

based on mortality rates for annuitants paid from the Civil Service Retirement and Disability Fund; and increases each year at an assumed rate of cost-of-living adjustment. Assumed rates of interest, mortality, and cost-of-living adjustments used in computing the present value are those used by the Board of Actuaries of the Civil Service Retirement System for valuation of the Federal Employees' Retirement System based on dynamic assumptions. The present value factors are unisex factors obtained as a composite of sex-distinct present value factors. Separate present value factors apply for FERS annuities that receive cost-of-living adjustments before the retiree attains age 62, versus FERS annuities that do not receive cost-of-living adjustments before the retiree attains age 62.

* * * * *

■ 21. Add § 842.616 to subpart F to read as follows:

§ 842.616 Publication of present value factors.

When OPM publishes in the **Federal Register** notice of normal cost percentages under § 841.407 of this chapter, it will also publish updated present value factors.

■ 22. Amend § 842.702 by revising the definition of “present value factor” to read as follows:

§ 842.702 Definitions.

* * * * *

Present value factor has the same meaning in this subpart as defined in § 842.602.

* * * * *

PART 847—ELECTIONS OF RETIREMENT COVERAGE BY CURRENT AND FORMER EMPLOYEES OF NONAPPROPRIATED FUND INSTRUMENTALITIES

■ 23. The authority citation for part 847 continues to read as follows:

Authority: 5 U.S.C. 8332(b)(17) and 8411(b)(6) and sections 1131 and 1132 of Pub. L. 107–107, December 28, 2001, 115 Stat 1242; 5 U.S.C. 8347(a) and 8461(g) and section 1043(b) of Pub. L. 104–106, Div. A, Title X, Feb. 10, 1996, 110 Stat. 434. Subpart B also issued under 5 U.S.C. 8347(q) and 8461(n).

Subpart A—General Provisions

■ 24. Amend § 847.103(b) by revising the definition of “actuarial present value” and adding the definition of “present value factor” in alphabetical order as follows:

§ 847.103 Definitions.

* * * * *

(b) * * *

Actuarial present value means the amount of monthly annuity at time of retirement multiplied by the applicable present value factor.

* * *

Present value factor has the same meaning in this part as defined in § 842.602.

* * * * *

§ 847.602 [Removed and Reserved]

■ 25. Remove and reserve § 847.602.

[FR Doc. 2016–30487 Filed 12–21–16; 8:45 am]

BILLING CODE 6325–38–P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA–2013–0740; Directorate Identifier 2013–NE–24–AD]

RIN 2120–AA64

Airworthiness Directives; Pratt & Whitney Turbofan Engines

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: We propose to supersede Airworthiness Directive (AD) 2014–05–32, which applies to all Pratt & Whitney (PW) PW2037, PW2037D, PW2037M, PW2040, PW2040D, PW2043, PW2143, PW2643, and F117–PW–100 turbofan engines. AD 2014–05–32 currently requires one-time eddy current inspection (ECI) of affected engines with certain diffuser and HPT cases installed. AD 2014–05–32 also requires a fluorescent-penetrant inspection (FPI) of the diffuser case rear flange and the HPT case front flange. Since we issued AD 2014–05–32, the manufacturer determined through analysis that the inspections required by AD 2014–05–32 are not adequate to maintain safety. This proposed AD would add additional repetitive, on-wing ECI inspections. We are proposing this AD to correct the unsafe condition on these products.

DATES: We must receive comments on this proposed AD by February 6, 2017.

ADDRESSES: You may send comments, using the procedures found in 14 CFR 11.43 and 11.45, by any of the following methods:

- *Federal eRulemaking Portal:* Go to <http://www.regulations.gov>. Follow the instructions for submitting comments.

- *Fax:* 202–493–2251.

- *Mail:* U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room

W12–140, 1200 New Jersey Avenue SE., Washington, DC 20590.

- *Hand Delivery:* Deliver to Mail address above between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

For service information identified in this NPRM, contact Pratt & Whitney, 400 Main St., East Hartford, CT 06108; phone: 860–565–8770; fax: 860–565–4503. You may view this service information at the FAA, Engine & Propeller Directorate, 1200 District Avenue, Burlington, MA. For information on the availability of this material at the FAA, call 781–238–7125.

Examining the AD Docket

You may examine the AD docket on the Internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA–2013–0740; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this proposed AD, the regulatory evaluation, any comments received, and other information. The street address for the Docket Office (phone: 800–647–5527) is in the **ADDRESSES** section. Comments will be available in the AD docket shortly after receipt.

FOR FURTHER INFORMATION CONTACT:

Brian Kierstead, Aerospace Engineer, Engine Certification Office, FAA, Engine & Propeller Directorate, 1200 District Avenue, Burlington, MA 01803; phone: 781–238–7772; fax: 781–238–7199; email: brian.kierstead@faa.gov.

SUPPLEMENTARY INFORMATION:

Comments Invited

We invite you to send any written relevant data, views, or arguments about this proposed AD. Send your comments to an address listed under the **ADDRESSES** section. Include “Docket No. FAA–2013–0740; Directorate Identifier 2013–NE–24–AD” at the beginning of your comments. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of this proposed AD. We will consider all comments received by the closing date and may amend this proposed AD because of those comments.

We will post all comments we receive, without change, to <http://www.regulations.gov>, including any personal information you provide. We will also post a report summarizing each substantive verbal contact we receive about this proposed AD.

Discussion

On March 6, 2014, we issued AD 2014–05–32, Amendment 39–17804 (79 FR 17856, March 31, 2014), (“AD 2014–05–32”), for all PW PW2037, PW2037D, PW2037M, PW2040, PW2040D, PW2043, PW2143, PW2643, and F117–PW–100 turbofan engines. AD 2014–05–32 requires a one-time ECI of affected engines with certain diffuser and HPT cases installed. AD 2014–05–32 also requires an FPI of the diffuser case rear flange and HPT case front flange. AD 2014–05–32 resulted from a rupture of the diffuser-to-HPT case flange. We issued AD 2014–05–32 to prevent failure of the diffuser-to-HPT case flange, which could lead to uncontained engine failure and damage to the airplane.

Actions Since AD 2014–05–32 Was Issued

Since we issued AD 2014–05–32, the manufacturer identified a subpopulation of diffuser cases installed on the affected engines with a repaired flange that has a lower fatigue capability. The repaired flange cannot be distinguished

from non-repaired flanges on diffuser cases installed on the affected engines. We determined, therefore, that the inspections required by AD 2014–05–32 are not adequate to maintain safety. To correct this unsafe condition, we are now proposing additional, repetitive ECI inspections.

Related Service Information Under 1 CFR Part 51

We reviewed PW Service Bulletin No. PW2000 72–763, Revision No. 1, dated August 30, 2013; and PW Alert Service Bulletin No. PW2000 A72–765, Revision No. 1, dated July 13, 2016. This service information describes procedures for a one-time ECI inspection of the engine diffuser case and the HPT case, and repetitive on-wing ECIs of the engine diffuser case assembly, respectively. This service information is reasonably available because the interested parties have access to it through their normal course of business or by the means identified in the **ADDRESSES** section.

FAA’s Determination

We are proposing this AD because we evaluated all the relevant information and determined the unsafe condition described previously is likely to exist or develop in other products of the same type design.

Proposed AD Requirements

This proposed AD would retain the requirements of AD 2014–05–32 except it would eliminate the Prohibition Statement. We determined that this statement is unnecessary for compliance with the AD. In addition, this proposed AD would require repetitive, on-wing ECI inspections. This proposed AD would also remove the PW2240 and PW2337 engines from the applicability section since these engines were removed from PW Type Certificate Number E17NE.

Costs of Compliance

We estimate that this proposed AD will affect 910 engines installed on airplanes of U.S. registry.

We estimate the following costs to comply with this proposed AD:

ESTIMATED COSTS

Action	Labor cost	Parts cost	Cost per product	Cost on U.S. operators
On-wing/module ECI Inspection.	8 work-hours × \$85 per hour = \$680.	\$0	\$680	\$230,520 per inspection cycle.
FPI Inspection	3 work-hours × \$85 per hour = \$255.	20	\$275 per inspection cycle	\$250,250 per inspection cycle.

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA’s authority to issue rules on aviation safety. Subtitle I, Section 106, describes the authority of the FAA Administrator. Subtitle VII, Aviation Programs, describes in more detail the scope of the Agency’s authority.

We are issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section 44701, “General requirements.” Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

We have determined that this proposed AD would not have federalism

implications under Executive Order 13132. This proposed AD would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that the proposed regulation:

- (1) Is not a “significant regulatory action” under Executive Order 12866,
- (2) Is not a “significant rule” under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979),
- (3) Will not affect intrastate aviation in Alaska to the extent that it justifies making a regulatory distinction, and
- (4) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

The Proposed Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA proposes to amend 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

- 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

- 2. The FAA amends § 39.13 by removing Airworthiness Directive (AD) 2014–05–32, Amendment 39–17804 (79 FR 17856, March 31, 2014), and adding the following new AD:

Pratt & Whitney: Docket No. FAA–2013–0740; Directorate Identifier 2013–NE–24–AD.

(a) Comments Due Date

The FAA must receive comments on this AD action by February 6, 2017.

(b) Affected ADs

This AD replaces AD 2014–05–32, Amendment 39–17804 (79 FR 17856, March 31, 2014).

(c) Applicability

This AD applies to all Pratt & Whitney (PW) PW2037, PW2037D, PW2037M, PW2040, PW2040D, PW2043, PW2143, PW2643, and F117–PW–100 turbofan engines.

(d) Subject

Joint Aircraft System Component (JASC) Code 72, Turbine/Turboprop Engine.

(e) Unsafe Condition

This AD was prompted by a rupture of the diffuser-to-high-pressure turbine (HPT) case flange. We are issuing this AD to prevent failure of the diffuser-to-HPT case flange, which could lead to uncontained engine failure and damage to the airplane.

(f) Compliance

Unless already done, comply with this AD within the compliance times specified.

(1) For diffuser case, part number (P/N) 1B7461, serial numbers (S/Ns) DGGUAK1306 and DGGUAK1308, and HPT case, P/N 1B2440, S/N DKLBCS1032:

(i) Within 100 flight cycles or 30 days after May 5, 2014, whichever is later, eddy current inspect the diffuser case and the HPT case M-flange. Use PW Service Bulletin (SB) No. PW2000 72–763, Revision No. 1, dated August 30, 2013, to do the inspection.

(ii) Reserved.

(2) For all diffuser and HPT cases, at the next piece-part opportunity and every piece-part opportunity thereafter, perform a high sensitivity fluorescent-penetrant inspection (FPI) of the entire diffuser case rear flange (M-flange) and bolt holes, and the entire HPT case forward flange (M-flange) and bolt holes.

(3) For diffuser cases that have not incorporated PW SB PW2000–72–364 or have incorporated either PW SB PW2000–72–700 or PW2000 Series Engine Manual, Repair-28, Task 72–41–01–300–028 (M-flange replacement), perform initial and repetitive eddy current inspections (ECIs) of the M-flange of the diffuser case in accordance with paragraph (f)(4) of this AD.

(4) Use, as applicable, either the Accomplishment Instructions, “For Engines Installed on the Aircraft,” paragraphs 3.(I) through 3.(J), or the Accomplishment Instructions, “For Engines Removed from the Aircraft,” paragraphs 3.(D) through 3.(E), of PW Alert Service Bulletin (ASB) No. PW2000 A72–765, Revision No. 1, dated July 13, 2016, to do the ECI as follows:

(i) Perform an initial inspection within the following period, whichever occurs later:

(A) Within 5,500 cycles since new or since M-flange replacement, or

(B) Within 2,500 cycles since last piece-part FPI inspection, or

(C) Within 500 cycles from the effective date of this AD.

(ii) If no crack indications are found, re-inspect within 2,500 cycles since last ECI or last piece-part FPI inspection, whichever occurs first.

(iii) If crack indications are found, measure the crack length and determine the re-inspect interval in accordance with:

(A) Paragraphs 5.(C) through 5.(D) of PW ASB No. PW2000 A72–765, Revision No. 1, dated July 13, 2016, “For Engines Installed on the Aircraft”; or

(B) Paragraphs 4.(C) through 4.(D) of PW ASB No. PW2000 A72–765, Revision No. 1, dated July 13, 2016, “For Engines Removed from the Aircraft.”

(iv) Remove from service diffuser cases with cracks exceeding 0.170 inches.

(g) Definition

For the purpose of this AD, piece-part opportunity is defined as when the part is completely disassembled.

(h) Credit for Previous Actions

If you performed an ECI of the diffuser case and HPT case M-flange using the Accomplishment Instructions of PW SB No. PW2000 72–763, Revision No. 1, dated August 13, 2013, or an earlier version, or you performed a high sensitivity FPI of the diffuser case and HPT case at the piece-part opportunity after January 1, 2010, you met the requirements of paragraph (f)(1) of this AD.

(i) Alternative Methods of Compliance (AMOCs)

The Manager, Engine Certification Office, FAA, may approve AMOCs for this AD. Use the procedures found in 14 CFR 39.19 to make your request. You may email your request to: ANE-AD-AMOC@faa.gov.

(j) Related Information

(1) For more information about this proposed AD, contact Brian Kierstead, Aerospace Engineer, Engine Certification Office, FAA, Engine & Propeller Directorate, 1200 District Avenue, Burlington, MA 01803; phone: 781–238–7772; fax: 781–238–7199; email: brian.kierstead@faa.gov.

(2) PW SB No. PW2000 72–763, Revision No. 1, dated August 30, 2013; and PW ASB No. PW2000 A72–765, Revision No. 1, dated July 13, 2016, can be obtained from PW using the contact information in paragraph (j)(3) of this AD.

(3) For service information identified in this proposed AD, contact Pratt & Whitney, United Technologies Corporation, 400 Main St., East Hartford, CT 06108; phone: 860–565–8770; fax: 860–565–4503.

(4) You may view this service information at the FAA, Engine & Propeller Directorate, 1200 District Avenue, Burlington, MA. For information on the availability of this material at the FAA, call 781–238–7125.

Issued in Burlington, Massachusetts, on December 1, 2016.

Colleen M. D'Alessandro,

Manager, Engine & Propeller Directorate, Aircraft Certification Service.

[FR Doc. 2016–30114 Filed 12–21–16; 8:45 am]

BILLING CODE 4910–13–P

OFFICE OF THE UNITED STATES TRADE REPRESENTATIVE**15 CFR Parts 2004 and 2005**

[Docket Number USTR–2016–0027]

RIN 0350–AA09

Privacy Act Policies and Procedures

AGENCY: Office of the United States Trade Representative.

ACTION: Proposed rule.

SUMMARY: As part of a comprehensive review of agency practices related to the disclosure of records and information, the Office of the United States Trade Representative (USTR) is updating both its systems of records and implementing rule under the Privacy Act of 1974 (Privacy Act). This proposed rule describes how individuals can find out if a USTR system of records contains information about them and, if so, how to access or amend a record. The proposed rule would move the Privacy Act regulation from part 2005 into a new subpart C to part 2004. USTR previously renamed and reorganized part 2004 to include all of the rules governing disclosure of USTR records and information. Elsewhere in this issue of the **Federal Register**, USTR is publishing a notice concerning updates to its Privacy Act systems of records. **DATES:** We must receive your written comments on or before January 23, 2017.

ADDRESSES: You should submit written comments through the Federal eRulemaking Portal: <http://www.regulations.gov>. The docket number for this rulemaking is USTR–2016–0027. USTR invites comments on all aspects of the proposed rule, and will revise the language as appropriate after taking all timely comments into consideration. Copies of all comments will be available for public viewing at www.regulations.gov upon completion of processing. You can view a submission by entering the docket number USTR–2016–0027 in the search field at <http://www.regulations.gov>. We will post comments without change and will include any personal information you provide, such as your name, mailing address, email address, and telephone number.

FOR FURTHER INFORMATION CONTACT:

Janice Kaye, Monique Ricker or Melissa Keppel, Office of General Counsel, Office of the US Trade Representative, Anacostia Naval Annex, Building 410/Door 123, 250 Murray Lane SW., Washington DC 20509, jkaye@ustr.eop.gov; mricker@ustr.eop.gov; mkeppel@ustr.eop.gov; 202–395–3150.