

flight testing of these new ice shapes indicated that they do not adversely affect the handling characteristics of the Brasilia, the testing did indicate that the stick-shaker-to-stick-pusher speed margins for the intercycle ice shapes may be reduced below the minimum standard values set forth in the applicable CTA and FAA Regulations. In order to preserve the original certification stick-shaker-to-stick-pusher margins when operating under the newly defined intercycle icing conditions, an upgraded Stall Warning Computer with new settings for shaker firing angle-of-attack (AOA) is required to be installed. The unsafe condition is reduced ability of the flightcrew to maintain the safe flight and landing of the airplane. The corrective action includes modification of certain electrical wiring and installation of a new Stall Warning Computer.

**Actions and Compliance**

(f) Within 36 months after the effective date of this AD, unless already done, do the following actions.

(1) Replace the current Stall Warning Computers with new improved ones in accordance with detailed instructions and procedures described in the EMBRAER Service Bulletin 120-27-0092, Revision 01, dated December 29, 2006.

(2) Before installing the improved Stall Warning Computers, accomplish the detailed instructions and procedures described in the

EMBRAER Service Bulletin 120-27-0091, Change 02, dated September 29, 2003.

(3) As of 36 months after the effective date of this AD, no person may install a Stall Warning Computer; part number C-81806-1 or -2, Mod. A, or C-81806-3, on any airplane.

**FAA AD Differences**

**Note:** This AD differs from the MCAI and/or service information as follows: No differences.

**Other FAA AD Provisions**

(g) The following provisions also apply to this AD:

(1) *Alternative Methods of Compliance (AMOCs):* The Manager, International Branch, ANM-116, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. Send information to ATTN: Dan Rodina, Aerospace Engineer, International Branch, ANM-116, FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, Washington 98057-3356; telephone (425) 227-2125; fax (425) 227-1149. Before using any approved AMOC on any airplane to which the AMOC applies, notify your appropriate principal inspector (PI) in the FAA Flight Standards District Office (FSDO), or lacking a PI, your local FSDO.

(2) *Airworthy Product:* For any requirement in this AD to obtain corrective actions from

a manufacturer or other source, use these actions if they are FAA-approved. Corrective actions are considered FAA-approved if they are approved by the State of Design Authority (or their delegated agent). You are required to assure the product is airworthy before it is returned to service.

(3) *Reporting Requirements:* For any reporting requirement in this AD, under the provisions of the Paperwork Reduction Act, the Office of Management and Budget (OMB) has approved the information collection requirements and has assigned OMB Control Number 2120-0056.

**Related Information**

(h) Refer to MCAI Brazilian Airworthiness Directive 2007-03-03, effective April 10, 2007; and EMBRAER Service Bulletins 120-27-0091, Change 02, dated September 29, 2003; and 120-27-0092, Revision 01, dated December 29, 2006; for related information.

**Material Incorporated by Reference**

(i) You must use EMBRAER Service Bulletin 120-27-0091, Change 02, dated September 29, 2003; or EMBRAER Service Bulletin 120-27-0092, Revision 01, dated December 29, 2006; as applicable; to do the actions required by this AD, unless the AD specifies otherwise. EMBRAER Service Bulletin 120-27-0091, Change 02, contains the following list of effective pages:

Page Nos.	Change level shown on page	Date shown on page
1, 2, 51, 58 .....	02	September 29, 2003.
3-50, 52-57, 59-87 .....	01	October 15, 2002.

(1) The Director of the Federal Register approved the incorporation by reference of this service information under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) For service information identified in this AD, contact Empresa Brasileira de Aeronautica S.A. (EMBRAER), P.O. Box 343—CEP 12.225, Sao Jose dos Campos—SP, Brazil.

(3) You may review copies at the FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, Washington; or at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call (202) 741-6030, or go to: <http://www.archives.gov/federal-register/cfr/ibr-locations.html>.

Issued in Renton, Washington, on December 21, 2007.

**Ali Bahrami,**

Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. E8-170 Filed 1-15-08; 8:45 am]

**BILLING CODE 4910-13-P**

**DEPARTMENT OF TRANSPORTATION**

**Federal Aviation Administration**

**14 CFR Part 39**

[Docket No. FAA-2007-0171; Directorate Identifier 2007-NM-220-AD; Amendment 39-15330; AD 2008-01-05]

**RIN 2120-AA64**

**Airworthiness Directives; Airbus Model A310 Series Airplanes**

**AGENCY:** Federal Aviation Administration (FAA), Department of Transportation (DOT).

**ACTION:** Final rule.

**SUMMARY:** The FAA is superseding an existing airworthiness directive (AD), that applies to certain Airbus Model A310 series airplanes. That AD currently requires modification of certain wires in the right-hand (RH) wing. This new AD requires further modification by installing an additional protection sleeve and segregating route 2S in the RH pylon area. This AD results from analysis of wire routing that revealed that route 2S of the fuel

electrical circuit, located in the RH wing, does not provide adequate separation of fuel quantity indication wires from wires carrying 115-volt alternating current (AC). We are issuing this AD to ensure that fuel quantity indication wires are properly separated from wires carrying 115-volt AC. Improper separation of such wires, in the event of wire damage, could lead to a short circuit and a possible ignition source, which could result in a fire in the airplane.

**DATES:** This AD becomes effective February 20, 2008.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in the AD as of February 20, 2008.

On September 3, 2004 (69 FR 45578, July 30, 2004), the Director of the Federal Register approved the incorporation by reference of Airbus Service Bulletin A310-28-2148, Revision 01, dated October 29, 2002.

**ADDRESSES:** For service information identified in this AD, contact Airbus, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France.

**Examining the AD Docket**

You may examine the AD docket on the Internet at <http://www.regulations.gov>; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this AD, the regulatory evaluation, any comments received, and other information. The address for the Docket Office (telephone 800-647-5527) is the Document Management Facility, U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue, SE., Washington, DC 20590.

**FOR FURTHER INFORMATION CONTACT:** Tom Stafford, Aerospace Engineer, International Branch, ANM-116, Transport Airplane Directorate, FAA,

1601 Lind Avenue, SW., Renton, Washington 98057-3356; telephone (425) 227-1622; fax (425) 227-1149.

**SUPPLEMENTARY INFORMATION:**

**Discussion**

The FAA issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 to include an AD that supersedes AD 2004-15-16, amendment 39-13750 (69 FR 45578, July 30, 2004). The existing AD applies to certain Airbus Model A310 series airplanes. That NPRM was published in the **Federal Register** on November 9, 2007 (72 FR 63506). That NPRM proposed to continue to require modification of certain wires in the right-hand (RH) wing. That NPRM also proposed to require further modification by installing an additional protection

sleeve and segregating route 2S in the RH pylon area.

**Comments**

We provided the public the opportunity to participate in the development of this AD. No comments have been received on the NPRM or on the determination of the cost to the public.

**Conclusion**

We have carefully reviewed the available data and determined that air safety and the public interest require adopting the AD as proposed.

**Costs of Compliance**

The following table provides the estimated costs for U.S. operators to comply with this AD.

ESTIMATED COSTS

Action	Work hours	Average labor rate per hour	Parts	Cost per airplane	Number of U.S.-registered airplanes	Fleet cost
Modification (required by AD 2004-15-16) .....	35	\$80	\$4,459	\$7,259	68	\$493,612
Further Modification (new proposed action) .....	22	80	1,870	3,630	68	246,840

**Authority for This Rulemaking**

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, Section 106, describes the authority of the FAA Administrator. Subtitle VII, Aviation Programs, describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section 44701, "General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

**Regulatory Findings**

We have determined that this AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that this AD:

- (1) Is not a "significant regulatory action" under Executive Order 12866;
- (2) Is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and
- (3) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared a regulatory evaluation of the estimated costs to comply with this AD and placed it in the AD docket. See the **ADDRESSES** section for a location to examine the regulatory evaluation.

**List of Subjects in 14 CFR Part 39**

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

**Adoption of the Amendment**

Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

**PART 39—AIRWORTHINESS DIRECTIVES**

■ 1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

**§ 39.13 [Amended]**

■ 2. The Federal Aviation Administration (FAA) amends § 39.13 by removing amendment 39-13750 (69 FR 45578, July 30, 2004) and by adding the following new airworthiness directive (AD):

**2008-01-05 Airbus:** Amendment 39-15330. Docket No. FAA-2007-0171; Directorate Identifier 2007-NM-220-AD.

**Effective Date**

(a) This AD becomes effective February 20, 2008.

**Affected ADs**

(b) This AD supersedes AD 2004-15-16.

**Applicability**

(c) This AD applies to Model A310 series airplanes, certificated in any category, all certified models, all serial numbers, except airplanes on which Airbus Service Bulletin A310-28-2148, Revision 02, dated March 9, 2007, has been done (Airbus Modifications 12427 and 12435).

**Unsafe Condition**

(d) This AD results from analysis of wire routing that revealed that route 2S of the fuel electrical circuit, located in the right-hand (RH) wing, does not provide adequate separation of fuel quantity indication wires from wires carrying 115-volt alternating current (AC). We are issuing this AD to ensure that fuel quantity indication wires are properly separated from wires carrying 115-volt AC. Improper separation of such wires,

in the event of wire damage, could lead to a short circuit and a possible ignition source, which could result in a fire in the airplane.

#### Compliance

(e) You are responsible for having the actions required by this AD performed within the compliance times specified, unless the actions have already been done.

#### Restatement of Requirements of AD 2004–15–16

#### Modification

(f) Within 4,000 flight hours after September 3, 2004 (the effective date of AD 2004–15–16): Modify the routing of wires in the RH wing by installing cable sleeves, per the Accomplishment Instructions of Airbus Service Bulletin A310–28–2148, Revision 01, dated October 29, 2002; or Revision 02, dated March 9, 2007. As of the effective date of this AD, Revision 02 must be used.

#### Actions Accomplished Previously

(g) Modification of the routing of wires accomplished before September 3, 2004, per Airbus Service Bulletin A310–28–2148, dated January 23, 2002, is acceptable for compliance with the corresponding requirements of paragraph (f) of this AD.

#### New Requirements of This AD

#### Modification (Additional Work)

(h) For airplanes on which the actions specified in Airbus Service Bulletin A310–28–2148, dated January 23, 2002; or Airbus Service Bulletin A310–28–2148, Revision 01, dated October 29, 2002; have been done before the effective date of this AD: Within 6,000 flight hours or 30 months after the effective date of this AD, whichever occurs first, perform further modification by installing additional protection sleeves in the outer wing area near the cadensicon sensor and segregating wire route 2S in the RH pylon area, in accordance with the Accomplishment Instructions of Airbus Service Bulletin A310–28–2148, Revision 02, dated March 9, 2007.

#### Alternative Methods of Compliance (AMOCs)

(i)(1) The Manager, International Branch, ANM–116, Transport Airplane Directorate, FAA, has the authority to approve AMOCs for this AD, if requested in accordance with the procedures found in 14 CFR 39.19.

(2) To request a different method of compliance or a different compliance time for this AD, follow the procedures in 14 CFR 39.19. Before using any approved AMOC on any airplane to which the AMOC applies, notify your appropriate principal inspector (PI) in the FAA Flight Standards District Office (FSDO), or lacking a PI, your local FSDO.

#### Related Information

(j) European Aviation Safety Agency airworthiness directive 2007–0230, dated August 15, 2007, also addresses the subject of this AD.

#### Material Incorporated by Reference

(k) You must use Airbus Service Bulletin A310–28–2148, Revision 01, dated October

29, 2002; or Airbus Service Bulletin A310–28–2148, Revision 02, dated March 9, 2007; as applicable, to perform the actions that are required by this AD, unless the AD specifies otherwise.

(1) The Director of the Federal Register approved the incorporation by reference of Airbus Service Bulletin A310–28–2148, Revision 02, dated March 9, 2007, in accordance with 5 U.S.C. 552(a) and 1 CFR part 51.

(2) On September 3, 2004 (69 FR 45578, July 30, 2004), the Director of the Federal Register approved the incorporation by reference of Airbus Service Bulletin A310–28–2148, Revision 01, dated October 29, 2002.

(3) Contact Airbus, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France, for a copy of this service information. You may review copies at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call 202–741–6030, or go to: <http://www.archives.gov/federal-register/cfr/ibr-locations.html>.

Issued in Renton, Washington, on January 3, 2008.

Ali Bahrami,

Manager, Transport Airplane Directorate,  
Aircraft Certification Service.

[FR Doc. E8–370 Filed 1–15–08; 8:45 am]

BILLING CODE 4910–13–P

## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 39

[Docket No. FAA–2007–0129; Directorate Identifier 2007–NM–099–AD; Amendment 39–15331; AD 2008–02–01]

RIN 2120–AA64

#### Airworthiness Directives; EMBRAER Model EMB–135BJ Airplanes

**AGENCY:** Federal Aviation Administration (FAA), Department of Transportation (DOT).

**ACTION:** Final rule.

**SUMMARY:** We are adopting a new airworthiness directive (AD) for the products listed above. This AD results from mandatory continuing airworthiness information (MCAI) originated by an aviation authority of another country to identify and correct an unsafe condition on an aviation product. The MCAI describes the unsafe condition as:

It has been found that some adhesive tapes used in the interior furnishings do not comply with the applicable flammability requirements. In case of some nearby ignition source, fire may propagate to the aircraft.

We are issuing this AD to require actions to correct the unsafe condition on these products.

**DATES:** This AD becomes effective February 20, 2008.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in this AD as of February 20, 2008.

**ADDRESSES:** You may examine the AD docket on the Internet at <http://www.regulations.gov> or in person at the U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue, SE., Washington, DC.

**FOR FURTHER INFORMATION CONTACT:** Todd Thompson, Aerospace Engineer, International Branch, ANM–116, FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98057–3356; telephone (425) 227–1175; fax (425) 227–1149.

#### SUPPLEMENTARY INFORMATION:

#### Discussion

We issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 to include an AD that would apply to the specified products. That NPRM was published in the **Federal Register** on November 1, 2007 (72 FR 61822). That NPRM proposed to correct an unsafe condition for the specified products. The MCAI states:

It has been found that some adhesive tapes used in the interior furnishings do not comply with the applicable flammability requirements. In case of some nearby ignition source, fire may propagate to the aircraft.

The corrective actions include an inspection to determine the presence of cotton adhesive tape, and replacement of the tape with new tape if necessary. You may obtain further information by examining the MCAI in the AD docket.

#### Comments

We gave the public the opportunity to participate in developing this AD. We received no comments on the NPRM or on the determination of the cost to the public.

#### Conclusion

We reviewed the available data and determined that air safety and the public interest require adopting the AD as proposed.

#### Differences Between This AD and the MCAI or Service Information

We have reviewed the MCAI and related service information and, in general, agree with their substance. But we might have found it necessary to use different words from those in the MCAI