

(i) There are means to extinguish or control a fire without requiring a crewmember to enter the compartment.

(ii) There are means to exclude hazardous quantities of smoke, flames, or extinguishing agent from any compartment occupied by the crew or passengers.

(iii) There is a separate approved smoke detector or fire detector system to give warning at the pilot or flight engineer station.

(iv) Crew members must receive training in the use of the fire extinguishers and the cargo fire containment covers; they must also receive training in the use of the approved procedure for the elimination of smoke and fumes that is specified in the airplane flight manual (AFM).

(v) Two additional fire extinguishers must be carried on the airplane.

(vi) Limitations (f)(2)(i) through (f)(2)(v) must be documented as operating limitations in the Limitations section of the CASA C-212-CC AFM supplement.

#### Special Flight Permits

(g) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the airplane can be modified (if the operator elects to do so), provided no passengers are onboard.

#### Alternative Methods of Compliance (AMOCs)

(h)(1) The Manager, Anchorage ACO, FAA, has the authority to approve AMOCs for this AD, if requested in accordance with the procedures found in 14 CFR 39.19.

(2) Before using any AMOC approved in accordance with 14 CFR 39.19 on any airplane to which the AMOC applies, notify the appropriate principal inspector in the FAA Flight Standards Certificate Holding District Office.

#### Material Incorporated by Reference

(i) None.

Issued in Renton, Washington, on July 14, 2006.

**Ali Bahrami,**

*Manager, Transport Airplane Directorate,  
Aircraft Certification Service.*

[FR Doc. E6-11706 Filed 7-24-06; 8:45 am]

**BILLING CODE 4910-13-P**

## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 39

[Docket No. FAA-2006-24779; Directorate Identifier 2006-NM-044-AD; Amendment 39-14689; AD 2006-15-09]

RIN 2120-AA64

#### **Airworthiness Directives; Airbus Model A300 Airplanes; Model A310 Airplanes; and Model A300 B4-600, B4-600R, and F4-600R Series Airplanes, and Model C4-605R Variant F Airplanes (Collectively Called A300-600 Series Airplanes)**

**AGENCY:** Federal Aviation Administration (FAA), Department of Transportation (DOT).

**ACTION:** Final rule.

**SUMMARY:** The FAA is adopting a new airworthiness directive (AD) for all Airbus Model A300 airplanes and Model A310 airplanes, and for certain Airbus Model A300-600 series airplanes. This AD requires an inspection of the wing and center fuel tanks to determine if certain P-clips are installed and corrective action if necessary. This AD also requires an inspection of electrical bonding points of certain equipment in the center fuel tank for the presence of a blue coat and related investigative and corrective actions if necessary. This AD also requires installation of new bonding leads and electrical bonding points on certain equipment in the wing, center, and trim fuel tanks, as necessary. This AD results from fuel system reviews conducted by the manufacturer. We are issuing this AD to ensure continuous electrical bonding protection of equipment in the wing, center, and trim fuel tanks and to prevent damage to wiring in the wing and center fuel tanks, due to failed P-clips used for retaining the wiring and pipes, which could result in a possible fuel ignition source in the fuel tanks.

**DATES:** This AD becomes effective August 29, 2006.

The Director of the Federal Register approved the incorporation by reference of certain publications listed in the AD as of August 29, 2006.

**ADDRESSES:** You may examine the AD docket on the Internet at <http://dms.dot.gov> or in person at the Docket Management Facility, U.S. Department of Transportation, 400 Seventh Street, SW., Nassif Building, Room PL-401, Washington, DC.

Contact Airbus, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France,

for service information identified in this AD.

**FOR FURTHER INFORMATION CONTACT:** Tom Stafford, Aerospace Engineer, International Branch, ANM-116, Transport Airplane Directorate, FAA, 1601 Lind Avenue, SW., Renton, Washington 98057-3356; telephone (425) 227-1622; fax (425) 227-1149.

#### **SUPPLEMENTARY INFORMATION:**

##### **Examining the Docket**

You may examine the airworthiness directive (AD) docket on the Internet at <http://dms.dot.gov> or in person at the Docket Management Facility office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The Docket Management Facility office (telephone (800) 647-5227) is located on the plaza level of the Nassif Building at the street address stated in the **ADDRESSES** section.

##### **Discussion**

The FAA issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 to include an AD that would apply to all Airbus Model A300 airplanes and Model A310 airplanes, and for certain Airbus Model A300 B4-600, B4-600R, and F4-600R Series Airplanes, and Model C4-605R Variant F airplanes (collectively called A300-600 series airplanes). That NPRM was published in the **Federal Register** on May 17, 2006 (71 FR 28611). That NPRM proposed to require an inspection of the wing and center fuel tanks to determine if certain P-clips are installed and corrective action if necessary. That NPRM also proposed to require an inspection of electrical bonding points of certain equipment in the center fuel tank for the presence of a blue coat and related investigative and corrective actions if necessary. That NPRM also proposed to require installation of new bonding leads and electrical bonding points on certain equipment in the wing, center, and trim fuel tanks, as necessary.

##### **Comments**

We provided the public the opportunity to participate in the development of this AD. We received no comments on the NPRM or on the determination of the cost to the public.

##### **Conclusion**

We have carefully reviewed the available data and determined that air safety and the public interest require adopting the AD as proposed.

##### **Costs of Compliance**

There are about 29 Model A300 airplanes, 63 Model A310 airplanes, and

102 Model A300–600 series airplanes of the affected design in the U.S. fleet. The following table provides the estimated costs, at an average labor rate of \$80 per hour, for U.S. operators to comply with this AD. For some actions, the estimated work hours and cost of parts in the following table depend on the airplane configuration.

Model	Action	Work hours	Parts	Cost per airplane	Number of U.S.-registered airplanes	Fleet cost
A300 airplanes .....	Inspect wing and center fuel tanks for P-clips.	40	None	\$3,200	29	\$92,800
	Install bonding leads/points in wing and center fuel tank.	136–155	\$3,800–5,200	14,680–17,600	29	425,720–510,400
A310 airplanes .....	Inspect wing and center fuel tanks for P-clips.	40	None	3,200	63	201,600
	Install bonding leads/points in wing and center fuel tank.	248–285	\$8,840–9,190	28,680–31,990	63	1,806,840–2,015,370
	Inspect and install bonding leads/points in the trim fuel tank.	53–61	\$50–70	4,290–4,950	63	270,270–311,850
A300–600 series airplanes	Inspect wing and center fuel tanks for P-clips.	40	None	3,200	102	326,400
	Install bonding leads/points in wing and center fuel tank.	157–185	\$8,840–9,190	21,400–23,990	102	2,182,800–2,446,980
	Inspect and install bonding leads/points in the trim fuel tank.	2–61	50–70	210–4,950	102	21,420–504,900

**Authority for This Rulemaking**

Title 49 of the United States Code specifies the FAA’s authority to issue rules on aviation safety. Subtitle I, Section 106, describes the authority of the FAA Administrator. Subtitle VII, Aviation Programs, describes in more detail the scope of the Agency’s authority.

We are issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section 44701, “General requirements.” Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

**Regulatory Findings**

We have determined that this AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that this AD:

(1) Is not a “significant regulatory action” under Executive Order 12866;

(2) Is not a “significant rule” under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and

(3) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared a regulatory evaluation of the estimated costs to comply with this AD and placed it in the AD docket. See the **ADDRESSES** section for a location to examine the regulatory evaluation.

**List of Subjects in 14 CFR Part 39**

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

**Adoption of the Amendment**

■ Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

**PART 39—AIRWORTHINESS DIRECTIVES**

■ 1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

**§ 39.13 [Amended]**

■ 2. The Federal Aviation Administration (FAA) amends § 39.13 by adding the following new airworthiness directive (AD):

**2006–15–09 Airbus:** Amendment 39–14689. Docket No. FAA–2006–24779; Directorate Identifier 2006–NM–044–AD.

**Effective Date**

(a) This AD becomes effective August 29, 2006.

**Affected ADs**

(b) None.

**Applicability**

(c) This AD applies to the Airbus airplanes identified in paragraphs (c)(1) and (c)(2) of this AD, certificated in any category.

(1) All Model A300 airplanes and Model A310 airplanes.

(2) Model A300 B4–601, B4–603, B4–620, and B4–622 airplanes; Model A300 B4–605R and B4–622R airplanes; Model A300 F4–605R and F4–622R airplanes; and Model A300 C4–605R Variant F airplanes; except those airplanes identified in paragraphs (c)(2)(i) and (c)(2)(ii) of this AD.

(i) Airplanes not equipped with trim fuel tanks on which Airbus Modifications 12226, 12365, and 12308 have been incorporated in production.

(ii) Airplanes equipped with trim fuel tanks on which Airbus Modifications 12226, 12365, 12308, 12294, and 12476 have been incorporated in production.

**Unsafe Condition**

(d) This AD results from fuel system reviews conducted by the manufacturer. We are issuing this AD to ensure continuous electrical bonding protection of equipment in the wing, center, and trim fuel tanks and to prevent damage to wiring in the wing and center fuel tanks, due to failed P-clips used for retaining the wiring and pipes, which could result in a possible fuel ignition source in the fuel tanks.

**Compliance**

(e) You are responsible for having the actions required by this AD performed within

the compliance times specified, unless the actions have already been done.

**Service Bulletin References**

(f) The term "service bulletin," as used in this AD, means the Accomplishment

Instructions of the service bulletin identified in Table 1 of this AD, as applicable.

**TABLE 1.—SERVICE BULLETIN REFERENCES**

For Airbus—	And the actions specified in—	Use Airbus Service Bulletin—	Dated—
Model A300 airplanes .....	Paragraph (g) of this AD .....	A300–28–0081 .....	July 20, 2005.
	Paragraph (h) of this AD .....	A300–28–0079 .....	September 29, 2005.
Model A310 airplanes .....	Paragraph (g) of this AD .....	A310–28–2143 .....	July 20, 2005.
	Paragraph (h) of this AD .....	A310–28–2142 .....	August 26, 2005.
	Paragraph (i) of this AD .....	A310–28–2153 .....	July 20, 2005.
Model A300 B4–601, B4–603, B4–620, and B4–622 airplanes; Model A300 B4–605R and B4–622R airplanes; Model A300 F4–605R and F4–622R airplanes; and Model A300 C4–605R Variant F airplanes.	Paragraph (g) of this AD .....	A300–28–6068 .....	July 20, 2005.
	Paragraph (h) of this AD .....	A300–28–6064 .....	July 28, 2005.
	Paragraph (i) of this AD .....	A300–28–6077 .....	July 25, 2005.

**Inspection and Corrective Actions**

(g) Within 59 months after the effective date of this AD: Do a general visual inspection of the right and left wing fuel tanks and center fuel tank, if applicable, to determine if any NSA5516–XXND and NSA5516–XXNJ type P-clips are installed for retaining wiring and pipes in any tank, and do all applicable corrective actions before further flight after the inspection, by accomplishing all the actions specified in the service bulletin.

**Note 1:** For the purposes of this AD, a general visual inspection is: "A visual examination of an interior or exterior area, installation, or assembly to detect obvious damage, failure, or irregularity. This level of inspection is made from within touching distance unless otherwise specified. A mirror may be necessary to ensure visual access to all surfaces in the inspection area. This level of inspection is made under normally available lighting conditions such as daylight, hangar lighting, flashlight, or droplight and may require removal or opening of access panels or doors. Stands, ladders, or platforms may be required to gain proximity to the area being checked."

**Installation of Bonding Leads and Points for Wing and Center Fuel Tanks**

(h) Within 59 months after the effective date of this AD: Do the actions specified in paragraphs (h)(1) and (h)(2) of this AD, by accomplishing all the actions specified in the service bulletin.

(1) In the center fuel tank, if applicable, do a general visual inspection of the electrical bonding points of the equipment identified in the service bulletin for the presence of a blue coat, and do all related investigative and corrective actions before further flight after the inspection.

(2) In the left and right wing fuel tanks and center fuel tank, if applicable, install bonding leads and electrical bonding points on the equipment identified in the service bulletin.

**Installation of Bonding Leads and Points for the Trim Fuel Tank**

(i) For Model A310 airplanes; Model A300 B4–601, B4–603, B4–620, and B4–622 airplanes; Model A300 B4–605R and B4–

622R airplanes; Model A300 F4–605R and F4–622R airplanes; and Model A300 C4–605R Variant F airplanes; equipped with a trim fuel tank: Within 59 months after the effective date of this AD, install a new bonding lead(s) on the water drain system of the trim fuel tank and install electrical bonding points on the equipment identified in the service bulletin in the trim fuel tank, by accomplishing all the actions specified in the service bulletin, as applicable.

**Parts Installation**

(j) As of the effective date of this AD, no person may install any NSA5516–XXND or NSA5516–XXNJ type P-clip for retaining wiring and pipes in any wing, center, or trim fuel tank, on any airplane.

**Alternative Methods of Compliance (AMOCs)**

(k)(1) The Manager, International Branch, ANM–116, Transport Airplane Directorate, FAA, has the authority to approve AMOCs for this AD, if requested in accordance with the procedures found in 14 CFR 39.19.

(2) Before using any AMOC approved in accordance with § 39.19 on any airplane to which the AMOC applies, notify the appropriate principal inspector in the FAA Flight Standards Certificate Holding District Office.

**Related Information**

(l) French airworthiness directive F–2006–031, dated February 1, 2006, also addresses the subject of this AD.

**Material Incorporated by Reference**

(m) You must use the Airbus service bulletins identified in Table 2 of this AD to perform the actions that are required by this AD, unless the AD specifies otherwise. The Director of the Federal Register approved the incorporation by reference of these documents in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Contact Airbus, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France, for a copy of this service information. You may review copies at the Docket Management Facility, U.S. Department of Transportation, 400 Seventh Street SW., Room PL–401, Nassif Building, Washington, DC; on the Internet at [http://](http://dms.dot.gov)

[dms.dot.gov](http://dms.dot.gov); or at the National Archives and Records Administration (NARA). For information on the availability of this material at the NARA, call (202) 741–6030, or go to [http://www.archives.gov/federal\\_register/code\\_of\\_federal\\_regulations/ibr\\_locations.html](http://www.archives.gov/federal_register/code_of_federal_regulations/ibr_locations.html).

**TABLE 2.—MATERIAL INCORPORATED BY REFERENCE**

Airbus Service Bulletin—	Dated—
A300–28–0079 .....	September 29, 2005.
A300–28–0081 .....	July 20, 2005.
A300–28–6064 .....	July 28, 2005.
A300–28–6068 .....	July 20, 2005.
A300–28–6077 .....	July 25, 2005.
A310–28–2142 .....	August 26, 2005.
A310–28–2143 .....	July 20, 2005.
A310–28–2153 .....	July 20, 2005.

Issued in Renton, Washington, on July 14, 2006.

**Ali Bahrami,**

*Manager, Transport Airplane Directorate, Aircraft Certification Service.*

[FR Doc. E6–11713 Filed 7–24–06; 8:45 am]

**BILLING CODE 4910–13–P**

**CONSUMER PRODUCT SAFETY COMMISSION**

**16 CFR Part 1115**

**Substantial Product Hazard Reports**

**AGENCY:** Consumer Product Safety Commission.

**ACTION:** Final interpretative rule.

**SUMMARY:** Section 15(b) of the Consumer Product Safety Act, 15 U.S.C. 2064(b), requires manufacturers, distributors, and retailers of consumer products to