

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

■ Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

■ 2. The FAA amends § 39.13 by adding the following new AD:

2009-09-08 Boeing: Amendment 39-15894. Docket No. FAA-2008-1239; Directorate Identifier 2008-NM-131-AD.

Effective Date

(a) This airworthiness directive (AD) is effective June 11, 2009.

Affected ADs

(b) None.

Applicability

(c) This AD applies to Boeing Model 747-100, 747-100B, 747-100B SUD, 747-200B, 747-200C, 747-200F, 747-300, 747-400, 747-400D, 747-400F, 747SR, and 747SP series airplanes, certificated in any category; as identified in Boeing Alert Service Bulletin 747-53A2651, dated June 12, 2008.

Unsafe Condition

(d) This AD results from reports of cracks in the radius detail of the upper lobe doublers. We are issuing this AD to detect and correct cracks in the upper lobe doublers. Such cracks could result in significant degradation of the fuselage structure and reduce its ability to carry flight loads from the vertical stabilizer, which could adversely affect the controllability of the airplane.

Compliance

(e) Comply with this AD within the compliance times specified, unless already done.

Inspection(s) and Corrective Action

(f) At the applicable times specified in paragraph 1.E., "Compliance," of Boeing Alert Service Bulletin 747-53A2651, dated June 12, 2008, except as required by paragraph (i) of this AD, do an external surface high frequency eddy current inspection to detect cracks in the radius detail of the upper lobe doubler on both sides of the airplane, and the applicable corrective action, by accomplishing all the applicable actions specified in the Accomplishment Instructions of the service bulletin, except as required by paragraphs (g) and (h) of this AD. The applicable corrective action must be done before further flight. As applicable,

repeat the inspection thereafter at the applicable times specified in paragraph 1.E., "Compliance," of Boeing Alert Service Bulletin 747-53A2651, dated June 12, 2008.

(g) Where Boeing Alert Service Bulletin 747-53A2651, dated June 12, 2008, paragraph 3.B., Work Instructions, PART 3, Step 1, specifies a sealant application "from STA 2520 to STA 2521," this AD requires a sealant application "from STA 2491 to STA 2521" on both sides of the airplane.

(h) Where Boeing Alert Service Bulletin 747-53A2651, dated June 12, 2008, specifies to contact Boeing for repair instructions instead of repairing or replacing any cracked upper lobe doubler in accordance with the service bulletin, this AD requires, before further flight, repairing any cracked upper lobe doubler using a method approved in accordance with the procedures specified in paragraph (j) of this AD.

(i) Where Boeing Alert Service Bulletin 747-53A2651, dated June 12, 2008, specifies a compliance time after the date on the service bulletin, this AD requires compliance within the specified compliance time after the effective date of this AD.

Alternative Methods of Compliance (AMOCs)

(j)(1) The Manager, Seattle Aircraft Certification Office (ACO), FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. Send information to ATTN: Ivan Li, Aerospace Engineer, Airframe Branch, ANM-120S, FAA, Seattle ACO, 1601 Lind Avenue, SW., Renton, Washington 98057-3356; telephone (425) 917-6437; fax (425) 917-6590.

(2) To request a different method of compliance or a different compliance time for this AD, follow the procedures in 14 CFR 39.19. Before using any approved AMOC on any airplane to which the AMOC applies, notify your principal maintenance inspector (PMI) or principal avionics inspector (PAI), as appropriate, or lacking a principal inspector, your local Flight Standards District Office. The AMOC approval letter must specifically reference this AD.

(3) An AMOC that provides an acceptable level of safety may be used for any repair required by this AD, if it is approved by an Authorized Representative for the Boeing Commercial Airplanes Delegation Option Authorization Organization who has been authorized by the Manager, Seattle ACO, to make those findings. For a repair method to be approved, the repair must meet the certification basis of the airplane, and the approval must specifically refer to this AD.

Material Incorporated by Reference

(k) You must use Boeing Alert Service Bulletin 747-53A2651, dated June 12, 2008, to do the actions required by this AD, unless the AD specifies otherwise.

(1) The Director of the Federal Register approved the incorporation by reference of this service information under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) For service information identified in this AD, contact Boeing Commercial Airplanes, Attention: Data & Services Management, P.O. Box 3707, MC 2H-65,

Seattle, Washington 98124-2207; telephone 206-544-5000, extension 1; fax 206-766-5680; e-mail me.boecom@boeing.com; Internet <https://www.myboeingfleet.com>.

(3) You may review copies of the service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington. For information on the availability of this material at the FAA, call 425-227-1221 or 425-227-1152.

(4) You may also review copies of the service information that is incorporated by reference at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call 202-741-6030, or go to: http://www.archives.gov/federal_register/code_of_federal_regulations/ibr_locations.html.

Issued in Renton, Washington, on April 22, 2009.

Stephen P. Boyd,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. E9-9925 Filed 5-6-09; 8:45 am]

BILLING CODE 4910-13-P

DEPARTMENT OF TRANSPORTATION**Federal Aviation Administration****14 CFR Part 39**

[Docket No. FAA-2008-1070; Directorate Identifier 2008-NM-087-AD; Amendment 39-15893; AD 2009-09-07]

RIN 2120-AA64

Airworthiness Directives; Boeing Model 737-100, -200, -200C, -300, -400, and -500 Series Airplanes

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final rule.

SUMMARY: We are adopting a new airworthiness directive (AD) for all Boeing Model 737-100, -200, -200C, -300, -400, and -500 series airplanes. For all airplanes, this AD requires repetitive overhaul of the retract actuator beam of the main landing gear (MLG). For certain airplanes, this AD requires repetitive inspections for damage of the retract actuator beam, and related investigative and corrective actions if necessary. This AD results from reports of broken retract actuator beams of the MLG and the subsequent failure of the MLG to fully retract. We are issuing this AD to detect and correct broken retract actuator beams of the MLG, which could cause damage to the beam arm, hydraulic tubing, and flight control cables. Damage to the flight control cables could result in loss of control of the airplane.

DATES: This AD is effective June 11, 2009.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in the AD as of June 11, 2009.

ADDRESSES: For service information identified in this AD, contact Boeing Commercial Airplanes, P.O. Box 3707, Seattle, Washington 98124-2207; telephone 206-544-9990; fax 206-766-5682; e-mail DDCS@boeing.com; Internet <https://www.myboeingfleet.com>.

Examining the AD Docket

You may examine the AD docket on the Internet at <http://www.regulations.gov>; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this AD, the regulatory evaluation, any comments received, and other information. The address for the Docket Office (telephone 800-647-5527) is the Document Management Facility, U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue, SE., Washington, DC 20590.

FOR FURTHER INFORMATION CONTACT: Nancy Marsh, Aerospace Engineer, Airframe Branch, ANM-120S, FAA, Seattle Aircraft Certification Office, 1601 Lind Avenue, SW., Renton, Washington 98057-3356; telephone (425) 917-6440; fax (425) 917-6590.

SUPPLEMENTARY INFORMATION:

Discussion

We issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 to include an airworthiness directive (AD) that would apply to all Boeing Model 737-100, -200, -200C, -300, -400, and -500 series airplanes. That NPRM was published in the **Federal Register** on October 8, 2008 (73 FR 58906). For all airplanes, that NPRM proposed to require repetitive overhaul of the retract actuator beam of the main landing gear (MLG). For certain airplanes, that NPRM proposed to require repetitive inspections for damage of the retract actuator beam, and

related investigative and corrective actions if necessary.

Comments

We gave the public the opportunity to participate in developing this AD. We considered the comments received from the commenters.

Support for the NPRM

Boeing concurs with the contents of the NPRM.

Request To Change Overhaul Requirements

Continental Airlines (CAL) asks that we not mandate overhaul of the retract actuator beam using the instructions specified in the Accomplishment Instructions of Boeing Service Bulletin 737-32A1355, Revision 2, dated March 5, 2008. CAL states that the reason for release of Boeing Service Bulletin 737-32A1355, Revision 2, is Boeing's concern regarding shop process deficiencies at some repair facilities where correct overhaul procedures were not followed. CAL finds this reasoning detrimental to all operators that follow correct overhaul procedures at their repair facilities.

CAL landing gear components, including the retract actuator beams of the left and right main landing gear, are time controlled per the Boeing 737-300/-500 Maintenance Program and are scheduled to be overhauled at 10-year intervals at an FAA-approved landing gear overhaul facility. CAL also makes the following recommendations regarding Boeing Service Bulletin 737-32A1355, Revision 2:

- The stripping of all chrome and nickel plating specified in Step 4 of Figure 2 should be included as an option, as in the Boeing 737 Component Maintenance Manual (CMM), Section 32-00-05.
- The nital etch inspection of machined surfaces specified only in Step 9 of Figure 2 should not be limited to machined surfaces.
- The stress relieving of the part specified in Step 10 of Figure 2 should be an optional step, as specified in Boeing 737 CMM, Section 32-00-05.

- The shot peening of the entire part specified in Step 12 of Figure 2 should be limited to the machined areas of the part since the geometry of the actuator beam does not allow for effective shot peening of the entire area.

- A caution note for arc burns associated with grounding (similar to Step 14) should be included in Step 13 of Figure 2.

We disagree that using the instructions specified in the Accomplishment Instructions of Boeing Service Bulletin 737-32A1355, Revision 2, dated March 5, 2008, to perform the overhaul of the retract actuator beam should not be mandated. Revision 2 of Boeing Service Bulletin 737-32A1355 references improved overhaul procedures, and those procedures are required by this AD. However, according to the provisions of paragraph (i) of this AD, we may approve a request for using different overhaul procedures if the request includes data that prove that those procedures would provide an acceptable level of safety. We have not changed the AD in this regard.

We acknowledge the request for changes to Boeing Service Bulletin 737-32A1355, Revision 2. At the present time, Boeing has not issued a revised service bulletin with the changes. However, if Boeing Service Bulletin 737-32A1355, Revision 2, is revised after issuance of this AD, we might consider approving the revised service bulletin as an alternative method of compliance (AMOC) with the requirements of this AD. We have not changed the AD in this regard.

Conclusion

We reviewed the relevant data, considered the comments received, and determined that air safety and the public interest require adopting the AD as proposed.

Costs of Compliance

We estimate that this AD affects 652 airplanes of U.S. registry. The following table provides the estimated costs for U.S. operators to comply with this AD.

ESTIMATED COSTS

| Action/airplane group | Work hours | Average labor rate per hour | Parts cost | Cost per product | Number of U.S.-registered airplanes | Fleet cost |
|---|------------|-----------------------------|------------|----------------------------------|-------------------------------------|-------------|
| Overhaul for Group 1; Configurations 1, 2, and 3. | 64 | \$80 | None | \$5,120, per overhaul cycle .. | 652 | \$3,338,240 |
| Inspection for Group 1, Configuration 3. | 1 | 80 | None | \$80, per inspection cycle | 525 | 42,000 |

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. "Subtitle VII: Aviation Programs," describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in "Subtitle VII, Part A, Subpart III, Section 44701: General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

This AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that this AD:

- (1) Is not a "significant regulatory action" under Executive Order 12866,
- (2) Is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979), and
- (3) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

You can find our regulatory evaluation and the estimated costs of compliance in the AD Docket.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

■ 2. The FAA amends § 39.13 by adding the following new AD:

2009-09-07 Boeing: Amendment 39-15893. Docket No. FAA-2008-1070; Directorate Identifier 2008-NM-087-AD.

Effective Date

(a) This airworthiness directive (AD) is effective June 11, 2009.

Affected ADs

(b) None.

Applicability

(c) This AD applies to all Boeing Model 737-100, -200, -200C, -300, -400, and -500 series airplanes, certificated in any category.

Unsafe Condition

(d) This AD results from reports of broken retract actuator beams of the main landing gear (MLG) and the subsequent failure of the MLG to fully retract. We are issuing this AD to detect and correct broken retract actuator beams of the MLG, which could result in damage to the beam arm, hydraulic tubing, and flight control cables. Damage to the flight control cables could result in loss of control of the airplane.

Compliance

(e) Comply with this AD within the compliance times specified, unless already done.

Inspection and Related Investigative and Corrective Actions/Overhaul

(f) Except as provided by paragraphs (g) and (h) of this AD: At the applicable times specified in paragraph 1.E. of Boeing Service Bulletin 737-32A1355, Revision 2, dated March 5, 2008, inspect for damage of the retract actuator beam of the MLG and overhaul the retract actuator beam, as applicable, by doing all the applicable actions specified in the Accomplishment Instructions of Boeing Service Bulletin 737-32A1355, Revision 2, dated March 5, 2008. Do all applicable related investigative and corrective actions before further flight. Repeat the applicable inspection or overhaul thereafter at the applicable time specified in paragraph 1.E. of Boeing Service Bulletin 737-32A1355, Revision 2, dated March 5, 2008.

Exceptions to Service Information

(g) Where Boeing Service Bulletin 737-32A1355, Revision 2, dated March 5, 2008, specifies a compliance time after " * * * the date on this service bulletin," this AD requires compliance within the specified compliance time after the effective date of this AD.

(h) Boeing Service Bulletin 737-32A1355, Revision 2, dated March 5, 2008, specifies that the actions are for airplanes with new MLG retract actuator beams that have not been overhauled having part number (P/N) 65-46108-15 and subsequent dash numbers; and new or overhauled MLG retract actuator beams having P/N 65-46108-14 and previous dash numbers. However, this AD is not limited to new or overhauled beams. This AD requires that the actions required by

paragraph (f) of this AD be done on airplanes with any MLG retract actuator beam having those P/Ns.

Alternative Methods of Compliance (AMOCs)

(i)(1) The Manager, Seattle Aircraft Certification Office (ACO), FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. Send information to ATTN: Nancy Marsh, Aerospace Engineer, Airframe Branch, ANM-120S, FAA, Seattle Aircraft Certification Office, 1601 Lind Avenue, SW., Renton, Washington 98057-3356; telephone (425) 917-6440; fax (425) 917-6590; or e-mail information to 9-ANM-Seattle-ACO-AMOC-Requests@faa.gov.

(2) To request a different method of compliance or a different compliance time for this AD, follow the procedures in 14 CFR 39.19. Before using any approved AMOC on any airplane to which the AMOC applies, notify your principal maintenance inspector (PMI) or principal avionics inspector (PAI), as appropriate, or lacking a principal inspector, your local Flight Standards District Office.

(3) An AMOC that provides an acceptable level of safety may be used for any repair required by this AD, if it is approved by an Authorized Representative for the Boeing Commercial Airplanes Delegation Option Authorization Organization who has been authorized by the Manager, Seattle ACO, to make those findings. For a repair method to be approved the repair must meet the certification basis of the airplane, and the approval must specifically refer to this AD.

Material Incorporated by Reference

(j) You must use Boeing Service Bulletin 737-32A1355, Revision 2, dated March 5, 2008, to do the actions required by this AD, unless the AD specifies otherwise.

(1) The Director of the Federal Register approved the incorporation by reference of this service information under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) For service information identified in this AD, contact Boeing Commercial Airplanes, P.O. Box 3707, Seattle, Washington 98124-2207; telephone 206-544-9990; fax 206-766-5682; e-mail DDCS@boeing.com; Internet <https://www.myboeingfleet.com>.

(3) You may review copies of the service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington. For information on the availability of this material at the FAA, call 425-227-1221 or 425-227-1152.

(4) You may also review copies of the service information that is incorporated by reference at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call 202-741-6030, or go to: http://www.archives.gov/federal_register/code_of_federal_regulations/ibr_locations.html.

Issued in Renton, Washington, on April 22, 2009.

Stephen P. Boyd,

Acting Manager, Transport Airplane
Directorate, Aircraft Certification Service.

[FR Doc. E9-9926 Filed 5-6-09; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA-2008-1275; Directorate
Identifier 2007-NM-167-AD; Amendment
39-15892; AD 2009-09-06]

RIN 2120-AA64

Airworthiness Directives; Boeing Model 737-100, -200, -200C, -300, -400, and -500 Series Airplanes

AGENCY: Federal Aviation
Administration (FAA), DOT.

ACTION: Final rule.

SUMMARY: We are adopting a new airworthiness directive (AD) for certain Boeing Model 737-100, -200, -200C, -300, -400, and -500 series airplanes. This AD requires repetitive detailed and high frequency eddy current inspections to detect cracks of the backup intercostals and the upper sill of the forward airstair doorway, and applicable corrective actions. This AD also provides for an optional terminating action, which would eliminate the need for repetitive inspections. This AD results from a report indicating that cracks were found in the backup intercostals and upper sill web of the forward airstair doorway. We are issuing this AD to detect and correct fatigue cracking of the backup intercostals and upper sill web of the forward airstair doorway, which could result in a rapid loss of cabin pressure.

DATES: This AD is effective June 11, 2009.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in the AD as of June 11, 2009.

ADDRESSES: For service information identified in this AD, contact Boeing Commercial Airplanes, Attention: Data & Services Management, P.O. Box 3707, MC 2H-65, Seattle, Washington 98124-2207; telephone 206-544-5000, extension 1; fax 206-766-5680; e-mail me.boecom@boeing.com; Internet & fml; <https://www.myboeingfleet.com>.

Examining the AD Docket

You may examine the AD docket on the Internet at <http://>

www.regulations.gov; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this AD, the regulatory evaluation, any comments received, and other information. The address for the Docket Office (telephone 800-647-5527) is the Document Management Facility, U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue, SE., Washington, DC 20590.

FOR FURTHER INFORMATION CONTACT:

Alan Pohl, Aerospace Engineer,
Airframe Branch, ANM-120S, FAA,
Seattle Aircraft Certification Office,
1601 Lind Avenue, SW., Renton,
Washington 98057-3356; telephone
(425) 917-6450; fax (425) 917-6590.

SUPPLEMENTARY INFORMATION:

Discussion

We issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 to include an airworthiness directive (AD) that would apply to certain Boeing Model 737-100, -200, -200C, -300, -400, and -500 series airplanes. That NPRM was published in the **Federal Register** on December 5, 2008 (73 FR 74080). That NPRM proposed to require repetitive detailed and high frequency eddy current inspections to detect cracks of the backup intercostals and the upper sill of the forward airstair doorway, and applicable corrective actions. That NPRM also provided an optional terminating action, which would eliminate the need for repetitive inspections.

Comments

We gave the public the opportunity to participate in developing this AD. We considered the comments received from the two commenters.

Support for the NPRM

Boeing concurs with the contents of the NPRM.

Request for Clarification

Southwest Airlines requests more information regarding alternative procedures for airplanes that have had the airstair door deactivated per Boeing Service Bulletin 737-52-1092. Southwest Airlines asks whether instructions developed by operators, for open and close of an airstair door after deactivation per Boeing Service Bulletin 737-52-1092, will be considered approved equivalent procedures.

No alternative procedures have been established that have general FAA approval; however, according to the

provisions of paragraph (h) of the final rule we may approve requests for different compliance methods if the requests include data that prove that the new methods would provide an acceptable level of safety.

Conclusion

We reviewed the relevant data, considered the comments received, and determined that air safety and the public interest require adopting the AD as proposed.

Costs of Compliance

There are 1,712 airplanes of the affected design in the worldwide fleet. This AD affects 509 airplanes of U.S. registry. The inspections take 2 work hours per airplane, at an average labor rate of \$80 per work hour. Based on these figures, the estimated cost of the AD for U.S. operators is \$81,440, or \$160 per airplane, per inspection cycle.

The optional terminating action, if done, would take 9 work hours, at an average labor rate of \$80 per work hour. Required parts cost between \$533 and \$566 per airplane, depending on the airplane configuration. Based on these figures, the estimated cost of the optional terminating action would range between \$1,253 and \$1,286 per airplane, depending on the airplane configuration.

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. "Subtitle VII: Aviation Programs," describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in "Subtitle VII, Part A, Subpart III, Section 44701: General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

This AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and