

For the reasons discussed above, I certify this proposed regulation:

1. Is not a "significant regulatory action" under Executive Order 12866;
2. Is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979);
3. Will not affect intrastate aviation in Alaska; and
4. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared a regulatory evaluation of the estimated costs to comply with this proposed AD and placed it in the AD docket.

#### List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

#### The Proposed Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA proposes to amend 14 CFR part 39 as follows:

#### PART 39—AIRWORTHINESS DIRECTIVES

- 1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

##### § 39.13 [Amended]

- 2. The FAA amends § 39.13 by adding the following new AD:

**BAE SYSTEMS (OPERATIONS) LIMITED:**  
Docket No. FAA-2013-0793; Directorate Identifier 2012-NM-138-AD.

##### (a) Comments Due Date

We must receive comments by November 12, 2013.

##### (b) Affected ADs

None.

##### (c) Applicability

This AD applies to BAE SYSTEMS (OPERATIONS) LIMITED Model BAe 146-100A, -200A, and -300A airplanes; and Model Avro 146-RJ70A, 146-RJ85A, and 146-RJ100A airplanes; certificated in any category; all models, all serial numbers.

##### (d) Subject

Air Transport Association (ATA) of America Code 26, Fire protection.

##### (e) Reason

This AD was prompted by reports of excess solder deposited during overhaul on the frangible plug of the extinguisher, which prevented the release of the extinguishant. We are issuing this AD to prevent the failure of a fire extinguisher to discharge, which reduces the ability of the fire protection system to extinguish fires in the engine or

APU fire zones, possibly resulting in damage to the airplane and injury to the passengers.

##### (f) Compliance

You are responsible for having the actions required by this AD performed within the compliance times specified, unless the actions have already been done.

##### (g) Inspection and Corrective Action

For airplanes equipped with fire extinguishers manufactured by Kidde Graviner Limited having part number (P/N) 57333 (all dash numbers): Within 12 months after the effective date of this AD, do an x-ray inspection to determine if there is solder between the operating head and container of the fire extinguishers in the engine and auxiliary power unit, in accordance with the Accomplishment Instructions of BAE SYSTEMS (OPERATIONS) LIMITED Service Bulletin 26-078, dated September 21, 2011; or Kidde Graviner Service Bulletin 26-080, Revision 1, dated July 27, 2011; as applicable.

(1) If any solder is found, before further flight, do the action specified in paragraph (g)(1)(i) or (g)(1)(ii) of this AD, in accordance with the Accomplishment Instructions of Kidde Graviner Service Bulletin 26-080, Revision 1, dated July 27, 2011.

(i) Overhaul the fire extinguisher and install. An overhaul includes the replacement of the operating head. Replacement of the pressure relief plug assembly only is not considered an overhaul.

(ii) Install a new fire extinguisher.

(2) If no solder is found, no further action is required by this paragraph.

##### (h) Parts Installation Limitation

As of the effective date of this AD, no person may install a Kidde Graviner Limited fire extinguisher having P/N 57333 (all dash numbers), on any airplane, unless the fire extinguisher is new, or it has been determined that there is no solder between the operating head and container of the fire extinguishers as required by paragraph (g) of this AD, or has been overhauled in accordance with the Accomplishment Instructions of Kidde Graviner Service Bulletin 26-080, Revision 1, dated July 27, 2011.

##### (i) Other FAA AD Provisions

The following provisions also apply to this AD:

(1) *Alternative Methods of Compliance (AMOCs):* The Manager, International Branch, ANM-116, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the International Branch, send it to ATTN: Todd Thompson, Aerospace Engineer, International Branch, ANM-116, Transport Airplane Directorate, FAA, 1601 Lind Avenue SW., Renton, WA 98057-3356; telephone 425-227-1175; fax 425-227-1149. Information may be emailed to: [9-ANM-116-AMOC-REQUESTS@faa.gov](mailto:9-ANM-116-AMOC-REQUESTS@faa.gov). Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal

inspector, the manager of the local flight standards district office/certificate holding district office. The AMOC approval letter must specifically reference this AD.

(2) *Airworthy Product:* For any requirement in this AD to obtain corrective actions from a manufacturer or other source, use these actions if they are FAA-approved. Corrective actions are considered FAA-approved if they are approved by the State of Design Authority (or their delegated agent). You are required to assure the product is airworthy before it is returned to service.

##### (j) Related Information

(1) Refer to Mandatory Continuing Airworthiness Information (MCAI) European Aviation Safety Agency (EASA) Airworthiness Directive 2012-0126R1, dated September 10, 2012, for related information, which can be found in the AD docket on the Internet at <http://www.regulations.gov>.

(2) For BAE SYSTEMS (OPERATIONS) LIMITED service information identified in this AD, contact BAE SYSTEMS (OPERATIONS) LIMITED, Customer Information Department, Prestwick International Airport, Ayrshire, KA9 2RW, Scotland, United Kingdom; telephone +44 1292 675207; fax +44 1292 675704; email [RApublications@baesystems.com](mailto:RApublications@baesystems.com); Internet <http://www.baesystems.com/Businesses/RegionalAircraft/index.htm>. For Kidde Graviner service information identified in this AD, contact Kidde Graviner Limited, Methisen Way, Colnbrook, Slough, Berkshire, SL3 0HB, United Kingdom; telephone +44 (0) 1753 683245; fax +44 (0) 1753 685040. You may review copies of the referenced service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, WA. For information on the availability of this material at the FAA, call 425-227-1221.

Issued in Renton, Washington, on September 13, 2013.

**Jeffrey E. Duven,**

*Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.*

[FR Doc. 2013-23276 Filed 9-24-13; 8:45 am]

**BILLING CODE 4910-13-P**

## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 39

[Docket No. FAA-2013-0790; Directorate Identifier 2013-NM-061-AD]

RIN 2120-AA64

#### Airworthiness Directives; The Boeing Company Airplanes

**AGENCY:** Federal Aviation Administration (FAA), DOT.

**ACTION:** Notice of proposed rulemaking (NPRM).

**SUMMARY:** We propose to supersede airworthiness directive (AD) 89-12-10, which applies to certain The Boeing Company Model 747 series airplanes.

AD 89-12-10 currently requires replacement of certain underwing fuel tank access doors with stronger, fire-resistant doors. Since we issued AD 89-12-10, we have received reports indicating that a standard access door was located where an impact-resistant access door was required, and stencils were missing from some impact-resistant access doors. Stencils are required to indicate that the door is impact-resistant and to indicate the correct location for installation of the impact-resistant door. This proposed AD would require an inspection of the left- and right-hand wing fuel tank access doors to determine whether impact-resistant access doors are installed in the correct locations, and replacement of any door with an impact-resistant access door if necessary. This proposed AD would also require an inspection for the presence of stencils and index markers on impact-resistant access doors, and application of new stencils or index markers if necessary. This proposed AD would also require revising the maintenance program to incorporate changes to the airworthiness limitations section. This proposed AD would also add airplanes to the applicability. We are proposing this AD to prevent foreign object penetration of the fuel tank, which could cause a fuel leak near an ignition source (e.g., hot brakes or engine exhaust nozzle), consequently leading to a fuel-fed fire.

**DATES:** We must receive comments on this proposed AD by November 12, 2013.

**ADDRESSES:** You may send comments, using the procedures found in 14 CFR 11.43 and 11.45, by any of the following methods:

- *Federal eRulemaking Portal:* Go to <http://www.regulations.gov>. Follow the instructions for submitting comments.
- *Fax:* 202-493-2251.
- *Mail:* U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE., Washington, DC 20590.
- *Hand Delivery:* Deliver to Mail address above between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

For service information identified in this proposed AD, contact Boeing Commercial Airplanes, Attention: Data & Services Management, P.O. Box 3707, MC 2H-65, Seattle, WA 98124-2207; telephone 206-544-5000, extension 1; fax 206-766-5680; Internet <https://www.myboeingfleet.com>. You may review copies of the referenced service information at the FAA, Transport

Airplane Directorate, 1601 Lind Avenue SW., Renton, WA. For information on the availability of this material at the FAA, call 425-227-1221.

#### Examining the AD Docket

You may examine the AD docket on the Internet at <http://www.regulations.gov>; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this proposed AD, the regulatory evaluation, any comments received, and other information. The street address for the Docket Office (phone: 800-647-5527) is in the **ADDRESSES** section. Comments will be available in the AD docket shortly after receipt.

**FOR FURTHER INFORMATION CONTACT:** Suzanne Lucier, Aerospace Engineer, Propulsion Branch, ANM-140S, FAA, Seattle Aircraft Certification Office, 1601 Lind Avenue SW., Renton, WA 98057-3356; phone: 425-917-6438; fax: 425-917-6590; email: [suzanne.lucier@faa.gov](mailto:suzanne.lucier@faa.gov).

#### SUPPLEMENTARY INFORMATION:

##### Comments Invited

We invite you to send any written relevant data, views, or arguments about this proposed AD. Send your comments to an address listed under the **ADDRESSES** section. Include "Docket No. FAA-2013-0790; Directorate Identifier 2013-NM-061-AD" at the beginning of your comments. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of this proposed AD. We will consider all comments received by the closing date and may amend this proposed AD because of those comments.

We will post all comments we receive, without change, to <http://www.regulations.gov>, including any personal information you provide. We will also post a report summarizing each substantive verbal contact we receive about this proposed AD.

##### Discussion

On May 22, 1989, we issued AD 89-12-10, Amendment 39-6230 (54 FR 23643, June 2, 1989) ("AD 89-12-10"), for certain The Boeing Company Model 747-100, 747-200, 747-300, and 747-SP series airplanes. AD 89-12-10 requires replacement of certain underwing fuel tank access doors with stronger, fire-resistant doors. AD 89-12-10 resulted from several incidents of door penetration by tire and engine debris, which resulted in spillage of significant quantities of fuel. We issued

AD 89-12-10 to prevent a fire in the wing fuel tank.

#### Actions Since AD 89-12-10 Was Issued

Since we issued AD 89-12-10, additional access doors, that may be installed on additional airplanes that were not identified in AD 89-12-10, have been identified that may be subject to the unsafe condition. Certain doors are addressed in other service bulletins. Additionally, due to repainting of the lower wing skin, stencils that identify the access door may no longer be in place to provide a visual confirmation that the correct door is installed.

#### Relevant Service Information

We reviewed the following service information. For information on the procedures and compliance times, see this service information at <http://www.regulations.gov> by searching for Docket No. FAA-2013-0790.

- Boeing Service Bulletin 747-28-2315, dated January 11, 2012.
- CDCCL Task 57-AWL-01, "Impact-Resistant Fuel Tank Access Doors," of Sub-section B, Airworthiness Limitations (AWLs)—Fuel Systems, of Section 9, Airworthiness Limitations (AWLs) and Certification Maintenance Requirements (CMRs) of Boeing 747-400 Maintenance Planning Data (MPD) Document D621U400, Revision August 2012.
- CDCCL Task 57-AWL-01, "Impact-Resistant Fuel Tank Access Doors," of Sub-section C, Airworthiness Limitations—Fuel Systems, of the Boeing 747-100/200/300/SP Airworthiness Limitations (AWLs) and Certification Maintenance Requirements (CMRs) Document D6-13747-CMR, Revision August 2012.

#### FAA's Determination

We are proposing this AD because we evaluated all the relevant information and determined the unsafe condition identified previously is likely to exist or develop in other products of these same type designs.

#### Proposed AD Requirements

This proposed AD would retain none of the requirements of AD 89-12-10. Since that AD was issued, the FAA issued section 121.316 of the Federal Aviation Regulations (14 CFR 121.316) requiring that each turbine-powered transport category airplane meet the requirements of section 25.963(e) of the Federal Aviation Regulations (14 CFR 25.963(e)). Section 25.963(e) outlines the certification requirements for fuel tank access covers on turbine powered transport category airplanes.

This proposed AD would require accomplishing the actions specified in the service information described previously. This proposed AD would add Models 747-400, 747-400D, 747-400F, and 747SR series airplanes to the applicability, and clarify the applicability of AD 89-12-10, Amendment 39-6230 (54 FR 23643, June 2, 1989) to identify models listed in the current type certification data sheets (TCDS).

This proposed AD would also require inspecting fuel tank access doors to determine that impact-resistant access doors are installed in the correct locations and replacing any door with an impact-resistant access door if necessary; inspecting application of stencils and index markers of impact-

resistant access doors and application of new stencils or index markers if necessary; and revising the maintenance program.

This proposed AD requires revisions to certain operator maintenance documents to include a new CDCCL. Compliance with CDCCLs is required by section 91.403(c) of the Federal Aviation Regulations (14 CFR 91.403(c)). For airplanes that have been previously modified, altered, or repaired in the areas addressed by this proposed AD, the operator might not be able to accomplish the actions described in the revisions. In this situation, to comply with 14 CFR 91.403(c), the operator must request approval for an alternative method of compliance according to the procedures specified in paragraph (j) of

this proposed AD. The request should include a description of changes to the required actions that will ensure the continued damage tolerance of the affected structure.

After accomplishment of the revision required by paragraph (h) of this AD, no alternative actions (e.g., inspections), intervals, and/or CDCCLs may be used unless the actions, intervals, and/or CDCCLs are approved as an alternative method of compliance (AMOC) in accordance with the procedures specified in paragraph (j) of this AD.

**Costs of Compliance**

We estimate that this proposed AD affects 189 airplanes of U.S. registry. We estimate the following costs to comply with this proposed AD:

**ESTIMATED COSTS**

Action	Labor cost	Parts cost	Cost per product	Cost on U.S. operators
Inspection .....	Up to 13 work-hours × \$85 per hour = \$1,105.	\$0	Up to \$1,105 .....	Up to \$208,845.
Maintenance program revision .....	1 work-hour × \$85 per hour = \$85.	0	85 .....	16,065.

We estimate the following costs to do any necessary replacements that would

be required based on the results of the proposed inspection. We have no way of

determining the number of aircraft that might need these replacements:

**ON-CONDITION COSTS**

Action	Labor cost	Parts cost	Cost per product
Replacement per door .....	3 work-hours × \$85 per hour = \$255 .....	\$8,000	\$8,255
Stencil and index marker (14 doors) .....	17 work-hours × \$85 per hour = \$1,445	0	1,445

According to the manufacturer, some of the costs of this proposed AD may be covered under warranty, thereby reducing the cost impact on affected individuals. We do not control warranty coverage for affected individuals. As a result, we have included all costs in our cost estimate.

**Authority for This Rulemaking**

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, Section 106, describes the authority of the FAA Administrator. Subtitle VII, Aviation Programs, describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section 44701, "General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures

the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

**Regulatory Findings**

We have determined that this proposed AD would not have federalism implications under Executive Order 13132. This proposed AD would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that the proposed regulation:

- (1) Is not a "significant regulatory action" under Executive Order 12866,
- (2) Is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979),

(3) Will not affect intrastate aviation in Alaska, and

(4) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

**List of Subjects in 14 CFR Part 39**

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

**The Proposed Amendment**

Accordingly, under the authority delegated to me by the Administrator, the FAA proposes to amend 14 CFR part 39 as follows:

**PART 39—AIRWORTHINESS DIRECTIVES**

- 1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

**§ 39.13 [Amended]**

■ 2. The FAA amends § 39.13 by removing airworthiness directive (AD) 89–12–10, Amendment 39–6230 (54 FR 23643, June 2, 1989), and adding the following new AD:

**The Boeing Company:** Docket No. FAA–2013–0790; Directorate Identifier 2013–NM–061–AD.

**(a) Comments Due Date**

The FAA must receive comments on this AD action by November 12, 2013.

**(b) Affected ADs**

This AD supersedes AD 89–12–10, Amendment 39–6230 (54 FR 23643, June 2, 1989).

**(c) Applicability**

This AD applies to The Boeing Company Model 747–100, 747–100B, 747–100B SUD, 747–200B, 747–200C, 747–200F, 747–300, 747–400, 747–400D, 747–400F, 747SR, and 747SP series airplanes; certificated in any category; as identified in Boeing Service Bulletin 747–28–2315, dated January 11, 2012.

**(d) Subject**

Joint Aircraft System Component (JASC)/ Air Transport Association (ATA) of America Code 28, Fuel.

**(e) Unsafe Condition**

This AD was prompted by reports indicating that a standard access door was located where an impact-resistant access door was required, and stencils were missing from some impact-resistant access doors. We are issuing this AD to prevent foreign object penetration of the fuel tank, which could cause a fuel leak near an ignition source (e.g., hot brakes or engine exhaust nozzle), consequently leading to a fuel-fed fire.

**(f) Compliance**

Comply with this AD within the compliance times specified, unless already done.

**(g) Inspection and Corrective Action**

Within 72 months after the effective date of this AD, do the actions specified in paragraphs (g)(1) and (g)(2) of this AD, in accordance with the Accomplishment Instructions of Boeing Service Bulletin 747–28–2315, dated January 11, 2012.

(1) Do either a general visual inspection or ultrasonic non-destructive test of the left- and right-hand wing fuel tank access doors to determine whether impact-resistant access doors are installed in the correct locations. If any standard access door is found, before further flight, replace with an impact-resistant access door, in accordance with the Accomplishment Instructions of Boeing Service Bulletin 747–28–2315, dated January 11, 2012.

(2) Do a general visual inspection of the left- and right-hand wing fuel tank impact resistant access doors to verify stencils and index markers are applied. If a stencil or index marker is missing, before further flight, apply a stencil or index marker, as applicable, in accordance with the

Accomplishment Instructions of Boeing Service Bulletin 747–28–2315, dated January 11, 2012.

**(h) Maintenance Program Revisions**

Within 60 days after the effective date of this AD do the actions specified in paragraphs (h)(1) and (h)(2) of this AD.

(1) Revise the maintenance program to incorporate CDCCL Task 57–AWL–01, “Impact-Resistant Fuel Tank Access Doors,” of Sub-section B, Airworthiness Limitations (AWLs)—Fuel Systems, of Section 9, Airworthiness Limitations (AWLs) and Certification Maintenance Requirements (CMRs) of Boeing 747–400 Maintenance Planning Data (MPD) Document D621U400, Revision August 2012.

(2) Revise the maintenance program to incorporate CDCCL Task 57–AWL–01, “Impact-Resistant Fuel Tank Access Doors,” of Sub-section C, Airworthiness Limitations—Fuel Systems, of the Boeing 747–100/200/300/SP Airworthiness Limitations (AWLs) and Certification Maintenance Requirements (CMRs) Document D6–13747–CMR, Revision August 2012.

**(i) No Alternative Actions, Intervals, and/or Critical Design Configuration Control Limitations (CDCCLs)**

After accomplishing the revision required by paragraph (h) of this AD, no alternative actions (e.g., inspections), intervals, and/or CDCCLs may be used unless the actions, intervals, and/or CDCCLs are approved as an alternative method of compliance (AMOC) in accordance with the procedures specified in paragraph (j) of this AD.

**(j) Alternative Methods of Compliance (AMOCs)**

(1) The Manager, Seattle Aircraft Certification Office (ACO), FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the manager of the ACO, send it to the attention of the person identified in the Related Information section of this AD. Information may be emailed to: [9-ANM-Seattle-ACO-AMOC-Requests@faa.gov](mailto:9-ANM-Seattle-ACO-AMOC-Requests@faa.gov).

(2) Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office.

(3) An AMOC that provides an acceptable level of safety may be used for any repair required by this AD if it is approved by the Boeing Commercial Airplanes Organization Designation Authorization (ODA) that has been authorized by the Manager, Seattle ACO to make those findings. For a repair method to be approved, the repair must meet the certification basis of the airplane, and the approval must specifically refer to this AD.

**(k) Related Information**

(1) For more information about this AD, contact Suzanne Lucier, Aerospace Engineer, Propulsion Branch, ANM–140S, FAA, Seattle Aircraft Certification Office, 1601 Lind

Avenue SW., Renton, WA 98057–3356; phone: 425–917–6438; fax: 425–917–6590; email: [suzanne.lucier@faa.gov](mailto:suzanne.lucier@faa.gov).

(2) For service information identified in this AD, contact Boeing Commercial Airplanes, Attention: Data & Services Management, P.O. Box 3707, MC 2H–65, Seattle, WA 98124–2207; telephone 206–544–5000, extension 1; fax 206–766–5680; Internet <https://www.myboeingfleet.com>. You may review copies of the referenced service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, WA. For information on the availability of this material at the FAA, call 425–227–1221.

Issued in Renton, Washington, on September 13, 2013.

**Jeffrey E. Duven,**

*Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.*

[FR Doc. 2013–23271 Filed 9–24–13; 8:45 am]

**BILLING CODE 4910–13–P**

**DEPARTMENT OF TRANSPORTATION****Federal Aviation Administration****14 CFR Part 39**

[Docket No. FAA–2013–0798; Directorate Identifier 2013–NM–087–AD]

**RIN 2120–AA64**

**Airworthiness Directives; Bombardier, Inc. Airplanes**

**AGENCY:** Federal Aviation Administration (FAA), DOT.

**ACTION:** Notice of proposed rulemaking (NPRM).

**SUMMARY:** We propose to adopt a new airworthiness directive (AD) for certain Bombardier, Inc. Model BD–100–1A10 (Challenger 300) airplanes. This proposed AD was prompted by multiple reports of erratic electrical status indications on the push button annunciators and the engine instrument and crew alerting system. Certain of those reported incidents resulted in the airplane experiencing a momentary loss of electrical power and loss of flight displays. This proposed AD would require modification of the direct current power centers. We are proposing this AD to prevent loss of electrical power, which could result in the loss of flight displays and reduced controllability of the airplane.

**DATES:** We must receive comments on this proposed AD by November 12, 2013.

**ADDRESSES:** You may send comments by any of the following methods:

- *Federal eRulemaking Portal:* Go to <http://www.regulations.gov>. Follow the instructions for submitting comments.
- *Fax:* (202) 493–2251.