Proposed Rules

This section of the FEDERAL REGISTER contains notices to the public of the proposed issuance of rules and regulations. The purpose of these notices is to give interested persons an opportunity to participate in the rule making prior to the adoption of the final rules.

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39


RIN 2120–AA64

Airworthiness Directives; Diamond Aircraft Industries GmbH Airplanes

AGENCY: Federal Aviation Administration (FAA), Department of Transportation (DOT).

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: We propose to adopt a new airworthiness directive (AD) for Diamond Aircraft Industries GmbH Model H–36, HK 36 R, HK 36 TS, and HK 36 TTS airplanes. This proposed AD results from mandatory continuing airworthiness information (MCAI) originated by an aviation authority of another country to identify and correct an unsafe condition on an aviation product. The MCAI describes the unsafe condition as installation of an unsuitable self-locking nut on the bell crank of the elevator push rod that can cause failure of the elevator, resulting in loss of control. We are issuing this proposed AD to require actions to address the unsafe condition on these products.

DATES: We must receive comments on this proposed AD by December 20, 2012.

ADDRESSES: You may send comments by any of the following methods:

• Federal eRulemaking Portal: Go to http://www.regulations.gov. Follow the instructions for submitting comments.
• Fax: (202) 493–2251.
• Hand Delivery: U.S. Department of Transportation, Docket Operations, M–30, Room W12–140, 1200 New Jersey Avenue SE., Washington, DC 20590, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

For service information identified in this proposed AD, contact Diamond Aircraft Industries GmbH, N.A. Otto-Straße 5, A–2700 Wiener Neustadt, Austria, telephone: +43 2622 26700; fax: +43 2622 26780; email: office@diamond-air.at; Internet: www.diamond-air.at/hk36_super_dimona+M52087573ab0.html. You may review copies of the referenced service information at the FAA, Small Airplane Directorate, 901 Locust, Kansas City, Missouri 64106. For information on the availability of this material at the FAA, call (816) 329–4148.

Examining the AD Docket

You may examine the AD docket on the Internet at http://www.regulations.gov; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this proposed AD, the regulatory evaluation, any comments received, and other information. The street address for the Docket Office is in the ADDRESSES section. Comments will be available in the AD docket shortly after receipt.

FOR FURTHER INFORMATION CONTACT:

Mike Kiesov, Aerospace Engineer, FAA, Small Airplane Directorate, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone: (816) 329–4144; fax: (816) 329–4090; email: mike.kiesov@faa.gov.

SUPPLEMENTARY INFORMATION:

Comments Invited

We invite you to send any written relevant data, views, or arguments about this proposed AD. Send your comments to an address listed under the ADDRESSES section. Include “Docket No. FAA–2012–1172; Directorate Identifier 2012–CE–040–AD” at the beginning of your comments. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of this proposed AD. We will consider all comments received by the closing date and may amend this proposed AD because of those comments.

We will post all comments we receive, without change, to http://regulations.gov, including any personal information you provide. We will also post a report summarizing each substantive verbal contact we receive about this proposed AD.

Discussion

The European Aviation Safety Agency (EASA), which is the Technical Agent for the Member States of the European Community, has issued EASA AD No.: 2012–0173, dated September 3, 2012 (referred to after this as “the MCAI”), to correct an unsafe condition for the specified products. The MCAI states:

A HK 36 R aeroplane recently experienced an in-flight elevator control failure after take-off which resulted in an uncontrolled landing. The results of the subsequent investigation revealed that the elevator control rod had disconnected from the elevator bell crank in the tail section of the fuselage, as a result of installation of a non-suitable self-locking nut.

The subsequent design review of the affected elevator bell crank joint with elevator control rod identified that its current configuration has a failure potential when components such as thin self-securing nuts and bearings are aging and original clearance of the control system cannot be maintained in service. Both the designs of elevator bell crank and elevator control rod are installed in DV 20 aeroplanes.

This condition, if not corrected, could lead to further cases of elevator control failure, likely resulting in reduced control of the aeroplane, consequent damage to the aeroplane and injury to the occupants.

To address this concern, Diamond Aircraft Industries (DAI) published Mandatory Service Bulletin (MSB) 36–108 and MSB 20–061/1 to improve the affected elevator control joint by embodiment of new design which prevents elevator bell crank and push rod disconnection.

For reasons described above, this AD requires replacement of aeroplane elevator bell cranks with improved parts and prohibits installation of any previous design elevator bell crank.

You may obtain further information by examining the MCAI in the AD docket.

Relevant Service Information

Diamond Aircraft Industries GmbH has issued Diamond Aircraft Industries GmbH Mandatory Service Bulletin MSB 36–108, and Diamond Aircraft Industries GmbH Work Instruction WI–MSB 36–108, both dated February 28, 2012. The actions described in this service information are intended to correct the unsafe condition identified in the MCAI.
FAA's Determination and Requirements of the Proposed AD

This product has been approved by the aviation authority of another country, and is approved for operation in the United States. Pursuant to our bilateral agreement with this State of Design Authority, they have notified us of the unsafe condition described in the MCAI and service information referenced above. We are proposing this AD because we evaluated all information and determined the unsafe condition exists and is likely to exist or develop on other products of the same type design.

Costs of Compliance

We estimate that this proposed AD will affect 25 products of U.S. registry. We also estimate that it would take about 2 work-hours per product to comply with the basic requirements of this proposed AD. The average labor rate is $85 per work-hour. Required parts would cost about $352 per product.

Based on these figures, we estimate the cost of the proposed AD on U.S. operators to be $13,050, or $522 per product.

Authority for This Rulemaking


We are issuing this rulemaking under the authority described in “Subtitle VII, Part A, Subpart III, Section 44701: General requirements.” Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

We determined that this proposed AD would not have federalism implications under Executive Order 13132. This proposed AD would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify this proposed regulation:

(1) Is not a “significant regulatory action” under Executive Order 12866,
(2) Is not a “significant rule” under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979),
(3) Will not affect intrastate aviation in Alaska, and
(4) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

The Proposed Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA proposes to amend 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

§39.13 [Amended]

2. The FAA amends §39.13 by adding the following new AD:


(a) Comments Due Date

We must receive comments by December 20, 2012.

(b) Affected ADs

None.

(c) Applicability

This AD applies to the following Diamond Aircraft Industries GmbH models and serial number (S/N) airplanes, certificated in any category: H–36 and HK 36 R airplanes, S/Ns 36.300 through 36.414, HK 36 TS airplanes, S/Ns 36.415 and 36.416; and HK 36 TTS airplane, S/N 36.393.

(d) Subject


(e) Reason

This AD was prompted by mandatory continuing airworthiness information (MCAI) originated by an aviation authority of another country to identify and correct an unsafe condition on an aviation product. The MCAI describes the unsafe condition as installation of an unsuitable self-locking nut on the bell crank of the elevator push rod that can cause failure of the elevator, resulting in loss of control. We are issuing this AD to prevent disconnection of the elevator bell crank and push rod.

(f) Actions and Compliance

Unless already done, do the following actions following Diamond Aircraft Industries GmbH Mandatory Service Bulletin MSB 36–108, and Diamond Aircraft Industries GmbH Work Instruction WI–MSB 36–108, both dated February 28, 2012:

(1) Within the next 200 hours time-in-service (TIS) after the effective date of this AD, or within the next 12 months after the effective date of this AD, whichever occurs first, replace each elevator bell crank assembly with part number (P/N) 820–2730–12–00, and replace each elevator bell crank mount with P/N 820–2730–11–00.

(2) After the effective date of this AD, only install on the airplane elevator bell crank assemblies with P/N 820–2730–12–00 and elevator bell crank mounts with P/N 820–2730–11–00.

(g) Other FAA AD Provisions

The following provisions also apply to this AD:

(1) Alternative Methods of Compliance (AMOCs): The Manager, Standards Office, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. Send information to ATTN: Mike Kiesov, Aerospace Engineer, FAA, Small Airplane Directorate, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone: (816) 329–4144; fax: (816) 329–4099; email: mike.kiesov@faa.gov. Before using any approved AMOC on any airplane to which the AMOC applies, notify your appropriate principal inspector (PI) in the FAA Flight Standards District Office (FSDO), or lacking a PI, your local FSDO.

(2) Airworthy Product: For any requirement in this AD to obtain corrective actions from a manufacturer or other source, use these actions if they are FAA-approved. Corrective actions are considered FAA-approved if they are approved by the State of Design Authority (or their delegated agent). You are required to assure the product is airworthy before it is returned to service.

(3) Reporting Requirements: For any reporting requirement in this AD, a federal agency may not conduct or sponsor, and a person is not required to respond to, nor shall a person be subject to a penalty for failure to comply with a collection of information subject to the requirements of the Paperwork Reduction Act unless that collection of information displays a current valid OMB Control Number. The OMB Control Number for this information collection is 2120–0056. Public reporting for this collection of information is estimated to be approximately 5 minutes per response, including the time for reviewing instructions, completing and reviewing the collection of information. All responses to this collection of information are mandatory. Comments concerning the accuracy of this burden and suggestions for reducing the burden should be directed to the FAA at: 800 Independence Ave. SW., Washington, DC 20591, Attn: Information Collection Clearance Officer, AES–200.

(h) Related Information

Refer to MCAI European Aviation Safety Agency (EASA) AD No.: 2012–0173, dated
September 3, 2012; Diamond Aircraft Industries GmbH Mandatory Service Bulletin MSB 36–108, dated February 28, 2012; and Diamond Aircraft Industries GmbH Work Instruction WI–MSB 36–108, dated February 28, 2012, for related information. For service information related to this AD, contact Diamond Aircraft Industries GmbH, N.A., Otto-Straße 5, A–2700 Wiener Neustadt, Austria, telephone: +43 2622 26700; fax: +43 2622 26780; email: office@diamond-air.at; Internet: www.diamond-air.at/hk36_super_dimona+MS2007753ab0.html. You may review copies of the referenced service information at the FAA, Small Airplane Directorate, 901 Locust, Kansas City, Missouri 64106. For information on the availability of this material at the FAA, call (816) 329–4148.

Issued in Kansas City, Missouri, on October 25, 2012.

James E. Jackson,
Acting Manager, Small Airplane Directorate, Aircraft Certification Service.

DEPARTMENT OF TRANSPORTATION
Federal Aviation Administration

14 CFR Part 39


RIN 2120–AA64

Airworthiness Directives; Hawker Beechcraft Corporation (Type Certificate Previously Held by Raytheon Aircraft Company; Beech Aircraft Corporation) Airplanes

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: We propose to adopt a new airworthiness directive (AD) for certain Hawker Beechcraft Corporation (Type Certificate previously held by Raytheon Aircraft Company; Beech Aircraft Corporation) Model 400A airplanes. This proposed AD was prompted by a report that the wiring for the 5-volt direct current (DC) system is undersized and does not have adequate circuit protection for the smaller gauge wire. This proposed AD would require installing an in-line fuse in the 5-volt DC system for each of the five instrument lighting control power supplies. We are proposing this AD to prevent failure of the wiring, which could result in smoke in the cockpit, loss of cockpit lighting, and potential damage to surrounding wiring for other cockpit equipment such as the stick shaker function or angle-of-attack indicators.

DATES: We must receive comments on this proposed AD by December 20, 2012.

ADDRESSES: You may send comments, using the procedures found in 14 CFR 11.43 and 11.45, by any of the following methods:

- Mail: U.S. Department of Transportation, Docket Operations, M–106, Room 100, Mid-Continent Airport, P.O. Box 85, Wichita, KS 67201–0085; phone: (316) 946–7734; email: tmdc@hawkerbeechcraft.com; Internet: https://www.hawkerbeechcraft.com/service_support/pubs. You may review copies of the referenced service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue SW, Renton, WA. For information on the availability of this material at the FAA, call 425–227–1221.

EXAMINING THE AD DOCKET

You may examine the AD docket on the Internet at http://www.regulations.gov; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this proposed AD, the regulatory evaluation, any comments received, and other information. The street address for the Docket Office (phone: 800–647–5527) is in the ADDRESSES section. Comments will be available in the AD docket shortly after receipt.

FOR FURTHER INFORMATION CONTACT:
Richard Rejniak, Aerospace Engineer, Electrical Systems and Avionics Branch, ACE–119W, FAA, Wichita Aircraft Certification Office, 1801 Airport Road, Room 100, Mid-Continent Airport, Wichita, Kansas 67209; phone: (316) 946–4128; fax: (316) 946–4107; email: richard.rejniak@faa.gov.

SUPPLEMENTARY INFORMATION:
Comments Invited

We invite you to send any written relevant data, views, or arguments about this proposal. Send your comments to an address listed under the ADDRESSES section. Include “Docket No. FAA–2012–1111; Directorate Identifier 2012–NM–114–AD” at the beginning of your comments. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of this proposed AD. We will consider all comments received by the closing date and may amend this proposed AD because of those comments.

We will post all comments we receive, without change, to http://www.regulations.gov, including any personal information you provide. We will also post a report summarizing each substantive verbal contact we receive about this proposed AD.

Discussion

We received a report that the wiring for the 5-volt DC system is undersized and does not have adequate circuit protection for the smaller gauge wire. The wire is adequate for normal electrical loads, but it cannot safely handle the power supply’s maximum current. This condition, if not corrected, could result in failure of the wiring, which could result in smoke in the cockpit, loss of cockpit lighting, and potential damage to surrounding wiring for other cockpit equipment such as the stick shaker function or angle-of-attack indicators.

Relevant Service Information

We reviewed Hawker Beechcraft Mandatory Service Bulletin SB 33–4002, dated October 2010. The service information describes procedures for installing an in-line fuse in the 5-volt DC system for each of the five instrument lighting control power supplies.

FAA’s Determination

We are proposing this AD because we evaluated all the relevant information and determined the unsafe condition described previously is likely to exist or develop in other products of the same type design.

Proposed AD Requirements

This proposed AD would require accomplishing the actions specified in the service information described previously.

Clarification of Service Bulletin Note

The Hawker Beechcraft Mandatory Service Bulletin SB 33–4002, dated October 2010, includes a note in the Accomplishment Instructions to inform operators to contact Hawker Beechcraft “should any difficulty be encountered”