

TABLE 1.—FIELD STRENGTH VOLTS/METER—Continued

Frequency	Peak	Average
1–2 GHz	5000	250
2–4 GHz	6000	490
4–6 GHz	7200	400
6–8 GHz	1100	170
8–12 GHz	5000	330
12–18 GHz	2000	330
18–40 GHz	1000	420

Applicability

As discussed previously, this special condition is applicable to Supplemental Type Certificate (STC) Project Number ST2902RC–R, for the installation of a Pratt & Whitney PT6–67D turbine engine in GHTI UH–1H military surplus helicopters type certificated under TC R00002RC. Should DynCorp International apply at a later date for a change to the STC to include another model incorporating the same novel or unusual design feature, the special condition would apply to that STC modification as well under the provisions of § 21.101.

Conclusion

This action affects only certain novel or unusual design features associated with this STC project. It is not a rule of general applicability and affects only the applicant who applied to the FAA for approval of these features on the helicopter.

The substance of this special condition has been subjected to a notice and comment period in several prior instances and has been derived without substantive change from those previously issued. It is unlikely that prior public comment would result in a significant change from the substance contained herein. For this reason, the FAA has determined that prior public notice and comment are unnecessary, and good cause exists for adopting this special condition upon issuance. The FAA is requesting comments to allow interested persons to submit views that may not have been submitted in response to the prior opportunities for comment.

List of Subjects in 14 CFR Parts 21 and 29

Aircraft, Air transportation, Aviation safety, Rotorcraft, Safety.

The authority citation for this special condition is as follows:

Authority: 42 U.S.C. 7572; 49 U.S.C. 106(g), 40105, 40113, 44701–44702, 44704, 44709, 44711, 44713, 44715, 45303.

The Special Condition

Accordingly, pursuant to the authority delegated to me by the Administrator, the following special condition is issued as part of the supplemental type certification basis for STC Project ST2902RC–R, installation of PT6–67D on Global Helicopter Technology, Inc. (GHTI), Model UH–1H, Restricted Category Helicopters, type certificated under TC R00002RC.

Protection for Electrical and Electronic Systems From High Intensity Radiated Fields.

1. Each system that performs critical control functions must be designed and installed to ensure that the operation and operational capabilities of these critical control functions are not adversely affected when the helicopter is exposed to high intensity radiated fields external to the helicopter.

2. For the purpose of this special condition, critical control functions are defined as those functions, whose failure would contribute to, or cause, an unsafe condition that would prevent the continued safe flight and landing of the aircraft.

Issued in Fort Worth, Texas, on November 7, 2007.

Mark R. Schilling,

Acting Manager, Aircraft Certification Service, Rotorcraft Directorate.

[FR Doc. 07–5698 Filed 11–15–07; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA–2007–0076; Directorate Identifier 2007–NM–241–AD; Amendment 39–15246; AD 2007–22–10]

RIN 2120–AA64

Airworthiness Directives; Airbus Model A330–200, A330–300, A340–200, A340–300, A340–500, and A340–600 Series Airplanes

AGENCY: Federal Aviation Administration (FAA), Department of Transportation (DOT).

ACTION: Final rule; correction.

SUMMARY: The FAA is correcting a typographical error in an existing airworthiness directive (AD) that was published in the **Federal Register** on November 1, 2007 (72 FR 61796). The error resulted in an error in an airplane series number identified in Table 2 of the AD. This AD applies to all Airbus Model A330–200, A330–300, A340–200, A340–300, A340–500, and A340–600

series airplanes. This AD requires repetitive detailed visual inspections for cracking of the LH (left hand) and RH (right hand) wing MLG (main landing gear) rib 6 aft bearing lugs, and repair or replacement of the MLG rib 6 fitting, if necessary.

DATES: Effective November 16, 2007.

ADDRESSES: You may examine the AD docket on the Internet at <http://www.regulations.gov>; or in person at the Docket Operations office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this AD, the regulatory evaluation, any comments received, and other information. The street address for the Docket Operations office (telephone (800) 647–5527) is the Document Management Facility, U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue, SE., Washington, DC.

FOR FURTHER INFORMATION CONTACT: Tim Backman, Aerospace Engineer, International Branch, ANM–116, Transport Airplane Directorate, FAA, 1601 Lind Avenue, SW., Renton, Washington 98057–3356; telephone (425) 227–2797; fax (425) 227–1149.

SUPPLEMENTARY INFORMATION: On October 24, 2007, the FAA issued AD 2007–22–10, amendment 39–15246 (72 FR 61796, November 1, 2007), for all Airbus Model A330–200, A330–300, A340–200, A340–300, A340–500, and A340–600 series airplanes. The AD requires repetitive detailed visual inspections for cracking of the LH (left hand) and RH (right hand) wing MLG (main landing gear) rib 6 aft bearing lugs, and repair or replacement of the MLG rib 6 fitting, if necessary.

As published, Table 2 of the AD states that certain repetitive inspection intervals apply to Model “A300–300 series airplanes, except WV27.” That sentence contains a typographical error and, instead, should state that those repetitive inspection intervals apply to Model “A340–300 series airplanes, except WV27.”

No other part of the regulatory information has been changed; therefore, the final rule is not republished in the **Federal Register**.

The effective date of this AD remains November 16, 2007.

§ 39.13 [Corrected]

■ In the **Federal Register** of November 1, 2007, on page 61799, Table 2 of AD 2007–22–10 is corrected to read as follows:

* * * * *

TABLE 2.—REPETITIVE INSPECTION INTERVALS

Model	Interval (whichever occurs first)
A330–200 series airplanes	300 flight cycles or 1,500 flight hours.
A330–300 series airplanes	300 flight cycles or 900 flight hours.
A340–200 series airplanes	200 flight cycles or 800 flight hours.
A340–300 Series airplanes, except WV27	200 flight cycles or 800 flight hours.
A340–300 series airplanes, WV27	200 flight cycles or 400 flight hours.
A340–500 and –600 series airplanes	100 flight cycles or 500 flight hours.

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Issued in Renton, Washington, on November 7, 2007.

Ali Bahrami,

Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. E7–22305 Filed 11–15–07; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 97

[Docket No. 30579; Amdt. No. 3244]

Standard Instrument Approach Procedures; Miscellaneous Amendments

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final rule.

SUMMARY: This rule amends Standard Instrument Approach Procedures (SIAPs) for operations at certain airports. These regulatory actions are needed because of changes in the National Airspace System, such as the commissioning of new navigational facilities, adding of new obstacles, or changing air traffic requirements. These changes are designed to provide safe and efficient use of the navigable airspace and to promote safe flight operations under instrument flight rules at the affected airports.

DATES: This rule is effective November 16, 2007. The compliance date for each SIAP is specified in the amendatory provisions.

The incorporation by reference of certain publications listed in the regulations is approved by the Director of the Federal Register as of November 16, 2007.

ADDRESSES: Availability of matter incorporated by reference in the amendment is as follows:

For Examination—

1. FAA Rules Docket, FAA Headquarters Building, 800 Independence Avenue, SW., Washington, DC 20591;

2. The FAA Regional Office of the region in which the affected airport is located;

3. The National Flight Procedures Office, 6500 South MacArthur Blvd., Oklahoma City, OK 73169 or,

4. The National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call 202–741–6030, or go to: http://www.archives.gov/federal_register/code_of_federal_regulations/ibr_locations.html.

Availability—All SIAPs are available online free of charge. Visit nfdc.faa.gov to register. Additionally, individual SIAP and Takeoff Minimums and ODP copies may be obtained from:

1. FAA Public Inquiry Center (APA–200), FAA Headquarters Building, 800 Independence Avenue, SW., Washington, DC 20591; or

2. The FAA Regional Office of the region in which the affected airport is located.

FOR FURTHER INFORMATION CONTACT:

Harry J. Hodges, Flight Procedure Standards Branch (AFS–420), Flight Technologies and Programs Division, Flight Standards Service, Federal Aviation Administration, Mike Monroney Aeronautical Center, 6500 South MacArthur Blvd., Oklahoma City, OK. 73169 (Mail Address: P.O. Box 25082 Oklahoma City, OK 73125) telephone: (405) 954–4164.

SUPPLEMENTARY INFORMATION: This rule amends Title 14, Code of Federal Regulations, Part 97 (14 CFR part 97) by amending the referenced SIAPs. The complete regulatory description of each SIAP is listed on the appropriate FAA Form 8260, as modified by the National Flight Data Center (FDC)/Permanent Notice to Airmen (P–NOTAM), and is incorporated by reference in the amendment under 5 U.S.C. 552(a), 1 CFR part 51, and § 97.20 of Title 14 of the Code of Federal Regulations.

The large number of SIAPs, their complex nature, and the need for a special format make their verbatim publication in the **Federal Register** expensive and impractical. Further, airmen do not use the regulatory text of

the SIAPs, but refer to their graphic depiction on charts printed by publishers of aeronautical materials. Thus, the advantages of incorporation by reference are realized and publication of the complete description of each SIAP contained in FAA form documents is unnecessary. This amendment provides the affected CFR sections and specifies the types of SIAP and the corresponding effective dates. This amendment also identifies the airport and its location, the procedure and the amendment number.

The Rule

This amendment to 14 CFR part 97 is effective upon publication of each separate SIAP as amended in the transmittal. For safety and timeliness of change considerations, this amendment incorporates only specific changes contained for each SIAP as modified by FDC/P–NOTAMs.

The SIAPs, as modified by FDC P–NOTAM, and contained in this amendment are based on the criteria contained in the U.S. Standard for Terminal Instrument Procedures (TERPS). In developing these changes to SIAPs, the TERPS criteria were applied only to specific conditions existing at the affected airports. All SIAP amendments in this rule have been previously issued by the FAA in a FDC NOTAM as an emergency action of immediate flight safety relating directly to published aeronautical charts. The circumstances which created the need for all these SIAP amendments requires making them effective in less than 30 days.

Because of the close and immediate relationship between these SIAPs and safety in air commerce, I find that notice and public procedure before adopting these SIAPs are impracticable and contrary to the public interest and, where applicable, that good cause exists for making these SIAPs effective in less than 30 days.

Conclusion

The FAA has determined that this regulation only involves an established body of technical regulations for which