

202, Revision 3, dated December 10, 2008, and concurrently conduct a bonding inspection at each grommet location in accordance with paragraph 2.C. of BAE Systems (Operations) Limited Inspection Service Bulletin ISB.53-202, Revision 3, dated December 10, 2008. If unsatisfactory bonding is detected, before further flight, apply electro-conductive paste in accordance with Appendix 4 of BAE Systems (Operations) Limited Inspection Service Bulletin ISB.53-202, Revision 3, dated December 10, 2008.

(4) For airplanes on which any new P/N SL5185 grommets have been installed without having a bonding inspection prior to the effective date of this AD: Before or during the next scheduled repetitive inspection in accordance with paragraph (f)(1) of this AD, conduct a bonding inspection in accordance with paragraph 2.C. of BAE Systems (Operations) Limited Inspection Service Bulletin ISB.53-202, Revision 3, dated December 10, 2008. If unsatisfactory bonding is detected, before further flight, apply electro-conductive paste in accordance with Appendix 4 of BAE Systems (Operations) Limited Inspection Service Bulletin ISB.53-202, Revision 3, dated December 10, 2008.

(5) Replacing all existing grommets with new P/N SL5185 grommets on all panels, including the corresponding bonding inspections and the application of the electro-conductive paste as applicable, in accordance with BAE Systems (Operations) Limited Inspection Service Bulletin ISB.53-202, Revision 3, dated December 10, 2008, terminates the repetitive inspections required by paragraph (f)(1) of this AD.

(6) Visual inspections, temporary repairs, and replacements of the grommets are also acceptable for compliance with the corresponding requirements of paragraphs (f)(1), (f)(2)(i), (f)(2)(ii), (f)(3), and (f)(5) of this AD if done before the effective date of this AD in accordance with BAE Systems (Operations) Limited Inspection Service Bulletin ISB.53-202, Revision 1, dated June 4, 2008.

(7) Visual inspections, temporary repairs, replacements of the grommets, bonding inspections, and applications of conductive paste are also acceptable for compliance with the corresponding requirements of paragraphs (f)(1), (f)(2)(i), (f)(2)(ii), (f)(3), (f)(4), and (f)(5) of this AD if done before the effective date of this AD in accordance with BAE Systems (Operations) Limited Inspection Service Bulletin ISB.53-202, Revision 2, dated October 24, 2008.

(8) Bonding inspections and applications of conductive paste are also acceptable for compliance with the corresponding requirement of paragraphs (f)(2)(i), (f)(3), (f)(4), and (f)(5) of this AD if done before the effective date of this AD in accordance with BAE Systems (Operations) Limited All Operator Message 08-015V, Issue 1, dated August 22, 2008.

#### FAA AD Differences

**Note 2:** This AD differs from the MCAI and/or service information as follows: No differences.

#### Other FAA AD Provisions

(g) The following provisions also apply to this AD:

(1) *Alternative Methods of Compliance (AMOCs):* The Manager, International Branch, ANM-116, Transport Airplane Directorate, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. Send information to ATTN: Todd Thompson, Aerospace Engineer, International Branch, ANM-116, Transport Airplane Directorate, FAA, 1601 Lind Avenue, SW., Renton, Washington 98057-3356; telephone (425) 227-1175; fax (425) 227-1149. Before using any approved AMOC on any airplane to which the AMOC applies, notify your principal maintenance inspector (PMI) or principal avionics inspector (PAI), as appropriate, or lacking a principal inspector, your local Flight Standards District Office.

(2) *Airworthy Product:* For any requirement in this AD to obtain corrective actions from a manufacturer or other source, use these actions if they are FAA-approved. Corrective actions are considered FAA-approved if they are approved by the State of Design Authority (or their delegated agent). You are required to ensure the product is airworthy before it is returned to service.

(3) *Reporting Requirements:* For any reporting requirement in this AD, under the provisions of the Paperwork Reduction Act, the Office of Management and Budget (OMB) has approved the information collection requirements and has assigned OMB Control Number 2120-0056.

#### Related Information

(h) Refer to MCAI European Aviation Safety Agency Airworthiness Directive 2008-0180, dated September 30, 2008; and BAE Systems (Operations) Limited Inspection Service Bulletin ISB.53-202, Revision 3, dated December 10, 2008; for related information.

#### Material Incorporated by Reference

(i) You must use BAE Systems (Operations) Limited Inspection Service Bulletin ISB.53-202, Revision 3, dated December 10, 2008, to do the actions required by this AD, unless the AD specifies otherwise.

(1) The Director of the Federal Register approved the incorporation by reference of this service information under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) For service information identified in this AD, contact BAE Systems Regional Aircraft, 13850 McLearen Road, Herndon, Virginia 20171; telephone 703-736-1080; e-mail [raebusiness@baesystems.com](mailto:raebusiness@baesystems.com); Internet <http://www.baesystems.com/Businesses/RegionalAircraft/index.htm>.

(3) You may review copies of the service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington. For information on the availability of this material at the FAA, call 425-227-1221 or 425-227-1152.

(4) You may also review copies of the service information that is incorporated by reference at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call 202-741-6030, or go

to: [http://www.archives.gov/federal\\_register/code\\_of\\_federal\\_regulations/ibr\\_locations.html](http://www.archives.gov/federal_register/code_of_federal_regulations/ibr_locations.html).

Issued in Renton, Washington, on July 2, 2009.

**Ali Bahrami,**

Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. E9-16932 Filed 7-20-09; 8:45 am]

**BILLING CODE 4910-13-P**

## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 39

[Docket No. FAA-2008-1365; Directorate Identifier 2008-NM-076-AD; Amendment 39-15970; AD 2009-15-07]

RIN 2120-AA64

#### Airworthiness Directives; Airbus Model A319, A320, and A321 Series Airplanes

**AGENCY:** Federal Aviation Administration (FAA), Department of Transportation (DOT).

**ACTION:** Final rule.

**SUMMARY:** We are adopting a new airworthiness directive (AD) for the products listed above. This AD results from mandatory continuing airworthiness information (MCAI) originated by an aviation authority of another country to identify and correct an unsafe condition on an aviation product. The MCAI describes the unsafe condition as:

In 2005 a lateral runway excursion occurred on an A320 aircraft. Such excursions are classified as hazardous, with a large reduction in safety margins. Investigation has shown that the aircraft landed with the nose wheels rotated nearly 20 degrees from center. During subsequent tests on the removed BSCU [Braking and Steering Control Unit], a BSCU hardware failure was found, affecting the monitoring function, including the system reconfiguration management, and leading to a runaway of [the] Nose Wheel Steering [uncommanded steering].

\* \* \* \* \*

The unsafe condition is an uncommanded steering condition during takeoff or landing, which could result in departure of the airplane from the runway. We are issuing this AD to require actions to correct the unsafe condition on these products.

**DATES:** This AD becomes effective August 25, 2009.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in this AD as of August 25, 2009.

**ADDRESSES:** You may examine the AD docket on the Internet at <http://www.regulations.gov> or in person at the U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue, SE., Washington, DC.

**FOR FURTHER INFORMATION CONTACT:** Tim Dulin, Aerospace Engineer, International Branch, ANM-116, Transport Airplane Directorate, FAA, 1601 Lind Avenue, SW., Renton, Washington 98057-3356; telephone (425) 227-2141; fax (425) 227-1149.

**SUPPLEMENTARY INFORMATION:**

**Discussion**

We issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 to include an AD that would apply to the specified products. That NPRM was published in the **Federal Register** on January 7, 2009 (74 FR 664). That NPRM proposed to correct an unsafe condition for the specified products. The MCAI states:

In 2005 a lateral runway excursion occurred on an A320 aircraft. Such excursions are classified as hazardous, with a large reduction in safety margins. Investigation has shown that the aircraft landed with the nose wheels rotated nearly 20 degrees from center. During subsequent tests on the removed BSCU [Braking and Steering Control Unit], a BSCU hardware failure was found, affecting the monitoring function, including the system reconfiguration management, and leading to a runaway of [the] Nose Wheel Steering [uncommanded steering].

DGAC [Direction Générale de l'Aviation Civile] Airworthiness Directive (AD) F-1992-117-025(B), Revision 1 [which corresponds to FAA AD 94-24-07], mandated the BSCU upgrade in order to improve the steering logic, but this modification has shown not to be sufficient to address the identified failure mechanism.

A software modification is now implemented in BSCU standard 10 which improves the system reconfiguration management when this failure mechanism is detected.

BSCU standard 10 also includes other improvements—as detailed in the associated Service Bulletin.

This AD therefore mandates the modification or replacement of the BSCU standard 7, 9 or 9.1, by the BSCU standard 10.

The unsafe condition is an uncommanded steering condition during takeoff or landing, which could result in departure of the airplane from the runway. The corrective action also includes replacement of certain DUNLOP tires that are not compatible with BSCU standard 10. You may obtain further information by examining the MCAI in the AD docket.

**Comments**

We gave the public the opportunity to participate in developing this AD. We considered the comments received.

**Support for the AD**

JetBlue Airways Corporation supports the actions specified in the NPRM.

**Request To Relocate Certain Language**

Airbus suggests that we relocate the following sentence in the Reason section: “An uncommanded steering condition during takeoff or landing could result in departure of the airplane from the runway.” Airbus states that relocating that sentence away from the previous sentence, which addresses replacing tires, would avoid misinterpretation by association.

We acknowledge the Airbus comment and we have relocated the subject sentence and clarified that it is the statement of the unsafe condition in both the Discussion and Reason sections of this AD.

**Conclusion**

We reviewed the available data, including the comments received, and determined that air safety and the public interest require adopting the AD with the change described previously. We determined that this change will not increase the economic burden on any operator or increase the scope of the AD.

**Differences Between This AD and the MCAI or Service Information**

We have reviewed the MCAI and related service information and, in general, agree with their substance. But we might have found it necessary to use different words from those in the MCAI to ensure the AD is clear for U.S. operators and is enforceable. In making these changes, we do not intend to differ substantively from the information provided in the MCAI and related service information.

We might also have required different actions in this AD from those in the MCAI in order to follow our FAA policies. Any such differences are highlighted in a NOTE within the AD.

**Costs of Compliance**

We estimate that this AD will affect 591 products of U.S. registry. We also estimate that it will take about 3 work-hours per product to comply with the basic requirements of this AD. The average labor rate is \$80 per work-hour. Required parts will cost about \$0 per product. Where the service information lists required parts costs that are covered under warranty, we have assumed that there will be no charge for these parts. As we do not control

warranty coverage for affected parties, some parties may incur costs higher than estimated here. Based on these figures, we estimate the cost of this AD to the U.S. operators to be \$141,840, or \$240 per product.

**Authority for This Rulemaking**

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. “Subtitle VII: Aviation Programs,” describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in “Subtitle VII, Part A, Subpart III, Section 44701: General requirements.” Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

**Regulatory Findings**

We determined that this AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify this AD:

1. Is not a “significant regulatory action” under Executive Order 12866;
2. Is not a “significant rule” under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and
3. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared a regulatory evaluation of the estimated costs to comply with this AD and placed it in the AD docket.

**Examining the AD Docket**

You may examine the AD docket on the Internet at <http://www.regulations.gov>; or in person at the Docket Operations office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains the NPRM, the regulatory evaluation, any comments received, and other information. The street address for the Docket Operations office (telephone

(800) 647-5527) is in the **ADDRESSES** section. Comments will be available in the AD docket shortly after receipt.

#### List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

#### Adoption of the Amendment

■ Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

#### PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

#### § 39.13 [Amended]

■ 2. The FAA amends § 39.13 by adding the following new AD:

**2009-15-07 Airbus:** Amendment 39-15970. Docket No. FAA-2008-1365; Directorate Identifier 2008-NM-076-AD.

#### Effective Date

(a) This airworthiness directive (AD) becomes effective August 25, 2009.

#### Affected ADs

(b) None.

#### Applicability

(c) This AD applies to Airbus Model A319-111, -112, -113, -114, -115, -131, -132, and -133; A320-111, -211, -212, -214, -231, -232, and -233; and A321-111, -112, -131, -211, -212, -213, -231, and -232 series airplanes; certificated in any category; equipped with one conventional pre-Enhanced Manufacture and Maintainability (pre-EMM) Braking and Steering Control Unit (BSCU), having the part numbers specified in paragraph (c)(1), (c)(2), or (c)(3) of this AD.

(1) C20216332292C (standard 7) installed by Airbus Modification 24449 in production, or by Airbus Service Bulletin A320-32-1124 in service.

(2) C202163372D32 (standard 9) installed by Airbus Modification 31106 in production, or by Airbus Service Bulletin A320-32-1227 or A320-32-1232 in service.

(3) C202163382D32 (standard 9.1) installed by Airbus Modification 32500 in production, or by Airbus Service Bulletin A320-32-1254 in service.

#### Subject

(d) Air Transport Association (ATA) of America Code 32: Landing gear.

#### Reason

(e) The mandatory continuing airworthiness information (MCAI) states:

In 2005 a lateral runway excursion occurred on an A320 aircraft. Such excursions are classified as hazardous, with a large reduction in safety margins. Investigation has shown that the aircraft landed with the nose wheels rotated nearly

20 degrees from center. During subsequent tests on the removed BSCU [Braking and Steering Control Unit], a BSCU hardware failure was found, affecting the monitoring function, including the system reconfiguration management, and leading to a runaway of [the] Nose Wheel Steering [uncommanded steering].

DGAC [Direction Générale de l'Aviation Civile] Airworthiness Directive (AD) F-1992-117-025(B), Revision 1 [which corresponds to FAA AD 94-24-07], mandated the BSCU upgrade in order to improve the steering logic, but this modification has shown not to be sufficient to address the identified failure mechanism.

A software modification is now implemented in BSCU standard 10 which improves the system reconfiguration management when this failure mechanism is detected.

BSCU standard 10 also includes other improvements—as detailed in the associated Service Bulletin.

This AD therefore mandates the modification or replacement of the BSCU standard 7, 9 or 9.1, by the BSCU standard 10.

The unsafe condition is an uncommanded steering condition during takeoff or landing, which could result in departure of the airplane from the runway. The corrective action also includes replacement of certain DUNLOP tires that are not compatible with BSCU standard 10.

#### Actions and Compliance

(f) Unless already done, do the following actions.

(1) Within 18 months after the effective date of this AD: Modify or replace the BSCU in accordance with the Accomplishment Instructions of Airbus Mandatory Service Bulletin A320-32-1336, Revision 01, dated January 10, 2008; and inspect the airplane to determine if DUNLOP tires 46x16-20 having part number (P/N) 11659 T or 11661 T are installed. If those tires are installed, before further flight, replace with acceptable tires using a method approved by either the Manager, International Branch, ANM-116, Transport Airplane Directorate, FAA; or the European Aviation Safety Agency (EASA) (or its delegated agent). Accomplishment of the applicable requirements in this paragraph terminates the requirements of AD 94-24-07, amendment 39-9080.

(2) Previous accomplishment of the modification or replacement of the BSCU before the effective date of this AD in accordance with Airbus Mandatory Service Bulletin A320-32-1336, dated September 19, 2007, meets the requirements of paragraph (f)(1) of this AD.

#### FAA AD Differences

**Note 1:** This AD differs from the MCAI and/or service information as follows: Although the MCAI and service information do not provide procedures for replacing the tires as specified in paragraph (f)(1) of this AD, this AD requires that you replace the tires using a method approved by either the Manager, International Branch, ANM-116, FAA, or the EASA (or its delegated agent).

#### Other FAA AD Provisions

(g) The following provisions also apply to this AD:

(1) *Alternative Methods of Compliance (AMOCs):* The Manager, International Branch, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. Send information to ATTN: Tim Dulin, Aerospace Engineer, International Branch, ANM-116, Transport Airplane Directorate, FAA, 1601 Lind Avenue, SW., Renton, Washington 98057-3356; telephone (425) 227-2141; fax (425) 227-1149. Before using any approved AMOC on any airplane to which the AMOC applies, notify your appropriate principal inspector (PI) in the FAA Flight Standards District Office (FSDO), or lacking a PI, your local FSDO.

(2) *Airworthy Product:* For any requirement in this AD to obtain corrective actions from a manufacturer or other source, use these actions if they are FAA-approved. Corrective actions are considered FAA-approved if they are approved by the State of Design Authority (or their delegated agent). You are required to assure the product is airworthy before it is returned to service.

(3) *Reporting Requirements:* For any reporting requirement in this AD, under the provisions of the Paperwork Reduction Act, the Office of Management and Budget (OMB) has approved the information collection requirements and has assigned OMB Control Number 2120-0056.

#### Related Information

(h) Refer to MCAI EASA Airworthiness Directive 2008-0048, dated February 28, 2008; and Airbus Mandatory Service Bulletin A320-32-1336, Revision 01, dated January 10, 2008; for related information.

#### Material Incorporated by Reference

(i) You must use Airbus Mandatory Service Bulletin A320-32-1336, Revision 01, dated January 10, 2008, to do the actions required by this AD, unless the AD specifies otherwise.

(1) The Director of the Federal Register approved the incorporation by reference of this service information under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) For service information identified in this AD, contact Airbus, Airworthiness Office—EAS, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France; telephone +33 5 61 93 36 96; fax +33 5 61 93 44 51; e-mail: [account.airworth-eas@airbus.com](mailto:account.airworth-eas@airbus.com); Internet <http://www.airbus.com>.

(3) You may review copies of the service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington. For information on the availability of this material at the FAA, call 425-227-1221 or 425-227-1152.

(4) You may also review copies of the service information that is incorporated by reference at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call 202-741-6030, or go to: [http://www.archives.gov/federal\\_register/code\\_of\\_federal\\_regulations/ibr\\_locations.html](http://www.archives.gov/federal_register/code_of_federal_regulations/ibr_locations.html).

Issued in Renton, Washington, on July 2, 2009.

**Ali Bahrami,**

*Manager, Transport Airplane Directorate,  
Aircraft Certification Service.*

[FR Doc. E9-16937 Filed 7-20-09; 8:45 am]

BILLING CODE 4910-13-P

## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 39

[Docket No. FAA-2008-1201; Directorate Identifier 2008-NM-007-AD; Amendment 39-15922; AD 2009-11-12]

RIN 2120-AA64

#### Airworthiness Directives; Airbus Model A310 Series Airplanes

**AGENCY:** Federal Aviation Administration (FAA), Department of Transportation (DOT).

**ACTION:** Final rule.

**SUMMARY:** The FAA is superseding an existing airworthiness directive (AD), which applies to certain Airbus Model A310 series airplanes. That AD currently requires repetitive inspections of the fuselage skin to detect corrosion or fatigue cracking around and under the chafing plates of the wing root; repetitive inspections for fatigue cracking of frame 39, stringer 35; and corrective actions if necessary. The existing AD also provides for an optional terminating action for certain repetitive inspections, except for certain areas where corrosion was detected and reworked. This new AD reduces the intervals for accomplishing repetitive inspections in a certain area. This AD results from mandatory continuing airworthiness information originated by an aviation authority of another country to identify and correct an unsafe condition on an aviation product. We are issuing this AD to detect and correct fatigue cracks and corrosion around and under the chafing plates of the wing root, which could result in reduced structural integrity of the airplane.

**DATES:** This AD becomes effective August 25, 2009.

The Director of the Federal Register approved the incorporation by reference of certain publications listed in the AD as of August 25, 2009.

The Director of the Federal Register previously approved the incorporation by reference of Airbus Service Bulletin A310-53-2070, dated October 3, 1994, on June 3, 1998 (63 FR 23377, April 29, 1998).

**ADDRESSES:** For service information identified in this AD, contact Airbus

SAS—EAW (Airworthiness Office), 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France; telephone +33 5 61 93 36 96; fax +33 5 61 93 44 51; e-mail: [account.airworth-eas@airbus.com](mailto:account.airworth-eas@airbus.com); Internet <http://www.airbus.com>.

#### Examining the AD Docket

You may examine the AD docket on the Internet at <http://www.regulations.gov>; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this AD, the regulatory evaluation, any comments received, and other information. The address for the Docket Office (telephone 800-647-5527) is the Document Management Facility, U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue, SE., Washington, DC 20590.

#### FOR FURTHER INFORMATION CONTACT:

Vladimir Ulyanov, Aerospace Engineer, International Branch, ANM-116, Transport Airplane Directorate, FAA, 1601 Lind Avenue, SW., Renton, Washington 98057-3356; telephone (425) 227-1138; fax (425) 227-1149.

#### SUPPLEMENTARY INFORMATION:

##### Discussion

The FAA issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 to include an AD that supersedes AD 2004-14-06, amendment 39-13715 (69 FR 41401, July 9, 2004). The existing AD applies to certain Airbus Model A310 series airplanes. That NPRM was published in the **Federal Register** on November 13, 2008 (73 FR 67110). That NPRM proposed to continue to require repetitive inspections of the fuselage skin to detect corrosion or fatigue cracking around and under the chafing plates of the wing root; repetitive inspections for fatigue cracking of frame 39, stringer 35; and corrective actions if necessary. That NPRM also proposed to continue to provide for an optional terminating action for certain repetitive inspections, except for certain areas where corrosion was detected and reworked. In addition, that NPRM proposed to reduce the intervals for accomplishing the repetitive inspections in a certain area.

##### Comments

We provided the public the opportunity to participate in the development of this AD. We have considered the comment that has been received on the NPRM. FedEx supports the NPRM.

#### Revisions to Paragraphs (f), (h), (i), and (j) of This AD

We have revised paragraph (h) of this AD to give credit for actions done in accordance with Airbus Service Bulletin A310-53-2070, Revision 1, dated September 23, 1996. We also added Airbus Service Bulletin A310-53-2069, Revision 06, dated May 22, 2007, as an acceptable source of service information for compliance with the requirements of paragraphs (f) and (i) of this AD. We have also revised paragraph (j) of this AD to include a new Table 1 to specify the number, revision, and date of each service bulletin for which no reporting is required.

#### Conclusion

We reviewed the relevant data, considered the comment received, and determined that air safety and the public interest require adopting the AD with the changes described previously. We also determined that these changes will not increase the economic burden on any operator or increase the scope of the AD.

#### Costs of Compliance

This AD affects about 69 Model A310 series airplanes of U.S. registry. The actions that are required by AD 2004-14-06 and retained in this AD take about 68 work hours per airplane, at an average labor rate of \$80 per work hour. Based on these figures, the estimated cost of the currently required actions is \$375,360, or \$5,440 per airplane, per inspection cycle.

#### Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, Section 106, describes the authority of the FAA Administrator. Subtitle VII, Aviation Programs, describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section 44701, "General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

#### Regulatory Findings

We have determined that this AD will not have federalism implications under