

**Material Incorporated by Reference**

(i) You must use the service information specified in Table 4 of this AD to do the actions required by this AD, unless the AD specifies otherwise.

(1) The Director of the Federal Register approved the incorporation by reference of this service information under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) For service information identified in this AD, contact Airbus SAS—Airworthiness Office—EAL, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France; fax +33 5 61 93 45 80, e-mail [airworthiness.A330-A340@airbus.com](mailto:airworthiness.A330-A340@airbus.com); Internet <http://www.airbus.com>.

(3) You may review copies of the service information that is incorporated by reference at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton,

Washington. For information on the availability of this material at the FAA, call 425-227-1221 or 425-227-1152.

(4) You may also review copies of the service information at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call (202) 741-6030, or go to: <http://www.archives.gov/federal-register/cfr/ibr-locations.html>.

TABLE 4—MATERIAL INCORPORATED BY REFERENCE

Airbus Mandatory Service Bulletin	Revision level	Date
A330-28-3082, including Appendix 1 .....	05	May 27, 2008.
A330-28-3092, excluding Appendix 01 .....	01	December 14, 2005.
A330-28-3101 .....	01	October 11, 2006.
A330-55-3016 .....	02	March 16, 2007.
A340-28-4073 .....	02	March 8, 2007.
A340-28-4078 .....	01	January 25, 2007.
A340-28-4097, including Appendix 01 .....	03	July 3, 2007.
A340-28-4107, excluding Appendix 01 .....	01	December 14, 2005.
A340-28-4118 .....	02	July 10, 2007.
A340-55-4017 .....	02	March 16, 2007.

Issued in Renton, Washington, on November 26, 2008.

**Ali Bahrami,**

Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. E8-29076 Filed 12-10-08; 8:45 am]

BILLING CODE 4910-13-P

**DEPARTMENT OF TRANSPORTATION**

**Federal Aviation Administration**

**14 CFR Part 39**

[Docket No. FAA-2008-1007; Directorate Identifier 2008-NM-135-AD; Amendment 39-15761; AD 2008-25-03]

RIN 2120-AA64

**Airworthiness Directives; Bombardier Model CL-600-2C10 (Regional Jet Series 700, 701 & 702) Airplanes and Model CL-600-2D24 (Regional Jet Series 900) Airplanes**

**AGENCY:** Federal Aviation Administration (FAA), Department of Transportation (DOT).

**ACTION:** Final rule.

**SUMMARY:** We are adopting a new airworthiness directive (AD) for the products listed above. This AD results from mandatory continuing airworthiness information (MCAI) originated by an aviation authority of another country to identify and correct an unsafe condition on an aviation product. The MCAI describes the unsafe condition as:

Bombardier Aerospace has completed a system safety review of the CL-600-2C10/

CL-600-2D24 aircraft fuel system against new fuel tank safety standards. \* \* \*

The assessment showed that due to the close proximity of intrinsically safe fuel system wiring with other wiring, a single failure from wire chafing at various locations of the fuselage could result in an ignition source inside the fuel tank. In addition, chafing of the temperature sensor wiring against the high power wiring in the avionics compartment could lead to overheating of the temperature sensor and hot surface ignition. The presence of an ignition source inside the fuel tank could result in a fuel tank explosion.

We are issuing this AD to require actions to correct the unsafe condition on these products.

**DATES:** This AD becomes effective January 15, 2009.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in this AD as of January 15, 2009.

**ADDRESSES:** You may examine the AD docket on the Internet at <http://www.regulations.gov> or in person at the U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue, SE., Washington, DC.

**FOR FURTHER INFORMATION CONTACT:** Rocco Viselli, Aerospace Engineer, Airframe and Propulsion Branch, ANE-171, FAA, New York Aircraft Certification Office, 1600 Stewart Avenue, Suite 410, Westbury, New York 11590; telephone (516) 228-7331; fax (516) 794-5531.

**SUPPLEMENTARY INFORMATION:**

**Discussion**

We issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 to include an AD that would apply to the specified products. That NPRM was published in the **Federal Register** on September 23, 2008 (73 FR 54749). That NPRM proposed to correct an unsafe condition for the specified products. The MCAI states:

Bombardier Aerospace has completed a system safety review of the CL-600-2C10/CL-600-2D24 aircraft fuel system against new fuel tank safety standards, introduced in Chapter 525 of the Airworthiness Manual through Notice of Proposed Amendment (NPA) 2002-043. The identified non-compliances were assessed using Transport Canada Policy Letter No. 525-001 to determine if mandatory corrective action was required.

The assessment showed that due to the close proximity of intrinsically safe fuel system wiring with other wiring, a single failure from wire chafing at various locations of the fuselage could result in an ignition source inside the fuel tank. In addition, chafing of the temperature sensor wiring against the high power wiring in the avionics compartment could lead to overheating of the temperature sensor and hot surface ignition. The presence of an ignition source inside the fuel tank could result in a fuel tank explosion.

To correct the unsafe condition, this directive mandates the installation of conduit and the addition of spacers to protect fuel tank wiring.

You may obtain further information by examining the MCAI in the AD docket.

**Comments**

We gave the public the opportunity to participate in developing this AD. We received no comments on the NPRM or

on the determination of the cost to the public.

### Conclusion

We reviewed the available data and determined that air safety and the public interest require adopting the AD as proposed.

### Differences Between This AD and the MCAI or Service Information

We have reviewed the MCAI and related service information and, in general, agree with their substance. But we might have found it necessary to use different words from those in the MCAI to ensure the AD is clear for U.S. operators and is enforceable. In making these changes, we do not intend to differ substantively from the information provided in the MCAI and related service information.

We might also have required different actions in this AD from those in the MCAI in order to follow our FAA policies. Any such differences are highlighted in a Note within the AD.

### Costs of Compliance

We estimate that this AD will affect about 159 products of U.S. registry. We also estimate that it will take about 38 work-hours per product to comply with the basic requirements of this AD. The average labor rate is \$80 per work-hour. Required parts will cost about \$2,914 per product. Where the service information lists required parts costs that are covered under warranty, we have assumed that there will be no charge for these parts. As we do not control warranty coverage for affected parties, some parties may incur costs higher than estimated here. Based on these figures, we estimate the cost of this AD to the U.S. operators to be \$946,686, or \$5,954 per product.

### Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. "Subtitle VII: Aviation Programs," describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in "Subtitle VII, Part A, Subpart III, Section 44701: General Requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition

that is likely to exist or develop on products identified in this rulemaking action.

### Regulatory Findings

We determined that this AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify this AD:

1. Is not a "significant regulatory action" under Executive Order 12866;
2. Is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and
3. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared a regulatory evaluation of the estimated costs to comply with this AD and placed it in the AD docket.

### Examining the AD Docket

You may examine the AD docket on the Internet at <http://www.regulations.gov>; or in person at the Docket Operations office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains the NPRM, the regulatory evaluation, any comments received, and other information. The street address for the Docket Operations office (telephone (800) 647-5527) is in the **ADDRESSES** section. Comments will be available in the AD docket shortly after receipt.

### List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

### Adoption of the Amendment

■ Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

### PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

#### § 39.13 [Amended]

■ 2. The FAA amends § 39.13 by adding the following new AD:

**2008-25-03 Bombardier, Inc. (Formerly Canadair):** Amendment 39-15761. Docket No. FAA-2008-1007; Directorate Identifier 2008-NM-135-AD.

### Effective Date

(a) This airworthiness directive (AD) becomes effective January 15, 2009.

### Affected ADs

(b) None.

### Applicability

(c) This AD applies to the airplanes identified in paragraphs (c)(1) and (c)(2) of this AD, certificated in any category.

(1) Bombardier Model CL-600-2C10 (Regional Jet Series 700, 701, & 702) airplanes, serial numbers 10003 through 10169 inclusive.

(2) Bombardier Model CL-600-2D24 (Regional Jet Series 900) airplanes, serial numbers 15001 through 15030 inclusive.

### Subject

(d) Air Transport Association (ATA) of America Code 24: Electrical Power.

### Reason

(e) The mandatory continuing airworthiness information (MCAI) states:

Bombardier Aerospace has completed a system safety review of the CL-600-2C10/CL-600-2D24 aircraft fuel system against new fuel tank safety standards, introduced in Chapter 525 of the Airworthiness Manual through Notice of Proposed Amendment (NPA) 2002-043. The identified non-compliances were assessed using Transport Canada Policy Letter No. 525-001 to determine if mandatory corrective action was required.

The assessment showed that due to the close proximity of intrinsically safe fuel system wiring with other wiring, a single failure from wire chafing at various locations of the fuselage could result in an ignition source inside the fuel tank. In addition, chafing of the temperature sensor wiring against the high power wiring in the avionics compartment could lead to overheating of the temperature sensor and hot surface ignition. The presence of an ignition source inside the fuel tank could result in a fuel tank explosion.

To correct the unsafe condition, this directive mandates the installation of conduit and the addition of spacers to protect fuel tank wiring.

### Actions and Compliance

(f) Unless already done, do the following actions.

(1) Within 4,500 flight hours after the effective date of this AD, modify the fuel system wiring along the fuselage and in the avionics compartment by installing protective conduit and spacers, in accordance with the Accomplishment Instructions of Bombardier Service Bulletin 670BA-24-011, Revision C, dated November 28, 2005.

(2) Actions done before the effective date of this AD in accordance with Bombardier Service Bulletin 670BA-24-011, dated September 7, 2004; Revision A, dated December 14, 2004; or Revision B, dated February 28, 2005; are acceptable for compliance with the corresponding requirements of this AD.

**FAA AD Differences**

**Note:** This AD differs from the MCAI and/or service information as follows: No differences.

**Other FAA AD Provisions**

(g) The following provisions also apply to this AD:

(1) Alternative Methods of Compliance (AMOCs): The Manager, New York Aircraft Certification Office (ACO), FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. Send information to ATTN: Rocco Viselli, Aerospace Engineer, Airframe and Propulsion Branch, ANE-171, FAA, New York ACO, 1600 Stewart Avenue, Suite 410, Westbury, New York 11590; telephone (516) 228-7331; fax (516) 794-5531. Before using any approved AMOC on any airplane to which the AMOC applies, notify your appropriate principal inspector (PI) in the FAA Flight Standards District Office (FSDO), or lacking a PI, your local FSDO.

(2) Airworthy Product: For any requirement in this AD to obtain corrective actions from a manufacturer or other source, use these actions if they are FAA-approved. Corrective actions are considered FAA-approved if they are approved by the State of Design Authority (or their delegated agent). You are required to assure the product is airworthy before it is returned to service.

(3) Reporting Requirements: For any reporting requirement in this AD, under the provisions of the Paperwork Reduction Act, the Office of Management and Budget (OMB) has approved the information collection requirements and has assigned OMB Control Number 2120-0056.

**Related Information**

(h) Refer to MCAI Canadian Airworthiness Directive CF-2008-25, dated July 3, 2008; and Bombardier Service Bulletin 670BA-24-011, Revision C, dated November 28, 2005; for related information.

**Material Incorporated by Reference**

(i) You must use Bombardier Service Bulletin 670BA-24-011, Revision C, dated November 28, 2005, to do the actions required by this AD, unless the AD specifies otherwise.

(1) The Director of the Federal Register approved the incorporation by reference of this service information under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) For service information identified in this AD, contact Bombardier, Inc., 400 Côte-Vertu Road West, Dorval, Québec H4S 1Y9, Canada; telephone 514-855-5000; fax 514-855-7401; e-mail [thd.crj@aero.bombardier.com](mailto:thd.crj@aero.bombardier.com); Internet <http://www.bombardier.com>.

(3) You may review copies of the service information that is incorporated by reference at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington. For information on the availability of this material at the FAA, call 425-227-1221 or 425-227-1152.

(4) You may also review copies of the service information at the National Archives and Records Administration (NARA). For information on the availability of this

material at NARA, call 202-741-6030, or go to: [http://www.archives.gov/federal\\_register/code\\_of\\_federal\\_regulations/ibr\\_locations.html](http://www.archives.gov/federal_register/code_of_federal_regulations/ibr_locations.html).

Issued in Renton, Washington, on November 26, 2008.

**Ali Bahrami,**

*Manager, Transport Airplane Directorate, Aircraft Certification Service.*

[FR Doc. E8-29077 Filed 12-10-08; 8:45 am]

**BILLING CODE 4910-13-P**

**DEPARTMENT OF TRANSPORTATION****Federal Aviation Administration****14 CFR Part 39**

**[Docket No. FAA-2007-0175; Directorate Identifier 2007-NM-184-AD; Amendment 39-15766; AD 2008-25-08]**

**RIN 2120-AA64**

**Airworthiness Directives; Boeing Model 757 Airplanes**

**AGENCY:** Federal Aviation Administration (FAA), DOT.

**ACTION:** Final rule.

**SUMMARY:** We are adopting a new airworthiness directive (AD) for certain Boeing Model 757 airplanes. This AD requires changing the wiring of the fuel boost pump and doing other specified actions. This AD results from reports of short circuits in an electrical connector at the wing-to-body electrical disconnect panel. We are issuing this AD to prevent a short circuit of the electrical connector for the fuel boost pump, which could cause the instruments for fuel, flap, slat, and aileron systems to malfunction and create a potential ignition source inside the fuel tanks. A potential ignition source inside the fuel tank in combination with flammable fuel vapors could result in a fuel tank explosion and consequent loss of the airplane.

**DATES:** This AD is effective January 15, 2009.

The Director of the Federal Register approved the incorporation by reference of certain publications listed in this AD as of January 15, 2009.

**ADDRESSES:** For service information identified in this AD, contact Boeing Commercial Airplanes, Attention: Data & Services Management, P.O. Box 3707, MC 2H-65, Seattle, Washington 98124-2207; telephone 206-544-5000, extension 1, fax 206-766-5680; e-mail [me.boecom@boeing.com](mailto:me.boecom@boeing.com); Internet <https://www.myboeingfleet.com>.

**Examining the AD Docket**

You may examine the AD docket on the Internet at <http://www.regulations.gov>; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this AD, the regulatory evaluation, any comments received, and other information. The address for the Docket Office (telephone 800-647-5527) is the Document Management Facility, U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue, SE., Washington, DC 20590.

**FOR FURTHER INFORMATION CONTACT:** Philip Sheridan, Aerospace Engineer, Systems and Equipment Branch, ANM-130S, FAA, Seattle Aircraft Certification Office, 1601 Lind Avenue, SW., Renton, Washington 98057-3356; telephone (425) 917-6441; fax (425) 917-6590.

**FOR FURTHER INFORMATION CONTACT:**

Philip Sheridan, Aerospace Engineer, Systems and Equipment Branch, ANM-130S, FAA, Seattle Aircraft Certification Office, 1601 Lind Avenue, SW., Renton, Washington 98057-3356; telephone (425) 917-6441; fax (425) 917-6590.

**SUPPLEMENTARY INFORMATION:****Discussion**

We issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 to include an airworthiness directive (AD) that would apply to certain Boeing Model 757 airplanes. That NPRM was published in the **Federal Register** on November 9, 2007 (72 FR 63512). That NPRM proposed to require changing the wiring of the fuel boost pump and doing other specified actions.

**Comments**

We gave the public the opportunity to participate in developing this AD. We considered the comments received.

**Request To Incorporate Revised Service Information**

Boeing asks that we refer to Boeing Special Attention Service Bulletins 757-28-0095 and 757-28-0096, both Revision 1, both dated June 4, 2008, in the final rule. Boeing Special Attention Service Bulletins 757-28-0095 and 757-28-0096, both dated June 18, 2007, were referred to in the NPRM as the appropriate sources of service information for accomplishing the actions specified.

We have reviewed Revision 1 of these referenced service bulletins and we agree with the commenter since no additional work is necessary on airplanes changed in accordance with the original issue of the referenced service information. Revision 1 of these service bulletins clarifies certain procedures and certain routing and splice locations. We have added Revision 1 of these service bulletins to the applicability specified in paragraphs (c)(1) and (c)(2) of this AD, and to paragraph (f) of this AD, as the appropriate sources of service