

tree nuts, and less than 7 percent of the world's hazelnut production.

Last season, 85 percent of the domestically produced hazelnut kernels were marketed in the domestic market and 15 percent were exported. Domestically produced kernels generally command a higher price in the domestic market than imported kernels. The industry is continuing its efforts to develop and expand other markets with emphasis on the domestic kernel market. Small business entities, both producers and handlers, benefit from the expansion efforts resulting from this program.

Inshell hazelnuts produced under the order compete well in export markets because of their high quality. Based on Board statistics, Europe has historically been the primary export market for U.S. produced inshell hazelnuts. Shipments have also been relatively consistent, not varying much from the 10 year average of 4,958 tons. Recent years, though, have seen a significant increase in export destinations. Last season, inshell shipments to Europe totaled 4,622 tons, representing just 38 percent of exports, with the largest share going to Germany. Inshell shipments to Southwest Pacific countries, and Hong Kong in particular, have increased dramatically in the past few years, rising to 50 percent of total exports of 12,042 tons for the 2005–2006 marketing year. The industry continues to pursue export opportunities.

There are some reporting, recordkeeping, and other compliance requirements under the order. The reporting and recordkeeping burdens are necessary for compliance purposes and for developing statistical data for maintenance of the program. The information collection requirements have been previously approved by the Office of Management and Budget under OMB No. 0581–0178. The forms require information which is readily available from handler records and which can be provided without data processing equipment or trained statistical staff. As with all Federal marketing order programs, reports and forms are periodically reviewed to reduce information requirements and duplication by industry and public sector agencies. This rule does not change those requirements.

The AMS is committed to complying with the E-Government Act, to promote the use of the Internet and other information technologies to provide increased opportunities for citizen access to Government information and services, and for other purposes.

In addition, USDA has not identified any relevant Federal rules that

duplicate, overlap, or conflict with this rule.

Further, the Board's meetings were widely publicized throughout the hazelnut industry and all interested persons were invited to attend the meetings and participate in Board deliberations. Like all Board meetings, those held on August 24 and November 15, 2006, were public meetings and all entities, both large and small, were able to express their views on this issue. Finally, interested persons are invited to submit information on the regulatory and informational impacts of this action on small businesses.

A small business guide on complying with fruit, vegetable, and specialty crop marketing agreements and orders may be viewed at: <http://www.ams.usda.gov/fv/moab.html>. Any questions about the compliance guide should be sent to Jay Guerber at the previously mentioned address in the **FOR FURTHER INFORMATION CONTACT** section.

This rule invites comments on the establishment of final free and restricted percentages for the 2006–2007 marketing year under the hazelnut marketing order. Any comments received will be considered prior to finalization of this rule.

After consideration of all relevant material presented, including the Board's recommendation, and other information, it is found that this interim final rule, as hereinafter set forth, will tend to effectuate the declared policy of the Act.

Pursuant to 5 U.S.C. 553, it is also found and determined upon good cause that it is impracticable, unnecessary, and contrary to the public interest to give preliminary notice prior to putting this rule into effect and that good cause exists for not postponing the effective date of this action until 30 days after publication in the **Federal Register** because: (1) The 2006–2007 marketing year began July 1, 2006, and the percentages established herein apply to all merchantable hazelnuts handled from the beginning of the crop year; (2) the percentages make the full trade demand available so handlers can take advantage of inshell marketing opportunities; (3) handlers are aware of this rule, which was recommended at an open Board meeting, and need no additional time to comply with this rule; and (4) interested persons are provided a 60-day comment period in which to respond, and all comments timely received will be considered prior to finalization of this action.

List of Subjects in 7 CFR Part 982

Filberts, Hazelnuts, Marketing agreements, Nuts, Reporting and recordkeeping requirements.

■ For the reasons set forth in the preamble, 7 CFR part 982 is amended as follows:

PART 982—HAZELNUTS GROWN IN OREGON AND WASHINGTON

■ 1. The authority citation for 7 CFR part 982 continues to read as follows:

Authority: 7 U.S.C. 601–674.

■ 2. A new subpart and § 982.254 are added to read as follows:

Subpart—Free and Restricted Percentages

§ 982.254 Free and restricted percentages—2006–2007 marketing year.

The final free and restricted percentages for merchantable hazelnuts for the 2006–2007 marketing year shall be 8.2840 percent and 91.7160 percent, respectively.

Dated: January 16, 2007.

Lloyd C. Day,

Administrator, Agricultural Marketing Service.

[FR Doc. E7–763 Filed 1–19–07; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA–2005–19559; Directorate Identifier 2004–NE–03–AD; Amendment 39–14892; AD 2007–02–05]

RIN 2120–AA64

Airworthiness Directives; Rolls-Royce plc RB211 Trent 700 Series Turbofan Engines

AGENCY: Federal Aviation Administration (FAA), Department of Transportation (DOT).

ACTION: Final rule; request for comments.

SUMMARY: The FAA is superseding an existing airworthiness directive (AD) for Rolls-Royce plc (RR) RB211 Trent 700 series turbofan engines. That AD currently requires initial and repetitive borescope inspections of the high pressure-and-intermediate pressure (HP-IP) turbine internal and external oil vent tubes for coking and carbon buildup, and cleaning or replacing the vent tubes if necessary. This AD requires the same actions but uses more stringent tube

replacement criteria than the previous AD. This AD results from a recent incident where an RB211 Trent 700 series turbofan engine had an oil vent tube rupture as a result of blockage, leading to significant loss of engine oil. The incident indicates that further measures are necessary to control carbon buildup in the oil vent tubes. We are issuing this AD to prevent internal oil fires due to coking and carbon buildup, that could cause uncontained engine failure and damage to the airplane.

DATES: Effective February 6, 2007. The Director of the Federal Register approved the incorporation by reference of certain publications listed in the regulations as of February 6, 2007.

We must receive any comments on this AD by March 23, 2007.

ADDRESSES: Use one of the following addresses to comment on this proposed AD.

- *DOT Docket Web site:* Go to <http://dms.dot.gov> and follow the instructions for sending your comments electronically.

- *Government-wide rulemaking Web site:* Go to <http://www.regulations.gov> and follow the instructions for sending your comments electronically.

- *Mail:* Docket Management Facility; U.S. Department of Transportation, 400 Seventh Street, SW., Nassif Building, Room PL-401, Washington, DC 20590-0001.

- *Fax:* (202) 493-2251.

- *Hand Delivery:* Room PL-401 on the plaza level of the Nassif Building, 400 Seventh Street, SW., Washington, DC, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

Contact Rolls-Royce plc, PO Box 31, Derby, England; telephone: 011-44-1332-249428; fax: 011-44-1332-249223, for the service information identified in this AD.

FOR FURTHER INFORMATION CONTACT: Christopher Spinney, Aerospace Engineer, Engine Certification Office, FAA, Engine and Propeller Directorate, 12 New England Executive Park, Burlington, MA 01803-5299; telephone (781) 238-7175; fax (781) 238-7199.

SUPPLEMENTARY INFORMATION: On November 1, 2004, we issued AD 2004-23-03, Amendment 39-13858 (69 FR 64653, November 8, 2004). That AD requires initial and repetitive borescope inspections of the HP-IP turbine internal and external oil vent tubes for coking and carbon buildup, and cleaning or replacing the vent tubes if necessary. That AD was the result of a report of an RB211 Trent 700 series engine experiencing a disk shaft separation, overspeed of the IP turbine

rotor, and multiple blade release of IP turbine blades. The findings suggested these events resulted from an internal oil fire in the HP-IP turbine oil vent tubes due to coking and carbon buildup. This fire led to a second fire in the internal air cavity below the IP turbine disk drive shaft. That condition, if not corrected, could result in uncontained engine failure and damage to the airplane.

Actions Since AD 2004-23-03 Was Issued

Since AD 2004-23-03 was issued, the European Aviation Safety Agency (EASA), which is the airworthiness authority for the European Union, notified us that an unsafe condition may exist on RB211 Trent 700 series turbofan engines. EASA advises that recently an oil vent tube ruptured as a result of blockage, leading to significant loss of engine oil, on an RB211 Trent 700 series turbofan engine. This incident indicates that further measures are necessary to control carbon buildup in the oil vent tubes.

Relevant Service Information

We have reviewed and approved the technical contents of RR Alert Service Bulletin (ASB) No. RB.211-72-AE302, Revision 3, dated September 20, 2006. That ASB describes procedures for borescope inspections, cleaning, and replacement if necessary of the internal and external oil vent tubes. For internal oil vent tubes to pass inspection, they must allow cleaning tool, number HU80298 to pass through them. AD 2004-23-03 was less stringent in that it allowed tubes that an 8 mm or 6 mm diameter borescope could pass through, back into service. EASA classified this ASB as mandatory and issued AD 2006-0355, dated December 4, 2006, in order to ensure the airworthiness of these RB211 Trent 700 series turbofan engines in Europe.

Bilateral Airworthiness Agreement

These engine models are manufactured in the United Kingdom and are type certificated for operation in the United States under the provisions of section 21.29 of the Federal Aviation Regulations (14 CFR 21.29) and the applicable bilateral airworthiness agreement. Under this bilateral airworthiness agreement, EASA kept the FAA informed of the situation described above. We have examined the findings of EASA, reviewed all available information, and determined that AD action is necessary for products of this type design that are certificated for operation in the United States.

FAA's Determination and Requirements of This AD

Although no airplanes that are registered in the United States use these RB211 Trent 700 series turbofan engines, the possibility exists that the engines could be used on airplanes that are registered in the United States in the future. The unsafe condition described previously is likely to exist or develop on other RB211 Trent 700 series turbofan engines of the same type design. This AD requires initial and repetitive borescope inspections of the HP-IP turbine internal and external oil vent tubes for coking and carbon buildup, and cleaning or replacing the vent tubes if necessary. We are issuing this AD to prevent internal oil fires due to coking and carbon buildup, that could cause uncontained engine failure and damage to the airplane. You must use the service information described previously to perform the actions required by this AD.

FAA's Determination of the Effective Date

Since there are currently no domestic operators of this engine model, notice and opportunity for public comment before issuing this AD are unnecessary. Therefore, a situation exists that allows the immediate adoption of this regulation.

Comments Invited

This AD is a final rule that involves requirements affecting flight safety and was not preceded by notice and an opportunity for public comment; however, we invite you to send us any written relevant data, views, or arguments regarding this AD. Send your comments to an address listed under **ADDRESSES**. Include "AD Docket No. FAA-2005-19559; Directorate Identifier 2004-NE-03-AD" in the subject line of your comments. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of the rule that might suggest a need to modify it.

We will post all comments we receive, without change, to <http://dms.dot.gov>, including any personal information you provide. We will also post a report summarizing each substantive verbal contact with FAA personnel concerning this AD. Using the search function of the DMS Web site, anyone can find and read the comments in any of our dockets, including the name of the individual who sent the comment (or signed the comment on behalf of an association, business, labor union, etc.). You may review the DOT's complete Privacy Act Statement in the

Federal Register published on April 11, 2000 (65 FR 19477-78) or you may visit <http://dms.dot.gov>.

Examining the AD Docket

You may examine the docket that contains the AD, any comments received, and any final disposition in person at the Docket Management Facility Docket Office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The Docket Office (telephone (800) 647-5227) is located on the plaza level of the Department of Transportation Nassif Building at the street address stated in **ADDRESSES**. Comments will be available in the AD docket shortly after the DMS receives them.

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, Section 106, describes the authority of the FAA Administrator. Subtitle VII, Aviation Programs, describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section 44701, "General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

We have determined that this AD will not have federalism implications under

Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that the regulation:

1. Is not a "significant regulatory action" under Executive Order 12866;
2. Is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and
3. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared a summary of the costs to comply with this AD and placed it in the AD Docket. You may get a copy of this summary by sending a request to us at the address listed under **ADDRESSES**.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

■ Under the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. The FAA amends § 39.13 by removing Amendment 39-13858 (69 FR 64653, November 8, 2004), and by

adding a new airworthiness directive, Amendment 39-14892, to read as follows:

2007-02-05 Rolls-Royce plc: Amendment 39-14892. Docket No. FAA-2005-19559; Directorate Identifier 2004-NE-03-AD.

Effective Date

(a) This airworthiness directive (AD) becomes effective February 6, 2007.

Affected ADs

(b) This AD supersedes AD 2004-23-03.

Applicability

(c) This AD applies to Rolls-Royce plc (RR) RB211 Trent 768-60, RB211 Trent 772-60, and RB211 Trent 772B-60 series turbofan engines. These engines are installed on, but not limited to, Airbus A330-243, -341, -342 and -343 series airplanes.

Unsafe Condition

(d) This AD results from a recent incident where an RB211 Trent 700 series turbofan engine had an oil vent tube rupture as a result of blockage, leading to significant loss of engine oil. The incident indicates that further measures are necessary to control carbon buildup in the oil vent tubes. We are issuing this AD to prevent internal oil fires due to coking and carbon buildup, that could cause uncontained engine failure and damage to the airplane.

Compliance

(e) You are responsible for having the actions required by this AD performed within the compliance times specified unless the actions have already been done.

Initial Inspections, Cleaning, and Replacements

(f) Using the schedule in Table 1 of this AD, borescope-inspect and clean as necessary, the high pressure-and-intermediate pressure (HP-IP) turbine internal oil vent tubes, external oil vent tubes, and bearing chamber.

TABLE 1.—INITIAL INSPECTION SCHEDULE

If the engine or the 05 Module:	Then initially inspect:
Has reached 10,000 hours time-since-new (TSN) or reached 2,500 cycles-since-new (CSN) on the effective date of this AD.	Within 3 months after the effective date of this AD.
Has fewer than 10,000 hours TSN or fewer than 2,500 CSN on the effective date of this AD.	Within 3 months after reaching 10,000 hours TSN or 2,500 CSN, whichever occurs first.
Is returned for a shop visit	Before returning to service.

(g) If after cleaning, there is still carbon in the vent tube that prevents cleaning tool, number HU80298, from passing through the tube, then replace the internal oil vent tube within 10 cycles-in-service (CIS).

(h) If after cleaning, there is still carbon of visible thickness in either of the two external oil vent tubes, then replace the external oil vent tube before further flight.

Repetitive Inspections, Cleaning, and Replacements

(i) Within 6,400 hours time-in-service since last inspection and cleaning, or within 1,600 cycles-since-last inspection and cleaning, or at the next engine shop visit, whichever occurs first, borescope-inspect the HP-IP turbine internal and external oil vent tubes and bearing chamber, and clean the oil vent tubes as necessary.

(j) If after cleaning there is still carbon in the internal oil vent tube that prevents cleaning tool, number HU80298, from passing through the tube, then replace the internal oil vent tube within 10 CIS.

(k) If after cleaning there is still carbon of visible thickness, in either of the two external oil vent tubes, then replace the external oil vent tube before further flight.

Inspection and Cleaning Procedures

(l) Use paragraphs 3.A. through 3.A.(4)(b) of the Accomplishment Instructions of Rolls-Royce plc Alert Service Bulletin No. RB.211-72-AE302, Revision 3, dated September 20, 2006, to do borescope inspections, and cleaning of the oil vent tubes and bearing chamber.

Alternative Methods of Compliance

(m) The Manager, Engine Certification Office, has the authority to approve alternative methods of compliance for this AD if requested using the procedures found in 14 CFR 39.19.

Material Incorporated by Reference

(n) You must use Rolls-Royce plc Alert Service Bulletin No. RB.211-72-AE302, Revision 3, dated September 20, 2006, to perform the inspections and cleaning required by this AD. The Director of the Federal Register approved the incorporation by reference of this service bulletin in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Contact Rolls-Royce plc, PO Box 31, Derby, England; telephone: 011-44-1332-249428; fax: 011-44-1332-249223, for a copy of this service information. You may review copies at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call 202-741-6030, or go to: <http://www.archives.gov/federal-register/cfr/ibr-locations.html>.

Related Information

(o) European Aviation Safety Agency airworthiness directive No. 2006-0355, dated December 4, 2006, also addresses the subject of this AD.

(p) Contact Christopher Spinney, Aerospace Engineer, Engine Certification Office, FAA, Engine & Propeller Directorate, 12 New England Executive Park, Burlington, MA 01803; telephone (781) 238-7175; fax (781) 238-7199, for more information about this AD.

Issued in Burlington, Massachusetts, on January 12, 2007.

Francis A. Favara,

Manager, Engine and Propeller Directorate, Aircraft Certification Service.

[FR Doc. E7-684 Filed 1-19-07; 8:45 am]

BILLING CODE 4910-13-P

DEPARTMENT OF TRANSPORTATION**Federal Aviation Administration****14 CFR Part 39**

[Docket No. FAA-2006-26236 Directorate Identifier 2006-CE-66-AD; Amendment 39-14891; AD 2007-02-04]

RIN 2120-AA64

Airworthiness Directives; SOCATA-Groupe Aerospatiale TB 20 and TB 21 Airplanes

AGENCY: Federal Aviation Administration (FAA), Department of Transportation (DOT).

ACTION: Final rule.

SUMMARY: We are adopting a new airworthiness directive (AD) for the products listed above. This AD results from mandatory continuing airworthiness information (MCAI) issued by an aviation authority of another country to identify and correct an unsafe condition on an aviation product. The MCAI describes the unsafe condition as reports of interference between the wing spar lower boom and the wheel fairing attaching screw. We are issuing this AD to require actions to correct the unsafe condition on these products.

DATES: This AD becomes effective February 26, 2007.

The Director of the Federal Register approved the incorporation by reference of certain publications listed in this AD as of February 26, 2007.

ADDRESSES: You may examine the AD docket on the Internet at <http://dms.dot.gov> or in person at the Docket Management Facility, U.S. Department of Transportation, 400 Seventh Street, SW., Nassif Building, Room PL-401, Washington, DC.

FOR FURTHER INFORMATION CONTACT: Doug Rudolph, Aerospace Engineer, FAA, Small Airplane Directorate, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone: (816) 329-4059; fax: (816) 329-4090.

SUPPLEMENTARY INFORMATION:**Streamlined Issuance of AD**

The FAA is implementing a new process for streamlining the issuance of ADs related to MCAI. The streamlined process will allow us to adopt MCAI safety requirements in a more efficient manner and will reduce safety risks to the public. This process continues to follow all FAA AD issuance processes to meet legal, economic, Administrative Procedure Act, and **Federal Register** requirements. We also continue to meet our technical decision-making responsibilities to identify and correct unsafe conditions on U.S.-certificated products.

This AD references the MCAI and related service information that we considered in forming the engineering basis to correct the unsafe condition. The AD contains text copied from the MCAI and for this reason might not follow our plain language principles.

Discussion

We issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 to include an AD that would apply to the specified products. That NPRM was published in the **Federal**

Register on November 22, 2006 (71 FR 67506). That NPRM proposed to correct an unsafe condition for the specified products. The MCAI states that there are reports of interference between the wing spar lower boom and the wheel fairing attaching screw causing an unsafe condition. The interference could, if left uncorrected, reduce the fatigue life of the wing spar with potentially catastrophic results.

Comments

We gave the public the opportunity to participate in developing this AD. We have considered the comment received.

Comment Issue: Cost of Compliance

EADS SOCATA states:

Application of SB10-148-57 does not require specific part. So, the cost is negligible. EADS SOCATA estimates that it would take 1 work-hour to inspect and displace the screw. If repair is necessary, the cost depends on the damage.

Our cost estimate included both the inspection and screw displacement costs as well as repair costs. We developed the repair cost estimate based on the information provided and assumed the worst case scenario if a repair was required. Since EADS SOCATA did not provide an estimate (work-hours or parts cost) if a repair is required and the FAA is required to provide this estimate to the public, we are keeping the language the same as the NPRM to account for worst case repair situations.

Conclusion

We reviewed the available data, including the comment received, and determined that air safety and the public interest require adopting the AD as proposed.

Differences Between This AD and the MCAI or Service Information

We have reviewed the MCAI and related service information and, in general, agree with their substance. But we might have found it necessary to use different words from those in the MCAI to ensure the AD is clear for U.S. operators and is enforceable. In making these changes, we do not intend to differ substantively from the information provided in the MCAI and related service information.

We might also have required different actions in this AD from those in the MCAI in order to follow FAA policies. Any such differences are described in a separate paragraph of the AD, and take precedence over the actions copied from the MCAI.