

NPRM, the regulatory evaluation, any comments received, and other information. The street address for the Docket Office (telephone (800) 647-5227) is in the **ADDRESSES** section. Comments will be available in the AD docket shortly after receipt.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. The FAA amends § 39.13 by adding the following new AD:

2007-06-11 EADS SOCATA Model TBM 700 Airplanes: Amendment 39-14992; Docket No. FAA-2006-26166; Directorate Identifier 2006-CE-58-AD.

Effective Date

(a) This airworthiness directive (AD) becomes effective April 20, 2007.

Affected ADs

(b) None.

Applicability

(c) This AD applies to SOCATA TBM 700 airplanes, serial numbers 1 through 308, plus the serial number 310, certificated in any category.

Note 1: This AD does not apply to airplanes in which both modifications No. MOD70-127-55 and MOD70-129-53 have been factory installed.

Reason

(d) The mandatory continuing airworthiness information (MCAI) states that: Cracks on a vertical stabilizer attachment fitting due to corrosion have been found on an aircraft in service.

Actions and Compliance

(e) Unless already done, do the following actions.

(1) Within the next 600 hours time-in-service (TIS) or the next 12 months, whichever occurs first, after the effective date of this AD, inspect the vertical stabilizer attachment fittings and bolts for cracks or corrosion, and, if necessary, repair or replace the damaged part and then apply a corrosion protection reinforcement, following EADS SOCATA Service Bulletin SB 70-104, Amendment 1, dated August 2004 or EADS SOCATA TBM Aircraft Mandatory Service Bulletin SB 70-104, Amendment 2, dated January 2007.

(2) Repeat the actions of paragraph (e)(1) every 1,200 hours TIS or every 24 months, whichever occurs first, following EADS SOCATA Service Bulletin SB 70-104, Amendment 1, dated August 2004 or EADS SOCATA Service Bulletin SB 70-104, Amendment 2, dated January 2007.

FAA AD Differences

Note 2: This AD differs from the MCAI and/or service information as follows: This AD permits Amendment 2 of the SB to be used.

Other FAA AD Provisions

(f) The following provisions also apply to this AD:

(1) *Alternative Methods of Compliance (AMOCs):* The Manager, Standards Staff, FAA, ATTN: Albert J. Mercado, Aerospace Engineer, FAA, Small Airplane Directorate, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone: (816) 329-4119; fax: (816) 329-4090, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. Before using any approved AMOC on any airplane to which the AMOC applies, notify your appropriate principal inspector (PI) in the FAA Flight Standards District Office (FSDO), or lacking a PI, your local FSDO.

(2) *Airworthy Product:* For any requirement in this AD to obtain corrective actions from a manufacturer or other source, use these actions if they are FAA-approved. Corrective actions are considered FAA-approved if they are approved by the State of Design Authority (or their delegated agent). You are required to assure the product is airworthy before it is returned to service.

(3) *Reporting Requirements:* For any reporting requirement in this AD, under the provisions of the Paperwork Reduction Act (44 U.S.C. 3501 et seq.), the Office of Management and Budget (OMB) has approved the information collection requirements and has assigned OMB Control Number 2120-0056.

Related Information

(g) Refer to Direction générale de l'aviation civile (DGAC) AD No F-2003-366 R1, dated November 24, 2004; EADS SOCATA TBM Aircraft Mandatory Service Bulletin SB 70-104, Amendment 1, dated August 2004; and EADS SOCATA TBM Aircraft Mandatory Service Bulletin SB 70-104, Amendment 2, dated January 2007 for related information.

Material Incorporated by Reference

(h) You must use EADS SOCATA TBM Aircraft Mandatory Service Bulletin No. SB 70-104, Amendment 1, dated August 2004, or EADS SOCATA TBM Aircraft Mandatory Service Bulletin No. SB 70-104, Amendment 2, dated January 2007 to do the actions required by this AD, unless the AD specifies otherwise.

(1) The Director of the Federal Register approved the incorporation by reference of this service information under 5 U.S.C. 552(a) and 1 CFR part 51.

(3) For service information identified in this AD, contact EADS SOCATA, Direction des Services, 65921 Tarbes Cedex 9, France.

(4) You may review copies at the FAA, Central Region, Office of the Regional

Counsel, 901 Locust, Room 506, Kansas City, Missouri 64106; or at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call 202-741-6030, or go to: <http://www.archives.gov/federal-register/cfr/ibr-locations.html>.

Issued in Kansas City, Missouri, on March 7, 2007.

David R. Showers,

Acting Manager, Small Airplane Directorate, Aircraft Certification Service.

[FR Doc. E7-4724 Filed 3-15-07; 8:45 am]

BILLING CODE 4910-13-P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA-2006-26180; Directorate Identifier 2006-CE-59-AD; Amendment 39-14995; AD 2007-06-14]

RIN 2120-AA64

Airworthiness Directives; EADS SOCATA Model TBM 700 Airplanes

AGENCY: Federal Aviation Administration (FAA), Department of Transportation (DOT).

ACTION: Final rule.

SUMMARY: We are adopting a new airworthiness directive (AD) for the products listed above. This AD results from mandatory continuing airworthiness information (MCAI) issued by an aviation authority of another country to identify and correct an unsafe condition on an aviation product. The MCAI describes the unsafe condition as the discovery of propeller control cables with a defective crimping. Two cable ends were found uncrimped at the factory after an engine run-up test, and one cable end was also found uncrimped on the first 100-hour aircraft maintenance check. We are issuing this AD to require actions to correct the unsafe condition on these products.

DATES: This AD becomes effective April 20, 2007.

The Director of the Federal Register approved the incorporation by reference of certain publications listed in this AD as of April 20, 2007.

ADDRESSES: You may examine the AD docket on the Internet at <http://dms.dot.gov> or in person at the Docket Management Facility, U.S. Department of Transportation, 400 Seventh Street, SW., Nassif Building, Room PL-401, Washington, DC.

FOR FURTHER INFORMATION CONTACT: Albert J. Mercado, Aerospace Engineer,

FAA, Small Airplane Directorate, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone: (816) 329-4119; fax: (816) 329-4090.

SUPPLEMENTARY INFORMATION:

Streamlined Issuance of AD

The FAA is implementing a new process for streamlining the issuance of ADs related to MCAI. The streamlined process will allow us to adopt MCAI safety requirements in a more efficient manner and will reduce safety risks to the public. This process continues to follow all FAA AD issuance processes to meet legal, economic, Administrative Procedure Act, and **Federal Register** requirements. We also continue to meet our technical decision-making responsibilities to identify and correct unsafe conditions on U.S.-certificated products.

This AD references the MCAI and related service information that we considered in forming the engineering basis to correct the unsafe condition. The AD contains text copied from the MCAI and for this reason might not follow our plain language principles.

Discussion

We issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 to include an AD that would apply to the specified products. That NPRM was published in the **Federal Register** on November 28, 2006 (71 FR 68764). That NPRM proposed to require you to check the batch number and/or replace the control cables.

Comments

We gave the public the opportunity to participate in developing this AD. We have considered the comment received.

EADS SOCATA states the costs of the required parts and the necessary work-hours to replace the control cables are under warranty coverage. Therefore, the cost of work-hours to operators/owners is only the one work-hour for inspection.

The FAA will revise the cost of compliance to reflect only the above costs of work-hours.

Conclusion

We reviewed the available data, including the comment received, and determined that air safety and the public interest require adopting the AD as proposed.

Differences Between This AD and the MCAI or Service Information

We have reviewed the MCAI and related service information and, in general, agree with their substance. But we might have found it necessary to use

different words from those in the MCAI to ensure the AD is clear for U.S. operators and is enforceable in a U.S. court of law. In making these changes, we do not intend to differ substantively from the information provided in the MCAI and related service information.

We might also have required different actions in this AD from those in the MCAI in order to follow FAA policies. Any such differences are described in a separate paragraph of the AD. These requirements, if any, take precedence over the actions copied from the MCAI.

Costs of Compliance

We estimate that this AD will affect 20 products of U.S. registry. We also estimate that it will take about 1 work-hour per product to comply with this AD. The average labor rate is \$80 per work-hour. Where the service information lists required parts costs that are covered under warranty, we have assumed that there will be no charge for these parts. As we do not control warranty coverage for affected parties, some parties may incur costs higher than estimated here. Based on these figures, we estimate the cost of this AD to the U.S. operators to be \$1,600, or \$80 per product.

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. "Subtitle VII: Aviation Programs," describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in "Subtitle VII, Part A, Subpart III, Section 44701: General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

We determined that this AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify this AD:

- (1) Is not a "significant regulatory action" under Executive Order 12866;
- (2) Is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and
- (3) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared a regulatory evaluation of the estimated costs to comply with this AD and placed it in the AD Docket.

Examining the AD Docket

You may examine the AD docket on the Internet at <http://dms.dot.gov>; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains the NPRM, the regulatory evaluation, any comments received, and other information. The street address for the Docket Office (telephone (800) 647-5227) is in the **ADDRESSES** section. Comments will be available in the AD docket shortly after receipt.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

■ Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

■ 2. The FAA amends § 39.13 by adding the following new AD:

2007-06-14 EADS SOCATA: Amendment 39-14995; Docket No. FAA-2006-26180; Directorate Identifier 2006-CE-59-AD.

Effective Date

(a) This airworthiness directive (AD) becomes effective April 20, 2007.

Affected ADs

(b) None.

Applicability

(c) This AD applies to Model TBM 700 airplanes, serial numbers 285 through 304 and 307, certificated in any category.

Reason

(d) The mandatory continuing airworthiness information (MCAI) states the

discovery of propeller control cables with a defective crimping. Two cable ends were found uncrimped at the factory after an engine run-up test, and one cable end was also found uncrimped on the first 100-hour time-in-service aircraft maintenance check. If not corrected, an incorrect crimping of the propeller control lever cable could generate a decrease of the propeller revolutions per minute which could result in loss of power.

Actions and Compliance

(e) Unless already done, within the next 50 hours time-in-service (TIS) after April 20, 2007 (the effective of this AD), inspect for the batch number identification and replace defective control cables as necessary in accordance with the paragraph B. of the "ACCOMPLISHMENT INSTRUCTIONS" of EADS SOCATA TBM Aircraft Mandatory Alert Service Bulletin SB 70-123, dated October 2004.

FAA AD Differences

Note: This AD differs from the MCAI, the French Direction générale de l'aviation civile (DGAC) AD No. F-2004-175, dated November 10, 2004, as follows:

(1) The requirement of paragraph 3.1 of DGAC AD No. F-2004-175, dated November 10, 2004, was based on the urgency in November 2004. However, in 2007, this action is not necessary.

(2) For the requirement of paragraph 3.2 of DGAC AD No. F-2004-175, dated November 10, 2004, the FAA has determined the seriousness of the condition does not warrant a compliance time of 25 hours TIS; we require instead a compliance time of 50 hours TIS.

Other FAA AD Provisions

(f) The following provisions also apply to this AD:

(1) *Alternative Methods of Compliance (AMOCs):* The Manager, Standards Staff, FAA, ATTN: Albert J. Mercado, Aerospace Safety Engineer, FAA, Small Airplane Directorate, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone: (816) 329-4119; fax: (816) 329-4090, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. Before using any approved AMOC on any airplane to which the AMOC applies, notify your appropriate principal inspector (PI) in the FAA Flight Standards District Office (FSDO), or lacking a PI, your local FSDO.

(2) *Airworthy Product:* For any requirement in this AD to obtain corrective actions from a manufacturer or other source, use these actions if they are FAA-approved. Corrective actions are considered FAA-approved if they are approved by the State of Design Authority (or their delegated agent). You are required to assure the product is airworthy before it is returned to service.

(3) *Reporting Requirements:* For any reporting requirement in this AD, under the provisions of the Paperwork Reduction Act (44 U.S.C. 3501 et seq.), the Office of Management and Budget (OMB) has approved the information collection requirements and has assigned OMB Control Number 2120-0056.

Related Information

(g) Refer to Direction Générale de l'Aviation Civile Airworthiness Directive No. F-2004-175, dated November 10, 2004 (the MCAI); and EADS SOCATA TBM Aircraft Mandatory Alert Service Bulletin SB 70-123, dated October 2004, for related information.

Material Incorporated by Reference

(h) You must use EADS SOCATA TBM Aircraft Mandatory Alert Service Bulletin SB 70-123, dated October 2004, to do the actions required by this AD, unless the AD specifies otherwise.

(1) The Director of the Federal Register approved the incorporation by reference of this service information under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) For service information identified in this AD, contact EADS SOCATA, Direction des Services, 65921 Tarbes Cedex 9, France; telephone: 33 (0)5 62.41.73.00; fax: 33 (0)5 62.41.76.54; or SOCATA AIRCRAFT, INC., North Perry Airport, 7501 Airport Road, Pembroke Pines, Florida 33023; telephone: (954) 893-1400; fax (954) 964-4141.

(3) You may review copies at the FAA, Central Region, Office of the Regional Counsel, 901 Locust, Room 506, Kansas City, Missouri 64106; or at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call 202-741-6030, or go to: <http://www.archives.gov/federal-register/cfr/ibr-locations.html>.

Issued in Kansas City, Missouri, on March 9, 2007.

David R. Showers,

Acting Manager, Small Airplane Directorate, Aircraft Certification Service.

[FR Doc. E7-4736 Filed 3-15-07; 8:45 am]

BILLING CODE 4910-13-P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA-2005-22036; Directorate Identifier 2005-NM-009-AD; Amendment 39-14994; AD 2007-06-13]

RIN 2120-AA64

Airworthiness Directives; Airbus Model A300 B4-600, B4-600R, and F4-600R Series Airplanes, and Model C4-605R Variant F Airplanes (Collectively Called A300-600 Series Airplanes); and Model A310 Airplanes; Equipped With General Electric CF6-80A3 or CF6-80C2 Engines

AGENCY: Federal Aviation Administration (FAA), Department of Transportation (DOT).

ACTION: Final rule.

SUMMARY: The FAA is adopting a new airworthiness directive (AD) for certain Airbus airplane models, as specified above. This AD requires installing

electro-pneumatic locking bar devices (TRAS lock systems) in the engine nacelles, installing a dedicated and shielded electrical circuit that is segregated from the existing thrust reverser control system, and performing related investigative/corrective actions if necessary. This AD results from the manufacturer's reassessment of the thrust reverser systems in the Airbus airplane models specified above, which showed that the thrust reverser could inadvertently deploy in flight under certain conditions. We are issuing this AD to prevent inadvertent deployment of thrust reversers in flight, which could result in reduced controllability of the airplane.

DATES: This AD becomes effective April 20, 2007.

The Director of the Federal Register approved the incorporation by reference of certain publications listed in the AD as of April 20, 2007.

ADDRESSES: You may examine the AD docket on the Internet at <http://dms.dot.gov> or in person at the Docket Management Facility, U.S. Department of Transportation, 400 Seventh Street, SW., Nassif Building, Room PL-401, Washington, DC.

Contact Airbus, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France, for service information identified in this AD.

FOR FURTHER INFORMATION CONTACT: Tim Backman, Aerospace Engineer, International Branch, ANM-116, FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98057-3356; telephone (425) 227-2797; fax (425) 227-1149.

SUPPLEMENTARY INFORMATION:

Examining the Docket

You may examine the airworthiness directive (AD) docket on the Internet at <http://dms.dot.gov> or in person at the Docket Management Facility office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The Docket Management Facility office (telephone (800) 647-5227) is located on the plaza level of the Nassif Building at the street address stated in the **ADDRESSES** section.

Discussion

The FAA issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 to include an AD that would apply to certain Airbus Model A300 B2 and B4 series airplanes; Model A300 B4-600, B4-600R, and F4-600R series airplanes, and Model C4-605R Variant F airplanes (collectively called A300-600 series airplanes); and Model A310 series airplanes; equipped with General