washington, DC 20590.

FOR FURTHER INFORMATION CONTACT:

SUPPLEMENTARY INFORMATION:
Discussion
We issued a notice of proposed rulemaking (NPRM) to amend 14 CFR