

effective date of AD 2009-22-08, Amendment 39-16059 (74 FR 55763, October 29, 2009), replace the control switches of cargo doors 1 and 2, in accordance with the Accomplishment Instructions of Boeing Special Attention Service Bulletin 757-52-0090, dated September 21, 2007.

#### (h) New Replacement

For Group 3 airplanes identified in Boeing Special Attention Service Bulletin 747-52-2286, Revision 1, dated October 28, 2010: Within 72 months from the date of issuance of the original certificate of airworthiness or the original export certificate of airworthiness, or within 12 months after the effective date of this AD, whichever occurs later, replace the control switches of the forward, aft, and nose cargo doors, as applicable, in accordance with the Accomplishment Instructions of Boeing Special Attention Service Bulletin 747-52-2286, Revision 1, dated October 28, 2010. Repeat the replacements thereafter at intervals not to exceed 72 months.

#### (i) Alternative Methods of Compliance (AMOCs)

(1) The Manager, Seattle Aircraft Certification Office (ACO), FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the manager of the ACO, send it to the attention of the person identified in the Related Information section of this AD. Information may be emailed to: [9-ANM-Seattle-ACO-AMOC-Requests@faa.gov](mailto:9-ANM-Seattle-ACO-AMOC-Requests@faa.gov).

(2) Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office.

(3) An AMOC that provides an acceptable level of safety may be used for any repair required by this AD if it is approved by the Boeing Commercial Airplanes Organization Designation Authorization (ODA) that has been authorized by the Manager, Seattle ACO, to make those findings. For a repair method to be approved, the repair must meet the certification basis of the airplane, and the approval must specifically refer to this AD.

(4) AMOCs approved previously in accordance with AD 2009-22-08, Amendment 39-16059 (74 FR 55763, October 29, 2009), are approved as AMOCs for the corresponding provisions of this AD.

#### (j) Related Information

For more information about this AD, contact Francis Smith, Aerospace Engineer, Cabin Safety & Environmental Systems Branch, ANM-150S, FAA, Seattle Aircraft Certification Office, 1601 Lind Avenue SW., Renton, Washington 98057-3356; phone: 425-917-6596; fax: 425-917-6590; email: [francis.smith@faa.gov](mailto:francis.smith@faa.gov).

#### (k) Material Incorporated by Reference

(1) The Director of the Federal Register approved the incorporation by reference (IBR) of the service information listed in this

paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) You must use this service information as applicable to do the actions required by this AD, unless the AD specifies otherwise.

(3) The following service information was approved for IBR on February 22, 2013.

(i) Boeing Special Attention Service Bulletin 747-52-2286, Revision 1, dated October 28, 2010.

(ii) Reserved.

(4) The following service information was approved for IBR on December 3, 2009 (74 FR 55763, October 29, 2009).

(i) Boeing Special Attention Service Bulletin 747-52-2286, dated September 28, 2007.

(ii) Boeing Special Attention Service Bulletin 757-52-0090, dated September 21, 2007.

(5) For Boeing service information identified in this AD, contact Boeing Commercial Airplanes, Attention: Data & Services Management, P.O. Box 3707, MC 2H-65, Seattle, WA 98124-2207; telephone 206-544-5000, extension 1; fax 206-766-5680; Internet <https://www.myboeingfleet.com>.

(6) You may view this service information at FAA, 1601 Lind Avenue SW., Renton, Washington 98057-3356. For information on the availability of this material at the FAA, call 425-227-1221.

(7) You may view this service information that is incorporated by reference at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call 202-741-6030, or go to: <http://www.archives.gov/federal-register/cfr/ibr-locations.html>.

Issued in Renton, Washington, on January 4, 2013.

**Kalene C. Yanamura,**

*Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.*

[FR Doc. 2013-00559 Filed 1-17-13; 8:45 am]

**BILLING CODE 4910-13-P**

## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 39

**[Docket No. FAA-2013-0025; Directorate Identifier 2012-CE-048-AD; Amendment 39-17320; AD 2013-01-06]**

**RIN 2120-AA64**

#### **Airworthiness Directives; PILATUS Aircraft Ltd. Airplanes**

**AGENCY:** Federal Aviation Administration (FAA), DOT.

**ACTION:** Final rule; request for comments.

**SUMMARY:** We are adopting a new airworthiness directive (AD) for PILATUS Aircraft Ltd. Model PC-7 airplanes. This AD results from

mandatory continuing airworthiness information (MCAI) issued by the aviation authority of another country to identify and correct an unsafe condition on an aviation product. The MCAI describes the unsafe condition as cracks in the engine mount fittings caused by stress corrosion. We are issuing this AD to require actions to address the unsafe condition on these products.

**DATES:** This AD is effective February 7, 2013.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in the AD as of February 7, 2013.

We must receive comments on this AD by March 4, 2013.

**ADDRESSES:** You may send comments by any of the following methods:

- *Federal eRulemaking Portal:* Go to <http://www.regulations.gov>. Follow the instructions for submitting comments.

- *Fax:* (202) 493-2251.

- *Mail:* U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE., Washington, DC 20590.

- *Hand Delivery:* U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE., Washington, DC 20590, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

For service information identified in this AD, contact PILATUS AIRCRAFT LTD., Customer Technical Support (MCC), P.O. Box 992, CH-6371 Stans, Switzerland; telephone: +41 (0)41 619 67 74; fax: 41 (0)41 619 67 73; Internet: <http://www.pilatus-aircraft.com> or email: [Techsupport@pilatus-aircraft.com](mailto:Techsupport@pilatus-aircraft.com). You may review copies of the referenced service information at the FAA, Small Airplane Directorate, 901 Locust, Kansas City, Missouri 64106. For information on the availability of this material at the FAA, call (816) 329-4148.

#### **Examining the AD Docket**

You may examine the AD docket on the Internet at <http://www.regulations.gov>; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this AD, the regulatory evaluation, any comments received, and other information. The street address for the Docket Office (telephone (800) 647-5527) is in the **ADDRESSES** section. Comments will be available in the AD docket shortly after receipt.

**FOR FURTHER INFORMATION CONTACT:** Doug Rudolph, Aerospace Engineer,

FAA, Small Airplane Directorate, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone: (816) 329-4059; fax: (816) 329-4090; email: [doug.rudolph@faa.gov](mailto:doug.rudolph@faa.gov).

#### SUPPLEMENTARY INFORMATION:

##### Discussion

The Federal Office of Civil Aviation (FOCA), which is the aviation authority for Switzerland, has issued FOCA AD HB-2012-009, dated December 20, 2012 (referred to after this as “the MCAI”), to correct an unsafe condition for the specified products. The MCAI states:

This Airworthiness Directive (AD) is prompted due to the discovery of cracks in the engine mount fittings. The cracks are caused by stress corrosion. It is possible for stress corrosion cracks to occur on engine mount fittings initially made of aluminium alloy AA2024-T351. Later in production, the material specification was changed to aluminium alloy AA2124-T851 to decrease the risk of stress corrosion. The Part Number (P/N) of the engine mount fittings remained the same.

Such a condition, if left uncorrected, could lead to failure of the engine mount fittings and possible failure of the engine attachment.

In order to correct and control the situation, this AD requires a one-time check to identify the material specification and inspect those affected engine mount fittings that are made of aluminium alloy AA2024-T351. Any engine mount fittings found to be cracked must be reported to Pilatus prior to further flight.

You may obtain further information by examining the MCAI in the AD docket.

##### Relevant Service Information

PILATUS Aircraft Ltd. has issued PILATUS PC-7 Service Bulletin No. 53-008, dated November 30, 2012. The actions described in this service information are intended to correct the unsafe condition identified in the MCAI.

##### FAA’s Determination and Requirements of the AD

This product has been approved by the aviation authority of another country, and is approved for operation in the United States. Pursuant to our bilateral agreement with this State of Design Authority, they have notified us of the unsafe condition described in the MCAI and service information referenced above. We are issuing this AD because we evaluated all information provided by the State of Design Authority and determined the unsafe condition exists and is likely to exist or develop on other products of the same type design.

##### Interim Action

We consider this AD interim action. The type certificate holder is looking at

repetitive inspection intervals to be done through the maintenance program. When these intervals are established, we may take additional rulemaking action to mandate the repetitive inspections.

##### FAA’s Determination of the Effective Date

An unsafe condition exists that requires the immediate adoption of this AD. The FAA has found that the risk to the flying public justifies waiving notice and comment prior to adoption of this rule because stress corrosion cracking in the engine mount area presents a critical safety of flight condition that requires inspection to ensure that the unsafe condition is mitigated. Therefore, we determined that notice and opportunity for public comment before issuing this AD are impracticable and that good cause exists for making this amendment effective in fewer than 30 days.

##### Comments Invited

This AD is a final rule that involves requirements affecting flight safety, and we did not precede it by notice and opportunity for public comment. We invite you to send any written relevant data, views, or arguments about this AD. Send your comments to an address listed under the **ADDRESSES** section. Include “Docket No. FAA-2013-0025; Directorate Identifier 2012-CE-048-AD” at the beginning of your comments. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of this AD. We will consider all comments received by the closing date and may amend this AD because of those comments.

We will post all comments we receive, without change, to <http://www.regulations.gov>, including any personal information you provide. We will also post a report summarizing each substantive verbal contact we receive about this AD.

##### Costs of Compliance

We estimate that this AD will affect 15 products of U.S. registry. We also estimate that it would take about 7 work-hours per product to comply with the basic requirements of this AD. The average labor rate is \$85 per work-hour. Required parts would cost about \$0 per product.

Based on these figures, we estimate the cost of the AD on U.S. operators to be \$8,925, or \$595 per product.

In addition, we estimate that any necessary follow-on actions would take about 10 work-hours and require parts costing \$5,000, for a cost of \$5,850 per product. We have no way of

determining the number of products that may need these actions.

##### Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA’s authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. “Subtitle VII: Aviation Programs,” describes in more detail the scope of the Agency’s authority.

We are issuing this rulemaking under the authority described in “Subtitle VII, Part A, Subpart III, Section 44701: General requirements.” Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

##### Regulatory Findings

We determined that this AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that this AD:

- (1) Is not a “significant regulatory action” under Executive Order 12866,
- (2) Is not a “significant rule” under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979),
- (3) Will not affect intrastate aviation in Alaska, and
- (4) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

##### List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

##### Adoption of the Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

##### PART 39—AIRWORTHINESS DIRECTIVES

- 1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

### § 39.13 [Amended]

■ 2. The FAA amends § 39.13 by adding the following new AD:

**2013-01-06 PILATUS Aircraft Ltd.:**  
Amendment 39-17320; Docket No.  
FAA-2013-0025; Directorate Identifier  
2012-CE-048-AD.

#### (a) Effective Date

This airworthiness directive (AD) becomes effective February 7, 2013.

#### (b) Affected ADs

None.

#### (c) Applicability

This AD applies to PILATUS Aircraft Ltd. Models PC-7 airplanes, serial numbers 101 through 618, certificated in any category.

#### (d) Subject

Air Transport Association of America (ATA) Code 53: Fuselage.

#### (e) Reason

This AD was prompted by mandatory continuing airworthiness information (MCAI) issued by the aviation authority of another country to identify and correct an unsafe condition on an aviation product. We are issuing this AD to detect and correct cracks in the engine mount fittings.

#### (f) Actions and Compliance

Unless already done, do the following actions.

(1) Within the next 90 days after February 7, 2013 (the effective date of this AD), perform a conductivity test to identify the material specification of the engine mount fittings (part number (P/N) 112.35.07.152) following paragraph 3.B. of PILATUS Aircraft Ltd. PILATUS PC-7 Service Bulletin No. 53-008, dated November 30, 2012.

(2) If during the conductivity test required by paragraph (f)(1) of this AD, engine mount fittings made of aluminum alloy AA2024-T351 are found, within the next 90 days after February 7, 2013 (the effective date of this AD), do the inspection following paragraph 3.C. of PILATUS Aircraft Ltd. PILATUS PC-7 Service Bulletin No. 53-008, dated November 30, 2012.

(3) If during the inspection required by paragraph (f)(2) of this AD, any crack is found in the engine mount fittings, before further flight, contact Pilatus Customer Technical Support (MCC) for further instructions at P.O. Box 992, CH-6371 Stans, Switzerland; telephone: +41 (0)41 619 67 74; fax: 41 (0)41 619 67 73; Internet: <http://www.pilatus-aircraft.com> or email: [Techsupport@pilatus-aircraft.com](mailto:Techsupport@pilatus-aircraft.com).

#### (g) Other FAA AD Provisions

The following provisions also apply to this AD:

(1) *Alternative Methods of Compliance (AMOCs):* The Manager, Standards Office, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. Send information to ATTN: Doug Rudolph, Aerospace Engineer, FAA, Small Airplane Directorate, 901 Locust,

Room 301, Kansas City, Missouri 64106; telephone: (816) 329-4059; fax: (816) 329-4090; email: [doug.rudolph@faa.gov](mailto:doug.rudolph@faa.gov). Before using any approved AMOC on any airplane to which the AMOC applies, notify your appropriate principal inspector (PI) in the FAA Flight Standards District Office (FSDO), or lacking a PI, your local FSDO.

(2) *Airworthy Product:* For any requirement in this AD to obtain corrective actions from a manufacturer or other source, use these actions if they are FAA-approved. Corrective actions are considered FAA-approved if they are approved by the State of Design Authority (or their delegated agent). You are required to assure the product is airworthy before it is returned to service.

(3) *Reporting Requirements:* For any reporting requirement in this AD, a federal agency may not conduct or sponsor, and a person is not required to respond to, nor shall a person be subject to a penalty for failure to comply with a collection of information subject to the requirements of the Paperwork Reduction Act unless that collection of information displays a current valid OMB Control Number. The OMB Control Number for this information collection is 2120-0056. Public reporting for this collection of information is estimated to be approximately 5 minutes per response, including the time for reviewing instructions, completing and reviewing the collection of information. All responses to this collection of information are mandatory. Comments concerning the accuracy of this burden and suggestions for reducing the burden should be directed to the FAA at: 800 Independence Ave. SW., Washington, DC 20591, Attn: Information Collection Clearance Officer, AES-200.

#### (h) Related Information

Refer to Swiss MCAI Federal Office of Civil Aviation (FOCA) AD HB-2012-009, dated December 20, 2012; and PILATUS Aircraft Ltd. PILATUS PC-7 Service Bulletin No. 53-008, dated November 30, 2012, for related information.

#### (i) Material Incorporated by Reference

(1) The Director of the **Federal Register** approved the incorporation by reference (IBR) of the service information listed in this paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) You must use this service information as applicable to do the actions required by this AD, unless the AD specifies otherwise.

(i) PILATUS Aircraft Ltd. PILATUS PC-7 Service Bulletin No. 53-008, dated November 30, 2012.

(ii) Reserved.

(3) For PILATUS Aircraft Ltd. service information identified in this AD, contact PILATUS AIRCRAFT LTD., Customer Technical Support (MCC), P.O. Box 992, CH-6371 Stans, Switzerland; telephone: +41 (0)41 619 67 74; fax: 41 (0)41 619 67 73; Internet: <http://www.pilatus-aircraft.com> or email: [Techsupport@pilatus-aircraft.com](mailto:Techsupport@pilatus-aircraft.com).

(4) You may view this service information at FAA, Small Airplane Directorate, 901 Locust, Kansas City, Missouri 64106. For information on the availability of this material at the FAA, call (816) 329-4148.

(5) You may view this service information that is incorporated by reference at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call 202-741-6030, or go to: <http://www.archives.gov/federal-register/cfr/index.html>.

Issued in Kansas City, Missouri, on January 11, 2013.

**John Colomy,**

*Acting Manager, Small Airplane Directorate, Aircraft Certification Service.*

[FR Doc. 2013-00894 Filed 1-17-13; 8:45 am]

**BILLING CODE 4910-13-P**

## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 39

[Docket No. FAA-2012-0987; Directorate Identifier 2012-NM-130-AD; Amendment 39-17317; AD 2013-01-03]

RIN 2120-AA64

#### Airworthiness Directives; The Boeing Company Airplanes

**AGENCY:** Federal Aviation Administration (FAA), DOT.

**ACTION:** Final rule.

**SUMMARY:** We are adopting a new airworthiness directive (AD) for certain The Boeing Company Model 737-300, 737-400, 737-500, and 757-200 series airplanes. This AD was prompted by a report of damage caused by electrical arcing to the wires that connect seat electronics boxes (SEBs). This AD requires installing a new relay and doing certain wiring changes of the entertainment control switch. We are issuing this AD to prevent power from being supplied to passenger seats when the entertainment control switch is in the OFF position, which could cause an electrical shock hazard resulting in serious or fatal injury to maintenance personnel.

**DATES:** This AD is effective February 22, 2013.

The Director of the **Federal Register** approved the incorporation by reference of certain publications listed in the AD as of February 22, 2013.

**ADDRESSES:** For service information identified in this AD, contact Boeing Commercial Airplanes, Attention: Data & Services Management, P.O. Box 3707, MC 2H-65, Seattle, WA 98124-2207; telephone 206-544-5000, extension 1; fax 206-766-5680; Internet <https://www.myboeingfleet.com>. You may review copies of the referenced service information at the FAA, Transport