(LPC STG1 blade), part numbers (P/Ns) 1B6531, 1B6231–001, or 1A9031–001 (LPC STG1 blade set P/Ns 1B6521, 1B6221–001, and 1A9721–001), with a cutback leading edge, installed. These engines are installed on, but not limited to, Boeing 757 airplanes.

Unsafe Condition

d) This AD results from reports from PW that fan blade leading edge erosion can result in a fan thrust deterioration mode (FTDM) condition, which reduces the engine’s capability of producing full rated take-off thrust. We are issuing this AD to prevent loss of engine thrust from an FTDM condition, which could result in an inability to maintain safe flight.

Compliance

(e) You are responsible for having the actions required by this AD performed within the compliance times specified unless the actions have already been done.

Restoring the Fan Blade Leading Edge Contour

(f) Within 500 cycles-in-service after the effective date of this AD, restore the fan blade leading edge contour using one of the following:


2. For engines that are not installed on the airplane, use the Accomplishment Instructions, For Engines Not Installed on Aircraft, paragraphs 1. through 1.S of PW ASB PW2000 A72–729, Revision 1, dated December 8, 2009.

(g) Thereafter, repeat paragraphs (f)(1) or (f)(2) of this AD, within intervals of 1,000 cycles-since-last repair.

Alternative Methods of Compliance

(h) Pratt & Whitney PW2037, PW2040, PW2240, PW2337 Turbofan Engine Manual, Part No. 1A6231, Chapter/Section 72–31–12, Repair 14, is an approved alternative method of compliance to paragraphs (f)(1) and (f)(2) of this AD.

(i) Boeing 757 Airplane Flight Manual Document D631N002, Appendix 24, (Performance For Operation Of PW2000 Series Engines With Cutback Fan Blades Installed), is an approved alternative method of compliance to paragraphs (f)(1) and (f)(2) of this AD.

(j) The Manager, Engine Certification Office, has the authority to approve alternative methods of compliance for this AD if requested using the procedures found in 14 CFR 39.19.

Related Information

(k) Contact Mark Riley, Aerospace Engineer, Engine Certification Office, FAA, Engine and Propeller Directorate, 12 New England Executive Park, Burlington, MA 01803; e-mail: mark.riley@faa.gov; telephone (781) 238–7758, fax (781) 238–7199, for more information about this AD.

(l) Pratt & Whitney ASB PW2000 A72–729, Revision 1, dated December 8, 2009, pertains to the subject of this AD. Contact Pratt & Whitney, 400 Main St., East Hartford, CT 06108; telephone (860) 565–8770; fax (860) 565–4503, for a copy of this service information.

Issued in Burlington, Massachusetts, on March 18, 2010.

Francis A. Favara,
Manager, Engine and Propeller Directorate, Aircraft Certification Service.

[FR Doc. 2010–6583 Filed 3–24–10; 8:45 am]
BILLING CODE 4910–13–P

DEPARTMENT OF TRANSPORTATION
Federal Aviation Administration

14 CFR Part 39


RIN 2120–AA64


AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: We propose to adopt a new airworthiness directive (AD) for the products listed above. This proposed AD results from mandatory continuing airworthiness information (MCAI) issued by an aviation authority of another country to identify and correct an unsafe condition on an aviation product. The MCAI describes the unsafe condition as: Following a review of operational data of the Tay 651–54 engine, it has been found that the actual stress levels in the Tay 651–54 engine High Pressure Compressor (HPC) stages 1, 3, 6, 7, and 12 discs were higher than those originally assumed and therefore the approved lives needed to be reduced.

We are proposing this AD to prevent HPC stages 1, 3, 6, 7, and 12 discs from exceeding the approved reduced life limits, which could result in an uncontained failure of a disc and damage to the airplane.

DATES: We must receive comments on this proposed AD by April 26, 2010.

ADDRESSES: You may send comments by any of the following methods:

• Federal eRulemaking Portal: Go to http://www.regulations.gov and follow the instructions for sending your comments electronically.


• Hand Delivery: Deliver to Mail address above between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

• Fax: (202) 493–2251.

Contact Rolls-Royce Deutschland Ltd & Co KG; Eschenweg 11, D–15827 Blankenfelde-Mahlow, Germany; telephone +49 (0) 33 7086 1768; fax +49 (0) 33 7086 3356 for the service information identified in this proposed AD.

Examining the AD Docket

You may examine the AD docket on the Internet at http://www.regulations.gov or in person at the Docket Operations office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this proposed AD, the regulatory evaluation, any comments received, and other information. The street address for the Docket Operations office (telephone (800) 647–5527) is the same as the Mail address provided in the ADDRESSES section. Comments will be available in the AD docket shortly after receipt.

FOR FURTHER INFORMATION CONTACT: Tara Chaidez, Aerospace Engineer, Engine Certification Office, FAA, Engine and Propeller Directorate, 12 New England Executive Park, Burlington, MA 01803; e-mail: tara.chaidez@faa.gov; telephone (781) 238–7773; fax (781) 238–7199.

SUPPLEMENTARY INFORMATION:

Comments Invited

We invite you to send any written relevant data, views, or arguments about this proposed AD. Send your comments to an address listed under the ADDRESSES section. Include “Docket No. FAA–2010–0301; Directorate Identifier 2009–NE–22–AD” at the beginning of your comments. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of this proposed AD. We will consider all comments received by the closing date and may amend this proposed AD based on those comments.

We will post all comments we receive, without change, to http://www.regulations.gov, including any personal information you provide. We will also post a report summarizing each substantive verbal contact with FAA personnel concerning this proposed AD. Using the search function of the Web site, anyone can find and read the comments in any of our dockets, including, if provided, the name of the individual who sent the comment (or signed the comment on behalf of an association, business, labor union, etc.).

You may review the DOT’s complete

Discussion

The European Aviation Safety Agency (EASA), which is the Technical Agent for the Member States of the European Community, has issued EASA Airworthiness Directive 2009–0092, dated April 17, 2009 (referred to after this as “the MCAI”), to correct an unsafe condition for the specified products. The MCAI states:

Following a review of operational data of the Tay 651–54 engine, it has been found that the actual stress levels in the Tay 651–54 engine High Pressure Compressor (HPC) stages 1, 3, 6, 7 and 12 discs were higher than those originally assumed and therefore the approved lives needed to be reduced.

As Tay 651–54 service run HPC discs may be installed on Tay 620–15 and Tay 650–15 engine models, it is necessary to reduce the maximum approved lives of the affected HPC disc serial numbers installed on Tay 620–15 and Tay 650–15 engines as well.

The approved lives of the affected HPC stages 1, 3, 6, 7 and 12 discs specified in this Airworthiness Directive supersede the approved lives given in the Time Limits Manuals, Chapter 05–10–01.

Exceeding of the approved life limits could potentially result in non-contained disc failure.

You may obtain further information by examining the MCAI in the AD docket.

Relevant Service Information

Rolls-Royce Deutschland Ltd & Co KG has issued Alert Service Bulletin TAY–72–A1740, dated February 11, 2009. The actions described in this service information are intended to correct the unsafe condition identified in the MCAI.

FAA’s Determination and Requirements of This Proposed AD

This product has been approved by the aviation authority of Germany, and is approved for operation in the United States. Pursuant to our bilateral agreement with Germany, they have notified us of the unsafe condition described in the MCAI and service information referenced above. We are issuing this AD because we evaluated all information provided by EASA, and determined the unsafe condition exists and is likely to exist or develop on other products of the same type design.

Costs of Compliance

Based on the service information, we estimate that this proposed AD would affect about 10 products of U.S. registry. We also estimate that it would take about 1 work-hour per product to comply with this proposed AD. The average labor rate is $85 per work-hour. Required parts would cost about $100,000 per product. Based on these figures, we estimate the cost of the proposed AD on U.S. operators to be $1,000,850.

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA’s authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. “Subtitle VII: Aviation Programs,” describes in more detail the scope of the Agency’s authority.

We are issuing this rulemaking under the authority described in “Subtitle VII, Part A, Subpart III, Section 44701: General requirements.” Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

We determined that this proposed AD would not have federalism implications under Executive Order 13132. This proposed AD would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify this proposed regulation:
1. Is not a “significant regulatory action” under Executive Order 12866; and
2. Is not a “significant rule” under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and
3. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared a regulatory evaluation of the estimated costs to comply with this proposed AD and placed it in the AD docket.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

The Proposed Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA proposes to amend 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. The FAA amends § 39.13 by adding the following new AD:


Comments Due Date

(a) We must receive comments by April 26, 2010.

Affected Airworthiness Directives (ADs)

(b) None.

Applicability

(c) This AD applies to Rolls-Royce Deutschland Ltd & Co KG (RRD) models Tay 620–15, Tay 650–15, and Tay 651–54 turbofan engines. These engines are installed on, but not limited to, Fokker F28 Mark 0070 and Mark 0100 airplanes and Boeing 727 series airplanes.

Reason

(d) Following a review of operational data of the Tay 651–54 engine, it has been found that the actual stress levels in the Tay 651–54 engine High Pressure Compressor (HPC) stages 1, 3, 6, 7 and 12 discs were higher than those originally assumed and therefore the approved lives needed to be reduced. We are issuing this AD to prevent HPC stages 1, 3, 6, 7, and 12 discs from exceeding the approved reduced life limits, which could result in an uncontained failure of a disc and damage to the airplane.

Actions and Compliance

(e) Unless already done, within 30 days after the effective date of this AD, amend the approved Airworthiness Limitation Section to incorporate the new, reduced life limits as follows:

For Tay 651–54 Engines

(1) The maximum approved lives (MAL) of the High Pressure Compressor (HPC) rotor discs are reduced to the MALs specified in the following Table 1 of this AD:

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. The FAA amends § 39.13 by adding the following new AD:


Comments Due Date

(a) We must receive comments by April 26, 2010.

Affected Airworthiness Directives (ADs)

(b) None.

Applicability

(c) This AD applies to Rolls-Royce Deutschland Ltd & Co KG (RRD) models Tay 620–15, Tay 650–15, and Tay 651–54 turbofan engines. These engines are installed on, but not limited to, Fokker F28 Mark 0070 and Mark 0100 airplanes and Boeing 727 series airplanes.

Reason

(d) Following a review of operational data of the Tay 651–54 engine, it has been found that the actual stress levels in the Tay 651–54 engine High Pressure Compressor (HPC) stages 1, 3, 6, 7 and 12 discs were higher than those originally assumed and therefore the approved lives needed to be reduced. We are issuing this AD to prevent HPC stages 1, 3, 6, 7, and 12 discs from exceeding the approved reduced life limits, which could result in an uncontained failure of a disc and damage to the airplane.

Actions and Compliance

(e) Unless already done, within 30 days after the effective date of this AD, amend the approved Airworthiness Limitation Section to incorporate the new, reduced life limits as follows:

For Tay 651–54 Engines

(1) The maximum approved lives (MAL) of the High Pressure Compressor (HPC) rotor discs are reduced to the MALs specified in the following Table 1 of this AD:
For Tay 620–15 and Tay 650–15 Engines

(2) The MAL of certain High Pressure Compressor (HPC) rotor discs are reduced. The affected disc serial numbers and the reduced MAL are defined in Rolls-Royce Deutschland Non-Modification Service Bulletin TAY–72–A1740, dated February 11, 2009.

(3) Thereafter, except as provided in paragraph (f) of this AD, no alternative replacement times may be approved for these parts.

Other FAA AD Provisions

(f) Alternative Methods of Compliance (AMOCs): The Manager, Engine Certification Office, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19.

Related Information

(g) Refer to MCAI EASA Airworthiness Directive 2009–0092, dated April 17, 2009, and Rolls-Royce Deutschland Ltd & Co KG Alert Service Bulletin TAY–72–A1740, dated February 11, 2009, for related information. Contact Rolls-Royce Deutschland Ltd & Co KG; Eschenweg 11, D–15827 Blankenfelde-Mahlow, Germany; telephone +49 (0) 33 7086 3356, for a copy of this service information.

(h) Contact Tara Chaidez, Aerospace Engineer, Engine Certification Office, FAA, Engine and Propeller Directorate, 12 New England Executive Park, Burlington, MA 01803; e-mail: tara.chaidez@faa.gov; telephone (781) 238–7773; fax (781) 238–7199, for more information about this AD.

Issued in Burlington, Massachusetts, on March 18, 2010.

Francis A. Favara,
Manager, Engine and Propeller Directorate, Aircraft Certification Service.

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 71


AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: This action proposes to amend Class E Airspace at Mount Airy, NC, to accommodate the additional airspace needed for the Standard Instrument Approach Procedures (SIAPs) developed for Mount Airy-Surry County Airport. This action enhances the safety and airspace management of Instrument Flight Rules (IFR) operations at the airport.

DATES: Comments must be received on or before May 10, 2010.


FOR FURTHER INFORMATION CONTACT: Melinda Giddens, Operations Support Group, Eastern Service Center, Federal Aviation Administration, P.O. Box 20636, Atlanta, Georgia 30320; telephone (404) 305–5610.

SUPPLEMENTARY INFORMATION:

Comments Invited

Interested persons are invited to comment on this rule by submitting such written data, views, or arguments, as they may desire. Comments that provide the factual basis supporting the views and suggestions presented are particularly helpful in developing reasoned regulatory decisions on the proposal. Comments are specifically invited on the overall regulatory, aeronautical, economic, environmental, and energy-related aspects of the proposal.

Communications should identify both docket numbers (FAA Docket No. FAA–2010–0070; Airspace Docket No. 10–ASO–14) and be submitted in triplicate to the Docket Management System (see ADDRESSES section for address and phone number). You may also submit comments through the Internet at http://www.regulations.gov.

Comments wishing the FAA to acknowledge receipt of their comments on this action must submit with those comments a self-addressed stamped postcard on which the following statement is made: “Comments to Docket No. FAA–2010–0070; Airspace Docket No. 10–ASO–14.” The postcard will be date/time stamped and returned to the commenter.

All communications received before the specified closing date for comments will be considered before taking action on the proposed rule. The proposal contained in this notice may be changed in light of the comments received. A report summarizing each substantive public contact with FAA personnel concerned with this rulemaking will be filed in the docket.

Availability of NPRMs

An electronic copy of this document may be downloaded from and comments submitted through http://www.regulations.gov. Recently published rulemaking documents can also be accessed through the FAA’s Web page at http://www.faa.gov/airports_airtraffic/air_traffic/publications/airspace_amendments/.

You may review the public docket containing the proposal, any comments received, and any final disposition in person in the Dockets Office (see the ADDRESSES section for address and phone number) between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. An informal docket may also be examined during normal business hours at the office of the Eastern Service Center, Federal Aviation Administration, room 210, 1701 Columbus Avenue, College Park, Georgia 30337.

Persons interested in being placed on a mailing list for future NPRM’s should contact the FAA’s Office of Rulemaking, (202) 267–9677, to request a copy of Advisory circular No. 11–2A, Notice of Proposed Rulemaking distribution System, which describes the application procedure.