

The FAA issues special conditions, as defined in § 11.19, under § 11.38, and they become part of the type certification basis under the provisions of § 21.101.

Novel or Unusual Design Features

As noted earlier, the McDonnell Douglas Model DC-10 airplanes modified by Canard Aerospace will incorporate the Astronautics Electronic Flight Information System (EFIS) that will perform critical functions. This system may be vulnerable to high-intensity radiated fields external to the airplane. Current airworthiness standards of part 25 do not contain adequate or appropriate safety standards for protecting this equipment from adverse effects of HIRF. So this system is considered to be a novel or unusual design feature.

Discussion

As previously stated, there is no specific regulation that addresses protection for electrical and electronic systems from HIRF. Increased power levels from radio frequency transmitters and the growing use of sensitive avionics/electronics and electrical systems to command and control airplanes have made it necessary to provide adequate protection.

To ensure that a level of safety is achieved equivalent to that intended by the regulations incorporated by reference, special conditions are needed for the McDonnell Douglas Model DC-10 airplanes modified by Canard Aerospace Corporation. These special conditions require that new avionics/electronics and electrical systems that perform critical functions be designed and installed to preclude component damage and interruption of function because of HIRF.

High-Intensity Radiated Fields (HIRF)

High-power radio frequency transmitters for radio, radar, television, and satellite communications can adversely affect operation of airplane electric and electronic systems. Therefore, the immunity of critical avionics/electronics and electrical systems to HIRF must be established.

Based on surveys and an analysis of existing HIRF emitters, an adequate level of protection exists when airplane system immunity is demonstrated when exposed to the HIRF environments in either paragraph 1 OR 2 below:

1. A minimum environment of 100 volts rms (root-mean-square) per meter electric field strength from 10 KHz to 18 GHz.

a. System elements and their associated wiring harnesses must be

exposed to the environment without benefit of airframe shielding.

b. Demonstration of this level of protection is established through system tests and analysis.

2. An environment external to the airframe of the field strengths shown in the table below for the frequency ranges indicated. Immunity to both peak and average field strength components from the table must be demonstrated.

Frequency	Field strength (volts per meter)	
	Peak	Average
10 kHz–100 kHz	50	50
100 kHz–500 kHz	50	50
500 kHz–2 MHz	50	50
2 MHz–30 MHz	100	100
30 MHz–70 MHz	50	50
70 MHz–100 MHz	50	50
100 MHz–200 MHz	100	100
200 MHz–400 MHz	100	100
400 MHz–700 MHz	700	50
700 MHz–1 GHz	700	100
1 GHz–2 GHz	2000	200
2 GHz–4 GHz	3000	200
4 GHz–6 GHz	3000	200
6 GHz–8 GHz	1000	200
8 GHz–12 GHz	3000	300
12 GHz–18 GHz	2000	200
18 GHz–40 GHz	600	200

The field strengths are expressed in terms of peak of the root-mean-square (rms) over the complete modulation period.

The environment levels identified above are the result of an FAA review of existing studies on the subject of HIRF and of the work of the Electromagnetic Effects Harmonization Working Group of the Aviation Rulemaking Advisory Committee.

Applicability

These special conditions are applicable to McDonnell Douglas Model DC-10 airplanes modified by Canard Aerospace Corporation. Should Canard Aerospace apply at a later date for a supplemental type certificate to modify any other model included on Type Certificate No. A22WE to incorporate the same or similar novel or unusual design feature, these special conditions would apply to that model as well under provisions of § 21.101.

Conclusion

This action affects only certain novel or unusual design features on McDonnell Douglas Model DC-10 airplanes modified by Canard Aerospace Corporation. It is not a rule of general applicability and affects only the applicant who applied to the FAA for approval of these features on the airplane.

List of Subjects in 14 CFR Part 25

Aircraft, Aviation safety, Reporting and recordkeeping requirements.

■ The authority citation for these special conditions is as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701, 44702, 44704.

The Special Conditions

■ Therefore, under the authority delegated to me by the Administrator, the following special conditions are issued as part of the supplemental type certification basis for the McDonnell Douglas Models DC-10-10, 10-15, 10-30, 10-30F, 10-40, and 10-40F airplanes modified by Canard Aerospace Corporation.

1. *Protection from Unwanted Effects of High-Intensity Radiated Fields (HIRF).* Each electrical and electronic system that performs critical functions must be designed and installed to ensure that the operation and operational capability of these systems to perform critical functions are not adversely affected when the airplane is exposed to high-intensity radiated fields.

2. For the purpose of these special conditions, the following definition applies:

Critical Functions: Functions whose failure would contribute to or cause a failure condition that would prevent continued safe flight and landing of the airplane.

Issued in Renton, Washington, on April 16, 2007.

Ali Bahrami,

Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. E7-7699 Filed 4-20-07; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA-2007-27866; Directorate Identifier 2007-NM-055-AD; Amendment 39-15027; AD 2007-08-09]

RIN 2120-AA64

Airworthiness Directives; Short Brothers Model SD3-60 SHERPA, SD3-SHERPA, SD3-30, and SD3-60 Airplanes

AGENCY: Federal Aviation Administration (FAA), Department of Transportation (DOT).

ACTION: Final rule; request for comments.

SUMMARY: We are adopting a new airworthiness directive (AD) for the products listed above. This AD results from mandatory continuing airworthiness information (MCAI) originated by an aviation authority of another country to identify and correct an unsafe condition on an aviation product. The MCAI describes the unsafe condition as:

An analysis of the cable operated control system installed on the SD3 aircraft types that use MS 21260 type end fittings has identified a number of potentially unsafe conditions due to a combination of failures * * *.

The failure of certain control cables could result in the loss of certain critical systems. For example, the loss of the low pressure (LP) fuel control cable in combination with a single failure of a fuel condition control cable on the same engine can cause the loss of the capability to shut down the engine in the event of an engine fire. This AD requires actions that are intended to address the unsafe condition described in the MCAI.

DATES: This AD becomes effective May 8, 2007.

The Director of the Federal Register approved the incorporation by reference of certain publications listed in the AD as of May 8, 2007.

We must receive comments on this AD by May 23, 2007.

ADDRESSES: You may send comments by any of the following methods:

- *DOT Docket Web site:* Go to <http://dms.dot.gov> and follow the instructions for sending your comments electronically.

- *Fax:* (202) 493-2251.

- *Mail:* Docket Management Facility, U.S. Department of Transportation, 400 Seventh Street, SW., Nassif Building, Room PL-401, Washington, DC 20590-0001.

- *Hand Delivery:* Room PL-401 on the plaza level of the Nassif Building, 400 Seventh Street, SW., Washington, DC, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

- *Federal eRulemaking Portal:* <http://www.regulations.gov>. Follow the instructions for submitting comments.

Examining the AD Docket

You may examine the AD docket on the Internet at <http://dms.dot.gov>; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this AD, the regulatory evaluation, any comments received, and other information. The street address for the Docket Office (telephone (800) 647-

5227) is in the **ADDRESSES** section. Comments will be available in the AD docket shortly after receipt.

FOR FURTHER INFORMATION CONTACT: Todd Thompson, Aerospace Engineer, International Branch, ANM-116, FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98057-3356; telephone (425) 227-1175; fax (425) 227-1149.

SUPPLEMENTARY INFORMATION:

Streamlined Issuance of AD

The FAA is implementing a new process for streamlining the issuance of ADs related to MCAI. This streamlined process will allow us to adopt MCAI safety requirements in a more efficient manner and will reduce safety risks to the public. This process continues to follow all FAA AD issuance processes to meet legal, economic, Administrative Procedure Act, and **Federal Register** requirements. We also continue to meet our technical decision-making responsibilities to identify and correct unsafe conditions on U.S.-certificated products.

This AD references the MCAI and related service information that we considered in forming the engineering basis to correct the unsafe condition. The AD contains text copied from the MCAI and for this reason might not follow our plain language principles.

Discussion

The European Aviation Safety Agency (EASA), which is the technical agent for the Member States of the European Community, has issued EASA Emergency Airworthiness Directive 2007-0039-E, dated February 16, 2007 (referred to after this as "the MCAI"), to correct an unsafe condition for the specified products. The MCAI states:

Following the identification of a failed propeller RPM (revolutions per minute) cable end fitting and an LP (low pressure) fuel lever cable end fitting on an SD3 aircraft, several subsequent occurrences of control cable end fittings (type MS21260) with signs of pitting corrosion or cracking have been reported to Bombardier Shorts. All reported instances being identified during ground maintenance inspections on the SD3 fleet. Bombardier Shorts have performed examinations on the failed cable end fittings and established the root cause of failure as stress corrosion cracking of the SAE-AISI 303 stainless steel material they are manufactured from, initiated by pitting corrosion on the surface. The root cause of the stress corrosion is sustained tensile stress in a corrosive (warm, humid and salty) atmosphere.

An analysis of the cable operated control systems installed on the SD3 aircraft types that use MS 21260 type end fittings has identified a number of potentially unsafe

conditions due to a combination of failures that may result from this common mode cause.

The failure of certain control cables could result in the loss of certain critical systems. For example, the loss of the LP fuel control cable in combination with a single failure of a fuel condition control cable on the same engine can cause the loss of the capability to shut down the engine in the event of an engine fire. You may obtain further information by examining the MCAI in the AD docket.

Relevant Service Information

Shorts has issued the following service bulletins. The actions described in this service information are intended to correct the unsafe condition identified in the MCAI.

- Shorts Alert Service Bulletin SD3 Sherpa-76-A02, Revision 1, dated January 24, 2007.
- Shorts Alert Service Bulletin SD330-76-A09, Revision 1, dated January 24, 2007.
- Shorts Alert Service Bulletin SD360 Sherpa-76-A03, Revision 1, dated January 24, 2007.
- Shorts Alert Service Bulletin SD360-76-A12, Revision 1, dated January 24, 2007.

FAA's Determination and Requirements of This AD

This product has been approved by the aviation authority of another country, and is approved for operation in the United States. Pursuant to our bilateral agreement with the State of Design Authority, we have been notified of the unsafe condition described in the MCAI and service information referenced above. We are issuing this AD because we evaluated all pertinent information and determined the unsafe condition exists and is likely to exist or develop on other products of the same type design.

Differences Between the AD and the MCAI or Service Information

We have reviewed the MCAI and related service information and, in general, agree with their substance. But we might have found it necessary to use different words from those in the MCAI to ensure the AD is clear for U.S. operators and is enforceable. In making these changes, we do not intend to differ substantively from the information provided in the MCAI and related service information.

We might also have required different actions in this AD from those in the MCAI in order to follow FAA policies. Any such differences are highlighted in a NOTE within the AD.

FAA's Determination of the Effective Date

An unsafe condition exists that requires the immediate adoption of this AD. The FAA has found that the risk to the flying public justifies waiving notice and comment prior to adoption of this rule because of the number of control cable assemblies that were deemed not to be airworthy when evaluated against specific inspection criteria. Therefore, we determined that notice and opportunity for public comment before issuing this AD are impracticable and that good cause exists for making this amendment effective in fewer than 30 days.

Comments Invited

This AD is a final rule that involves requirements affecting flight safety, and we did not precede it by notice and opportunity for public comment. We invite you to send any written relevant data, views, or arguments about this AD. Send your comments to an address listed under the **ADDRESSES** section. Include "Docket No. FAA-2007-27866; Directorate Identifier 2007-NM-055-AD" at the beginning of your comments. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of this AD. We will consider all comments received by the closing date and may amend this AD because of those comments.

We will post all comments we receive, without change, to <http://dms.dot.gov>, including any personal information you provide. We will also post a report summarizing each substantive verbal contact we receive about this AD.

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. "Subtitle VII: Aviation Programs," describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in "Subtitle VII, Part A, Subpart III, Section 44701: General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition

that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

We determined that this AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify this AD:

1. Is not a "significant regulatory action" under Executive Order 12866;
2. Is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and
3. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared a regulatory evaluation of the estimated costs to comply with this AD and placed it in the AD docket.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

■ Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

■ 2. The FAA amends § 39.13 by adding the following new AD:

2007-08-09 Short Brothers PLC:
Amendment 39-15027. Docket No. FAA-2007-27866; Directorate Identifier 2007-NM-055-AD.

Effective Date

(a) This airworthiness directive (AD) becomes effective May 8, 2007.

Affected ADs

(b) None.

Applicability

(c) This AD applies to all Shorts Model SD3-60 SHERPA, SD3-SHERPA, SD3-30, and SD3-60 airplanes, certificated in any category.

Subject

(d) Engine controls.

Reason

(e) The mandatory continued airworthiness information (MCAI) states:

Following the identification of a failed propeller RPM (revolutions per minute) cable end fitting and an LP (low pressure) fuel lever cable end fitting on an SD3 aircraft, several subsequent occurrences of control cable end fittings (type MS21260) with signs of pitting corrosion or cracking have been reported to Bombardier Shorts. All reported instances being identified during ground maintenance inspections on the SD3 fleet. Bombardier Shorts have performed examinations on the failed cable end fittings and established the root cause of failure as stress corrosion cracking of the SAE-AISI 303 stainless steel material they are manufactured from, initiated by pitting corrosion on the surface. The root cause of the stress corrosion is sustained tensile stress in a corrosive (warm, humid and salty) atmosphere.

An analysis of the cable operated control systems installed on the SD3 aircraft types that use MS 21260 type end fittings has identified a number of potentially unsafe conditions due to a combination of failures that may result from this common mode cause.

The failure of certain control cables could result in the loss of certain critical systems. For example, the loss of the low pressure (LP) fuel control cable in combination with a single failure of a fuel condition control cable on the same engine can cause the loss of the capability to shut down the engine in the event of an engine fire.

Actions and Compliance

(f) Unless already done, do the following actions.

(1) Within 30 days after the effective date of this AD, inspect the affected cable assembly end fittings in accordance with the applicable service bulletin specified in Table 1 of this AD.

(2) If no pitting/corrosion or cracking is found, within 12 months after the inspection required by paragraph (f)(1) of this AD, and thereafter at intervals not to exceed 12 months, repeat the inspection of the cable assembly end fittings in accordance with the applicable service bulletin specified in Table 1 of this AD. Replacing the cable assembly with a new cable assembly in accordance with the applicable service bulletin terminates the repetitive inspection intervals of this paragraph for the replaced cable assembly.

(3) When pitting/corrosion or cracking is found during any inspection required by this AD, before further flight, replace the affected cable assembly with a new cable assembly in accordance with the applicable service bulletin specified in Table 1 of this AD.

(4) After any replacement done in accordance with paragraph (f)(2) or (f)(3) of this AD, repeat the inspection required by paragraph (f)(1) of this AD for the replaced

cable assembly at intervals not to exceed 180 months.

(5) Do the actions in paragraphs (f)(1), (f)(2), (f)(3), and (f)(4) of this AD in accordance with the Accomplishment

Instructions of the applicable Shorts Alert Service Bulletin listed in Table 1 of this AD.

TABLE 1.—SERVICE BULLETINS FOR APPLICABLE ACTIONS

Shorts Alert Service Bulletin	Revision level	Date
SD3 Sherpa-76-A02	1	January 24, 2007.
SD330-76-A09	1	January 24, 2007.
SD360 Sherpa-76-A03	1	January 24, 2007.
SD360-76-A12	1	January 24, 2007.

FAA AD Differences

Note: This AD differs from the MCAI and/or service information as follows: No differences.

Other FAA AD Provisions

(g) The following provisions also apply to this AD:

(1) *Alternative Methods of Compliance (AMOCs):* The Manager, International Branch, ANM-116, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. Send information to ATTN: Todd Thompson, Aerospace Engineer, International Branch,

ANM-116, FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98057-3356; telephone (425) 227-1175; fax (425) 227-1149. Before using any AMOC approved in accordance with § 39.19 on any airplane to which the AMOC applies, notify the appropriate principal inspector in the FAA Flight Standards Certificate Holding District Office.

(2) *Airworthy Product:* For any requirement in this AD to obtain corrective actions from a manufacturer or other source, use these actions if they are FAA-approved. Corrective actions are considered FAA-approved if they are approved by the State of Design Authority (or their delegated agent). You are required

to assure the product is airworthy before it is returned to service.

(3) *Reporting Requirements:* For any reporting requirement in this AD, under the provisions of the Paperwork Reduction Act, the Office of Management and Budget (OMB) has approved the information collection requirements and has assigned OMB Control Number 2120-0056.

Related Information

(h) Refer to MCAI European Aviation Safety Agency (EASA) Emergency Airworthiness Directive 2007-0039-E, dated February 16, 2007, and the Shorts service information listed in Table 2 of this AD.

TABLE 2.—RELATED INFORMATION

Shorts Alert Service Bulletin	Revision level	Date
SD3 Sherpa-76-A02	1	January 24, 2007.
SD330-76-A09	1	January 24, 2007.
SD360 Sherpa-76-A03	1	January 24, 2007.
SD360-76-A12	1	January 24, 2007.

Material Incorporated by Reference

(i) You must use the service information specified in Table 3 of this AD to do the

actions required by this AD, unless the AD specifies otherwise.

TABLE 3.—MATERIAL INCORPORATED BY REFERENCE

Shorts Alert Service Bulletin	Page No.	Revision level	Date
SD3 Sherpa-76-A02	1, 6	1	January 24, 2007.
	2-5, 7-16	Original	January 10, 2007.
SD330-76-A09	1, 6	1	January 24, 2007.
	2-5, 7-19	Original	January 10, 2007.
SD360 Sherpa-76-A03	1, 6	1	January 24, 2007.
	2-5, 7-16	Original	January 10, 2007.
SD360-76-A12	1, 6	1	January 24, 2007.
	2-5, 7-16	Original	January 10, 2007.

(1) The Director of the Federal Register approved the incorporation by reference of this service information under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) For service information identified in this AD, contact Short Brothers, Airworthiness & Engineering Quality, P.O. Box 241, Airport Road, Belfast BT3 9DZ, Northern Ireland.

(3) You may review copies at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call (202) 741-6030, or go to: <http://www.archives.gov/federal-register/cfr/ibr-locations.html>.

Issued in Renton, Washington, on April 6, 2007.

Ali Bahrami,

*Manager, Transport Airplane Directorate,
Aircraft Certification Service.*

[FR Doc. E7-7118 Filed 4-20-07; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Office of the Secretary

14 CFR Parts 204 and 399

[Docket No. OST-2003-15759]

RIN 2105-AD25

Review of Data Filed by Certificated or Commuter Air Carriers To Support Continuing Fitness Determinations Involving Citizenship Issues

AGENCY: Office of the Secretary, DOT.

ACTION: Final rule.

SUMMARY: The Department is adopting its proposed editorial changes to its rules on Data to Support Fitness Determinations, 14 CFR part 204, and has determined to maintain its existing procedures for conducting reviews of the continuing fitness of air carriers. These actions complete this rulemaking. The Department had earlier withdrawn a proposal made in this rulemaking to modify the Department's standards for determining whether carriers remain under the actual control of U.S. citizens.

EFFECTIVE DATE: The rule is effective May 23, 2007.

FOR FURTHER INFORMATION CONTACT: William M. Bertram, Chief, Air Carrier Fitness Division (X-56), Office of Aviation Analysis, U.S. Department of Transportation, 400 7th Street, SW., Washington, DC 20590; (202) 366-9721.

SUPPLEMENTARY INFORMATION:

Introduction

By statute, only citizens of the United States may obtain and hold certificate

authority under 49 U.S.C. 41102 or 41103 authorizing them to provide air transportation within the United States or operate as a U.S. air carrier on international routes. The statutory citizenship requirements require that at least 75 percent of the voting interest of a U.S. air carrier be owned and controlled by U.S. citizens, that the president and two-thirds of the board of directors and managing officers be U.S. citizens, and that U.S. carriers be subject to the actual control of U.S. citizens. 49 U.S.C. 40102(a)(15). In this proceeding, we invited public comment on three matters related to our consideration of citizenship issues: (i) We proposed technical changes to our rules governing citizenship and fitness determinations, 14 CFR part 204; (ii) we considered whether we should modify our procedures for reviewing whether a carrier is complying with the continuing citizenship requirement; and (iii) we proposed to modify the standards used for determining whether a carrier is actually controlled by U.S. citizens. We have withdrawn the proposal to modify our standards on actual control. 71 FR 71106 (December 8, 2006). In this final rule, we are resolving the other two matters. We are adopting the proposed technical changes to part 204, and we explain why we have decided to continue following our procedural practices in continuing fitness cases.

Background

We examine carrier citizenship primarily in two situations. First, when a firm applies for authority to operate as a U.S. carrier, we conduct an initial fitness review, which necessarily includes a review of the carrier's citizenship. We conduct initial fitness reviews through docketed proceedings, where a public record of the pleadings is maintained; we publish all Department decisions in the case; and we give interested persons an opportunity to comment on the application. Second, we conduct a continuing fitness review if an existing carrier undergoes a substantial change in ownership, operations, or management. We usually conduct continuing fitness investigations without a public proceeding and therefore do not create a docket containing record material, publish a final decision, or provide an opportunity for public comment. In some continuing fitness cases, we may decide to use more formal public procedures. See 71 FR 26426-26427.

Rulemaking Notices

We issued a Notice of Proposed Rulemaking (NPRM) that proposed to

update our interpretation of actual control and to continue using our informal procedures in most continuing fitness reviews. 70 FR 67389 (November 7, 2005). We also proposed changes to part 204 to correct minor typographical errors, update statutory references, and clarify some language. 70 FR 67395. We thereafter issued a Supplemental Notice of Proposed Rulemaking (SNPRM) to address the comments made on the NPRM, and to propose additional refinements to our proposed modification of our actual control standard. 71 FR 26425 (May 5, 2006). We again proposed to continue using our informal procedures in most continuing fitness reviews.

In the NPRM and SNPRM, we stated that we had tentatively determined to continue using the same informal procedures for continuing fitness reviews that we have always used. 71 FR 26436; 70 FR 67392. We believed that significant potential harm could result if we made all substantial foreign investment cases subject to public notice and comment, and that using public proceedings in all significant cases appeared to be unnecessary for the protection of interested persons. We stated that we would have the option of beginning a public proceeding in any case if we found that doing so would be useful. 71 FR 26436.

Comments

The comments on the NPRM and SNPRM focused on our proposed change to our standard for defining when U.S. citizens had actual control of a U.S. carrier. None of the commenters opposed our proposed changes to part 204. While several commenters discussed the procedural issues in their responses to our NPRM, only Continental commented in any detail on our SNPRM's proposed decision to continue using informal procedures in most continuing fitness reviews. Continental asserted that the informal procedures enable us to resolve citizenship matters after negotiating only with the carrier and its foreign investors, not with other persons affected by the transaction. Continental Comments at 9.

Decision on Procedures

We have determined to continue following our existing procedures for continuing fitness reviews for the reasons stated in our earlier notices. We can, of course, always choose to use public procedures in any continuing fitness review, and interested persons have the right to ask us to do so. See 71 FR 26436.