

9, 2008, as applicable. Before an airplane may be dispatched with one spring missing or damaged, the instructions contained in Airbus A310 Operations Engineering Bulletin 160/2, dated October 2008; or A300-600 Operations Engineering Bulletin 121/1, dated May 2008; as applicable; must be accomplished by the flightcrew.

(ii) If two springs are missing or found damaged during any inspection required by paragraph (f)(1) of this AD, before further flight, replace the springs in accordance with Airbus Mandatory Service Bulletin A310-25A2199 or A300-25A6210, both dated July 9, 2008, as applicable.

(2) Replacing parts in accordance with Airbus Mandatory Service Bulletin A310-25A2199 or A300-25A6210, both dated July 9, 2008, as applicable, is not a terminating action for the repetitive inspections required in paragraph (f)(1) of this AD.

(3) As of the effective date of this AD, do not install an IPECO pilot or co-pilot mechanical seat P/N 3A218-000X-01-1 on any airplane, unless the seat has been inspected and modified, as applicable in accordance with Airbus Mandatory Service Bulletin A310-25A2199 or A300-25A6210, both dated July 9, 2008, as applicable.

(4) Within 6 months after the effective date of this AD, modify the airplane by replacing the pilot and co-pilot mechanical seats P/N 3A218-000X-01-1 with P/N 3A218-000X-01-2 seats, in accordance with Airbus Mandatory Service Bulletin A310-25-2202

or A300-25-6214, both dated February 3, 2009, as applicable.

(5) Installing both pilot and co-pilot seats P/N 3A218-000X-01-2 in accordance with Airbus Mandatory Service Bulletin A310-25-2202 or A300-25-6214, both dated February 3, 2009, as applicable, on any airplane is a terminating action for the repetitive inspections required by paragraph (f)(1) of this AD for that airplane.

(6) As of 6 months after the effective date of this AD, do not install an IPECO pilot or co-pilot mechanical seat P/N 3A218-000X-01-1 on any airplane.

(7) Although Airbus Mandatory Service Bulletins A310-25A2199 and A300-25A6210, both dated July 9, 2008, specify to submit certain information to the manufacturer, this AD does not include that requirement.

**FAA AD Differences**

**Note 1:** This AD differs from the MCAI and/or service information as follows: Although the MCAI and service information request to submit reporting information to Airbus, paragraph (f)(7) of this AD specifies that such submittal is not required.

**Other FAA AD Provisions**

(g) The following provisions also apply to this AD:

(1) Alternative Methods of Compliance (AMOCs): The Manager, International

Branch, ANM-116, Transport Airplane Directorate, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. Send information to ATTN: Dan Rodina, Aerospace Engineer, International Branch, ANM-116, Transport Airplane Directorate, FAA, 1601 Lind Avenue, SW., Renton, Washington 98057-3356; telephone (425) 227-2125; fax (425) 227-1149. Before using any approved AMOC on any airplane to which the AMOC applies, notify your principal maintenance inspector (PMI) or principal avionics inspector (PAI), as appropriate, or lacking a principal inspector, your local Flight Standards District Office. The AMOC approval letter must specifically reference this letter.

(2) Airworthy Product: For any requirement in this AD to obtain corrective actions from a manufacturer or other source, use these actions if they are FAA-approved. Corrective actions are considered FAA-approved if they are approved by the State of Design Authority (or their delegated agent). You are required to assure the product is airworthy before it is returned to service.

**Related Information**

(h) Refer to MCAI European Aviation Safety Agency Airworthiness Directive 2009-0045, dated February 27, 2009, and the service information listed in Table 1 of this AD for related information.

TABLE 1—SERVICE INFORMATION

Airbus service information	Date
A300-600 Operations Engineering Bulletin 121/1 .....	May 2008.
A310 Operations Engineering Bulletin 160/2 .....	October 2008.
Mandatory Service Bulletin A300-25-6214 .....	February 3, 2009.
Mandatory Service Bulletin A300-25A6210 .....	July 9, 2008.
Mandatory Service Bulletin A310-25-2202 .....	February 3, 2009.
Mandatory Service Bulletin A310-25A2199 .....	July 9, 2008.

Issued in Renton, Washington, on July 6, 2009.

Ali Bahrami,

Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. E9-16939 Filed 7-15-09; 8:45 am]

BILLING CODE 4910-13-P

**DEPARTMENT OF TRANSPORTATION**

**Federal Aviation Administration**

**14 CFR Part 39**

[Docket No. FAA-2009-0616; Directorate Identifier 2009-NM-070-AD]

RIN 2120-AA64

**Airworthiness Directives; 328 Support Services GmbH Dornier Model 328-100 and -300 Airplanes**

**AGENCY:** Federal Aviation Administration (FAA), DOT.

**ACTION:** Notice of proposed rulemaking (NPRM).

**SUMMARY:** We propose to adopt a new airworthiness directive (AD) for the products listed above. This proposed AD results from mandatory continuing airworthiness information (MCAI) originated by an aviation authority of another country to identify and correct an unsafe condition on an aviation product. The MCAI describes the unsafe condition as:

A recent incident has been reported with a Dornier 328-100 aeroplane, where the right-hand (RH) power lever jammed in flight-idle position during the landing roll-out. The aeroplane was stopped by excessive braking.

The investigation by the operator revealed that the cockpit door locking device \* \* \* had fallen off the RH cockpit wall and blocked the RH power/condition lever pulley/cable cluster below the door. \* \* \*

This condition, if not corrected, could cause interference with the engine- and/or

flight control cables, possibly resulting in reduced control of the aeroplane.

\* \* \* \* \*

The proposed AD would require actions that are intended to address the unsafe condition described in the MCAI.

**DATES:** We must receive comments on this proposed AD by August 17, 2009.

**ADDRESSES:** You may send comments by any of the following methods:

- *Federal eRulemaking Portal:* Go to <http://www.regulations.gov>. Follow the instructions for submitting comments.

- *Fax:* (202) 493-2251.

- *Mail:* U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue, SE., Washington, DC 20590.

- *Hand Delivery:* U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-40, 1200 New Jersey Avenue, SE., Washington, DC, between 9 a.m. and 5

p.m., Monday through Friday, except Federal holidays.

For service information identified in this proposed AD, contact 328 Support Services GmbH, Global Support Center, P.O. Box 1252, D-82231 Wessling, Federal Republic of Germany; telephone +49 8153 88111 6666; fax +49 8153 88111 6565; e-mail [gsc.op@328support.de](mailto:gsc.op@328support.de); Internet <http://www.328support.de>. You may review copies of the referenced service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington. For information on the availability of this material at the FAA, call 425-227-1221 or 425-227-1152.

### Examining the AD Docket

You may examine the AD docket on the Internet at <http://www.regulations.gov>; or in person at the Docket Operations office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this proposed AD, the regulatory evaluation, any comments received, and other information. The street address for the Docket Operations office (telephone (800) 647-5527) is in the **ADDRESSES** section. Comments will be available in the AD docket shortly after receipt.

**FOR FURTHER INFORMATION CONTACT:** Tom Groves, Aerospace Engineer, International Branch, ANM-116, Transport Airplane Directorate, FAA, 1601 Lind Avenue, SW., Renton, Washington 98057-3356; telephone (425) 227-1503; fax (425) 227-1149.

### SUPPLEMENTARY INFORMATION:

#### Comments Invited

We invite you to send any written relevant data, views, or arguments about this proposed AD. Send your comments to an address listed under the **ADDRESSES** section. Include "Docket No. FAA-2009-0616; Directorate Identifier 2009-NM-070-AD" at the beginning of your comments. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of this proposed AD. We will consider all comments received by the closing date and may amend this proposed AD based on those comments.

We will post all comments we receive, without change, to <http://www.regulations.gov>, including any personal information you provide. We will also post a report summarizing each substantive verbal contact we receive about this proposed AD.

#### Discussion

The European Aviation Safety Agency (EASA), which is the Technical Agent

for the Member States of the European Community, has issued EASA Airworthiness Directive 2009-0082, dated April 7, 2009 (referred to after this as "the MCAI"), to correct an unsafe condition for the specified products. The MCAI states:

A recent incident has been reported with a Dornier 328-100 aeroplane, where the right-hand (RH) power lever jammed in flight-idle position during the landing roll-out. The aeroplane was stopped by excessive braking.

The investigation by the operator revealed that the cockpit door locking device Part Number 001A252A3914012 had fallen off the RH cockpit wall and blocked the RH power/condition lever pulley/cable cluster below the door. Although the affected aeroplane had been modified, the technical investigation showed that a loose Cockpit Door Locking device could also occur on 328-100 and 328-300 aeroplanes with a standard installation.

This condition, if not corrected, could cause interference with the engine- and/or flight control cables, possibly resulting in reduced control of the aeroplane.

For the reasons described above, this AD requires a one-time inspection of the cockpit door locking device and the surrounding area [for proper installation] and the reporting of all findings to the TC [type certificate] holder. This AD is considered to be an interim action and the retrofit of a new design may be implemented later.

The corrective action is re-torquing the attachment screws. You may obtain further information by examining the MCAI in the AD docket.

#### Relevant Service Information

328 Support Services GmbH has issued Service Bulletins SB-328-25-485 and SB-328J-25-235, both dated January 28, 2009. The actions described in this service information are intended to correct the unsafe condition identified in the MCAI.

#### FAA's Determination and Requirements of This Proposed AD

This product has been approved by the aviation authority of another country, and is approved for operation in the United States. Pursuant to our bilateral agreement with the State of Design Authority, we have been notified of the unsafe condition described in the MCAI and service information referenced above. We are proposing this AD because we evaluated all pertinent information and determined an unsafe condition exists and is likely to exist or develop on other products of the same type design.

#### Differences Between This AD and the MCAI or Service Information

We have reviewed the MCAI and related service information and, in

general, agree with their substance. But we might have found it necessary to use different words from those in the MCAI to ensure the AD is clear for U.S. operators and is enforceable. In making these changes, we do not intend to differ substantively from the information provided in the MCAI and related service information.

We might also have proposed different actions in this AD from those in the MCAI in order to follow FAA policies. Any such differences are highlighted in a NOTE within the proposed AD.

#### Costs of Compliance

Based on the service information, we estimate that this proposed AD would affect about 69 products of U.S. registry. We also estimate that it would take about 1 work-hour per product to comply with the basic requirements of this proposed AD. The average labor rate is \$80 per work-hour. Based on these figures, we estimate the cost of the proposed AD on U.S. operators to be \$5,520, or \$80 per product.

#### Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. "Subtitle VII: Aviation Programs," describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in "Subtitle VII, Part A, Subpart III, Section 44701: General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

#### Regulatory Findings

We determined that this proposed AD would not have federalism implications under Executive Order 13132. This proposed AD would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify this proposed regulation:

1. Is not a "significant regulatory action" under Executive Order 12866;

2. Is not a “significant rule” under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and

3. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared a regulatory evaluation of the estimated costs to comply with this proposed AD and placed it in the AD docket.

#### List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

#### The Proposed Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA proposes to amend 14 CFR part 39 as follows:

#### PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

##### § 39.13 [Amended]

2. The FAA amends § 39.13 by adding the following new AD:

**328 Support Services GmbH (Formerly, AvCraft Aerospace GmbH, Formerly Fairchild Dornier GmbH, Formerly Dornier Luftfahrt GmbH):** Docket No. FAA–2009–0616; Directorate Identifier 2009–NM–070–AD.

#### Comments Due Date

(a) We must receive comments August 17, 2009.

#### Affected ADs

(b) None.

#### Applicability

(c) This AD applies to all 328 Support Services GmbH Dornier Model 328–100 and –300 airplanes, certificated in any category.

#### Subject

(d) Air Transport Association (ATA) of America Code 25: Equipment/furnishings.

#### Reason

(e) The mandatory continuing airworthiness information (MCAI) states:

A recent incident has been reported with a Dornier 328–100 aeroplane, where the right-hand (RH) power lever jammed in flight-idle position during the landing roll-out. The aeroplane was stopped by excessive braking.

The investigation by the operator revealed that the cockpit door locking device Part Number 001A252A3914012 had fallen off the RH cockpit wall and blocked the RH power/condition lever pulley/cable cluster below the door. Although the affected aeroplane had been modified, the technical

investigation showed that a loose Cockpit Door Locking device could also occur on 328–100 and 328–300 aeroplanes with a standard installation.

This condition, if not corrected, could cause interference with the engine- and/or flight control cables, possibly resulting in reduced control of the aeroplane.

For the reasons described above, this AD requires a one-time inspection of the cockpit door locking device and the surrounding area [for proper installation] and the reporting of all findings to the TC [type certificate] holder. This AD is considered to be an interim action and the retrofit of a new design may be implemented later.

The corrective action is re-torquing the attachment screws.

#### Actions and Compliance

(f) Unless already done, do the following actions.

(1) Within 3 months after the effective date of this AD, do a detailed visual inspection of the cockpit door locking device and the surrounding area for proper installation, in accordance with the Accomplishment Instructions of 328 Support Services Service Bulletin SB–328–25–485 or SB–328J–25–235, both dated January 28, 2009, as applicable.

(2) If any discrepancy is found during the inspection specified in paragraph (f)(1) of this AD, before further flight, do the corrective action in accordance with the Accomplishment Instructions of 328 Support Services Service Bulletin SB–328–25–485 or SB–328J–25–235, both dated January 28, 2009, as applicable.

(3) Submit a report of the findings (both positive and negative) of the inspection required by paragraph (f)(1) of this AD to the Manager, Attention Dept P1, 328 Support Services GmbH, Customer Services, P.O.B. 1252, D–82231 Wessling, Fed. Rep. of Germany; Fax +49 (0) 8153 88111–6565, at the applicable time specified in paragraph (f)(3)(i) or (f)(3)(ii) of this AD. The report must include the inspection results, a description of any discrepancies found, the airplane serial number, and the number of landings and flight hours on the airplane.

(i) If the inspection was done on or after the effective date of this AD: Submit the report within 30 days after the inspection.

(ii) If the inspection was accomplished prior to the effective date of this AD: Submit the report within 30 days after the effective date of this AD.

#### FAA AD Differences

**Note 1:** This AD differs from the MCAI and/or service information as follows: No differences.

#### Other FAA AD Provisions

(g) The following provisions also apply to this AD:

(1) *Alternative Methods of Compliance (AMOCs):* The Manager, International Branch, ANM–116, Transport Standards, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. Send information to *Attn:* Tom Groves, Aerospace Engineer, International Branch, ANM–116, Transport Airplane Directorate, FAA, 1601 Lind

Avenue, SW., Renton, Washington 98057–3356; telephone (425) 227–1503; fax (425) 227–1149. Before using any approved AMOC on any airplane to which the AMOC applies, notify your principal maintenance inspector (PMI) or principal avionics inspector (PAI), as appropriate, or lacking a principal inspector, your local Flight Standards District Office. The AMOC approval letter must specifically reference this AD.

(2) *Airworthy Product:* For any requirement in this AD to obtain corrective actions from a manufacturer or other source, use these actions if they are FAA-approved. Corrective actions are considered FAA-approved if they are approved by the State of Design Authority (or their delegated agent). You are required to assure the product is airworthy before it is returned to service.

(3) *Reporting Requirements:* For any reporting requirement in this AD, under the provisions of the Paperwork Reduction Act, the Office of Management and Budget (OMB) has approved the information collection requirements and has assigned OMB Control Number 2120–0056.

#### Related Information

(h) Refer to MCAI European Aviation Safety Agency (EASA) Airworthiness Directive 2009–0082, dated April 7, 2009; and 328 Support Services Service Bulletins SB–328–25–485 and SB–328J–25–235, both dated January 28, 2009; for related information.

Issued in Renton, Washington, on July 6, 2009.

**Ali Bahrami,**

*Manager, Transport Airplane Directorate, Aircraft Certification Service.*

[FR Doc. E9–16940 Filed 7–15–09; 8:45 am]

**BILLING CODE 4910–13–P**

## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 39

[Docket No. FAA–2009–0642; Directorate Identifier 2009–NM–001–AD]

RIN 2120–AA64

#### Airworthiness Directives; Boeing Model 767 Airplanes

**AGENCY:** Federal Aviation Administration (FAA), DOT.

**ACTION:** Notice of proposed rulemaking (NPRM).

**SUMMARY:** We propose to adopt a new airworthiness directive (AD) for certain Boeing Model 767 series airplanes. This proposed AD would require inspections for scribe lines in the fuselage skin at skin lap joints, the skin at certain external approved repairs, the skin around external features such as antennas, and the skin at decals; and related investigative and corrective actions if necessary. This proposed AD