Proposed Rules

This section of the FEDERAL REGISTER contains notices to the public of the proposed issuance of rules and regulations. The purpose of these notices is to give interested persons an opportunity to participate in the rule making prior to the adoption of the final rules.

DEPARTMENT OF ENERGY
10 CFR Part 431
RIN 1904–AC36

Energy Conservation Program: Public Meeting and Availability of the Framework Document for High-Intensity Discharge Lamps


ACTION: Extension of public comment period.

SUMMARY: This document announces an extension of the time period for submitting comments on the Notice of Public Meeting (NOPM) and Availability of Framework Document Regarding Energy Conservation Standards for High-Intensity Discharge (HID) Lamps. The comment period is extended to April 12, 2012.

DATES: The comment period for the HID Lamps Framework NOPM published on February 28, 2012 at 77 FR 11785 is extended to April 12, 2012.

ADDRESSES: Any comments submitted must identify the request for information on set-top boxes and network equipment and provide docket number EERE–2010–BT–STD–0043 and/or RIN number 1904–AC36. Comments may be submitted using any of the following methods:


• Email: to HIDLamps-2010-STD-0043@ee.doe.gov. Include EERE–2010–BT–STD–0043 in the subject line of the message.


Docket: For access to the docket to read background documents, or comments received, go to the Federal eRulemaking Portal at http://www.regulations.gov.


SUPPLEMENTARY INFORMATION: On February 28, 2012, the U.S. Department of Energy (DOE) published a notice of public meeting (NOPM) and availability of framework document for high-intensity discharge (HID) lamps, initiating the rulemaking and data collection process to consider establishing energy conservation standards (77 FR 11785). The NOPM requested public comment from interested parties and provided for the submission of comments by April 5, 2012. The National Electrical Manufacturers Association (NEMA), on behalf of itself and its members, requested an extension of the public comment period by one week. NEMA stated that the additional time is necessary to more fully evaluate and prepare the information requested in the NOPM. NEMA stated that the extension will allow NEMA to conduct necessary meetings before providing its comments.

Based on NEMA’s request, DOE believes that extending the comment period to allow additional time for interested parties to prepare and submit comments is appropriate. Therefore, DOE is extending the comment period until April 12, 2012. Accordingly, DOE will consider any comments received by April 12, 2012 to be timely submitted.

Issued in Washington, DC, on March 23, 2012.


[FR Doc. 2012–7543 Filed 3–28–12; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39
RIN 2120–AA64

Airworthiness Directives; MD Helicopters, Inc.

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: We propose to supersede an existing airworthiness directive (AD) for MD Helicopters, Inc. (MDHI) Model MD900 helicopters. The existing AD requires a visual inspection, and if necessary, an eddy current inspection of the main rotor lower hub assembly (lower hub) for a crack. If a crack exists, the AD requires replacing the lower hub with an airworthy lower hub before further flight. Because that AD was immediately effective, we did not include in its requirements certain long-term actions that did not necessitate adoption prior to public comment. This proposed AD would require those long-term actions. This proposed AD would require the same inspections as the existing AD but would also require recurring inspections and replacing the lower hub with an airworthy lower hub. We are proposing this AD to detect a crack in the lower hub and prevent failure of the lower hub and subsequent loss of control of the helicopter.

DATES: We must receive comments on this proposed AD by May 29, 2012.

ADDRESSES: You may send comments by any of the following methods:

• Federal eRulemaking Docket: Go to http://www.regulations.gov. Follow the online instructions for sending your comments electronically.

• Fax: 202–493–2251.
We reviewed MDHI Service Bulletin SB900–117, dated January 14, 2011 (SB). The SB specifies an initial 100-hour and recurring 300-hour visual and eddy current inspections of the lower hub for a crack and, if a crack exists, replacement of the lower hub with an airworthy lower hub. The SB requires the inspections at the stated intervals or during the next annual inspection, whichever occurs first. The SB also specifies replacing the lower hub within three years.

Proposed AD Requirements

This proposed AD would require the same inspections currently required by AD 2011–14–05 (76 FR 41662, July 15, 2011), but would require that those inspections also be performed at intervals not to exceed 300 hours TIS or during the next annual inspection, whichever occurs first. This proposed AD would also require, within three years, replacing the lower hub with an airworthy hub not included in the Applicability section of this AD. This replacement would provide terminating action for this AD's requirements.

Differences Between the Proposed AD and the Service Information

This proposed AD would not require you to contact the manufacturer nor to return the lower hub assembly with a certain report.

Costs of Compliance

We estimate that this proposed AD would affect 12 helicopters of U.S. registry.

We estimate the following costs to comply with this proposed AD:

- Visually inspecting the hub. We estimate that would take one work-hour at $85 an hour, for a total cost per helicopter of $85 and a total cost of $1,020 for the fleet.
- Eddy current inspecting the lower hub. We estimate that would take one work-hour at $85 an hour, for a total cost per helicopter of $85 and a total cost of $1,020 for the fleet.
- Replacing the lower hub. We estimate that would take 11 work-hours at $85 an hour for a total labor cost of $935, and that parts would cost $12,480 per hub, for a total cost of $13,415 per helicopter and a total cost of $160,980 for the fleet.
- The total of all of the costs listed for U.S. operators would be $163,020, assuming that the lower hubs for the entire fleet would get replaced.

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA’s authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. “Subtitle VII:
Aviation Programs,” describes in more detail the scope of the Agency’s authority.

We are issuing this rulemaking under the authority described in “Subtitle VII, Part A, Subpart III, Section 44701: General requirements.” Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

We determined that this proposed AD would not have federalism implications under Executive Order 13132. This proposed AD would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed, I certify this proposed regulation:

1. Is not a “significant regulatory action” under Executive Order 12866;
2. Is not a “significant rule” under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979);
3. Will not affect intrastate aviation in Alaska to the extent that it justifies making a regulatory distinction; and
4. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared an economic evaluation of the estimated costs to comply with this proposed AD and placed it in the AD docket.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

The Proposed Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA proposes to amend 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. The FAA amends § 39.13 by removing airworthiness directive (AD) 2011–14–05, Amendment 39–16740 (76 FR 41662, July 15, 2011), and adding the following new AD:


(a) Applicability

This AD applies to MDHI Model MD900 helicopters with main rotor lower hub assembly (lower hub) part number 900R2101008–107, with serial numbers beginning with 5009, certified in any category.

(b) Unsafe Condition

This AD is prompted by the determination that a certain manufacturer had incorrectly inserted flanged bushings into the lower hub bore. This condition could result in failure of the lower hub and subsequent loss of control of the helicopter.

(c) Other Affected ADs.

This AD supersedes AD 2011–14–05, Amendment 39–16740 (76 FR 41662, July 15, 2011).

(d) Compliance

You are responsible for performing each action required by this AD within the specified compliance time unless it has already been accomplished prior to that time.

(e) Required Actions

(1) Within 100 hours time-in-service (TIS) or during the next annual inspection, whichever occurs first, unless done within the last 200 hours TIS, and thereafter at intervals not to exceed 300 hours TIS or during the next annual inspection, whichever occurs first:

(i) Visually inspect the sides and bottom of the area between the arms for the centering bearing and the areas adjacent to the bushings of the lower hub assembly for a crack. If there is a crack, before further flight, replace the lower hub with an airworthy lower hub.

(ii) If the lower hub is not replaced as a result of the visual inspection required by paragraph (e)(1)(i) of this AD, eddy current inspect the lower hub for a crack by following the Accomplishment Instructions, paragraphs 2.A(2) through 2.A.(10), of MD Helicopters Inc. Service Bulletin SB960–117, dated January 14, 2011 (SB). If there is a crack, before further flight, replace the lower hub with an airworthy hub.

(2) The eddy current inspection required by paragraph (e)(1)(ii) of this AD must be done by a Level II technician with ASNT–TC–1–1A, CEN EN 4179, MIL–STD–410, NASA410, or equivalent certification in eddy current inspections. The technician must have done an eddy current inspection in the last 12 months.

(3) Within 3 years, replace the lower hub with an airworthy lower hub not included in the Applicability section of this AD. This replacement is terminating action for the requirements of this AD.

(f) Alternative Methods of Compliance (AMOCs)

(1) The Manager, Los Angeles Aircraft Certification Office (LAACO), FAA, may approve AMOCs for this AD. Send your proposal to: Eric Schrieber, Aviation Safety Engineer, Los Angeles Aircraft Certification Office, Transport Airplane Directorate, FAA, 3960 Paramount Blvd., Lakewood, CA 90712; telephone (562) 627–5348; email eric.schrieber@faa.gov.

(2) For operations conducted under a Part 119 operating certificate or under Part 91, Subpart K, we suggest that you notify your principal inspector, or lacking a principal inspector, the manager of the local flight standards district office or certificate holding district office before operating any aircraft complying with this AD through an AMOC.

(g) Additional Information

For service information identified in this AD, contact MD Helicopters Inc., Attn: Customer Support Division, 4555 E. McDowell Rd. Mail Stop M615, Mesa, AZ 85215–9734, telephone 1–800–338–3378, fax 480–346–6813, or at http://www.mdhelicopters.com. You may review copies of this information at the FAA, Office of the Regional Counsel, Southwest Region, 2601 Meacham Blvd., Room 663, Fort Worth, Texas 76137.

(h) Subject

Joint Aircraft Service Component (JASC) Code: 6220, Main Rotor Head.

Issued in Fort Worth, Texas, on March 21, 2012.

Kim Smith, Manager, Rotorcraft Directorate, Aircraft Certification Service.

[FR Doc. 2012–7535 Filed 3–28–12; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA–2012–0338; Director Identifier 2009–SW–51–AD]

RIN 2120–AA64

Airworthiness Directives; Eurocopter France Helicopters

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: We propose to adopt a new airworthiness directive (AD) for Eurocopter France Model SA341G helicopters. This proposed AD is prompted by an analysis and tests performed by the manufacturer that indicate that the life limit of the rotating star should be 12,000 hours time-in-service (TIS). The proposed actions are intended to prevent failure of the