

amendments because the Board for good cause determined that delaying implementation of the new primary and secondary credit rates in order to allow notice and public comment would be unnecessary and contrary to the public interest in fostering price stability and sustainable economic growth. For these same reasons, the Board also has not provided 30 days prior notice of the effective date of the rule under section 553(d).

**List of Subjects in 12 CFR Part 201**

Banks, Banking, Federal Reserve System, Reporting and recordkeeping.

**Authority and Issuance**

■ For the reasons set forth in the preamble, the Board is amending 12 CFR Chapter II to read as follows:

**PART 201—EXTENSIONS OF CREDIT BY FEDERAL RESERVE BANKS (REGULATION A)**

■ 1. The authority citation for part 201 continues to read as follows:

**Authority:** 12 U.S.C. 248(i)–(j), 343 *et seq.*, 347a, 347b, 347c, 348 *et seq.*, 357, 374, 374a, and 461.

■ 2. In § 201.51, paragraphs (a) and (b) are revised to read as follows:

**§ 201.51 Interest rates applicable to credit extended by a Federal Reserve Bank.<sup>1</sup>**

(a) *Primary credit.* The interest rates for primary credit provided to depository institutions under § 201.4(a) are:

Federal Reserve Bank	Rate	Effective
Boston .....	5.75	August 17, 2007.
New York .....	5.75	August 17, 2007.
Philadelphia .....	5.75	August 17, 2007.
Cleveland .....	5.75	August 17, 2007.
Richmond .....	5.75	August 17, 2007.
Atlanta .....	5.75	August 17, 2007.
Chicago .....	5.75	August 17, 2007.
St. Louis .....	5.75	August 20, 2007.
Minneapolis .....	5.75	August 17, 2007.
Kansas City .....	5.75	August 17, 2007.
Dallas .....	5.75	August 17, 2007.
San Francisco ....	5.75	August 17, 2007.

(b) *Secondary credit.* The interest rates for secondary credit provided to depository institutions under 201.4(b) are:

Federal Reserve Bank	Rate	Effective
Boston .....	6.25	August 17, 2007.

<sup>1</sup> The primary, secondary, and seasonal credit rates described in this section apply to both advances and discounts made under the primary, secondary, and seasonal credit programs, respectively.

Federal Reserve Bank	Rate	Effective
New York .....	6.25	August 17, 2007.
Philadelphia .....	6.25	August 17, 2007.
Cleveland .....	6.25	August 17, 2007.
Richmond .....	6.25	August 17, 2007.
Atlanta .....	6.25	August 17, 2007.
Chicago .....	6.25	August 17, 2007.
St. Louis .....	6.25	August 20, 2007.
Minneapolis .....	6.25	August 17, 2007.
Kansas City .....	6.25	August 17, 2007.
Dallas .....	6.25	August 17, 2007.
San Francisco ....	6.25	August 17, 2007.

\* \* \* \* \*

By order of the Board of Governors of the Federal Reserve System, August 20, 2007.

**Jennifer J. Johnson,**

*Secretary of the Board.*

[FR Doc. E7–16764 Filed 8–23–07; 8:45 am]

**BILLING CODE 6210–02–P**

**DEPARTMENT OF TRANSPORTATION**

**Federal Aviation Administration**

**14 CFR Part 39**

**[Docket No. FAA–2006–23742; Directorate Identifier 2005–NE–53–AD; Amendment 39–15180; AD 2007–17–21]**

**RIN 2120–AA64**

**Airworthiness Directives; Pratt & Whitney (PW) JT9D–7R4 Series Turbofan Engines**

**AGENCY:** Federal Aviation Administration (FAA), Department of Transportation (DOT).

**ACTION:** Final rule.

**SUMMARY:** The FAA is adopting a new airworthiness directive (AD) for PW JT9D–7R4 series turbofan engines. This AD requires removing reduced cooling flow 2nd stage high pressure turbine (HPT) vane assemblies, part numbers (P/Ns) 797282, 796972, 800082, 800072, 803182, 803282, and 822582, installed in 2nd stage HPT vane cluster assemblies P/Ns 797592, 797372, 799872, 799782, and 822572. It also requires a visual and a fluorescent penetrant inspection (FPI) of the 2nd stage HPT air seal assembly, P/N 815097. This AD results from a report of an uncontained failure of the 2nd stage HPT air seal assembly, caused by the air seal assembly brace disengaging from the air seal, due to insufficient cooling air flow. We are issuing this AD to prevent uncontained failure of the 2nd stage HPT air seal assembly, leading to engine in-flight shutdown and damage to the airplane.

**DATES:** This AD becomes effective September 28, 2007. The Director of the

Federal Register approved the incorporation by reference of certain publications listed in the regulations as of September 28, 2007.

**ADDRESSES:** You can get the service information identified in this AD from Pratt & Whitney, 400 Main St., East Hartford, CT 06108; telephone (860) 565–8770; fax (860) 565–4503.

You may examine the AD docket on the Internet at <http://dms.dot.gov> or at the U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue, SE., Washington, DC 20590.

**FOR FURTHER INFORMATION CONTACT:**

Mark Riley, Aerospace Engineer, Engine Certification Office, FAA, Engine and Propeller Directorate, 12 New England Executive Park, Burlington, MA 01803; telephone (781) 238–7758, fax (781) 238–7199.

**SUPPLEMENTARY INFORMATION:** The FAA proposed to amend 14 CFR part 39 with a proposed AD. The proposed AD applies to PW JT9D–7R4E1, –7R4E4, –7R4G2, and –7R4H1 turbofan engines. We published the proposed AD in the **Federal Register** on February 2, 2007 (72 FR 4964). That action proposed to require removal of reduced cooling flow 2nd stage HPT vane assemblies. It also proposed to require a visual and an FPI of the 2nd stage HPT air seal assembly.

**Examining the AD Docket**

You may examine the AD docket on the Internet at <http://dms.dot.gov>; or in person at the Docket Operations office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains the NPRM, the regulatory evaluation, any comments received, and other information. The street address for the Docket Operations office (telephone (800) 647–5527) is provided in the **ADDRESSES** section. Comments will be available in the AD docket shortly after receipt.

**Comments**

We provided the public the opportunity to participate in the development of this AD. We have considered the comments received.

**Revise the Cost Impact**

FEDEX requests that we revise the estimated cost impact to implement the AD because they believe that the actual cost of modification (according to PW Alert Service Bulletin (ASB) JT9D–7R4–A72–596, dated September 15, 2005) for one set of HPT 2nd stage vane assemblies will be approximately \$23,000, instead of \$5,400 projected by the proposed AD. We do not agree. We

provided the estimated material cost of \$5,400 per engine from the engine manufacturer in the Costs of Compliance section in the proposed AD. Parts included new 2nd stage HPT vane covers, part number (P/N) 822734, and rivets, P/N ST1219–213, that are required to modify the 2nd stage HPT vane cluster assemblies, as specified in ASB JT9D–7R4–A72–596, dated September 15, 2005. However, based on latest pricing information in the PW spare parts price catalog, we increased the estimated material cost to \$6,700. We also estimate that it will take about 65.5 work-hours per engine to perform the actions, and that the average labor rate is \$80 per work-hour. Based on these figures, we estimate the total cost of the AD to be \$11,940 per engine.

#### Revise the Applicability Section

Boeing Company asks that we revise paragraph (c) of the AD to remove the reference to the Boeing model 747–200B, –200C, –200F, and –300 airplanes. Boeing states that the JT9D–7R4E1, –7R4E4, and –7R4H1 are not used on any model Boeing 747 and the description implies that these engines may be installed on the Boeing 747 airplane. We partially agree. We removed the letter designations for the 747–200 series airplanes. However, we do not agree that the airplane listing in paragraph (c) implies that the JT9D–7R4E1, –7R4E4, and –7R4H1 engines may be installed on the Boeing 747. The airplane references in paragraph (c) are informative only. The Type Certificate Data Sheet for an airplane specifies the engines that are installed on the airplane, not the applicability paragraph of an AD.

PW asks that we revise the Applicability section to clarify the affected engine models. They also request that we add the following aircraft to the list of airplanes that can have these engines installed: Boeing 767–200, Airbus A300–600, and A310–300.

We partially agree and revised the Applicability section of the AD to add Boeing 767–200, and Airbus A300–600 and A310–300 airplane models to paragraph (c) of the AD. We do not agree that we need to clarify the engine models with Pratt & Whitney internal model designation. We specify the affected engines by the model designations specified in the Type Certificate Data Sheet for the engine.

#### Revise Paragraph (f) of the Compliance Section

PW asks that we revise paragraph (f) of the Compliance section to delete the specific lenticular seal part number

reference, as there are other lenticular seal part numbers in the field that will also require visual and FPI inspections at the next HPT module exposure. We partially agree. The specific lenticular seal part number should be removed from paragraph (f) of the Compliance section of the AD because there are other lenticular seal part numbers in the field that will also require inspections. However, to prevent further delay in addressing the unsafe condition, we are issuing this AD with the specific lenticular seal part number listed. Removal of the part number will expand the number of parts requiring inspection and we must give the public an opportunity to comment on this change before we can incorporate it. We will supersede this AD to remove the lenticular seal part number.

#### List Required Lenticular Seal Inspections

PW asks that we revise paragraph (f) of the Compliance section to list all of the specific inspections required for the lenticular seal. We agree and listed each inspection procedure in the PW engine manual required for the lenticular seal. We also changed the paragraph to (g).

#### Revise Paragraph (g) of the Compliance Section

PW also asks that we revise paragraph (g) of the Compliance section to reference the associated 2nd stage HPT vane cluster assemblies (higher level assembly) that have reduced cooling flow 2nd stage HPT vane assemblies installed, because airlines may not recognize individual 2nd stage vane assemblies. We agree. We revised paragraph (g) of the Compliance section to state the following: “At the next HPT module exposure, remove reduced cooling flow 2nd stage HPT vane assemblies, P/Ns 797282, 796972, 800082, 800072, 803182, 803282, and 822582, installed in 2nd stage HPT vane cluster assemblies, P/Ns 797592, 797372, 799872, 799782, and 822572.” We also designated the paragraph as (f).

#### Correct Errors and Remove Duplicate P/N Reference

Boeing Company asks that we correct typographical errors for the engine designations in paragraph (c) in the Applicability section. Boeing states that the JT9D–7R4E1, –7R4E4, –7R4G2, and –7R4H1 should be JT9D–7R4E1, –7R4E4, –7R4G2, and –7R4H1, respectively.

Boeing also asks that we revise paragraph (g) in the Compliance section to remove a duplicate reference to part number 803182. We agree and corrected the Applicability section and the

duplicate P/N reference in paragraph (g).

#### Conclusion

We have carefully reviewed the available data, including the comments received, and determined that air safety and the public interest require adopting the AD with the changes described previously. We have determined that these changes will neither increase the economic burden on any operator nor increase the scope of the AD.

#### Costs of Compliance

We estimate that this proposed AD would affect 85 PW JT9D–7R4 series turbofan engines installed on airplanes U.S. registry. We also estimate that it will take about 65.5 work-hours per engine to perform the actions, and that the average labor rate is \$80 per work-hour. Required parts will cost about \$6,700 per engine. Based on these figures, we estimate the total cost of the AD to U.S. operators to be \$1,014,900.

#### Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA’s authority to issue rules on aviation safety. Subtitle I, Section 106, describes the authority of the FAA Administrator. Subtitle VII, Aviation Programs, describes in more detail the scope of the Agency’s authority.

We are issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section 44701, “General requirements.” Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

#### Regulatory Findings

We have determined that this AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that this AD:

- (1) Is not a “significant regulatory action” under Executive Order 12866;
- (2) Is not a “significant rule” under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and

(3) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared a summary of the costs to comply with this AD and placed it in the AD Docket. You may get a copy of this summary at the address listed under **ADDRESSES**.

#### List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

#### Adoption of the Amendment

Accordingly, under the authority delegated to me by the Administrator, the Federal Aviation Administration amends 14 CFR part 39 as follows:

#### PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

#### § 39.13 [Amended]

2. The FAA amends § 39.13 by adding the following new airworthiness directive:

**2007-17-21 Pratt & Whitney:** Amendment 39-15180. Docket No. FAA-2006-23742; Directorate Identifier 2005-NE-53-AD.

#### Effective Date

(a) This airworthiness directive (AD) becomes effective September 28, 2007.

#### Affected ADs

(b) None.

#### Applicability

(c) This AD applies to Pratt & Whitney (PW) JT9D-7R4G2, -7R4E1, -7R4E4, and -7R4H1 series turbofan engines. These engines are installed on, but not limited to, Boeing 747-200, -300, 767-200, and Airbus A300-600 and A310-300 series airplanes.

#### Unsafe Condition

(d) This AD results from a report of an uncontained failure of the 2nd stage high pressure turbine (HPT) air seal assembly, caused by the air seal assembly brace disengaging from the air seal, due to insufficient cooling air flow. We are issuing this AD to prevent uncontained failure of the 2nd stage HPT air seal assembly, leading to engine in-flight shutdown and damage to the airplane.

#### Compliance

(e) You are responsible for having the actions required by this AD performed at the next HPT module exposure after the effective date of this AD, unless the actions have already been done.

(f) At the next HPT module exposure, remove reduced cooling flow 2nd stage HPT vane assemblies part numbers (P/Ns): 797282, 796972, 800082, 800072, 803182,

803282, and 822582, installed in 2nd stage HPT vane cluster assemblies: P/Ns 797592, 797372, 799872, 799782, and 822572.

(g) For 2nd stage HPT air seals that have operated in an engine with reduced cooling flow HPT vane assemblies, at the next HPT module exposure do the following:

(1) Perform a onetime visual inspection of the 2nd stage HPT air seal assembly, P/N 815097, using instructions in JT9D-7R4 engine manual, Section 72-51-22, Inspection/Check-01, paragraphs 1.D.(1), 1.D.(4), and 1.D.(6).

(2) Perform a fluorescent penetrant inspection (FPI) of the 2nd stage HPT air seal assembly for cracks, using instructions in JT9D-7R4 engine manual, Section 71-51-00, Inspection/Check-03.

#### Definition

(h) For the purpose of this AD, an HPT module exposure is defined as removing the 1st stage HPT rotor or the 2nd stage HPT rotor from the HPT case.

#### Alternative Methods of Compliance

(i) The Manager, Engine Certification Office, has the authority to approve alternative methods of compliance for this AD if requested using the procedures found in 14 CFR 39.19.

#### Related Information

(j) Pratt & Whitney Alert Service Bulletin JT9D-7R4-A72-596, dated September 15, 2005, contains information for modifying the reduced cooling flow 2nd stage HPT vane assemblies.

Issued in Burlington, Massachusetts, on August 17, 2007.

**Peter A. White,**

*Acting Manager, Engine and Propeller Directorate, Aircraft Certification Service.*

[FR Doc. E7-16665 Filed 8-23-07; 8:45 am]

**BILLING CODE 4910-13-P**

## DEPARTMENT OF THE TREASURY

### Internal Revenue Service

#### 26 CFR Parts 1 and 602

[TD 9357]

RIN 1545-BE09

#### Elimination of Country-by-Country Reporting to Shareholders of Foreign Taxes Paid by Regulated Investment Companies

**AGENCY:** Internal Revenue Service (IRS), Treasury.

**ACTION:** Final regulations.

**SUMMARY:** This document contains final regulations that generally eliminate country-by-country reporting by a regulated investment company (RIC) to its shareholders of foreign source income that the RIC takes into account and foreign taxes that it pays. This change is necessary to conform the

regulations to changes in the tax law relating to the foreign tax credit. These final regulations will affect certain RICs that pay foreign taxes and the shareholders of those RICs.

**DATES:** *Effective Date:* These regulations are effective August 24, 2007.

*Applicability Date:* These regulations are applicable for RIC taxable years ending on or after December 31, 2007. For reporting purposes, however, a taxpayer may rely on the current regulations for a taxable year ending on or after December 31, 2007, and beginning before August 24, 2007.

**FOR FURTHER INFORMATION CONTACT:** Richard C. LaFalce, (202) 622-3930 (not a toll-free number).

#### SUPPLEMENTARY INFORMATION:

##### Paperwork Reduction Act

The collection of information contained in these final regulations has been reviewed and approved by the Office of Management and Budget in accordance with the Paperwork Reduction Act of 1995 (44 U.S.C. 3507(d)) under control number 1545-2035. Comments on the accuracy of the estimated burden and suggestions for reducing the burden should be sent to the Internal Revenue Service, Attn: IRS Reports Clearance Officer, SE:W:CAR:MP:T:T:SP, Washington, DC 20224.

The collection of information in these final regulations is in § 1.853-4(c) and (d). A RIC is required to notify the IRS of amounts of income received from sources within foreign countries and possessions of the United States and taxes paid to each such foreign country or possession in order that the IRS may monitor shareholder compliance with the foreign tax credit provisions. The collection of information is required if a RIC elects to pass through the benefits of the foreign tax credit to its shareholders.

*Estimated total annual recordkeeping burden:* 80 hours.

*Estimated average annual burden per recordkeeper:* 2 hours.

*Estimated number of recordkeepers:* 40.

*Estimated frequency of recordkeeping:* Annually.

An agency may not conduct or sponsor, and a person is not required to respond to, a collection of information unless it displays a valid control number.

Books or records relating to a collection of information must be retained as long as their contents may become material in the administration of any internal revenue law. Generally, tax returns and tax return information