

2nd stage HPT vane assembly. The proposed AD would require you to use the service information described previously to perform these actions.

#### Interim Action

These actions are interim actions and we may take further rulemaking actions in the future.

#### Costs of Compliance

We estimate that this proposed AD would affect 240 engines installed on airplanes of U.S. registry. We also estimate that it would take about 5 work-hours per engine to perform the proposed actions, that each engine might require two inspections, and that the average labor rate is \$80 per work-hour. Based on these figures, we estimate the total cost of the proposed AD to U.S. operators to be \$192,000.

#### Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, Section 106, describes the authority of the FAA Administrator. Subtitle VII, Aviation Programs, describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section 44701, "General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

#### Regulatory Findings

We have determined that this proposed AD would not have federalism implications under Executive Order 13132. This proposed AD would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that the proposed AD:

1. Is not a "significant regulatory action" under Executive Order 12866;
2. Is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and
3. Would not have a significant economic impact, positive or negative, on a substantial number of small entities

under the criteria of the Regulatory Flexibility Act.

We prepared a regulatory evaluation of the estimated costs to comply with this proposed AD. You may get a copy of this summary at the address listed under **ADDRESSES**.

#### List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

#### The Proposed Amendment

Under the authority delegated to me by the Administrator, the Federal Aviation Administration proposes to amend 14 CFR part 39 as follows:

#### PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

#### § 39.13 [Amended]

2. The FAA amends § 39.13 by adding the following new airworthiness directive:

**Pratt & Whitney:** Docket No. FAA-2008-0759; Directorate Identifier 2008-NE-02-AD.

#### Comments Due Date

- (a) The Federal Aviation Administration (FAA) must receive comments on this airworthiness directive (AD) action by September 8, 2008.

#### Affected ADs

- (b) None.

#### Applicability

- (c) This AD applies to Pratt & Whitney (PW) JT9D-7, -7A, -7AH, -7H, -7F, and -7J turbofan engines. These engines are installed on, but not limited to, Boeing 747 series airplanes.

#### Unsafe Condition

- (d) This AD results from an uncontained failure of a 2nd stage high-pressure turbine (HPT) rotor disk that caused the engine to separate from the airplane. We are issuing this AD to prevent failure of the 2nd stage HPT rotor disk, which could result in uncontained engine failure, damage to the airplane, and the engine separating from the airplane.

#### Compliance

- (e) You are responsible for having the actions required by this AD performed within the compliance times specified unless the actions have already been done.

#### Initial Borescope Inspection

- (f) Within 100 cycles-in-service (CIS) after the effective date of this AD, or within 1,000 CIS after the last HPT module overhaul, whichever occurs later, do the following:

- (1) Use the Accomplishment Instructions of PW Alert Service Bulletin (ASB) JT9D A6488, Revision 1, dated April 18, 2008, to

borescope-inspect the 2nd stage HPT rotor and stator assembly either on-wing or in the shop.

- (2) If you see any damage or contact between the 2nd stage HPT vanes and the 2nd stage HPT rotor, remove the engine from service.

#### Repetitive Borescope Inspection

- (g) Thereafter, within 1,000 cycles-since-last inspection, do the following:

- (1) Use the Accomplishment Instructions of PW ASB JT9D A6488 Revision 1, dated April 18, 2008, to borescope-inspect the 2nd stage HPT rotor and stator assembly either on-wing or in the shop.

- (2) If you see any damage or contact between the 2nd stage HPT vanes and the 2nd stage HPT rotor, remove the engine from service.

#### Optional Terminating Action

- (h) Installing the 2nd stage HPT vanes as specified in the JT9D-7 Engine Manual Revision 122, dated February 15, 2008, terminates the repetitive inspection requirement specified in paragraph (g) of this AD.

#### Alternative Methods of Compliance

- (i) The Manager, Engine Certification Office, has the authority to approve alternative methods of compliance for this AD if requested using the procedures found in 14 CFR 39.19.

#### Related Information

- (j) PW ASB JT9D A6488, Revision 1, dated April 18, 2008, pertains to the subject of this AD.

- (k) Contact Mark Riley, Aerospace Engineer, Engine Certification Office, FAA, Engine and Propeller Directorate, 12 New England Executive Park, Burlington, MA 01803; e-mail: mark.riley@faa.gov; telephone (781) 238-7758; fax (781) 238-7199, for more information about this AD.

Issued in Burlington, Massachusetts, on July 3, 2008.

**Diane Cook,**

*Acting Manager, Engine and Propeller Directorate, Aircraft Certification Service.*

[FR Doc. E8-15682 Filed 7-9-08; 8:45 am]

**BILLING CODE 4910-13-P**

## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 39

[Docket No. FAA-2008-0732; Directorate Identifier 2008-NM-053-AD]

RIN 2120-AA64

#### Airworthiness Directives; Dassault Model Mystere-Falcon 50 Airplanes

**AGENCY:** Federal Aviation Administration (FAA), DOT.

**ACTION:** Notice of proposed rulemaking (NPRM).

**SUMMARY:** We propose to adopt a new airworthiness directive (AD) for the products listed above. This proposed AD results from mandatory continuing airworthiness information (MCAI) originated by an aviation authority of another country to identify and correct an unsafe condition on an aviation product. The MCAI describes the unsafe condition as:

\* \* \* [S]ome aircraft could have experienced wing overpressure consecutive to the latent failure of both [pressure relief] valve units. Overpressure although not sufficient to cause static damages could have impaired the fatigue damage tolerance of the wing structure. \* \* \*

The proposed AD would require actions that are intended to address the unsafe condition described in the MCAI.

**DATES:** We must receive comments on this proposed AD by August 11, 2008.

**ADDRESSES:** You may send comments by any of the following methods:

- *Federal eRulemaking Portal:* Go to <http://www.regulations.gov>. Follow the instructions for submitting comments.
- *Fax:* (202) 493-2251.
- *Mail:* U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue, SE., Washington, DC 20590.
- *Hand Delivery:* U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-40, 1200 New Jersey Avenue, SE., Washington, DC, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

#### Examining the AD Docket

You may examine the AD docket on the Internet at <http://www.regulations.gov>; or in person at the Docket Operations office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this proposed AD, the regulatory evaluation, any comments received, and other information. The street address for the Docket Operations office (telephone (800) 647-5527) is in the **ADDRESSES** section. Comments will be available in the AD docket shortly after receipt.

**FOR FURTHER INFORMATION CONTACT:** Tom Rodriguez, Aerospace Engineer, International Branch, ANM-116, Transport Airplane Directorate, FAA, 1601 Lind Avenue, SW., Renton, Washington 98057-3356; telephone (425) 227-1137; fax (425) 227-1149.

#### SUPPLEMENTARY INFORMATION:

##### Comments Invited

We invite you to send any written relevant data, views, or arguments about

this proposed AD. Send your comments to an address listed under the

**ADDRESSES** section. Include “Docket No. FAA-2008-0732; Directorate Identifier 2008-NM-053-AD” at the beginning of your comments. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of this proposed AD. We will consider all comments received by the closing date and may amend this proposed AD based on those comments.

We will post all comments we receive, without change, to <http://www.regulations.gov>, including any personal information you provide. We will also post a report summarizing each substantive verbal contact we receive about this proposed AD.

#### Discussion

The European Aviation Safety Agency (EASA), which is the Technical Agent for the Member States of the European Community, has issued EASA Airworthiness Directive 2008-0021, dated January 31, 2008 (referred to after this as “the MCAI”), to correct an unsafe condition for the specified products. The MCAI states:

Analysed in-service events revealed that corrosion of pressure relief valves in wing fuel tanks was likely to occur well before reaching their Time Between Overhaul (TBO) and could make the valves stick in the closed position.

Therefore some aircraft could have experienced wing overpressure consecutive to the latent failure of both valve units. Overpressure although not sufficient to cause static damages could have impaired the fatigue damage tolerance of the wing structure. Consequently this Airworthiness Directive (AD) mandates introduction of a new repetitive inspection of the wing structure.

The repetitive ultrasonic inspection is intended to detect incipient cracking on the stiffeners of the right-hand and left-hand wing lower panels between ribs 13 and 17 (the inspection area extends to just beyond rib 16). The corrective actions if any cracking is found include contacting Dassault for repair instructions, and doing the repair. You may obtain further information by examining the MCAI in the AD docket.

#### Relevant Service Information

Dassault has issued Temporary Revision 74, dated November 2007, to the Dassault Falcon 50 Maintenance Manual, Maintenance Procedure 57-401, “Non-Destructive Check of the Wing Lower Panels Stiffeners between Ribs 13 and 16 (ATA 57-00-21).” The actions described in this service information are intended to correct the unsafe condition identified in the MCAI.

#### FAA’s Determination and Requirements of This Proposed AD

This product has been approved by the aviation authority of another country, and is approved for operation in the United States. Pursuant to our bilateral agreement with the State of Design Authority, we have been notified of the unsafe condition described in the MCAI and service information referenced above. We are proposing this AD because we evaluated all pertinent information and determined an unsafe condition exists and is likely to exist or develop on other products of the same type design.

#### Differences Between This AD and the MCAI or Service Information

We have reviewed the MCAI and related service information and, in general, agree with their substance. But we might have found it necessary to use different words from those in the MCAI to ensure the AD is clear for U.S. operators and is enforceable. In making these changes, we do not intend to differ substantively from the information provided in the MCAI and related service information.

We might also have proposed different actions in this AD from those in the MCAI in order to follow FAA policies. Any such differences are highlighted in a NOTE within the proposed AD.

#### Costs of Compliance

Based on the service information, we estimate that this proposed AD would affect about 247 products of U.S. registry. We also estimate that it would take about 6 work-hours per product to comply with the basic requirements of this proposed AD. The average labor rate is \$80 per work-hour. Based on these figures, we estimate the cost of the proposed AD on U.S. operators to be \$118,560, or \$480 per product.

#### Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA’s authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. “Subtitle VII: Aviation Programs,” describes in more detail the scope of the Agency’s authority.

We are issuing this rulemaking under the authority described in “Subtitle VII, Part A, Subpart III, Section 44701: General requirements.” Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation

is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

### Regulatory Findings

We determined that this proposed AD would not have federalism implications under Executive Order 13132. This proposed AD would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify this proposed regulation:

1. Is not a "significant regulatory action" under Executive Order 12866;
2. Is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and
3. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared a regulatory evaluation of the estimated costs to comply with this proposed AD and placed it in the AD docket.

### List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

### The Proposed Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA proposes to amend 14 CFR part 39 as follows:

### PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

#### § 39.13 [Amended]

2. The FAA amends § 39.13 by adding the following new AD:

**Dassault Aviation:** Docket No. FAA-2008-0732; Directorate Identifier 2008-NM-053-AD.

#### Comments Due Date

(a) We must receive comments by August 11, 2008.

#### Affected ADs

(b) None.

#### Applicability

(c) This AD applies to all Dassault Model Mystere-Falcon 50 airplanes, certificated in any category.

### Subject

(d) Air Transport Association (ATA) of America Code 57: Wings.

### Reason

(e) The mandatory continuing airworthiness information (MCAI) states:

Analysed in-service events revealed that corrosion of pressure relief valves in wing fuel tanks was likely to occur well before reaching their Time Between Overhaul (TBO) and could make the valves stick in the closed position.

Therefore some aircraft could have experienced wing overpressure consecutive to the latent failure of both valve units. Overpressure although not sufficient to cause static damages could have impaired the fatigue damage tolerance of the wing structure. Consequently this Airworthiness Directive (AD) mandates introduction of a new repetitive inspection of the wing structure.

The repetitive ultrasonic inspection is intended to detect incipient cracking on the stiffeners of the right-hand and left-hand wing lower panels between ribs 13 and 17 (the inspection area extends to just beyond rib 16). The corrective actions if any cracking is found include contacting Dassault for repair instructions, and doing the repair.

### Actions and Compliance

(f) Unless already accomplished, do the following actions: Prior to the accumulation of 14,200 total flight cycles, or within 160 flight cycles after the effective date of this AD, whichever occurs later, do the ultrasonic inspection described in Temporary Revision 74, dated November 2007, to the Dassault Falcon 50 Maintenance Manual, Maintenance Procedure 57-401, "Non-Destructive Check of the Wing Lower Panels Stiffeners between Ribs 13 and 16 (ATA 57-00-21)." Do all applicable corrective actions before further flight. Repeat the inspection thereafter at intervals not to exceed 5,350 flight cycles.

### FAA AD Differences

**Note:** This AD differs from the MCAI and/or service information as follows: No differences.

### Other FAA AD Provisions

(g) The following provisions also apply to this AD:

(1) *Alternative Methods of Compliance (AMOCs):* The Manager, ANM-116, International Branch, Transport Airplane Directorate, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. Send information to ATTN: Tom Rodriguez, Aerospace Engineer, 1601 Lind Avenue, SW., Renton, Washington 98057-3356; telephone (425) 227-1137; fax (425) 227-1149. Before using any approved AMOC on any airplane to which the AMOC applies, notify your appropriate principal inspector (PI) in the FAA Flight Standards District Office (FSDO), or lacking a PI, your local FSDO.

(2) *Airworthy Product:* For any requirement in this AD to obtain corrective actions from a manufacturer or other source, use these actions if they are FAA-approved. Corrective actions are considered FAA-approved if they

are approved by the State of Design Authority (or their delegated agent). You are required to assure the product is airworthy before it is returned to service.

(3) *Reporting Requirements:* For any reporting requirement in this AD, under the provisions of the Paperwork Reduction Act, the Office of Management and Budget (OMB) has approved the information collection requirements and has assigned OMB Control Number 2120-0056.

### Related Information

(h) Refer to MCAI European Aviation Safety Agency (EASA) Airworthiness Directive 2008-0021, dated January 31, 2008, and Temporary Revision 74, dated November 2007, to the Dassault Falcon 50 Maintenance Manual, Maintenance Procedure 57-401, "Non-Destructive Check of the Wing Lower Panels Stiffeners between Ribs 13 and 16 (ATA 57-00-21)," for related information.

Issued in Renton, Washington, on June 27, 2008.

### Ali Bahrami,

Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. E8-15714 Filed 7-9-08; 8:45 am]

BILLING CODE 4910-13-P

## DEPARTMENT OF THE TREASURY

### Internal Revenue Service

#### 26 CFR Part 1

[REG-142040-07]

RIN 1545-BH53

### Reasonable Good Faith Interpretation of Required Minimum Distribution Rules by Governmental Plans

**AGENCY:** Internal Revenue Service (IRS), Treasury.

**ACTION:** Notice of proposed rulemaking.

**SUMMARY:** This document contains proposed regulations under sections 401(a)(9) and 403(b) of the Internal Revenue Code (Code) to permit a governmental plan to comply with the required minimum distribution rules by using a reasonable and good faith interpretation of the statute. These proposed regulations will affect administrators of, employers maintaining, participants in, and beneficiaries of governmental plans.

**DATES:** Written or electronic comments and requests for a public hearing must be received by October 8, 2008.

**ADDRESSES:** Send submissions to: CC:PA:LPD:PR (REG-142040-07), room 5203, Internal Revenue Service, PO Box 7604, Ben Franklin Station, Washington DC 20044. Submissions may be hand-delivered Monday through Friday between the hours of 8 a.m. and 4 p.m. to: CC:PA:LPD:PR (REG-142040-07),