

Proposed Rules

Federal Register

Vol. 82, No. 2

Wednesday, January 4, 2017

This section of the FEDERAL REGISTER contains notices to the public of the proposed issuance of rules and regulations. The purpose of these notices is to give interested persons an opportunity to participate in the rule making prior to the adoption of the final rules.

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA-2016-9524; Directorate Identifier 2016-NM-049-AD]

RIN 2120-AA64

Airworthiness Directives; Airbus Airplanes

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: We propose to supersede Airworthiness Directive (AD) 2014-16-19, for all Airbus Model A330-200 Freighter, -200, and -300 series airplanes. AD 2014-16-19 currently requires revision of the maintenance or inspection program to include certain fuel airworthiness limitations. Since we issued AD 2014-16-19, Airbus has issued more restrictive fuel airworthiness limitations. This proposed AD would require revision of the maintenance or inspection program, as applicable, to include new fuel airworthiness limitations. The proposed AD also removes certain airplanes from the applicability of AD 2014-16-19. We are proposing this AD to address the unsafe condition on these products.

DATES: We must receive comments on this proposed AD by February 21, 2017.

ADDRESSES: You may send comments, using the procedures found in 14 CFR 11.43 and 11.45, by any of the following methods:

- *Federal eRulemaking Portal:* Go to <http://www.regulations.gov>. Follow the instructions for submitting comments.
- *Fax:* 202-493-2251.
- *Mail:* U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE., Washington, DC 20590.
- *Hand Delivery:* Deliver to Mail address above between 9 a.m. and 5

p.m., Monday through Friday, except Federal holidays.

For service information identified in this NPRM, contact Airbus SAS, Airworthiness Office—EAL, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France; telephone: +33 5 61 93 36 96; fax: +33 5 61 93 45 80; email: airworthiness.A330-A340@airbus.com; Internet: <http://www.airbus.com>. You may view this referenced service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, WA. For information on the availability of this material at the FAA, call 425-227-1221.

Examining the AD Docket

You may examine the AD docket on the Internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA-2016-9524; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this proposed AD, the regulatory evaluation, any comments received, and other information. The street address for the Docket Operations office (telephone: 800-647-5527) is in the **ADDRESSES** section. Comments will be available in the AD docket shortly after receipt.

FOR FURTHER INFORMATION CONTACT: Vladimir Ulyanov, Aerospace Engineer, International Branch, ANM-116, Transport Airplane Directorate, FAA, 1601 Lind Avenue SW., Renton, WA 98057-3356; telephone: 425-227-1138; fax: 425-227-1149.

SUPPLEMENTARY INFORMATION:

Comments Invited

We invite you to send any written relevant data, views, or arguments about this proposed AD. Send your comments to an address listed under the **ADDRESSES** section. Include “Docket No. FAA-2016-9524; Directorate Identifier 2016-NM-049-AD” at the beginning of your comments. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of this proposed AD. We will consider all comments received by the closing date and may amend this proposed AD based on those comments.

We will post all comments we receive, without change, to <http://www.regulations.gov>, including any personal information you provide. We

will also post a report summarizing each substantive verbal contact we receive about this proposed AD.

Discussion

On August 4, 2014, we issued AD 2014-16-19, Amendment 39-17943 (79 FR 49449, August 21, 2014) (“AD 2014-16-19”). AD 2014-16-19 requires actions intended to correct an unsafe condition for all Airbus Model A330-200 Freighter, -200, and -300 series airplanes.

Since we issued AD 2014-16-19, Airbus has issued more restrictive fuel airworthiness limitations.

The European Aviation Safety Agency (EASA), which is the Technical Agent for the Member States of the European Union, has issued EASA AD 2016-0065, dated April 5, 2016 (referred to after this as the Mandatory Continuing Airworthiness Information, or “the MCAI”), to correct an unsafe condition for all Airbus Model A330-200 Freighter series airplanes, Model A330-200 series airplanes, Model A330-300 series airplanes; and Model A340-200 series airplanes, Model A340-300 series airplanes, Model A340-500 series airplanes, and Model A340-600 series airplanes. The MCAI states:

Prompted by an accident * * *, the Federal Aviation Authority (FAA) published Special Federal Aviation Regulation (SFAR) 88, and the Joint Aviation Authorities (JAA) published Interim Policy INT/POL/25/12. A design review was conducted by Airbus to develop Fuel Airworthiness Limitations (FAL) for Airbus A330 and A340 aeroplanes in response to these regulations.

The FAL, which are approved by EASA, are defined and published in Airbus A330 and A340 Airworthiness Limitations Section (ALS) documents known as Part 5. Failure to comply with these instructions could result in a fuel tank explosion and consequent loss of the aeroplane.

EASA issued AD 2012-0168 [which corresponds with FAA AD 2014-16-19 for Model A330 airplanes, and FAA AD 2013-26-03, Amendment 39-17712 (78 FR 79292, December 30, 2013) for Model A340 airplanes] to require compliance with the FAL as specified in the A330 and A340 ALS Part 5 Revision 00.

Since that [EASA] AD was issued, Airbus issued Revision 01 of both ALS Parts 5 for Airbus A330 and A340 to introduce more restrictive maintenance requirements and/or airworthiness limitations.

For the reason described above, this [EASA] AD retains the requirements of EASA AD 2012-0168, which is superseded, and requires accomplishment of the actions

specified in Airbus A330 ALS Part 5 Revision 01, A340 ALS Part 5 Revision 01, as applicable (hereafter collectively referred to as 'the ALS' in this AD).

You may examine the MCAI in the AD docket on the Internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA–2016–9524.

This proposed AD would require revisions to certain operator maintenance documents to include new actions (e.g., inspections) and critical design configuration control limitations (CDCCLs). Compliance with these actions and CDCCLs is required by 14 CFR 91.403(c). For airplanes that have been previously modified, altered, or repaired in the areas addressed by this proposed AD, the operator may not be able to accomplish the actions described in the revisions. In this situation, to comply with 14 CFR 91.403(c), the operator must request approval for an alternative method of compliance according to paragraph (l)(1) of this proposed AD. The request should include a description of changes to the required actions that will ensure the continued damage tolerance of the affected structure.

Related Service Information Under 14 CFR Part 51

Airbus has issued Airbus A330 Airworthiness Limitations Section (ALS) Part 5—Fuel Airworthiness Limitations (FAL), Revision 01, dated October 28, 2015. The airworthiness limitations introduce more restrictive fuel airworthiness limitations. This service information is reasonably available because the interested parties have access to it through their normal course of business or by the means identified in the **ADDRESSES** section.

FAA's Determination and Requirements of This Proposed AD

This product has been approved by the aviation authority of another country, and is approved for operation in the United States. Pursuant to our bilateral agreement with the State of Design Authority, we have been notified of the unsafe condition described in the MCAI and service information referenced above. We are proposing this AD because we evaluated all pertinent information and determined an unsafe condition exists and is likely to exist or develop on other products of these same type designs.

Differences Between This Proposed AD and the MCAI or Service Information

This proposed AD does not include the Airbus Model A340 airplanes that are specified in the MCAI. We have

added the MCAI to the required airworthiness actions list (RAAL) for the Model A340 airplanes.

The MCAI specifies that if there are findings from the ALS inspection tasks, corrective actions must be accomplished in accordance with Airbus maintenance documentation. However, this proposed AD does not include that requirement. Operators of U.S.-registered airplanes are required by general airworthiness and operational regulations to perform maintenance using methods that are acceptable to the FAA. We consider those methods to be adequate to address any corrective actions necessitated by the findings of ALS inspections required by this proposed AD.

Airworthiness Limitations Based on Type Design

The FAA recently became aware of an issue related to the applicability of ADs that require incorporation of an ALS revision into an operator's maintenance or inspection program.

Typically, when these types of ADs are issued by civil aviation authorities of other countries, they apply to all airplanes covered under an identified type certificate (TC). The corresponding FAA AD typically retains applicability to all of those airplanes.

In addition, U.S. operators must operate their airplanes in an airworthy condition, in accordance with 14 CFR 91.7(a). Included in this obligation is the requirement to perform any maintenance or inspections specified in the ALS, and in accordance with the ALS as specified in 14 CFR 43.16 and 91.403(c), unless an alternative has been approved by the FAA.

When a type certificate is issued for a type design, the specific ALS, including revisions, is a part of that type design, as specified in 14 CFR 21.31(c).

The sum effect of these operational and maintenance requirements is an obligation to comply with the ALS defined in the type design referenced in the manufacturer's conformity statement. This obligation may introduce a conflict with an AD that requires a specific ALS revision if new airplanes are delivered with a later revision as part of their type design.

To address this conflict, the FAA has approved alternative methods of compliance (AMOCs) that allow operators to incorporate the most recent ALS revision into their maintenance/inspection programs, in lieu of the ALS revision required by the AD. This eliminates the conflict and enables the operator to comply with both the AD and the type design.

However, compliance with AMOCs is normally optional, and we recently

became aware that some operators choose to retain the AD-mandated ALS revision in their fleet-wide maintenance/inspection programs, including those for new airplanes delivered with later ALS revisions, to help standardize the maintenance of the fleet. To ensure that operators comply with the applicable ALS revision for newly delivered airplanes containing a later revision than that specified in an AD, we plan to limit the applicability of ADs that mandate ALS revisions to those airplanes that are subject to an earlier revision of the ALS, either as part of the type design or as mandated by an earlier AD.

This proposed AD therefore would apply to Model A330 series airplanes with an original certificate of airworthiness or original export certificate of airworthiness that was issued on or before the date of approval of the ALS revision identified in this proposed AD. Operators of airplanes with an original certificate of airworthiness or original export certificate of airworthiness issued after that date must comply with the airworthiness limitations specified as part of the approved type design and referenced on the type certificate data sheet.

Costs of Compliance

We estimate that this proposed AD affects 104 airplanes of U.S. registry.

The actions required by AD 2014–16–19, and retained in this proposed AD take about 1 work-hour per product, at an average labor rate of \$85 per work-hour. Based on these figures, the estimated cost of the actions that are required by AD 2014–16–19 is \$85 per product.

We also estimate that it would take about 1 work-hour per product to comply with the basic requirements of this proposed AD. The average labor rate is \$85 per work-hour. Based on these figures, we estimate the cost of this proposed AD on U.S. operators to be \$8,840, or \$85 per product.

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. "Subtitle VII: Aviation Programs," describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in "Subtitle VII, Part A, Subpart III, Section 44701: General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in

air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

We determined that this proposed AD would not have federalism implications under Executive Order 13132. This proposed AD would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify this proposed regulation:

1. Is not a "significant regulatory action" under Executive Order 12866;
2. Is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979);
3. Will not affect intrastate aviation in Alaska; and
4. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

The Proposed Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA proposes to amend 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

- 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

- 2. The FAA amends § 39.13 by removing Airworthiness Directive (AD) 2014–16–19, Amendment 39–17943 (79 FR 49449, August 21, 2014), and adding the following new AD:

Airbus: Docket No. FAA–2016–9524; Directorate Identifier 2016–NM–049–AD.

(a) Comments Due Date

We must receive comments by February 21, 2017.

(b) Affected ADs

This AD replaces AD 2014–16–19, Amendment 39–17943 (79 FR 49449, August 21, 2014) ("AD 2014–16–19").

(c) Applicability

This AD applies to the Airbus airplanes identified in paragraphs (c)(1) through (c)(3) of this AD, certificated in any category, with an original certificate of airworthiness or original export certificate of airworthiness issued on or before October 28, 2015.

(1) Airbus Model A330–223F and –243F airplanes.

(2) Airbus Model A330–201, –202, –203, –223, and –243 airplanes.

(3) Airbus Model A330–301, –302, –303, –321, –322, –323, –341, –342, and –343 airplanes.

(d) Subject

Air Transport Association (ATA) of America Code 05, Time Limits/Maintenance Checks.

(e) Reason

This AD was prompted by the issuance of more restrictive fuel airworthiness limitations. We are issuing this AD to prevent the potential of ignition sources inside fuel tanks, which, in combination with flammable fuel vapors, could result in fuel tank explosions and consequent loss of the airplane.

(f) Compliance

Comply with this AD within the compliance times specified, unless already done.

(g) Retained Maintenance Program Revision and Airworthiness Limitations Compliance, With No Changes

This paragraph restates the requirements of paragraph (g) of AD 2014–16–19, with no changes.

(1) Within 3 months after September 25, 2014 (the effective date of AD 2014–16–19), revise the maintenance or inspection program, as applicable, by incorporating Airbus A330 Airworthiness Limitations Section (ALS) Part 5—Fuel Airworthiness Limitations (FAL), dated November 16, 2011.

(2) Comply with all applicable instructions and airworthiness limitations included in Airbus A330 ALS Part 5—FAL, dated November 16, 2011. The initial compliance times for the actions specified in Airbus A330 ALS Part 5—FAL, dated November 16, 2011, are at the later of the times specified in paragraphs (g)(2)(i) and (g)(2)(ii) of this AD, except as required by paragraphs (h) and (i) of this AD.

(i) Within the applicable compliance times specified in Airbus A330 ALS Part 5—FAL, dated November 16, 2011.

(ii) Within 3 months after accomplishing the actions required by paragraph (g)(1) of this AD.

(h) Retained Exceptions to Compliance Times for Design Changes, With No Changes

This paragraph restates the exceptions specified in paragraph (h) of AD 2014–16–19, with no changes.

(1) For type design changes specified in "Sub-part 5–2 Changes to Type Design," of Airbus A330 ALS Part 5—FAL, dated November 16, 2011, the compliance times are defined as "Embodiment Limits," except as defined in paragraph (h)(2) of this AD.

(2) Where Airbus A330 ALS Part 5—FAL, dated November 16, 2011, specifies a compliance time based on a calendar date for modifying the control circuit for the fuel pump of the center fuel tank (installing ground fault interrupters to the center tank fuel pump control circuit), the compliance date is September 18, 2016 (48 months after the effective date of AD 2012–16–05, Amendment 39–17152 (77 FR 48425, August 14, 2012)).

(i) Retained No Alternative Actions, Intervals, or Critical Design Configuration Control Limitations (CDCCLs), With Added Exception

This paragraph restates the requirements of paragraph (i) of AD 2014–16–19, with an added exception. Except as required by paragraph (j) of this AD: After accomplishing the revision required by paragraph (g)(1) of this AD, no alternative actions (e.g., inspections), intervals, or CDCCLs may be used; except as specified in paragraph (h) of this AD; or unless the actions, intervals, or CDCCLs are approved as an alternative method of compliance (AMOC) in accordance with the procedures specified in paragraph (l)(1) of this AD.

(j) New Requirement of This AD: Revise the Maintenance or Inspection Program

Within 3 months after the effective date of this AD, revise the maintenance or inspection program, as applicable, to incorporate Airbus A330 ALS Part 5—FAL, Revision 01, dated October 28, 2015. The compliance times for accomplishing the initial tasks specified in Airbus A330 ALS Part 5—FAL, Revision 01, dated October 28, 2015, or within 3 months after revising the maintenance or inspection program as required by paragraph (j) of this AD, whichever occurs later. Accomplishing the revision required by this paragraph terminates the actions required by paragraph (g) of this AD.

(k) New Requirement of This AD: No Alternative Actions, Intervals, or CDCCLs

After accomplishing the revision required by paragraph (j) of this AD, no alternative actions (e.g., inspections), intervals, or CDCCLs may be used unless the actions, intervals, or CDCCLs are approved as an alternative method of compliance (AMOC) in accordance with the procedures specified in paragraph (l)(1) of this AD.

(l) Other FAA AD Provisions

The following provisions also apply to this AD:

(1) *Alternative Methods of Compliance (AMOCs):* The Manager, International Branch, ANM–116, Transport Airplane Directorate, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the International Branch, send it to ATTN: Vladimir Ulyanov, Aerospace Engineer, International Branch, ANM–116, Transport

Airplane Directorate, FAA, 1601 Lind Avenue SW., Renton, WA 98057-3356; telephone: 425-227-1138; fax: 425-227-1149. Information may be emailed to: *9-ANM-116-AMOC-REQUESTS@faa.gov*. Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office.

(2) *Contacting the Manufacturer:* As of the effective date of this AD, for any requirement in this AD to obtain corrective actions from a manufacturer, the action must be accomplished using a method approved by the Manager, International Branch, ANM-116, Transport Airplane Directorate, FAA; or the European Aviation Safety Agency (EASA); or Airbus's EASA Design Organization Approval (DOA). If approved by the DOA, the approval must include the DOA-authorized signature.

(m) Related Information

(1) Refer to Mandatory Continuing Airworthiness Information (MCAI) EASA AD 2016-0065, dated April 5, 2016, for related information. This MCAI may be found in the AD docket on the Internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA-2016-9524.

(2) For service information identified in this AD, contact Airbus SAS, Airworthiness Office—EAL, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France; telephone: +33 5 61 93 36 96; fax: +33 5 61 93 45 80; email: *airworthiness.A330-A340@airbus.com*; Internet: <http://www.airbus.com>. You may view this service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, WA. For information on the availability of this material at the FAA, call 425-227-1221.

Issued in Renton, Washington, on December 16, 2016.

Ross Landes,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 2016-31366 Filed 1-3-17; 8:45 am]

BILLING CODE 4910-13-P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA-2016-9522; Directorate Identifier 2016-NM-144-AD]

RIN 2120-AA64

Airworthiness Directives; Gulfstream Aerospace Corporation Airplanes

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: We propose to supersede Airworthiness Directive (AD) 2009-17-01, which applies to certain Gulfstream Model G-IV, GIV-X, GV-SP airplanes

and Model GV airplanes. AD 2009-17-01 currently requires, an inspection for sealant applied to the exterior of the auxiliary power unit (APU) enclosure (firewall), and a revision of the airplane flight manual (AFM), as applicable. Since we issued AD 2009-17-01, we received a report indicating that the type design sealant applied to the APU enclosure is flammable and failed certain tests. This proposed AD would require revising the AFM, and revising the applicability to include additional airplanes. We are proposing this AD to address the unsafe condition on these products.

DATES: We must receive comments on this proposed AD by February 21, 2017.

ADDRESSES: You may send comments, using the procedures found in 14 CFR 11.43 and 11.45, by any of the following methods:

- *Federal eRulemaking Portal:* Go to <http://www.regulations.gov>. Follow the instructions for submitting comments.
- *Fax:* 202-493-2251.
- *Mail:* U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE., Washington, DC 20590.
- *Hand Delivery:* Deliver to Mail address above between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

For service information identified in this NPRM, contact Gulfstream Aerospace Corporation, Technical Publications Dept., P.O. Box 2206, Savannah, GA 31402-2206; telephone 800-810-4853; fax 912-965-3520; email *pubs@gulfstream.com*; Internet http://www.gulfstream.com/product_support/technical_pubs/pubs/index.htm. You may view this referenced service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, WA. For information on the availability of this material at the FAA, call 425-227-1221.

Examining the AD Docket

You may examine the AD docket on the Internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA-2016-9522; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this proposed AD, the regulatory evaluation, any comments received, and other information. The street address for the Docket Office (phone: 800-647-5527) is in the **ADDRESSES** section. Comments will be available in the AD docket shortly after receipt.

FOR FURTHER INFORMATION CONTACT: Ky Phan, Aerospace Engineer, Propulsion and Services Branch, ACE-118A, FAA, Atlanta Aircraft Certification Office (ACO) 1701 Columbia Avenue, College Park, GA 30337; phone: 404-474-5536; fax: 404-474-5606; email: *ky.phan@faa.gov*.

SUPPLEMENTARY INFORMATION:

Comments Invited

We invite you to send any written relevant data, views, or arguments about this proposed AD. Send your comments to an address listed under the **ADDRESSES** section. Include "Docket No. FAA-2016-9522; Directorate Identifier 2016-NM-144-AD" at the beginning of your comments. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of this proposed AD. We will consider all comments received by the closing date and may amend this proposed AD because of those comments.

We will post all comments we receive, without change, to <http://www.regulations.gov>, including any personal information you provide. We will also post a report summarizing each substantive verbal contact we receive about this proposed AD.

Discussion

On July 31, 2009, we issued AD 2009-17-01, Amendment 39-15991 (74 FR 40061, August 11, 2009) ("AD 2009-17-01"), for certain Gulfstream Model G-IV, GIV-X, GV-SP airplanes, and Model GV airplanes. AD 2009-17-01 requires, for certain airplanes, a one-time inspection for sealant applied to the exterior of the APU enclosure, and for airplanes with the subject sealant and certain other airplanes, a revision of the AFM to prohibit operation of the APU during certain ground and flight operations. AD 2009-17-01 resulted from notification by the airplane manufacturer that an improper, flammable sealant was used on the interior and exterior of the APU enclosure. We issued AD 2009-17-01 to prevent this flammable sealant from igniting the exterior surfaces of the APU enclosure under certain anomalous conditions, such as an APU failure/APU compartment fire, which could result in propagation of an uncontained fire to other critical areas of the airplane.

Actions Since AD 2009-17-01 Was Issued

Since we issued AD 2009-17-01, the manufacturer has notified us that the type design sealant (AMS 3374) applied to the APU enclosure is flammable and failed a certification test and a company