

and -200 IGW airplanes: Within 3,000 flight hours after the effective date of this AD, replace the shear pins having part number (P/N) 190-15178-003 and P/N 190-15181-003 in the rear outboard and inboard shear pin assembly in the right- and left-hand pylons, with new shear pins having P/N 190-15178-005 and P/N 190-15181-005, respectively, in accordance with the Accomplishment Instructions of EMBRAER Service Bulletin 190-54-0010, dated May 19, 2010.

(h) For Model ERJ 190-100 ECJ airplanes: Within 3,000 flight hours or within 12 months after the effective date of this AD, whichever occurs first, replace the shear pins having P/N 190-15178-003 and P/N 190-15181-003, in the rear outboard and inboard shear pin assembly in the right- and left-hand pylons, with new shear pins having P/N 190-15178-005 and P/N 190-15181-005, respectively, in accordance with the Accomplishment Instructions of EMBRAER Service Bulletin 190LIN-54-0001, dated June 21, 2010.

FAA AD Differences

Note 1: This AD differs from the MCAI and/or service information as follows: The MCAI allows credit for previous installation of internal shear pins in accordance with EMBRAER 190 Aircraft Maintenance Manual Task 54-50-00-400, Revision 19, dated July 15, 2010. This AD does not allow credit for this task; however, under the provisions of paragraph (i) of this AD, we will consider requests for an alternative method of compliance.

Other FAA AD Provisions

(i) The following provisions also apply to this AD:

(1) *Alternative Methods of Compliance (AMOCs):* The Manager, International Branch, ANM-116, Transport Airplane Directorate, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the International Branch, send it to ATTN: Cindy Ashforth, Aerospace Engineer, International Branch, ANM-116, Transport Airplane Directorate, FAA, 1601 Lind Avenue, SW., Renton, Washington 98057-3356; telephone (425) 227-2768; fax (425) 227-1149. Information may be e-mailed to: 9-ANM-116-AMOC-REQUESTS@faa.gov. Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office. The AMOC approval letter must specifically reference this AD.

(2) *Airworthy Product:* For any requirement in this AD to obtain corrective actions from a manufacturer or other source, use these actions if they are FAA-approved. Corrective actions are considered FAA-approved if they are approved by the State of Design Authority (or their delegated agent). You are required to assure the product is airworthy before it is returned to service.

Related Information

(j) Refer to MCAI Agência Nacional de Aviação Civil (ANAC) Airworthiness Directive 2010-08-02, dated September 20, 2010; and EMBRAER Service Bulletins 190-54-0010, dated May 19, 2010, and 190LIN-54-0001, dated June 21, 2010; for related information.

Material Incorporated by Reference

(k) You must use EMBRAER Service Bulletin 190-54-0010, dated May 19, 2010; or EMBRAER Service Bulletin 190LIN-54-0001, dated June 21, 2010; as applicable; to do the actions required by this AD, unless the AD specifies otherwise.

(1) The Director of the Federal Register approved the incorporation by reference of this service information under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) For service information identified in this AD, contact Empresa Brasileira de Aeronáutica S.A. (EMBRAER), Technical Publications Section (PC 060), Av. Brigadeiro Faria Lima, 2170—Putim—12227-901 São Jose dos Campos—SP—BRASIL; telephone +55 12 3927-5852 or +55 12 3309-0732; fax +55 12 3927-7546; e-mail distrib@embraer.com.br; Internet <http://www.flyembraer.com>.

(3) You may review copies of the service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington. For information on the availability of this material at the FAA, call 425-227-1221.

(4) You may also review copies of the service information that is incorporated by reference at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call 202-741-6030, or go to: http://www.archives.gov/federal_register/code_of_federal_regulations/ibr_locations.html.

Issued in Renton, Washington, on August 19, 2011.

Ali Bahrami,

Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 2011-22028 Filed 9-8-11; 8:45 am]

BILLING CODE 4910-13-P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA-2011-0471; Directorate Identifier 2010-NM-219-AD; Amendment 39-16800; AD 2011-18-18]

RIN 2120-AA64

Airworthiness Directives; Bombardier, Inc. Model DHC-8-400 Series Airplanes

AGENCY: Federal Aviation Administration (FAA), Department of Transportation (DOT).

ACTION: Final rule.

SUMMARY: We are adopting a new airworthiness directive (AD) for the products listed above. This AD results from mandatory continuing airworthiness information (MCAI) originated by an aviation authority of another country to identify and correct an unsafe condition on an aviation product. The MCAI describes the unsafe condition as:

Several operators have reported pitch oscillations and/or elevator asymmetry caution lights illumination when flying with the autopilot engaged. Investigations revealed that loose rivets in the torque tube assemblies caused relative motion between the crank arms and torque tubes.

Loose rivets could result in excessive wear and subsequent significant backlash in the driving crank arms. This condition, if left uncorrected, will progressively get worse and degrade the controllability of the aeroplane.

We are issuing this AD to require actions to correct the unsafe condition on these products.

DATES: This AD becomes effective October 14, 2011.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in this AD as of October 14, 2011.

ADDRESSES: You may examine the AD docket on the Internet at <http://www.regulations.gov> or in person at the U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue, SE., Washington, DC.

FOR FURTHER INFORMATION CONTACT: Stephen Kowalski, Aerospace Engineer, Airframe and Mechanical Systems Branch, ANE-171, FAA, New York Aircraft Certification Office, 1600 Stewart Avenue, Suite 410, Westbury, New York 11590; telephone (516) 228-7327; fax (516) 794-5531.

SUPPLEMENTARY INFORMATION:

Discussion

We issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 to include an AD that would apply to the specified products. That NPRM was published in the **Federal Register** on May 12, 2011 (76 FR 27617). That NPRM proposed to correct an unsafe condition for the specified products. The MCAI states:

Several operators have reported pitch oscillations and/or elevator asymmetry caution lights illumination when flying with the autopilot engaged. Investigations revealed that loose rivets in the torque tube assemblies caused relative motion between the crank arms and torque tubes.

Loose rivets could result in excessive wear and subsequent significant backlash in the driving crank arms. This condition, if left

uncorrected, will progressively get worse and degrade the controllability of the aeroplane.

Required actions include doing an inspection for the part number of the left and right elevator torque tube assemblies and, if necessary, replacing the elevator torque tube assembly or replacing the elevator torque tube rivets, and re-identifying the assemblies. You may obtain further information by examining the MCAI in the AD docket.

Comments

We gave the public the opportunity to participate in developing this AD. We received no comments on the NPRM (76 FR 27617, May 12, 2011) or on the determination of the cost to the public.

Revised Service Information

Bombardier Service Bulletin 84-27-50, Revision ‘D,’ dated September 22, 2010, has been issued for clarification,

but adds no new actions. We have updated the references in paragraphs (g), (h), (i), and (m) of this AD to include Bombardier Service Bulletin 84-27-50, Revision ‘D,’ dated September 22, 2010. We have also updated Table 1 of this AD to allow credit for Bombardier Service Bulletin 84-27-50, Revision ‘C,’ dated July 26, 2010.

Conclusion

We reviewed the relevant data, and determined that air safety and the public interest require adopting the AD with the changes described previously. We also determined that these changes will not increase the economic burden on any operator or increase the scope of the AD.

Differences Between This AD and the MCAI or Service Information

We have reviewed the MCAI and related service information and, in

general, agree with their substance. But we might have found it necessary to use different words from those in the MCAI to ensure the AD is clear for U.S. operators and is enforceable. In making these changes, we do not intend to differ substantively from the information provided in the MCAI and related service information.

We might also have required different actions in this AD from those in the MCAI in order to follow our FAA policies. Any such differences are highlighted in a NOTE within the AD.

Costs of Compliance

We estimate that this AD will affect 66 products of U.S. registry. We also estimate the following costs to comply with this AD.

ESTIMATED COSTS

Action	Labor cost	Parts cost	Cost per product	Cost on U.S. operators
Inspection	2 × \$85 per hour = \$170	None	\$170	\$11,220

We estimate the following costs to do any necessary replacements that would

be required based on the results of the required inspection. We have no way of

determining the number of aircraft that might need these replacements:

ON-CONDITION COSTS

Action	Labor cost	Parts cost	Cost per product
Replace left torque tube	15 work-hours × \$85 per hour = \$1,275	\$4,354	\$5,629
Replace right torque tube	15 work-hours × \$85 per hour = \$1,275	5,913	7,188

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA’s authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. “Subtitle VII: Aviation Programs,” describes in more detail the scope of the Agency’s authority.

We are issuing this rulemaking under the authority described in “Subtitle VII, Part A, Subpart III, Section 44701: General requirements.” Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

We determined that this AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify this AD:

1. Is not a “significant regulatory action” under Executive Order 12866;
2. Is not a “significant rule” under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and
3. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared a regulatory evaluation of the estimated costs to comply with this AD and placed it in the AD docket.

Examining the AD Docket

You may examine the AD docket on the Internet at <http://www.regulations.gov>; or in person at the Docket Operations office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains the NPRM, the regulatory evaluation, any comments received, and other information. The street address for the Docket Operations office (telephone (800) 647-5527) is in the ADDRESSES section. Comments will be available in the AD docket shortly after receipt.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

■ 2. The FAA amends § 39.13 by adding the following new AD:

2011-18-18 Bombardier, Inc.: Amendment 39-16800. Docket No. FAA-2011-0471; Directorate Identifier 2010-NM-219-AD.

Effective Date

(a) This airworthiness directive (AD) becomes effective October 14, 2011.

Affected ADs

(b) None.

Applicability

(c) This AD applies to Bombardier, Inc. Model DHC-8-400, -401, and -402 airplanes; certificated in any category; serial numbers 4001 through 4305 inclusive.

Subject

(d) Air Transport Association (ATA) of America Code 27: Flight controls.

Reason

(e) The mandatory continuing airworthiness information (MCAI) states: Several operators have reported pitch oscillations and/or elevator asymmetry caution lights illumination when flying with the autopilot engaged. Investigations revealed that loose rivets in the torque tube assemblies caused relative motion between the crank arms and torque tubes.

Loose rivets could result in excessive wear and subsequent significant backlash in the

driving crank arms. This condition, if left uncorrected, will progressively get worse and degrade the controllability of the aeroplane.

Compliance

(f) You are responsible for having the actions required by this AD performed within the compliance times specified, unless the actions have already been done.

Inspection for Part Number

(g) At the applicable times identified in paragraphs (g)(1) and (g)(2) of this AD, do an inspection to determine the part numbers of the left and right elevator torque tubes, in accordance with the Accomplishment Instructions of Bombardier Service Bulletin 84-27-50, Revision 'D,' dated September 22, 2010. A review of airplane maintenance records is acceptable in lieu of this inspection if the part numbers of the left and right elevator torque tubes can be conclusively determined from that review.

(1) For airplanes that have accumulated 8,000 or more total flight hours as of the effective date of this AD: Within 2,000 flight hours after the effective date of this AD.

(2) For airplanes that have accumulated less than 8,000 total flight hours as of the effective date of this AD: Within 6,000 flight hours after the effective date of this AD, but before the accumulation of 10,000 total flight hours.

Corrective Actions

(h) If, as a result of the inspection required by paragraph (g) of this AD, any left elevator torque tube has part number (P/N) 82760709-009, at the applicable time in paragraph (g)(1) or (g)(2) of this AD, do the actions in paragraph (h)(1) or (h)(2) of this AD.

(1) Replace the elevator torque tube with a new elevator torque tube having P/N 82760709-011, in accordance with the

Accomplishment Instructions of Bombardier Service Bulletin 84-27-50, Revision 'D,' dated September 22, 2010.

(2) Replace the rivets in each elevator torque tube assembly with Hi Lite pins having P/N B0206001AG8 and collars having P/N HST1070CY, and re-identify the elevator torque tube assembly having P/N 82760709-009, in accordance with the Accomplishment Instructions of Bombardier Service Bulletin 84-27-50, Revision 'D,' dated September 22, 2010.

(i) If, as a result of the inspection required by paragraph (g) of this AD, any right elevator torque tube has P/N 82760757-009, at the applicable time in paragraph (g)(1) or (g)(2) of this AD, do the actions in paragraph (i)(1) or (i)(2) of this AD.

(1) Replace the elevator torque tube with a new elevator torque tube having P/N 82760757-011, in accordance with the Accomplishment Instructions of Bombardier Service Bulletin 84-27-50, Revision 'D,' dated September 22, 2010.

(2) Replace the rivets in each elevator torque tube assembly with Hi Lite pins having P/N B0206001AG8 and collars having P/N HST1070CY, and re-identify the elevator torque tube assembly having P/N 82760757-009, in accordance with the Accomplishment Instructions of Bombardier Service Bulletin 84-27-50, Revision 'D,' dated September 22, 2010.

Credit for Actions Accomplished in Accordance With Previous Service Information

(j) Actions done before the effective date of this AD, in accordance with the service bulletins listed in table 1 of this AD, are considered acceptable for compliance with the corresponding action specified in this AD.

TABLE 1—CREDIT SERVICE BULLETINS

Service Bulletin	Revision	Date
Bombardier Service Bulletin 84-27-50	Original	March 3, 2010.
Bombardier Service Bulletin 84-27-50	A	April 28, 2010.
Bombardier Service Bulletin 84-27-50	B	May 19, 2010.
Bombardier Service Bulletin 84-27-50	C	July 26, 2010.

Parts Installation

(k) As of the effective date of this AD, no person may install on any airplane an elevator torque tube assembly having P/N 82760709-009 or 82760757-009.

FAA AD Differences

Note 1: This AD differs from the MCAI and/or service information as follows: No differences.

Other FAA AD Provisions

(l) The following provisions also apply to this AD:

(1) *Alternative Methods of Compliance (AMOCs):* The Manager, New York Aircraft Certification Office (ACO), ANE-170, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR

39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the ACO, send it to ATTN: Program Manager, Continuing Operational Safety, FAA, New York ACO, 1600 Stewart Avenue, Suite 410, Westbury, New York 11590; telephone 516-228-7300; fax 516-794-5531. Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office. The AMOC approval letter must specifically reference this AD.

(2) *Airworthy Product:* For any requirement in this AD to obtain corrective actions from a manufacturer or other source, use these actions if they are FAA-approved. Corrective actions are considered FAA-approved if they are approved by the State of Design Authority

(or their delegated agent). You are required to assure the product is airworthy before it is returned to service.

Related Information

(m) Refer to MCAI Canadian Airworthiness Directive CF-2010-27, dated August 20, 2010; and Bombardier Service Bulletin 84-27-50, Revision 'D,' dated September 22, 2010; for related information.

Material Incorporated by Reference

(n) You must use Bombardier Service Bulletin 84-27-50, Revision 'D,' dated September 22, 2010, to do the actions required by this AD, unless the AD specifies otherwise.

(1) The Director of the Federal Register approved the incorporation by reference of this service information under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) For service information identified in this AD, contact Bombardier, Inc., Q-Series Technical Help Desk, 123 Garratt Boulevard, Toronto, Ontario M3K 1Y5, Canada; telephone 416-375-4000; fax 416-375-4539; e-mail thd.qseries@aero.bombardier.com; Internet <http://www.bombardier.com>.

(3) You may review copies of the service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington. For information on the availability of this material at the FAA, call 425-227-1221.

(4) You may also review copies of the service information that is incorporated by reference at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call 202-741-6030, or go to: http://www.archives.gov/federal_register/code_of_federal_regulations/ibr_locations.html.

Issued in Renton, Washington, on August 23, 2011.

Ali Bahrami,

Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 2011-22277 Filed 9-8-11; 8:45 am]

BILLING CODE 4910-13-P

SECURITIES AND EXCHANGE COMMISSION

17 CFR Parts 239, 249, 269 and 274

[Release Nos. 33-9256; 34-65244; 39-2478; IC-29780]

Amendments To Include New Applicant Types on Form ID

AGENCY: Securities and Exchange Commission.

ACTION: Final rule amendments.

SUMMARY: The Securities and Exchange Commission (“Commission”) is amending Form ID to include additional applicant types in order to facilitate processing of the form. Form ID is the application for access codes to file on the Commission’s Electronic Data Gathering, Analysis, and Retrieval (“EDGAR”) system. The purpose of introducing these new applicant types is to improve the Commission’s internal procedures for processing filings, including by routing Form ID filings to the appropriate internal office or division.

DATES: *Effective Date:* September 9, 2011.

FOR FURTHER INFORMATION CONTACT: Catherine Moore, Senior Special Counsel or Andrew Bernstein, Attorney-Adviser, Office of Clearance and Settlement, Division of Trading and Markets, Securities and Exchange Commission, 100 F Street, NE.,

Washington, DC 20549, at (202) 551-5710.

SUPPLEMENTARY INFORMATION:

I. Background

Form ID is filed by registrants, third party filers, or any of their respective agents, to whom the Commission previously has not assigned a Central Index Key (“CIK”) code, to request access codes in order to file in electronic format through EDGAR. EDGAR access codes include the CIK code, the CIK Confirmation Code (“CCC”), Password (“PW”), and Password Modification Authorization Code (“PMAC”).¹

Currently, Form ID does not differentiate applicants by specific type and simply lists as possible applicant types “filer,” “filing agent,” “training agent,” “transfer agent,” and “individual.” However, the number and type of persons that use EDGAR for submitting filings has increased since Form ID was first adopted by the Commission and may increase further following the adoption of various rules under the Dodd-Frank Wall Street Reform and Consumer Protection Act (“Dodd-Frank Act”).² Accordingly, the Commission is amending Form ID to list specific persons as applicant types on the form in order to allow the form to be assigned for processing within the Commission based on the type of applicant.

The new applicant types include persons that currently file on EDGAR but who are not separately listed on Form ID, persons that currently file forms with the Commission in paper but who may be required to file on EDGAR in the future, and persons who will be required to meet certain new filing obligations under the Securities Exchange Act of 1934 (“Exchange Act”), including provisions added by the Dodd-Frank Act. The amendments to Form ID also include corresponding definitions for each new applicant type.³ New applicants should select only one entity type when completing and submitting Form ID.⁴ If an applicant

¹ See EDGAR Filing Manual (Volume I) General Information (Section 2.4, Accessing EDGAR).

² The Dodd-Frank Wall Street Reform and Consumer Protection Act, Public Law 111-203, 124 Stat. 1376 (2010).

³ The definitions included in Form ID are to facilitate the correct selection of “applicant type” by a particular filer and are not intended to amend or otherwise change any provision of the federal securities laws or the regulations promulgated thereunder.

⁴ For purposes of Form ID, the term “person” includes either an individual or entity. If the applicant is also an “individual” as defined in the current Form ID, then the applicant must apply as both an “individual” as well as another appropriate applicant type that properly characterizes it.

qualifies as more than one of the applicant types listed on the form, it should select the applicant type related to the first filing it plans to submit on EDGAR. The access codes the applicant retrieves after Form ID is approved may be used to submit filings on EDGAR for any entity type (other than transfer agent) provided that such filing complies with all other applicable rules and regulations.⁵ Persons that have previously filed Form ID applications with the Commission are not required to re-file Form ID as a result of these amendments.

As more fully described below, the following applicant types and applicable definitions are being added to Form ID: Investment Company, Business Development Company or Insurance Company Separate Account, Institutional Investment Manager (Form 13F Filer), Non-Investment Company Applicant under the Investment Company Act of 1940, Large Trader, Clearing Agency, Municipal Advisor, Municipal Securities Dealer, Nationally Recognized Statistical Rating Organization, Security-Based Swap Data Repository, Security-Based Swap Dealer and Major Security-Based Swap Participant, and Security-Based Swap Execution Facility.

Investment Company, Business Development Company or Insurance Company Separate Account, Institutional Investment Manager (Form 13F Filer), and Non-Investment Company Applicant Under the Investment Company Act of 1940

Currently, a person that may fall within the applicant type of “Investment Company, Business Development Company or Insurance Company Separate Account,” “Institutional Investment Manager (Form 13F Filer),” or “Non-Investment Company Applicant under the Investment Company Act of 1940” may make submissions on EDGAR in electronic format without referencing the appropriate applicant type on Form ID. As such, the Commission is adding these specific applicant types to Form ID in order to facilitate processing of the form as filed by such persons. The applicant type of “Investment Company, Business Development Company or Insurance Company Separate Account” being added to Form ID includes persons that meet the definition of “investment company” in Section 3 of

⁵ Persons that are transfer agents must apply for a separate set of access codes even if they already submit filings on EDGAR in another capacity. See Securities Exchange Act Release No. 54865 (December 4, 2006), 71 FR 74698 (December 12, 2006) (File No. S7-14-06).