

Actions and Compliance

(e) Within 30 hours time-in-service (TIS) or 30 days, whichever occurs first, unless already accomplished, do the following:

(f) Remove the left-hand and right-hand side cockpit door windows and replace them with airworthy cockpit door windows, part number (P/N) 3P5211A10152A1 (left-hand side window) and P/N 3P5211A48131A1 (right-hand side window). Install an emergency jettison strap, P/N 3G5610A04751, with each cockpit door window, and install an external emergency exit placard, P/N 212-072-636-109, on each cockpit door external side in accordance with steps 1. through 11. of the Compliance Instructions in Bollettino Tecnico No. 139-129, dated June 3, 2008, except you are not required to contact the manufacturer.

Differences Between This AD and the MCAI

(g) The MCAI requires compliance within 200 flight hours or 6 months and uses the term "flight hours." This AD uses the term "hours TIS" rather than "flight hours," and requires compliance within 30 hours TIS or 30 days, whichever occurs first. Also, this AD references specific steps in BT 139-129 to use in complying with the AD. Finally, this AD does not require you to contact Agusta S.p.A. AW139 Customer Support Engineering.

Other Information

(h) Alternative Methods of Compliance (AMOCs): The Manager, Safety Management Group, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. Send information to ATTN: Sharon Miles, Aerospace Engineer, Regulations and Policy Group, Rotorcraft Directorate, FAA, 2601 Meacham Blvd., Fort Worth, TX 76137; telephone (817) 222-5122; fax (817) 222-5961.

Related Information

(i) European Aviation Safety Agency MCAI Airworthiness Directive No. 2008-0108, dated June 5, 2008, contains related information.

Air Transport Association of America (ATA) Tracking Code

(j) ATA Code 5610: Flight Compartment Windows.

Material Incorporated by Reference

(k) You must use the specified portions of Bollettino Tecnico No. 139-129, dated June 3, 2008, to do the actions required.

(1) The Director of the Federal Register approved the incorporation by reference of this service information under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) For service information identified in this AD, contact Agusta, Product Support Italy, Via per Tornavento, 15 21019 Somma Lombardo, Varese Italy, telephone 39 (0331) 711111, fax 39 (0331) 711397, or at http://customersupport.agusta.com/technical_advice.php.

(3) You may review copies at FAA, Office of the Regional Counsel, Southwest Region, 2601 Meacham Blvd., Room 663, Fort Worth, Texas or at the National Archives and

Records Administration (NARA). For information on the availability of this material at NARA, call (202) 741-6030, or go to: <http://www.archives.gov/federal-register/cfr/ibr-locations.html>.

Issued in Fort Worth, Texas on February 12, 2009.

Scott A. Horn,

Acting Manager, Rotorcraft Directorate, Aircraft Certification Service.

[FR Doc. E9-4941 Filed 3-13-09; 8:45 am]

BILLING CODE 4910-13-P

DEPARTMENT OF TRANSPORTATION**Federal Aviation Administration****14 CFR Part 39**

[Docket No. FAA-2008-1330; Directorate Identifier 2008-NM-138-AD; Amendment 39-15839; AD 2009-06-03]

RIN 2120-AA64

Airworthiness Directives; Viking Air Limited Model DHC-7 Airplanes

AGENCY: Federal Aviation Administration (FAA), Department of Transportation (DOT).

ACTION: Final rule.

SUMMARY: We are adopting a new airworthiness directive (AD) for the products listed above. This AD results from mandatory continuing airworthiness information (MCAI) originated by an aviation authority of another country to identify and correct an unsafe condition on an aviation product. The MCAI describes the unsafe condition as:

Transport Canada has received numerous service difficulty reports concerning Viking DHC-7 and Bombardier DHC-8 aircraft fluorescent lamp holder damage due to overheating. It has been determined that lamp holder overheating is a result of arcing between the fluorescent tube pins and the lamp holder contacts when the tube is not properly seated during installation. Overheating of lamp holders, if not corrected, could generate fumes and smoke * * *.

* * * * *

The unsafe condition could result in an in-flight fire. We are issuing this AD to require actions to correct the unsafe condition on these products.

DATES: This AD becomes effective April 20, 2009.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in this AD as of April 20, 2009.

ADDRESSES: You may examine the AD docket on the Internet at <http://www.regulations.gov> or in person at the U.S. Department of Transportation, Docket Operations, M-30, West

Building Ground Floor, Room W12-140, 1200 New Jersey Avenue, SE., Washington, DC.

FOR FURTHER INFORMATION CONTACT:

Wing Chan, Aerospace Engineer, Systems and Flight Test Branch, ANE-172, FAA, New York Aircraft Certification Office, 1600 Stewart Avenue, Suite 410, Westbury, New York 11590; telephone (516) 228-7311; fax (516) 794-5531.

SUPPLEMENTARY INFORMATION:**Discussion**

We issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 to include an AD that would apply to the specified products. That NPRM was published in the **Federal Register** on December 23, 2008 (73 FR 78673). That NPRM proposed to correct an unsafe condition for the specified products. The MCAI states:

Transport Canada has received numerous service difficulty reports concerning Viking DHC-7 and Bombardier DHC-8 aircraft fluorescent lamp holder damage due to overheating. It has been determined that lamp holder overheating is a result of arcing between the fluorescent tube pins and the lamp holder contacts when the tube is not properly seated during installation. Overheating of lamp holders, if not corrected, could generate fumes and smoke, causing concern to passengers and crew.

This directive mandates repetitive inspection[s] for proper installation [and functioning] of fluorescent tubes and prohibits installation of non-arc-protected replacement fluorescent lamp ballasts.

The unsafe condition could result in an in-flight fire. The corrective actions include replacing any lamps that are not properly seated in the lamp holder, and replacing any broken, non-functioning lamp holders. Replacing all affected fluorescent lamp ballasts would terminate the repetitive inspections. You may obtain further information by examining the MCAI in the AD docket.

Comments

We gave the public the opportunity to participate in developing this AD. We received no comments on the NPRM or on the determination of the cost to the public.

Conclusion

We reviewed the available data and determined that air safety and the public interest require adopting the AD as proposed.

Differences Between This AD and the MCAI or Service Information

We have reviewed the MCAI and related service information and, in general, agree with their substance. But we might have found it necessary to use

different words from those in the MCAI to ensure the AD is clear for U.S. operators and is enforceable. In making these changes, we do not intend to differ substantively from the information provided in the MCAI and related service information.

We might also have required different actions in this AD from those in the MCAI in order to follow our FAA policies. Any such differences are highlighted in a NOTE within the AD.

Costs of Compliance

We estimate that this AD will affect 21 products of U.S. registry. We also estimate that it will take about 1 work-hour per product to comply with the basic requirements of this AD. The average labor rate is \$80 per work-hour. Based on these figures, we estimate the cost of this AD to the U.S. operators to be \$1,680, or \$80 per product.

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. "Subtitle VII: Aviation Programs," describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in "Subtitle VII, Part A, Subpart III, Section 44701: General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

We determined that this AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify this AD:

1. Is not a "significant regulatory action" under Executive Order 12866;
2. Is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and
3. Will not have a significant economic impact, positive or negative, on a substantial number of small entities

under the criteria of the Regulatory Flexibility Act.

We prepared a regulatory evaluation of the estimated costs to comply with this AD and placed it in the AD docket.

Examining the AD Docket

You may examine the AD docket on the Internet at <http://www.regulations.gov>; or in person at the Docket Operations office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains the NPRM, the regulatory evaluation, any comments received, and other information. The street address for the Docket Operations office (telephone (800) 647-5527) is in the ADDRESSES section. Comments will be available in the AD docket shortly after receipt.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

■ Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

■ 2. The FAA amends § 39.13 by adding the following new AD:

2009-06-03 Viking Air Limited (Formerly Bombardier, Inc.): Amendment 39-15839. Docket No. FAA-2008-1330; Directorate Identifier 2008-NM-138-AD.

Effective Date

(a) This airworthiness directive (AD) becomes effective April 20, 2009.

Affected ADs

(b) None.

Applicability

(c) This AD applies to Viking Air Limited Model DHC-7-1, DHC-7-100, DHC-7-101, DHC-7-102, and DHC-7-103 airplanes, certificated in any category; serial numbers 1 through 113 inclusive, with Modifications 7/2444 and 7/2445 incorporated.

Subject

(d) Air Transport Association (ATA) of America Code 33: Lights.

Reason

(e) The mandatory continuing airworthiness information (MCAI) states: "Transport Canada has received numerous service difficulty reports concerning Viking DHC-7 and Bombardier DHC-8 aircraft

fluorescent lamp holder damage due to overheating. It has been determined that lamp holder overheating is a result of arcing between the fluorescent tube pins and the lamp holder contacts when the tube is not properly seated during installation. Overheating of lamp holders, if not corrected, could generate fumes and smoke, causing concern to passengers and crew.

This directive mandates repetitive inspection[s] for proper installation [and functioning] of fluorescent tubes and prohibits installation of non-arc-protected replacement fluorescent lamp ballasts." The unsafe condition could result in an in-flight fire. The corrective actions include replacing any lamps that are not properly seated in the lamp holder, and replacing any broken, non-functioning lamp holders. Replacing all affected fluorescent lamp ballasts would terminate the repetitive inspections.

Actions and Compliance

(f) Unless already done, do the following actions.

(1) Within 1,000 flight hours after the effective date of this AD: Perform a visual inspection to ensure proper installation and functioning of the fluorescent tubes in the lamp holders, and perform all applicable corrective actions before further flight, in accordance with the Accomplishment Instructions of Viking Service Bulletin V7-33-01, dated February 28, 2008. Repeat the inspection thereafter at intervals not to exceed 1,000 flight hours.

(2) Replacing all fluorescent lamp ballasts having part number (P/N) BAO8006-1 and BA[O]8006-28-1 with new fluorescent lamp ballasts having P/N BR9000-21, in accordance with the Accomplishment Instructions of Viking Service Bulletin V7-33-01, dated February 28, 2008, terminates the repetitive inspections required by paragraph (f)(1) of this AD.

FAA AD Differences

Note 1: This AD differs from the MCAI and/or service information as follows: No differences.

Other FAA AD Provisions

(g) The following provisions also apply to this AD:

(1) Alternative Methods of Compliance (AMOCs): The Manager, New York Aircraft Certification Office (ACO), FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. Send information to ATTN: Wing Chan, Aerospace Engineer, Systems and Flight Test Branch, ANE-172, FAA, New York ACO, 1600 Stewart Avenue, Suite 410, Westbury, New York 11590; telephone (516) 228-7311; fax (516) 794-5531. Before using any approved AMOC on any airplane to which the AMOC applies, notify your appropriate principal inspector (PI) in the FAA Flight Standards District Office (FSDO), or lacking a PI, your local FSDO.

(2) Airworthy Product: For any requirement in this AD to obtain corrective actions from a manufacturer or other source, use these actions if they are FAA-approved. Corrective actions are considered FAA-

approved if they are approved by the State of Design Authority (or their delegated agent). You are required to assure the product is airworthy before it is returned to service.

(3) Reporting Requirements: For any reporting requirement in this AD, under the provisions of the Paperwork Reduction Act, the Office of Management and Budget (OMB) has approved the information collection requirements and has assigned OMB Control Number 2120-0056.

Related Information

(h) Refer to MCAI Canadian Airworthiness Directive CF-2008-27, dated July 4, 2008; and Viking Service Bulletin V7-33-01, dated February 28, 2008; for related information.

Material Incorporated by Reference

(i) You must use Viking Service Bulletin V7-33-01, dated February 28, 2008, to do the actions required by this AD, unless the AD specifies otherwise.

(1) The Director of the Federal Register approved the incorporation by reference of this service information under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) For service information identified in this AD, contact Viking Air Limited, 9574 Hampden Road, Sidney, British Columbia V8L 8V5, Canada; telephone 250-656-7227; fax 250-656-0673; e-mail technical.publications@vikingair.com; Internet <http://www.vikingair.com>.

(3) You may review copies of the service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington. For information on the availability of this material at the FAA, call 425-227-1221 or 425-227-1152.

(4) You may also review copies of the service information that is incorporated by reference at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call 202-741-6030, or go to: http://www.archives.gov/federal_register/code_of_federal_regulations/ibr_locations.html.

Issued in Renton, Washington, on February 27, 2009.

Ali Bahrami,

*Manager, Transport Airplane Directorate,
Aircraft Certification Service.*

[FR Doc. E9-4970 Filed 3-13-09; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA-2008-1072; Directorate Identifier 2008-NM-109-AD; Amendment 39-15838; AD 2009-06-02]

RIN 2120-AA64

Airworthiness Directives; Boeing Model 747-100, 747-100B, 747-100B SUD, 747-200B, 747-200C, 747-200F, 747-300, 747-400, 747SR, and 747SP Series Airplanes

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final rule.

SUMMARY: We are adopting a new airworthiness directive (AD) for certain Boeing Model 747-100, 747-100B, 747-100B SUD, 747-200B, 747-200C, 747-200F, 747-300, 747-400, 747SR, and 747SP series airplanes. This AD requires inspecting for skin cracks at the shear tie end fastener locations of the fuselage frames, and repairing cracks if necessary. This AD results from a widespread fatigue damage assessment of Model 747 airplanes. We are issuing this AD to detect and correct cracks in the fuselage skin that can propagate and grow, resulting in a loss of structural integrity and a sudden decompression of the airplane during flight.

DATES: This AD is effective *April 20, 2009*.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in the AD as of April 20, 2009.

ADDRESSES: For service information identified in this AD, contact Boeing Commercial Airplanes, P.O. Box 3707, Seattle, Washington 98124-2207; telephone 206-544-9990; fax 206-766-5682; e-mail DDCS@boeing.com; Internet <https://www.myboeingfleet.com>.

Examining the AD Docket

You may examine the AD docket on the Internet at <http://www.regulations.gov>; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this AD, the regulatory evaluation, any comments received, and other information. The address for the Docket Office (telephone 800-647-5527) is the Document Management Facility, U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue, SE., Washington, DC 20590.

FOR FURTHER INFORMATION CONTACT: Ivan Li, Aerospace Engineer, Airframe Branch, ANM-120S, FAA, Seattle Aircraft Certification Office, 1601 Lind Avenue, SW., Renton, Washington 98057-3356; telephone (425) 917-6437; fax (425) 917-6590.

SUPPLEMENTARY INFORMATION:

Discussion

We issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 to include an airworthiness directive (AD) that would apply to certain Boeing model 747-100, 747-100B, 747-100B SUD, 747-200B, 747-200C, 747-200F, 747-300, 747-400, 747SR, and 747SP series airplanes. That NPRM was published in the **Federal Register** on October 8, 2008 (73 FR 58901). That NPRM proposed to require inspecting for skin cracks at the shear tie end fastener locations of the fuselage frames, and repairing cracks if necessary.

Comments

We gave the public the opportunity to participate in developing this AD. We considered the comment received. Boeing concurs with the contents of the NPRM.

Conclusion

We reviewed the relevant data, considered the comments received, and determined that air safety and the public interest require adopting the AD as proposed.

Costs of Compliance

We estimate that this AD affects 147 airplanes of U.S. registry. We also estimate that it takes 30 work-hours for the detailed inspection, or 49 work hours for the HFEC inspection, per product, to comply with this AD. The average labor rate is \$80 per work-hour. Based on these figures, we estimate the cost of this AD to the U.S. operators to be \$352,800 or \$576,240, or \$2,400 or \$3,920 per product.

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. "Subtitle VII: Aviation Programs," describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in "Subtitle VII, Part A, Subpart III, Section 44701: General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations