

doorstop does not meet integrity standards during any inspection required by this paragraph, before further flight, repair or replace the doorstop with a new or serviceable doorstop in accordance with the repair drawing.

(B) Within 5,000 flight hours after accomplishing the inspection described in paragraph (f)(1) of this AD, repair the support beam in accordance with the repair drawing or replace in accordance with the service bulletin. Doing the repair or replacement terminates the inspections required by paragraph (f)(1)(i)(A) of this AD.

(ii) If any support beam is damaged at one or two spring locations and any damage exceeds the limits defined in Bombardier Repair Drawing RD 8/4-52-202, Issue 1, dated December 2, 2005, prior to further flight, replace the damaged support beam with a new support beam in accordance with the service bulletin.

(iii) If any support beam is damaged at two spring locations and the damage is within the limits defined in Bombardier Repair Drawing RD 8/4-52-202, Issue 1, dated December 2, 2005, prior to further flight, repair the support beam in accordance with the repair drawing.

(2) Within 1,000 flight hours after the effective date of this AD, remove the nuts and washers at the bottom of the over-centering spring assemblies of the forward baggage door, aft service door, and aft passenger door by incorporating Modsum 4-155296, in accordance with Bombardier Service Bulletin 84-52-51, Revision A, dated September 8, 2006.

#### FAA AD Differences

**Note:** This AD differs from the MCAI and/or service information as follows: No differences.

#### Other FAA AD Provisions

(g) The following provisions also apply to this AD:

(1) *Alternative Methods of Compliance (AMOCs):* The Manager, New York Aircraft Certification Office (ACO), FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. Send information to ATTN: Pong K. Lee, Aerospace Engineer, New York ACO, FAA, 1600 Stewart Avenue, Suite 410, Westbury, New York 11590; telephone (516) 228-7324; fax (516) 794-5531. Before using any approved AMOC on any airplane to which the AMOC applies, notify your appropriate principal inspector (PI) in the FAA Flight Standards District Office (FSDO), or lacking a PI, your local FSDO.

(2) *Airworthy Product:* For any requirement in this AD to obtain corrective actions from a manufacturer or other source, use these actions if they are FAA-approved. Corrective actions are considered FAA-approved if they are approved by the State of Design Authority (or their delegated agent). You are required to assure the product is airworthy before it is returned to service.

(3) *Reporting Requirements:* For any reporting requirement in this AD, under the provisions of the Paperwork Reduction Act, the Office of Management and Budget (OMB) has approved the information collection

requirements and has assigned OMB Control Number 2120-0056.

#### Related Information

(h) Refer to MCAI Canadian Airworthiness Directive CF-2007-05, effective April 24, 2007; Bombardier Service Bulletin 84-52-51, Revision A, dated September 8, 2006, including Service Bulletin 8-MHI0084, Revision C, dated September 6, 2006; and Bombardier Repair Drawing RD 8/4-52-202, Issue 1, dated December 2, 2005, for related information.

Issued in Renton, Washington, on November 5, 2007.

**Ali Bahrami,**

Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. E7-22103 Filed 11-9-07; 8:45 am]

**BILLING CODE 4910-13-P**

## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 39

[Docket No. FAA-2007-0182; Directorate Identifier 2007-NM-138-AD]

RIN 2120-AA64

#### **Airworthiness Directives; Dassault Model Fan Jet Falcon, Fan Jet Falcon Series C, D, E, F, and G Airplanes; Model Mystere-Falcon 200 Airplanes; and Model Mystere-Falcon 20-C5, 20-D5, 20-E5, and 20-F5 Airplanes**

**AGENCY:** Federal Aviation Administration (FAA), DOT.

**ACTION:** Notice of proposed rulemaking (NPRM).

**SUMMARY:** We propose to adopt a new airworthiness directive (AD) for the products listed above. This proposed AD results from mandatory continuing airworthiness information (MCAI) originated by an aviation authority of another country to identify and correct an unsafe condition on an aviation product. The MCAI describes the unsafe condition as:

One occurrence has been reported where a maintenance operation had been performed on the elevator controls, and bellcrank \* \* \* located in the Right Hand MLG (main landing gear) wheel well was mistakenly installed upside down. This discrepancy and improper installation caused an unexpected 5° positioning offset of the elevator control surfaces leading to a hazardous condition on landing. [involving] the pilot being unable to flare the aircraft as needed \* \* \* [which resulted in a hard landing].

The unsafe condition is reduced controllability of the airplane. The proposed AD would require actions that are intended to address the unsafe condition described in the MCAI.

**DATES:** We must receive comments on this proposed AD by December 13, 2007.

**ADDRESSES:** You may send comments by any of the following methods:

- *DOT Docket Web Site:* Go to <http://dms.dot.gov> and follow the instructions for sending your comments electronically.

- *Fax:* (202) 493-2251.

- *Mail:* U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue, SE., Washington, DC 20590.

- *Hand Delivery:* Room W12-140 on the ground floor of the West Building, 1200 New Jersey Avenue, SE., Washington, DC, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

- *Federal eRulemaking Portal:* <http://www.regulations.gov>. Follow the instructions for submitting comments.

#### Examining the AD Docket

You may examine the AD docket on the Internet at <http://dms.dot.gov>; or in person at the Docket Operations office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this proposed AD, the regulatory evaluation, any comments received, and other information. The street address for the Docket Operations office (telephone (800) 647-5527) is in the **ADDRESSES** section. Comments will be available in the AD docket shortly after receipt.

**FOR FURTHER INFORMATION CONTACT:** Tom Rodriguez, Aerospace Engineer, International Branch, ANM-116, FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98057-3356; telephone (425) 227-1137; fax (425) 227-1149.

#### **SUPPLEMENTARY INFORMATION:**

##### **Comments Invited**

We invite you to send any written relevant data, views, or arguments about this proposed AD. Send your comments to an address listed under the **ADDRESSES** section. Include "Docket No. FAA-2007-0182; Directorate Identifier 2007-NM-138-AD" at the beginning of your comments. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of this proposed AD. We will consider all comments received by the closing date and may amend this proposed AD based on those comments.

We will post all comments we receive, without change, to <http://dms.dot.gov>, including any personal information you provide. We will also post a report summarizing each

substantive verbal contact we receive about this proposed AD.

### Discussion

The European Aviation Safety Agency (EASA), which is the Technical Agent for the Member State of the European Community, has issued EASA Airworthiness Directive 2006-0185, dated July 6, 2006 (referred to after this as "the MCAI"), to correct an unsafe condition for the specified products. The MCAI states:

One occurrence has been reported where a maintenance operation had been performed on the elevator controls, and bellcrank P/N (part number) MY20273017 or P/N MY20273017015 located in the Right Hand MLG (main landing gear) wheel well was mistakenly installed upside down. This discrepancy and improper installation caused an unexpected 5° positioning offset of the elevator control surfaces leading to a hazardous condition on landing, [involving] the pilot being unable to flare the aircraft as needed \* \* \* [which resulted in a hard landing].

The purpose of this AD is to prevent reoccurrence of this kind of incident introducing disabusing markings on the incriminated parts by applying SB (Service Bulletin) F20-768 or SB F200-122 as appropriate.

The unsafe condition is reduced controllability of the airplane. Corrective actions include verifying the correct assembly of the elevator bellcrank and re-installing if necessary. You may obtain further information by examining the MCAI in the AD docket.

### Relevant Service Information

Dassault has issued Service Bulletins F20-768, dated May 23, 2006, and F200-122, dated May 23, 2006. The actions described in this service information are intended to correct the unsafe condition identified in the MCAI.

### FAA's Determination and Requirements of This Proposed AD

This product has been approved by the aviation authority of another country, and is approved for operation in the United States. Pursuant to our bilateral agreement with the State of Design Authority, we have been notified of the unsafe condition described in the MCAI and service information referenced above. We are proposing this AD because we evaluated all pertinent information and determined an unsafe condition exists and is likely to exist or develop on other products of the same type design.

### Differences Between This AD and the MCAI or Service Information

We have reviewed the MCAI and related service information and, in

general, agree with their substance. But we might have found it necessary to use different words from those in the MCAI to ensure the AD is clear for U.S. operators and is enforceable. In making these changes, we do not intend to differ substantively from the information provided in the MCAI and related service information.

We might also have proposed different actions in this AD from those in the MCAI in order to follow FAA policies. Any such differences are highlighted in a NOTE within the proposed AD.

### Costs of Compliance

Based on the service information, we estimate that this proposed AD would affect about 255 products of U.S. registry. We also estimate that it would take about 3 work-hours per product to comply with the basic requirements of this proposed AD. The average labor rate is \$80 per work-hour. Required parts would cost about \$9 per product. Where the service information lists required parts costs that are covered under warranty, we have assumed that there will be no charge for these costs. As we do not control warranty coverage for affected parties, some parties may incur costs higher than estimated here. Based on these figures, we estimate the cost of the proposed AD on U.S. operators to be \$63,495, or \$249 per product.

### Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. "Subtitle VII: Aviation Programs," describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in "Subtitle VII, Part A, Subpart III, Section 44701: General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

### Regulatory Findings

We determined that this proposed AD would not have federalism implications under Executive Order 13132. This proposed AD would not have a substantial direct effect on the States, on

the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify this proposed regulation:

1. Is not a "significant regulatory action" under Executive Order 12866;
2. Is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and
3. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared a regulatory evaluation of the estimated costs to comply with this proposed AD and placed it in the AD docket.

### List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

### The Proposed Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA proposes to amend 14 CFR part 39 as follows:

### PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

#### § 39.13 [Amended]

2. The FAA amends § 39.13 by adding the following new AD:

**Dassault Aviation (Formerly Avions Marcel Dassault-Breguet Aviation (AMD/BA)):**  
Docket No. FAA-2007-0182; Directorate Identifier 2007-NM-138-AD.

#### Comments Due Date

- (a) We must receive comments by December 13, 2007.

#### Affected ADs

- (b) None.

#### Applicability

- (c) This AD applies to all Dassault Model Fan Jet Falcon, Fan Jet Falcon series C, D, E, F, and G airplanes; Model Mystere-Falcon 200 airplanes; and Model Mystere-Falcon 20-C5, 20-D5, 20-E5, and 20-F5 airplanes, all serial numbers, certificated in any category.

#### Subject

- (d) Air Transport Association (ATA) of America Code 27: Flight Controls.

#### Reason

- (e) The mandatory continuing airworthiness information (MCAI) states:

One occurrence has been reported where a maintenance operation had been performed on the elevator controls, and bellcrank P/N

(part number) MY20273017 or P/N MY20273017015 located in the Right Hand MLG (main landing gear) wheel well was mistakenly installed upside down. This discrepancy and improper installation caused an unexpected 5° positioning offset of the elevator control surfaces leading to a hazardous condition on landing, [involving] the pilot being unable to flare the aircraft as needed \* \* \* [which resulted in a hard landing].

The purpose of this AD is to prevent reoccurrence of this kind of incident introducing disabusing markings on the incriminated parts by applying SB (Service Bulletin) F20-768 or SB F200-122 as appropriate.

The unsafe condition is reduced controllability of the airplane. Corrective actions include verifying the correct assembly of the elevator bellcrank and re-installing if necessary.

#### Actions and Compliance

(f) Within 74 months from the effective date of this AD, unless already done, do the following actions.

(1) Verify the correct assembly of the elevator bellcrank P/N (part number) MY20273-17 or P/N MY20273-17-15 at frame 26, as instructed in Dassault Service Bulletin F20-768, dated May 23, 2006; or Dassault Service Bulletin F200-122, dated May 23, 2006; as applicable.

(2) If the elevator bellcrank is found in the reverse orientation, reinstall it prior to next flight in accordance with Dassault Service Bulletin F20-768, dated May 23, 2006; or Dassault Service Bulletin F200-122, dated May 23, 2006; as applicable.

(3) Label the elevator bellcrank as instructed in Dassault Service Bulletin F20-768, dated May 23, 2006; or Dassault Service Bulletin F200-122, dated May 23, 2006; as applicable.

#### FAA AD Differences

**Note:** This AD differs from the MCAI and/or service information as follows: No Differences.

#### Other FAA AD Provisions

(g) The following provisions also apply to this AD:

(1) Alternative Methods of Compliance (AMOCs): The Manager, International Branch, ANM-116, Transport Airplane Directorate, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. Send information to ATTN: Tom Rodriguez, Aerospace Engineer, International Branch, ANM-116, FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98057-3356; telephone (425) 227-1137; fax (425) 227-1149. Before using any approved AMOC on any airplane to which the AMOC applies, notify your appropriate principal inspector (PI) in the FAA Flight Standards District Office (FSDO), or lacking a PI, your local FSDO.

(2) Airworthy Product: For any requirement in this AD to obtain corrective actions from a manufacturer or other source, use these actions if they are FAA-approved. Corrective actions are considered FAA-

approved if they are approved by the State of Design Authority (or their delegated agent). You are required to assure the product is airworthy before it is returned to service.

(3) Reporting Requirements: For any reporting requirement in this AD, under the provisions of the Paperwork Reduction Act, the Office of Management and Budget (OMB) has approved the information collection requirements and has assigned OMB Control Number 2120-0056.

#### Related Information

(h) Refer to MCAI European Aviation Safety Agency Airworthiness Directive 2006-0185, dated July 6, 2006, and Dassault Service Bulletins F20-768 and F200-122, both dated May 23, 2006, for related information.

Issued in Renton, Washington, on October 23, 2007.

**Stephen P. Boyd,**

*Assistant Manager, Transport Airplane Directorate, Aircraft Certification Service.*

[FR Doc. E7-22102 Filed 11-9-07; 8:45 am]

**BILLING CODE 4910-13-P**

## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 39

[Docket No. FAA-2007-0184; Directorate Identifier 2007-NM-140-AD]

RIN 2120-AA64

#### Airworthiness Directives; Boeing Model 737-100, -200, -200C, -300, -400, and -500 Series Airplanes

**AGENCY:** Federal Aviation Administration (FAA), Department of Transportation (DOT).

**ACTION:** Notice of proposed rulemaking (NPRM).

**SUMMARY:** The FAA proposes to adopt a new airworthiness directive (AD) for certain Boeing Model 737-100, -200, -200C, -300, -400, and -500 series airplanes. This proposed AD would require various repetitive inspections for cracking of the upper frame to side frame splice of the fuselage, and other specified and corrective actions if necessary. This proposed AD also provides for an optional preventive modification, which would terminate the repetitive inspections. This proposed AD results from a report that the upper frame of the fuselage was severed between stringers S-13L and S-14L at station 747, and the adjacent frame at station 767 had a 1.3-inch-long crack at the same stringer location. We are proposing this AD to detect and correct fatigue cracking of the upper frame to side frame splice of the fuselage, which could result in reduced

structural integrity of the frame and adjacent lap joint. This reduced structural integrity can increase loading in the fuselage skin, which will accelerate skin crack growth and result in decompression of the airplane.

**DATES:** We must receive comments on this proposed AD by December 28, 2007.

**ADDRESSES:** You may send comments by any of the following methods:

- *Federal eRulemaking Portal:* Go to <http://www.regulations.gov>. Follow the instructions for submitting comments.

- *Fax:* 202-493-2251.

- *Mail:* U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue, SE., Washington, DC 20590.

*Hand Delivery:* U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE., Washington, DC 20590, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

For service information identified in this AD contact Boeing Commercial Airplanes, P.O. Box 3707, Seattle, Washington 98124-2207.

#### Examining the AD Docket

You may examine the AD docket on the Internet at <http://www.regulations.gov>; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this proposed AD, the regulatory evaluation, any comments received, and other information. The street address for the Docket Office (telephone 800-647-5527) is in the **ADDRESSES** section. Comments will be available in the AD docket shortly after receipt.

#### FOR FURTHER INFORMATION CONTACT:

Wayne Lockett, Aerospace Engineer, Airframe Branch, ANM-120S, FAA, Seattle Aircraft Certification Office, 1601 Lind Avenue, SW., Renton, Washington 98057-3356; telephone (425) 917-6447; fax (425) 917-6590.

#### SUPPLEMENTARY INFORMATION:

##### Comments Invited

We invite you to send any written relevant data, views, or arguments about this proposed AD. Send your comments to an address listed under the **ADDRESSES** section. Include "Docket No. FAA-2007-0184; Directorate Identifier 2007-NM-140-AD" at the beginning of your comments. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of this proposed AD. We will