

List of Subjects in 12 CFR Part 226

Advertising, Federal Reserve System, Mortgages, Reporting and recordkeeping requirements, Truth in lending.

For the reasons set forth in the preamble, the Board amends Regulation Z, 12 CFR part 226, as set forth below:

PART 226—TRUTH IN LENDING (REGULATION Z)

1. The authority citation for part 226 continues to read as follows:

Authority: 12 U.S.C. 3806; 15 U.S.C. 1604 and 1637(c)(5).

2. In Supplement I to Part 226, under Section 226.32—Requirements for Certain Closed-End Home Mortgages, under Paragraph 32(a)(1)(ii), paragraph 2.xvi. is added.

Supplement I to Part 226—Official Staff Interpretations

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Subpart E—Special Rules for Certain Home Mortgage Transactions

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Section 226.32—Requirements for Certain Closed-End Home Mortgages

32(a) Coverage

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Paragraph 32(a)(1)(ii)

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2. Annual adjustment of \$400 amount.

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xvi. For 2011, \$592, reflecting a 2.2 percent increase in the CPI-U from June 2009 to June 2010, rounded to the nearest whole dollar.

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By order of the Board of Governors of the Federal Reserve System, acting through the Director of the Division of Consumer and Community Affairs under delegated authority, July 29, 2010.

Jennifer J. Johnson, Secretary of the Board.

[FR Doc. 2010-19101 Filed 8-3-10; 8:45 am]

BILLING CODE 6210-01-P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 25

[Docket No. NM430; Special Conditions No. 25-408-SC]

Special Conditions: Embraer ERJ 190-100 Series Airplane Seats With Non-Traditional, Large, Non-Metallic Panels

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final special conditions; request for comments.

SUMMARY: These special conditions are issued for the Embraer ERJ 190-100 series airplane. This airplane will have novel or unusual design features that include non-traditional, large, non-metallic panels that would affect survivability during a post-crash fire event. The applicable airworthiness regulations do not contain adequate or appropriate safety standards for this design feature. These special conditions contain the additional safety standards that the Administrator considers necessary to establish a level of safety equivalent to that established by the existing airworthiness standards.

DATES: The effective date of these special conditions is June 29, 2010. We must receive your comments by September 3, 2010.

ADDRESSES: You must mail two copies of your comments to: Federal Aviation Administration, Transport Airplane Directorate, Attn: Rules Docket (ANM-113), Docket No. NM430, 1601 Lind Avenue, SW., Renton, Washington 98057-3356. You may deliver two copies to the Transport Airplane Directorate at the above address. You must mark your comments: Docket No. NM430. You can inspect comments in the Rules Docket weekdays, except Federal holidays, between 7:30 a.m. and 4 p.m.

FOR FURTHER INFORMATION CONTACT: Cindy Ashforth, FAA, International Branch, ANM-116, Transport Airplane Directorate, Aircraft Certification Service, 1601 Lind Avenue, SW., Renton, Washington 98057-3356; telephone (425) 227-2768; facsimile (425) 227-1320.

SUPPLEMENTARY INFORMATION:

Future Requests for Installation of Seats With Non-Traditional, Large, Non-Metallic Panels

The FAA has determined that notice of, and opportunity for prior public comment on, these special conditions are impracticable because these

procedures would significantly delay issuance of the design approval and thus delivery of the affected aircraft. In addition, the substance of these special conditions has been subject to the public-comment process in several prior instances with no substantive comments received. The FAA therefore finds that good cause exists for making these special conditions effective upon issuance.

We anticipate that seats with non-traditional, large, non-metallic panels will be installed in other makes and models of airplanes. We have made the determination to require special conditions for all applications requesting the installation of seats with non-traditional, large, non-metallic panels until the airworthiness requirements can be revised to address this issue. Having the same standards across the range of airplane makes and models will ensure consistent ruling for the aviation industry.

Comments Invited

We invite interested people to take part in this rulemaking by sending written comments, data, or views. The most helpful comments reference a specific portion of the special conditions, explain the reason for any recommended change, and include supporting data. We ask that you send us two copies of written comments.

We will file in the docket all comments we receive, as well as a report summarizing each substantive public contact with FAA personnel about these special conditions. You can inspect the docket before and after the comment closing date. If you wish to review the docket in person, go to the address in the ADDRESSES section of this preamble between 7:30 a.m. and 4 p.m., Monday through Friday, except Federal holidays.

We will consider all comments we receive by the closing date for comments. We will consider comments filed late if it is possible to do so without incurring expense or delay. We may change these special conditions based on the comments we receive.

If you want us to acknowledge receipt of your comments on these special conditions, include with your comments a self-addressed, stamped postcard on which you have written the docket number. We will stamp the date on the postcard and mail it back to you.

Background

On March 9, 2010, Embraer applied for a change to Type Certificate No. A57NM for a new interior arrangement of 112 slim passenger seats in the ERJ 190-100 STD, ERJ 190-100 LR, and ERJ

190–100 IGW. The Embraer ERJ 190–100 series airplanes, currently approved under Type Certificate No. A57NM, are low-wing, conventional-tail, twin-turbofan, transport-category airplanes.

The applicable regulations to airplanes currently approved under Type Certificate No. A57NM do not require seats to meet the more stringent flammability standards required of large, non-metallic panels in the cabin interior. At the time the applicable rules were written, seats were designed with a metal frame covered by fabric, not with large, non-metallic panels. Seats also met the then-recently adopted standards for flammability of seat cushions. With the seat design being mostly fabric and metal, their contribution to a fire in the cabin had been minimized and was not considered a threat. For these reasons, seats did not need to be tested to heat-release and smoke-emission requirements.

Seat designs have now evolved to occasionally include non-traditional, large, non-metallic panels. Taken in total, the surface area of these panels is on the same order as the sidewall and overhead-stowage-bin interior panels. To provide the level of passenger protection intended by the airworthiness standards, these non-traditional, large, non-metallic panels in the cabin must meet the standards of Title 14 Code of Federal Regulations (CFR), part 25, Appendix F, parts IV and V, heat-release and smoke-emission requirements.

Type Certification Basis

Under the provisions of § 21.101 Embraer must show that the ERJ 190–100, as changed, continues to meet the applicable provisions of the regulations incorporated by reference in Type Certificate No. A57NM or the applicable regulations in effect on the date of application for the change. The regulations incorporated by reference in the type certificate are commonly referred to as the “original type certification basis.” The regulations incorporated by reference in A57NM are as follows: Part 25, as amended by Amendment 25–1 through Amendment 25–101. In addition, the certification basis includes certain special conditions, exemptions, or later amended sections of the applicable part that are not relevant to these special conditions.

If the Administrator finds that the applicable airworthiness regulations (i.e., 14 CFR part 25) do not contain adequate or appropriate safety standards for the ERJ 190–100 because of a novel or unusual design feature, special

conditions are prescribed under the provisions of § 21.16.

In addition to the applicable airworthiness regulations and special conditions, the ERJ 190–100 must comply with the fuel-vent and exhaust-emission requirements of 14 CFR part 34 and the noise-certification requirements of 14 CFR part 36;

The FAA issues special conditions, as defined in 14 CFR 11.19, in accordance with § 11.38, and they become part of the type-certification basis under § 21.101.

Special conditions are initially applicable to the model for which they are issued. Should the type certificate for that model be amended later to include any other model that incorporates the same novel or unusual design feature, or should any other model already included on the same type certificate be modified to incorporate the same novel or unusual design feature, the special conditions would also apply to the other model.

Novel or Unusual Design Features

The ERJ 190–100 series aircraft will incorporate the following novel or unusual design features: These models offer interior arrangements that include passenger seats that incorporate non-traditional, large, non-metallic panels in lieu of the traditional metal frame covered by fabric. The flammability properties of these panels have been shown to significantly affect the survivability of the cabin in the case of fire. These seats are considered a novel design for transport-category airplanes that include Amendment 25–61 and Amendment 25–66 in the certification basis, and were not considered when those airworthiness standards were established. The existing regulations do not provide adequate or appropriate safety standards for seat designs that incorporate non-traditional, large, non-metallic panels in their designs. To provide a level of safety that is equivalent to that afforded to the balance of the cabin, additional airworthiness standards, in the form of special conditions, are necessary. These special conditions supplement § 25.853. The requirements contained in these special conditions consist of applying the identical test conditions, required of all other large panels in the cabin, to seats with non-traditional, large, non-metallic panels.

A non-traditional, large, non-metallic panel, in this case, is defined as a panel with exposed surface areas greater than 1.5 square feet installed per seat place. The panel may consist of either a single component or multiple components in a concentrated area. Examples of parts of

the seat where these non-traditional panels are installed include, but are not limited to: Seat backs, bottoms and leg/foot rests, kick panels, back shells, credenzas, and associated furniture. Examples of traditional exempted parts of the seat include: Arm caps, armrest close-outs such as end bays and armrest-styled center consoles, food trays, video monitors, and shrouds.

Clarification of “Exposed”

“Exposed” is considered to include panels that are directly exposed to the passenger cabin in the traditional sense, and panels that are enveloped, such as by a dress cover. Traditional fabrics or leathers currently used on seats are excluded from these special conditions. These materials must still comply with §§ 25.853(a) and 25.853(c) if used as a covering for a seat cushion, or § 25.853(a) if installed elsewhere on the seat. Non-traditional, large, non-metallic panels covered with traditional fabrics or leathers will be tested without their coverings or covering attachments.

Discussion

In the early 1980s, the Federal Aviation Administration (FAA) conducted extensive research on the effects of post-crash flammability in the passenger cabin. As a result of this research and service experience, the FAA adopted new standards for interior surfaces associated with larger-surface-area parts. Specifically, the rules require measurement of heat release and smoke emission (part 25, Appendix F, parts IV and V) for the affected parts. Heat release has been shown to have a direct correlation with post-crash, fire-survival time. The materials that comply with the standards (i.e., § 25.853, titled “Compartment Interiors,” as amended by Amendments 25–61 and 25–66) extended survival time by approximately 2 minutes over materials that do not comply.

At the time Amendment 25–61 was written, the potential application of the requirement to seats was explored. The seat frame itself was not a concern because it was primarily made of aluminum and included only small amounts of non-metallic materials (for example, a food-tray table and armrest closeout). It was determined that the overall effect on survivability was negligible, whether or not these panels met the heat-release and smoke-emission requirements. The requirements therefore did not address seats, and the preambles to both Notice of Proposed Rule Making (NPRM) 85–10 and the final rule (Amendment 25–61) specifically noted that they were excluded “because the recently-adopted

standards for flammability of seat cushions will greatly inhibit involvement of the seats.”

In the late 1990s, when it became clear that seat designs were evolving to include large non-metallic panels with surface areas that would impact survivability during a cabin-fire event compared to partitions or galleys, the FAA issued Policy Memorandum 97–112–39. This memo noted that large-surface-area panels must comply with heat-release and smoke-emission requirements, even if they were attached to a seat. If the FAA had not issued such policy, seat designs would have been viewed as a loophole to the airworthiness standards that would result in an unacceptable decrease in survivability during a cabin-fire event.

Applicability

As discussed above, these special conditions are applicable to the ERJ 190–100. Should Embraer apply at a later date for a change to the type certificate to include another model incorporating the same novel or unusual design feature, the special conditions would apply to that model as well.

As discussed above, these special conditions are applicable to Embraer ERJ 190–100 series airplanes. It is not our intent, however, to require seats with non-traditional, large, non-metallic panels to meet § 25.853, which calls out appendix F, parts IV and V, if they are installed in cabins of airplanes that otherwise are not required to meet these standards. Because the heat-release and smoke-emission testing requirements of § 25.853, per appendix F, parts IV and V, are not part of the type-certification basis of the Model ERJ 190–100, these special conditions are only applicable if the Model ERJ 190–100 series airplanes are in 14 CFR part 121 operations. Section 121.312 requires compliance with the heat-release and smoke-emission testing requirements of § 25.853, for certain airplanes, irrespective of the type-certification bases of those airplanes. For Model ERJ 190–100 series airplanes, these are the airplanes that would be affected by these special conditions. Should Embraer apply at a later date for a supplemental type certificate to modify any other model included on Type Certificate No. A57NM, to incorporate the same novel or unusual design feature, the special conditions would apply to that model as well.

Conclusion

This action affects only certain novel or unusual design features on one model-series of airplanes. It is not a rule of general applicability and affects only

the applicant who applied to the FAA for approval of these features on the airplane.

Under standard practice, the effective date of final special conditions would be 30 days after the date of publication in the **Federal Register**; however, as the return-to-service date for the Embraer ERJ 190–100 series airplane is imminent, the FAA finds that good cause exists to make these special conditions effective upon issuance.

List of Subjects in 14 CFR Part 25

Aircraft, Aviation safety, Reporting and recordkeeping requirements.

■ The authority citation for these special conditions is as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701, 44702, 44704.

The Special Conditions

Accordingly, pursuant to the authority delegated to me by the Administrator, the following special conditions are issued as part of the type-certification basis for Embraer ERJ 190–100 series airplanes.

1. Except as provided in paragraph 3 of these special conditions, compliance with 14 CFR part 25, appendix F, parts IV and V, heat release and smoke emission, is required for seats that incorporate non-traditional, large, non-metallic panels that may be either a single component or multiple components in a concentrated area in their design.

2. The applicant may designate up to and including 1.5 square feet of non-traditional, non-metallic panel material per seat place that does not have to comply with special condition (1), above. A triple-seat assembly may have a total of 4.5 square feet excluded on any portion of the assembly (e.g., outboard-seat place 1 square foot; middle, 1 square foot; and inboard, 2.5 square feet).

3. Seats do not have to meet the test requirements of 14 CFR part 25, appendix F, parts IV and V, when installed in compartments that are not otherwise required to meet these requirements. Examples include:

- Airplanes with passenger capacities of 19 or fewer,
- Airplanes that do not have § 25.853, Amendment 25–61 or later, in their certification basis and do not need to comply with the requirements of 14 CFR 121.312, and
- Airplanes exempted from § 25.853, Amendment 25–61 or later.

Issued in Renton, Washington, on June 29, 2010.

Ali Bahrami,

Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 2010–19071 Filed 8–3–10; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 25

[Docket No. NM431; Special Conditions No. 25–409–SC]

Special Conditions: Bombardier Inc. Model CL–600–2E25 Series Airplane; Passenger Seats With Non-Traditional, Large, Non-Metallic Panels

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final special conditions; request for comments.

SUMMARY: These special conditions are issued for the Bombardier Inc. Model CL–600–2E25 Series Airplane. These airplanes will have a novel or unusual design feature associated with seats that include non-traditional, large, non-metallic panels that would affect survivability during a post-crash fire event. The applicable airworthiness regulations do not contain adequate or appropriate safety standards for this design feature. These special conditions contain the additional safety standards that the Administrator considers necessary to establish a level of safety equivalent to that established by the existing airworthiness standards.

DATES: The effective date of these special conditions is July 27, 2010. We must receive your comments by September 20, 2010.

ADDRESSES: You must mail two copies of your comments to: Federal Aviation Administration, Transport Airplane Directorate, Attn: Rules Docket (ANM–113), Docket No. NM431, 1601 Lind Avenue, SW., Renton, Washington 98057–3356. You may deliver two copies to the Transport Airplane Directorate at the above address. You must mark your comments: Docket No. NM431. You can inspect comments in the Rules Docket weekdays, except Federal holidays, between 7:30 a.m. and 4 p.m.

FOR FURTHER INFORMATION CONTACT: Alan Sinclair, FAA, Airframe/Cabin Safety Branch, ANM–115, Transport Airplane Directorate, Aircraft Certification Service, 1601 Lind Avenue, SW., Renton, Washington 98057–3356; telephone (425) 227–2195;