

substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

*For the reasons discussed above, I certify this proposed regulation:*

1. Is not a "significant regulatory action" under Executive Order 12866;
2. Is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and
3. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared a regulatory evaluation of the estimated costs to comply with this proposed AD and placed it in the AD docket.

#### List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

#### The Proposed Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA proposes to amend 14 CFR part 39 as follows:

#### PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

##### § 39.13 [Amended]

2. The FAA amends § 39.13 by adding the following new AD:

**Bombardier, Inc.:** Docket No. FAA-2011-1088; Directorate Identifier 2011-NM-099-AD.

#### Comments Due Date

(a) We must receive comments by December 5, 2011.

#### Affected ADs

(b) None.

#### Applicability

(c) This AD applies to Bombardier, Inc. Model DHC-8-400, -401, and -402 airplanes; certificated in any category; serial numbers 4161 through 4296 inclusive.

#### Subject

(d) Air Transport Association (ATA) of America Code 53: Fuselage.

#### Reason

(e) The mandatory continuing airworthiness information (MCAI) states:

Several operators have reported difficulties in opening the airstair door. Investigation revealed that the airstair door gearbox drain paths were blocked by sealant, causing water

to accumulate and freeze in the gearbox assembly. An airstair door that is unable to be opened could hinder evacuation in the event of an emergency.

\* \* \* \* \*

#### Compliance

(f) You are responsible for having the actions required by this AD performed within the compliance times specified, unless the actions have already been done.

#### Actions

(g) Within 600 flight hours after the effective date of this AD, do a general visual inspection of the structure and gearbox drain paths for blockages by sealant, in accordance with the Accomplishment Instructions of Bombardier Service Bulletin 84-53-48, dated December 2, 2010. If any blockages are found, before further flight, remove blockages in accordance with the Accomplishment Instructions of Bombardier Service Bulletin 84-53-48, dated December 2, 2010.

#### FAA AD Differences

**Note 1:** This AD differs from the MCAI and/or service information as follows: No differences.

#### Other FAA AD Provisions

(h) The following provisions also apply to this AD:

(1) *Alternative Methods of Compliance (AMOCs):* The Manager, New York Aircraft Certification Office (ACO), ANE-170, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the ACO, send it to ATTN: Program Manager, Continuing Operational Safety, FAA, New York ACO, 1600 Stewart Avenue, Suite 410, Westbury, New York 11590; telephone 516-228-7300; fax 516-794-5531. Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office. The AMOC approval letter must specifically reference this AD.

(2) *Airworthy Product:* For any requirement in this AD to obtain corrective actions from a manufacturer or other source, use these actions if they are FAA-approved. Corrective actions are considered FAA-approved if they are approved by the State of Design Authority (or their delegated agent). You are required to assure the product is airworthy before it is returned to service.

#### Related Information

(i) Refer to MCAI Transport Canada Civil Aviation (TCCA) Airworthiness Directive CF-2011-06, dated April 26, 2011; and Bombardier Service Bulletin 84-53-48, dated December 2, 2010; for related information.

Issued in Renton, Washington, on October 6, 2011.

**Ali Bahrami,**

*Manager, Transport Airplane Directorate, Aircraft Certification Service.*

[FR Doc. 2011-27023 Filed 10-18-11; 8:45 am]

**BILLING CODE 4910-13-P**

## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 39

[Docket No. FAA-2011-1069; Directorate Identifier 2011-NM-025-AD]

RIN 2120-AA64

#### Airworthiness Directives; Learjet Inc. Model 45 Airplanes

**AGENCY:** Federal Aviation Administration (FAA), DOT.

**ACTION:** Notice of proposed rulemaking (NPRM).

**SUMMARY:** We propose to adopt a new airworthiness directive (AD) for the products listed above. This proposed AD would require revising the maintenance program to include new or more restrictive life-limits and inspections. This proposed AD was prompted by changes to the Airworthiness Limitations Section (ALS) of the maintenance manual, which adds life-limits, revises life-limits, or adds inspections not previously identified. We are proposing this AD to limit exposure of flight critical components to corrosion, cracking, or failure due to life-limits, which if not corrected, could result in loss of roll control, fatigue cracking, or loss of structural components.

**DATES:** We must receive comments on this proposed AD by December 5, 2011.

**ADDRESSES:** You may send comments by any of the following methods:

- *Federal eRulemaking Portal:* Go to <http://www.regulations.gov>. Follow the instructions for submitting comments.

- *Fax:* 202-493-2251.

- *Mail:* U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue, SE., Washington, DC 20590.

- *Hand Delivery:* Deliver to Mail address above between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

For service information identified in this proposed AD, contact Learjet, Inc., One Learjet Way, Wichita, Kansas 67209-2942; phone 316-946-2000; fax 316-946-2220; e-mail [ac.ict@aero.bombardier.com](mailto:ac.ict@aero.bombardier.com); Internet

<http://www.bombardier.com>. You may review copies of the referenced service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington. For information on the availability of this material at the FAA, call 425-227-1221.

**Examining the AD Docket**

You may examine the AD docket on the Internet at <http://www.regulations.gov>; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this proposed AD, the regulatory evaluation, any comments received, and other information. The street address for the Docket Office (phone: 800-647-5527) is in the ADDRESSES section. Comments will be available in the AD docket shortly after receipt.

**FOR FURTHER INFORMATION CONTACT:** William Griffith, Aerospace Engineer, Airframe Branch, ACE-118W, FAA, Wichita Aircraft Certification Office, 1801 Airport Road, Room 100, Mid-Continent Airport, Wichita, Kansas 67209; phone: 316-946-4116; fax: 316-946-4107; e-mail: [William.E.Griffith@faa.gov](mailto:William.E.Griffith@faa.gov).

**SUPPLEMENTARY INFORMATION:**

**Comments Invited**

We invite you to send any written relevant data, views, or arguments about

this proposal. Send your comments to an address listed under the ADDRESSES section. Include "Docket No. FAA-2011-1069; Directorate Identifier 2011-NM-025-AD" at the beginning of your comments. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of this proposed AD. We will consider all comments received by the closing date and may amend this proposed AD because of those comments.

We will post all comments we receive, without change, to <http://www.regulations.gov>, including any personal information you provide. We will also post a report summarizing each substantive verbal contact we receive about this proposed AD.

**Discussion**

We have reviewed the design approval holder's changes to the ALS of the maintenance manual, which adds life-limits, revises life-limits, or adds inspections not previously identified. These changes resulted from the design holder's analysis, testing, and in-service history of certain components. We are proposing this AD to limit exposure of flight critical components to corrosion, cracking, or failure due to life-limits, which if not corrected, could result in loss of roll control, fatigue cracking, or loss of structural components. The

corrective action is revising the maintenance program.

**Relevant Service Information**

We reviewed Chapter 04, Airworthiness Limitations, of the Bombardier Learjet 45 Maintenance Manual MM-104, Revision 53, dated January 10, 2011; and Bombardier Learjet 40 Maintenance Manual MM-105, Revision 21, dated January 10, 2011. This service information describes component and system checks and replacements and includes new or revised life-limits and new inspections.

**FAA's Determination**

We are proposing this AD because we evaluated all the relevant information and determined the unsafe condition described previously is likely to exist or develop in other products of the same type design.

**Proposed AD Requirements**

This proposed AD would require accomplishing the actions specified in the service information described previously.

**Costs of Compliance**

We estimate that this proposed AD affects 336 of U.S. registry.

We estimate the following costs to comply with this proposed AD:

**ESTIMATED COSTS**

Action	Labor cost	Parts cost	Cost per product	Cost on U.S. operators
Change ALS in maintenance manual .....	1 work-hour × \$85 per hour = \$85 .....	\$0	\$85	\$28,560

**Authority for This Rulemaking**

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. Subtitle VII: Aviation Programs, describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section 44701: "General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on

products identified in this rulemaking action.

**Regulatory Findings**

We determined that this proposed AD would not have federalism implications under Executive Order 13132. This proposed AD would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

*For the reasons discussed above, I certify this proposed regulation:*

- (1) Is not a "significant regulatory action" under Executive Order 12866,
- (2) Is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979),

(3) Will not affect intrastate aviation in Alaska, and

(4) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

**List of Subjects in 14 CFR Part 39**

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

**The Proposed Amendment**

Accordingly, under the authority delegated to me by the Administrator, the FAA proposes to amend 14 CFR part 39 as follows:

**PART 39—AIRWORTHINESS DIRECTIVES**

- 1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

**§ 39.13 [Amended]**

2. The FAA amends § 39.13 by adding the following new airworthiness directive (AD):

**Learjet Inc.:** Docket No. FAA–2011–1069; Directorate Identifier 2011–NM–025–AD.

**Comments Due Date**

(a) We must receive comments by December 5, 2011.

**Affected ADs**

(b) None

**Applicability**

(c) This AD applies to all Learjet Inc. Model 45 airplanes, certificated in any category.

**Note 1:** This AD requires revisions to certain operator maintenance documents to include new actions (e.g. inspections). Compliance with these actions is required by 14 CFR 91.403(c). For airplanes that have been previously modified, altered, or repaired in the areas addressed by these actions, the operator may not be able to

accomplish the inspections described in the revisions. In this situation, to comply with 14 CFR 91.403(c), the operator must request approval for an alternative method of compliance according to paragraph (i) of this AD. The request should include a description of changes to the required actions that will ensure the continued operational safety of the airplane. The FAA has provided guidance for this determination in Advisory Circular (AC) 25.1529–1A.

**Subject**

(d) Joint Aircraft System Component (JASC)/Air Transport Association (ATA) of America Code 05: Periodic Inspections.

**Unsafe Condition**

(e) This AD was prompted by changes to the Airworthiness Limitations Section (ALS) of the maintenance manual (MM), which adds life-limits, revises life-limits, or adds inspections not previously identified. We are issuing this AD to limit exposure of flight critical components to corrosion, cracking, or failure due to life-limits, which if not corrected, could result in loss of roll control, fatigue cracking, or loss of structural components.

**Compliance**

(f) Comply with this AD within the compliance times specified, unless already done.

**Maintenance Program Revision**

(g) Within 90 days after the effective date of this AD, revise the maintenance program by incorporating the applicable inspection reference number (IRN) tasks identified in table 1 of this AD as specified in Chapter 04, Airworthiness Limitations, of the Bombardier Learjet 45 Maintenance Manual MM–104, Revision 53, dated January 10, 2011; or Bombardier Learjet 40 Maintenance Manual MM–105, Revision 21, dated January 10, 2011; as applicable. The initial task compliance time is within 90 days after the effective date of this AD, or the applicable initial compliance time specified in Table 1 of this AD, whichever is later.

**Note 2:** IRN # R2710041 shown in table 1 of this AD is identified as IRN # N2710041 in prior revisions of Bombardier Learjet 45 Maintenance Manual MM–104; and Bombardier Learjet 40 Maintenance Manual MM–105.

TABLE 1—IRN TASK REVISION

Model—	IRN #—	Initial compliance time—	Chapter 04 of these documents—
Model 40, 45 .....	R2710041 .....	Within 10 years after the date of issuance of the original standard airworthiness certificate or the date of issuance of the original export certificate of airworthiness, or within 10 years after the most recent replacement, whichever occurs later.	Bombardier Learjet 45 Maintenance Manual MM–104, Revision 53, dated January 10, 2011; or Bombardier Learjet 40 Maintenance Manual MM–105, Revision 21, dated January 10, 2011; as applicable.
Model 40, 45 .....	Q5510091 .....	Within 600 flight hours after the most recent inspection done in accordance with IRN # Q5510091.	Bombardier Learjet 45 Maintenance Manual MM–104, Revision 53, dated January 10, 2011; or Bombardier Learjet 40 Maintenance Manual MM–105, Revision 21, dated January 10, 2011; as applicable.
Model 40, 45 .....	Q5530011 .....	Before the accumulation of 9,600 total flight hours.	Bombardier Learjet 45 Maintenance Manual MM–104, Revision 53, dated January 10, 2011; or Bombardier Learjet 40 Maintenance Manual MM–105, Revision 21, dated January 10, 2011; as applicable.
Model 40, 45 .....	P3220007 .....	Within 48 months after the most recent inspection done in accordance with IRN # P3220007.	Bombardier Learjet 45 Maintenance Manual MM–104, Revision 53, dated January 10, 2011; or Bombardier Learjet 40 Maintenance Manual MM–105, Revision 21, dated January 10, 2011; as applicable.
Model 40, 45 .....	P3220146 .....	Before the accumulation of 4,800 total landings.	Bombardier Learjet 45 Maintenance Manual MM–104, Revision 53, dated January 10, 2011; or Bombardier Learjet 40 Maintenance Manual MM–105, Revision 21, dated January 10, 2011; as applicable.
Model 40, 45 .....	N3220012, N3220023, N3220035, N3220036, and N3220037.	Before the accumulation of 10,000 total landings on the component.	Bombardier Learjet 45 Maintenance Manual MM–104, Revision 53, dated January 10, 2011; or Bombardier Learjet 40 Maintenance Manual MM–105, Revision 21, dated January 10, 2011; as applicable.

TABLE 1—IRN TASK REVISION—Continued

Model—	IRN #—	Initial compliance time—	Chapter 04 of these documents—
Model 40, 45 .....	N3220103, N3220104, N3220105, and N3220106.	Before the accumulation of 17,000 total landings on the component.	Bombardier Learjet 45 Maintenance Manual MM-104, Revision 53, dated January 10, 2011; or Bombardier Learjet 40 Maintenance Manual MM-105, Revision 21, dated January 10, 2011; as applicable.
Model 45 .....	N5710147, N5710171, and N5710173 ..	Before the accumulation of 6,500 total flight hours.	Bombardier Learjet 45 Maintenance Manual MM-104, Revision 53, dated January 10, 2011.
Model 45 .....	N5710175 .....	Before the accumulation of 6,900 total flight hours.	Bombardier Learjet 45 Maintenance Manual MM-104, Revision 53, dated January 10, 2011.
Model 45 .....	N5710177 .....	Before the accumulation of 7,000 total flight hours.	Bombardier Learjet 45 Maintenance Manual MM-104, Revision 53, dated January 10, 2011.

**No Alternative Intervals**

(h) After accomplishing the revisions required by paragraphs (g) of this AD, no alternative IRN task or IRN task interval may be used unless the IRN task or IRN task interval is approved as an AMOC in accordance with the procedures specified in paragraph (i)(1) of this AD.

**Alternative Methods of Compliance (AMOCs)**

(i)(1) The Manager, Wichita Aircraft Certification Office (ACO), FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the manager of the ACO, send it to the attention of the person identified in the Related Information section of this AD.

(2) Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office.

**Related Information**

(j) For more information about this AD, contact William Griffith, Aerospace Engineer, Airframe Branch, ACE-118W, FAA, Wichita Aircraft Certification Office, 1801 Airport Road, Room 100, Mid-Continent Airport, Wichita, Kansas 67209; phone: 316-946-4116; fax: 316-946-4107; e-mail: William.E.Griffith@faa.gov.

(k) For service information identified in this AD, contact Learjet, Inc., One Learjet Way, Wichita, Kansas 67209-2942; telephone 316-946-2000; fax 316-946-2220; e-mail ac.ict@aero.bombardier.com; Internet http://www.bombardier.com. You may review copies of the referenced service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington. For information on the availability of this material at the FAA, call 425-227-1221.

Issued in Renton, Washington, on October 5, 2011.

**Ali Bahrami,**

*Manager, Transport Airplane Directorate, Aircraft Certification Service.*

[FR Doc. 2011-27010 Filed 10-18-11; 8:45 am]

**BILLING CODE 4910-13-P**

**DEPARTMENT OF TRANSPORTATION**

**Federal Aviation Administration**

**14 CFR Part 39**

**[Docket No. FAA-2011-1087; Directorate Identifier 2011-NM-032-AD]**

**RIN 2120-AA64**

**Airworthiness Directives; Airbus Model A318, A319, A320, and A321 Series Airplanes**

**AGENCY:** Federal Aviation Administration (FAA), DOT.

**ACTION:** Notice of proposed rulemaking (NPRM).

**SUMMARY:** We propose to adopt a new airworthiness directive (AD) for the products listed above that would supersede two existing ADs. This proposed AD results from mandatory continuing airworthiness information (MCAI) originated by an aviation authority of another country to identify and correct an unsafe condition on an aviation product. The MCAI describes the unsafe condition as:

Following in-service experience, analyses of the failure to follow procedure or heed existing cockpit cues were conducted to assess the consequences of mismanagement of thrust levers during landing.

The investigation results identified the need for improvements in the identification of throttle mis-positioning and so providing further opportunity for the flight crew to identify an incorrect thrust lever configuration and to correct this. \* \* \* In addition, the analysis of the thrust lever

management issue shows two categories of scenarios that could lead to thrust asymmetry during landing with controllability and deceleration consequences [.]

\* \* \* \* \*

These thrust asymmetry conditions, if not corrected, could result in loss of control of the aeroplane during landing.

\* \* \* \* \*

The proposed AD would require actions that are intended to address the unsafe condition described in the MCAI.

**DATES:** We must receive comments on this proposed AD by December 5, 2011.

**ADDRESSES:** You may send comments by any of the following methods:

- *Federal eRulemaking Portal:* Go to <http://www.regulations.gov>. Follow the instructions for submitting comments.

- *Fax:* (202) 493-2251.

- *Mail:* U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue, SE., Washington, DC 20590.

- *Hand Delivery:* U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue, SE., Washington, DC, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

For service information identified in this proposed AD, contact Airbus, Airworthiness Office—EAS, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France; telephone +33 5 61 93 36 96; fax +33 5 61 93 44 51; e-mail: [account.airworth-eas@airbus.com](mailto:account.airworth-eas@airbus.com); Internet <http://www.airbus.com>. You may review copies of the referenced service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington. For information on the availability of this material at the FAA, call 425-227-1221.