§ 39.13 [Amended]

2. The FAA amends § 39.13 by adding the following new airworthiness directive (AD):


(a) Comments Due Date

We must receive comments by January 26, 2012.

(b) Affected ADs

None.

(c) Applicability

This AD applies to The Boeing Company Model 777–200 and –300 series airplanes, certificated in any category, equipped with Rolls-Royce RB211 Trent 800 engines, as identified in Boeing Service Bulletin 777–75A0002, Revision 1, dated October 26, 2011.

(d) Subject

Joint Aircraft System Component (JASC)/Air Transport Association (ATA) of America Code 78, Exhaust.

(e) Unsafe Condition

This AD was prompted by reports of events related to thermal damage of the thrust reverser (T/R) inner wall on Rolls-Royce RB211 Trent 800 engines. We are issuing this AD to eliminate T/R thermal damage caused by excessive heat downstream of the 8th stage IP8 exhaust ports, which could result in T/R structural failure. This failure could result in large pieces of the T/R or adjacent components departing the airplane. A separated T/R piece could result in a rejected takeoff and cause asymmetric thrust and consequent loss of control of the airplane during reverse thrust operations. Separated components could also cause structural damage to the airplane, damage to other airplanes, or injury to people on the ground.

(f) Compliance

Comply with this AD within the compliance times specified, unless already done.

(g) Replacement of Bleed Valve Parts and Tubing

Within 36 months after the effective date of this AD, replace the bleed valve parts and tubing with new parts and tubing on the left and right engines in accordance with the Accomplishment Instructions of Boeing Alert Service Bulletin 777–75A0002, dated January 12, 2011, before the effective date of this AD is acceptable for compliance with the corresponding replacements required by paragraph (g) of this AD.

(k) Alternative Methods of Compliance (AMOCs)

(1) The Manager, Seattle Certification Office (ACO), FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the manager of the ACO, send it to the attention of the person identified in the Related Information section of this AD. Information may be emailed to: 9–AM–Seattle–ACO–AMOC–Requests@faa.gov.

(2) Before using any approved AMOC, notify your appropriate principal inspector, or a principal inspector, the manager of the local flight standards district office/ certificate holding district office.

(1) Related Information

(1) For more information about this AD, contact Rebel Nichols, Aerospace Engineer, Propulsion Branch, ANM–140S, FAA, Seattle Aircraft Certification Office (ACO), 1601 Lind Avenue SW., Renton, Washington 98057–3356; phone: (425) 917–6509; fax: (425) 917–6590; email: Rebel.Nichols@faa.gov.

(2) For Boeing service information identified in this AD, contact Boeing Commercial Airplanes, Attention: Data & Services Management, P.O. Box 3707, MC 2H–65, Seattle, Washington 98124–2207; phone: (206) 544–5000, extension 1; fax: (206) 766–5680; email: me.boecom@boeing.com; Internet: https://www.myboeingfleet.com.

(3) For Rolls-Royce service information identified in this AD, contact Rolls-Royce plc, P.O. Box 31, DERBY, DE24 8BJ, UK; telephone 011 44 1332 242424; fax 011 44 1332 249936. You may review copies of the referenced service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, Washington. For information on the availability of this material at the FAA, call (425) 227–1221. Issued in Renton, Washington, on December 5, 2011.

Ali Bahrami,
Manager, Transport Airplane Directorate,
Aircraft Certification Service.

[FR Doc. 2011–31738 Filed 12–9–11; 8:45 am]

BILLING CODE 4910–13–P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39


RIN 2120–AA64

Airworthiness Directives; 328 Support Services GmbH Airplanes

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: We propose to adopt a new airworthiness directive (AD) for all 328 Support Services GmbH (Type Certificate previously held by AvCraft Aerospace GmbH; Fairchild Dornier GmbH; Dornier Luftfahrt GmbH) Model 328–100 and –300 airplanes that would supersede an existing AD. This proposed AD results from mandatory continuing airworthiness information (MCAI) originated by an aviation authority of another country to identify and correct an unsafe condition on an aviation product. The MCAI describes the unsafe condition as:

An incident has been reported with a Dornier 328–100 aeroplane, where the right-hand (RH) power lever jumped in flight-idle position during the landing roll-out. The aeroplane was stopped by excessive braking.

The reason for the jamming was that the cockpit door locking device * * * had fallen off the RH cockpit wall, blocking the RH power/condition lever pulley/cable cluster below the door. * * * This condition, if not corrected, could cause interference with the engine and/or flight control cables, possibly resulting in reduced control of the aeroplane.

The proposed AD would require actions that are intended to address the unsafe condition described in the MCAI.

DATES: We must receive comments on this proposed AD by January 26, 2012.

ADDRESSES: You may send comments by any of the following methods:

• Federal eRulemaking Portal: Go to http://www.regulations.gov. Follow the instructions for submitting comments.
We will post all comments we receive, without change, to http://www.regulations.gov, including any personal information you provide. We will also post a report summarizing each substantive verbal contact we receive about this proposed AD.

Discussion

On September 30, 2009, we issued AD 2009–21–06, Amendment 39–16043 (74 FR 53151, October 16, 2009). That AD required actions intended to address an unsafe condition on 328 Support Services GmbH (Type Certificate Previously Held by AvCraft Aerospace GmbH; Fairchild Dornier GmbH; Dornier Luftfahrten GmbH) Model 328–100 and –300 airplanes. Since we issued AD 2009–21–06, Amendment 39–16043 (74 FR 53151, October 16, 2009), the manufacturer has provided two options to fix the locking device, depending on airplane configuration: Installing an improved locking device for the cockpit door, or installing a gap filler between the cockpit door and the cockpit wall. We have determined these actions are necessary to address the identified unsafe condition. The European Aviation Safety Agency (EASA), which is the Technical Agent for the Member States of the European Community, has issued EASA Airworthiness Directive 2010–0169, dated August 13, 2010 (referred to after this as “the MCAI”), to correct an unsafe condition for the specified products. The MCAI states:

An incident has been reported with a Dornier 328–100 aeroplane, where the right-hand (RH) power lever jammed in flight-idle position during the roll-out. The aeroplane was stopped by excessive braking. The reason for the jamming was that the cockpit door locking device Part Number (P/N) 001A252A3914012 had fallen off the RH cockpit wall, blocking the RH power/condition lever pulley/cable cluster below the door. Although the affected aeroplane had been modified, the technical investigation showed that a loose Cockpit Door Locking device could also occur on 328–100 and 328–300 aeroplanes with a standard installation.

This condition, if not corrected, could cause interference with the engine and/or flight control cables, possibly resulting in reduced control of the aeroplane. To address that unsafe condition, EASA issued AD 2009–0082 (which corresponds to FAA AD 2009–21–06, Amendment 39–16043 [74 FR 53151, October 16, 2009]) as an interim solution, to require a one-time inspection of the cockpit door locking device and the surrounding area and the reporting of all findings to the TC (type certificate) holder.

Since that AD was issued, the TC holder has developed an improved cockpit door locking device, P/N 001A252A3914016. Consequently, this EASA AD retains the requirements of [EASA] AD 2009–0082 [FAA AD 2009–21–06, Amendment 39–16043 (74 FR 53151, October 16, 2009)], which is superseded, and requires the replacement of the current P/N 001A252A3914012 with new designed P/N 001A252A3914016 cockpit door locking device, or the removal of the cockpit door locking device P/N 001A252A3914012 and the installation of a gap filler, as applicable to aeroplane configuration.

The required actions include performing operational tests, and repair if necessary. You may obtain further information by examining the service information in the MCAI and the AD docket.

Relevant Service Information

328 Support Services has issued the following service bulletins:

- SB–328–25–492, dated March 18, 2010 (for Model 328–100 airplanes);
- SB–328–25–493, dated March 18, 2010 (for Model 328–300 airplanes);
- SB–328–25–494, dated March 18, 2010 (for Model 328–100 airplanes); and

The actions described in this service information are intended to correct the unsafe condition identified in the MCAI.

FAA’s Determination and Requirements of This Proposed AD

This product has been approved by the aviation authority of another country, and is approved for operation in the United States. Pursuant to our bilateral agreement with the State of Design Authority, we have been notified of the unsafe condition described in the MCAI and service information referenced above. We are proposing this AD because we evaluated all pertinent information and determined an unsafe condition exists and is likely to exist or develop on other products of the same type design.

Differences Between This AD and the MCAI or Service Information

We have reviewed the MCAI and related service information and, in general, agree with their substance. But we might have found it necessary to use different words from those in the MCAI to ensure the AD is clear for U.S. operators and is enforceable. In making these changes, we do not intend to differ substantively from the information provided in the MCAI and related service information.

We might also have proposed different actions in this AD from those in the MCAI in order to follow FAA policies. Any such differences are highlighted in a NOTE within the proposed AD.
Costs of Compliance

Based on the service information, we estimate that this proposed AD would affect about 59 products of U.S. registry. The estimates that are required by AD 2009–21–06, Amendment 39–16043 (74 FR 53151, October 16, 2009), and retained in this proposed AD take about 1 work-hour per product, at an average labor rate of $85 per work hour. Based on these figures, the estimated cost of the currently required actions is $85 per product.

We estimate that it would take 6 work-hours per product, depending on airplane configuration, to comply with the new basic requirements of this proposed AD. The average labor rate is $85 per work-hour. Required parts would cost $2,315 per product. Based on these figures, we estimate the cost of the proposed AD on U.S. operators to be $166,675, or $2,825 per product.

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA’s authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. “Subtitle VII: Aviation Programs,” describes in more detail the scope of the Agency’s authority.

We are issuing this rulemaking under the authority described in “Subtitle VII, Part A, Subpart III, Section 44701: General requirements.” Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

We determined that this proposed AD would not have federalism implications under Executive Order 13132. This proposed AD would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify this proposed regulation:
1. Is not a “significant regulatory action” under Executive Order 12866;
2. Is not a “significant rule” under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and
3. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared a regulatory evaluation of the estimated costs to comply with this proposed AD and placed it in the AD docket.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

The Proposed Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA proposes to amend 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§39.13 Amended

2. The FAA amends § 39.13 by removing Amendment 39–16043 (74 FR 53151, October 16, 2009) and adding the following new AD:


Comments Due Date

(a) We must receive comments by January 26, 2012.

Affected ADs

(b) This AD supersedes AD 2009–21–06, Amendment 39–16043 (74 FR 53151, October 16, 2009).

Applicability

(c) This AD applies to 328 Support Services GmbH (Type Certificate previously held by AvCraft Aerospace GmbH; Fairchild Dornier GmbH; Dornier Luftfahrt GmbH) Model 328–100 and –300 airplanes; certified in any category; all serial numbers.

Subject

(d) Air Transport Association (ATA) of America Code 25: Equipment/Furnishings.

Reason

(e) The mandatory continuing airworthiness information (MCAI) states:

An incident has been reported with a Dornier 328–100 airplane, where the right-hand (RH) power lever jammed in flight-idle position during the landing roll-out. The airplane was stopped by excessive braking. The reason for the jamming was that the cockpit door locking device * * * had fallen off the RH cockpit wall, blocking the RH power/condition lever pulley/cable cluster below the door. * * *

This condition, if not corrected, could cause interference with the engine and/or flight control cables, possibly resulting in reduced control of the aeroplane.

* * * * *

Compliance

(f) You are responsible for having the actions required by this AD performed within the compliance times specified, unless the actions have already been done.

Restatement of Certain Requirements of AD 2009–21–06, Amendment 39–16043 (74 FR 53151, October 16, 2009)

(g) Within 3 months after November 20, 2009 (the effective date of AD 2009–21–06, Amendment 39–16043 (74 FR 53151, October 16, 2009)), do a detailed visual inspection of the cockpit door locking device and the surrounding area for proper installation, in accordance with the Accomplishment Instructions of 328 Support Services Service Bulletin SB–328–25–245 or SB–328J–25–235, both dated January 28, 2009, as applicable.

(h) If any discrepancy is found during the inspection specified in paragraph (g) of this AD, before further flight, do the corrective action, in accordance with the Accomplishment Instructions of 328 Support Services Service Bulletin SB–328–25–485 or SB–328J–25–235, both dated January 28, 2009, as applicable.

New Requirements of This AD

(i) Within 4,000 flight hours or 24 months after the effective date of this AD, whichever occurs first, do the corrective actions specified in paragraph (j)(1) or (j)(2) of this AD:

(1) For airplanes on which a door locking device with Option 521K010 is installed: Remove the locking device of the cockpit door, part number (P/N) 001A252A3914012, install the gap filler parts, and do operational tests, in accordance with the Accomplishment Instructions of 328 Support Services Service Bulletin SB–328–25–492, dated March 18, 2010 (for Model 328–100 airplanes); or 328 Support Services Service Bulletin SB–328J–25–244, dated March 18, 2010 (for Model 328–300 airplanes).

(2) For airplanes on which a door locking device with Option 521K010 is not installed: Replace the locking device of the cockpit door, P/N 001A252A3914012, with a new locking device, P/N 001A252A3914016, and do operational tests, in accordance with the Accomplishment Instructions of 328 Support Services Service Bulletin SB–328–25–491, dated March 18, 2010 (for Model 328–100 airplanes); or 328 Support Services Service Bulletin SB–328J–25–243, dated March 18, 2010 (for Model 328–300 airplanes).

(j) If any operational test fails during the actions specified in paragraph (j)(1) or (j)(2) of this AD, before further flight, repair in accordance with a method approved by the Manager, International Branch, ANM–116, Transport Airplane Directorate, FAA, or EASA (or its delegated agent).

(k) As the effective date of this AD, no person may install a locking device of the cockpit door having P/N 001A252A3914012 on any airplane.
DEPARTMENT OF HOUSING AND URBAN DEVELOPMENT
24 CFR Parts 50, 55, and 58
[Docket No. FR–5423–P–01]
RIN 2501–ADS1

Floodplain Management and Protection of Wetlands

AGENCY: Office of the Secretary, HUD.

ACTION: Proposed rule.

SUMMARY: This proposed rule would update and modify HUD’s regulations governing the protection of wetlands and floodplains. With respect to wetlands, the proposed rule would codify existing procedures for Executive Order 11990 (E.O. 11990), Protection of Wetlands. HUD’s current policy is to require the use of E.O. 11990’s 8 Step Process for floodplains for actions performed by the Department or actions performed with HUD financial assistance. This rule will codify this policy and thereby improve consistency and increase transparency by placing the E.O. requirements in regulation. In certain instances, the new wetlands procedures will allow recipients of HUD assistance to use permits issued under section 404 of the Clean Water Act in lieu of five steps of the E.O. 11990’s 8 Step Process, thereby streamlining the wetlands decision-making processes. With respect to floodplains, the proposed rule would prohibit HUD funding (e.g., Community Development Block Grants, HOME Investment Partnerships Program, Choice Neighborhoods, etc.) or Federal Housing Administration (FHA) mortgage insurance for the construction of new structures in Coastal High Hazard Areas. The current regulations allow for such new construction so long as the construction, is in accordance with certain standards. This change is anticipated to have minimal effect, since HUD receives few requests to fund or insure mortgages for new construction in these areas.

The proposal would also make several other changes to HUD’s floodplain and wetland regulations; the changes are designed to streamline floodplain and wetland environmental procedures and avoid unnecessary delays in processing. The procedures proposed by this rule would apply to HUD and to state, tribal, and local governments when they are responsible for environmental reviews under HUD programs.

DATES: Comment Due Date: February 10, 2012.

ADDRESSES: Interested persons are invited to submit comments regarding this proposed rule to the Regulations Division, Office of General Counsel, Department of Housing and Urban Development, 451 7th Street SW., Room 10276, Washington, DC 20410–0500. Communications must refer to the above docket number and title. There are two methods for submitting public comments. All submissions must refer to the above docket number and title.

1. Submission of Comments by Mail. Comments may be submitted by mail to the Regulations Division, Office of General Counsel, Department of Housing and Urban Development, 451 7th Street SW., Room 10276, Washington, DC 20410–0500.

2. Electronic Submission of Comments. Interested persons may submit comments electronically through the Federal eRulemaking Portal at http://www.regulations.gov. HUD strongly encourages commenters to submit comments electronically. Electronic submission of comments allows the commenter maximum time to prepare and submit a comment, ensures timely receipt by HUD, and enables HUD to make them immediately available to the public. Comments submitted electronically through the http://www.regulations.gov Web site can be viewed by other commenters and interested members of the public. Commenters should follow the instructions provided on that site to submit comments electronically.

Note: To receive consideration as public comments, comments must be submitted through one of the two methods specified above. Again, all submissions must refer to the docket number and title of the rule.

No Facsimile Comments. Facsimile (FAX) comments are not acceptable.

Public Inspection of Public Comments. All properly submitted comments and communications submitted to HUD will be available for public inspection and copying between 8 a.m. and 5 p.m. weekdays at the above address. Due to security measures at the HUD Headquarters building, an appointment to review the public comments must be scheduled in advance by calling the Regulations Division at (202) 708–3055 (this is not a toll-free number). Individuals with speech or hearing impairments may access this number via TTY by calling the Federal Relay Service at (800) 877–8839. Copies of all comments submitted are available for inspection and downloading at http://www.regulations.gov.

FOR FURTHER INFORMATION CONTACT: Charles Bien, Acting Director Office of Environment and Energy, Office of Community Planning and Development,