

on the ground or an in-flight engine shutdown, accomplish the following:

Modification/Repetitive Inspections

(a) Within 12 months after the effective date of this AD: Modify the hot detection system of the tail pipe harness of the engine nacelles (including a general visual inspection of the heat shrink sleeve, thixotropic sealant, and connectors for damage and/or corrosion, and all applicable repairs), by doing all the actions specified in the Accomplishment Instructions of Saab Service Bulletin 340-26-030, Revision 01, dated November 14, 2003. All applicable repairs must be done before further flight in accordance with the service bulletin. Repeat the general visual inspection thereafter at intervals not to exceed 12 months.

Note 1: For the purposes of this AD, a general visual inspection is defined as: "A visual examination of an interior or exterior area, installation, or assembly to detect obvious damage, failure, or irregularity. This level of inspection is made from within touching distance unless otherwise specified. A mirror may be necessary to enhance visual access to all exposed surfaces in the inspection area. This level of inspection is made under normally available lighting conditions such as daylight, hangar lighting, flashlight, or droplight and may require removal or opening of access panels or doors. Stands, ladders, or platforms may be required to gain proximity to the area being checked."

(b) Accomplishing the modification/repetitive inspections specified in Saab Service Bulletin 340-26-030, dated October 28, 2002; or Saab Service Bulletins 340-26-018, Revision 02, and 340-26-029, both dated October 28, 2002; before the effective date of this AD, is considered acceptable for compliance with the modification required by paragraph (a) of this AD.

Reporting Requirement

(c) Within 30 days after any false warning indication to the flightcrew from the hot detection system of the tail pipe harness of the engine nacelles occurs: Submit a report containing the information specified in paragraphs (c)(1), (c)(2), (c)(3), and (c)(4) of this AD to the Swedish Civil Aviation Authority (Luftfartsstyrelsen)—Attn: Mr. Christer Sundqvist, SAAB 340 Certification Manager, SE-601 79, Norrköping, Sweden. Under the provisions of the Paperwork Reduction Act (44 U.S.C. 3501 *et seq.*), the Office of Management and Budget (OMB) has approved the information collection requirements contained in this AD and has assigned OMB Control Number 2120-0056.

(1) The date and time, weather conditions, and phase of flight of the warning.

(2) The action taken by the crew to address the warning (aborted takeoff, high speed/high energy abort requiring inspection, return for landing, in-flight diversion, declared emergency, ATC priority handling requested or given, or engine shutdown).

(3) The action taken by maintenance to address/correct the warning.

(4) Time-in-service on the airplane since the last inspection accomplished in accordance with paragraph (a) of this AD.

Alternative Methods of Compliance (AMOCs)

(d)(1) In accordance with 14 CFR 39.19, the Manager, International Branch, ANM-116, Transport Airplane Directorate, FAA, is authorized to approve AMOCs for this AD.

(2) Before using any AMOC approved in accordance with 14 CFR 39.19 on any airplane to which the AMOC applies, notify the appropriate principal inspector in the FAA Flight Standards Certificate Holding District Office.

Note 2: The subject of this AD is addressed in Swedish airworthiness directive 1-184, dated October 28, 2002.

Issued in Renton, Washington, on June 19, 2006.

Ali Bahrami,

Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. E6-10014 Filed 6-23-06; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA-2006-25174; Directorate Identifier 2005-NM-007-AD]

RIN 2120-AA64

Airworthiness Directives; Learjet Model 45 Airplanes

AGENCY: Federal Aviation Administration (FAA), Department of Transportation (DOT).

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: The FAA proposes to adopt a new airworthiness directive (AD) for certain Learjet Model 45 airplanes. This proposed AD would require revising the Airworthiness Limitations section of the airplane maintenance manual to incorporate certain inspections and compliance times to detect fatigue cracking of certain principal structural elements (PSEs). This proposed AD results from new and more restrictive life limits and inspection intervals for certain PSEs. We are proposing this AD to ensure that fatigue cracking of various PSEs is detected and corrected; such fatigue cracking could adversely affect the structural integrity of these airplanes.

DATES: We must receive comments on this proposed AD by August 10, 2006.

ADDRESSES: Use one of the following addresses to submit comments on this proposed AD.

- DOT Docket Web site: Go to <http://dms.dot.gov> and follow the instructions for sending your comments electronically.

- Government-wide rulemaking Web site: Go to <http://www.regulations.gov> and follow the instructions for sending your comments electronically.

- Mail: Docket Management Facility, U.S. Department of Transportation, 400 Seventh Street, SW., Nassif Building, Room PL-401, Washington, DC 20590.

- Fax: (202) 493-2251.

- Hand Delivery: Room PL-401 on the plaza level of the Nassif Building, 400 Seventh Street, SW., Washington, DC, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

Contact Learjet, Inc., One Learjet Way, Wichita, Kansas 67209-2942, for the service information identified in this proposed AD.

FOR FURTHER INFORMATION CONTACT:

Steve Litke, Aerospace Engineer, Airframe Branch, ACE-118W, FAA, Wichita Aircraft Certification Office, 1801 Airport Road, Room 100, Mid-Continent Airport, Wichita, Kansas 67209; telephone (316) 946-4127; fax (316) 946-4107.

SUPPLEMENTARY INFORMATION:

Comments Invited

We invite you to submit any relevant written data, views, or arguments regarding this proposed AD. Send your comments to an address listed in the **ADDRESSES** section. Include the docket number "FAA-2006-25174; Directorate Identifier 2005-NM-007-AD" at the beginning of your comments. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of the proposed AD. We will consider all comments received by the closing date and may amend the proposed AD in light of those comments.

We will post all comments we receive, without change, to <http://dms.dot.gov>, including any personal information you provide. We will also post a report summarizing each substantive verbal contact with FAA personnel concerning this proposed AD. Using the search function of that Web site, anyone can find and read the comments in any of our dockets, including the name of the individual who sent the comment (or signed the comment on behalf of an association, business, labor union, etc.). You may review DOT's complete Privacy Act Statement in the **Federal Register** published on April 11, 2000 (65 FR 19477-78), or you may visit <http://dms.dot.gov>.

Examining the Docket

You may examine the AD docket on the Internet at <http://dms.dot.gov>, or in person at the Docket Management

Facility office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The Docket Management Facility office (telephone (800) 647-5227) is located on the plaza level of the Nassif Building at the DOT street address stated in the **ADDRESSES** section. Comments will be available in the AD docket shortly after the Docket Management System receives them.

Discussion

As service experience is accumulated on airplanes or as the result of post-certification testing and evaluation, it may become necessary to revise removal limits for removal of certain life-limited components of the airplane or revise the interval for certain structural inspections in order to ensure the continued structural integrity of the airplane. The manufacturer may revise the Airworthiness Limitations document to include more restrictive life limits or revise repetitive intervals for certain non-destructive inspection (NDI) techniques and procedures for each principal structural element (PSE). For the purposes of this airworthiness directive, a PSE is defined as an element of structure that contributes significantly to carrying flight, ground, and pressurization loads. If a failure occurs on any of those PSEs, it could adversely affect the structural integrity of the airplane.

The actions specified by the proposed AD are intended to ensure that fatigue cracking of various PSEs is detected and corrected; such fatigue cracking could adversely affect the structural integrity of these airplanes.

New Revisions of Airworthiness Limitations Sections (ALS)

We have reviewed Chapter 4, "Airworthiness Limitations," of the Learjet 40 Maintenance Manual (MM), Revision 6, dated April 24, 2006; and Chapter 4, "Airworthiness Limitations," of the Learjet 45 MM, Revision 38, dated April 24, 2006. These MM chapters add new and more restrictive life limits and inspection intervals for certain PSEs. PSEs include, but are not limited to, door cutouts, windshields, skin sections, bolts, and attachment hardware. The MM chapters explicitly identify all of the PSEs that are to be inspected in accordance with the requirements of the Airworthiness Limitations section (ALS). Accomplishing the actions specified in these chapters is intended to adequately address the unsafe condition.

FAA's Determination and Requirements of the Proposed AD

We have evaluated all pertinent information and identified an unsafe condition that is likely to exist or develop on other airplanes of this same type design. For this reason, we are proposing this AD, which would require revising the ALS of the MM to incorporate certain inspections and compliance times to detect fatigue cracking of certain PSEs.

Clarification of Model Designations

Certain Learjet Model 45 airplanes are also referred to as Model 45 (Learjet 40) airplanes. Model 45 (Learjet 40) airplanes have serial numbers (S/Ns) 45-2001 through 45-4000 inclusive. The remainder of the Learjet Model 45 airplanes are referred to as Model 45 (Learjet 45) airplanes, and have S/Ns 45-002 through 45-2000 inclusive.

Costs of Compliance

There are about 230 airplanes of the affected design in the worldwide fleet. This proposed AD would affect about 171 airplanes of U.S. registry. The proposed actions would take about 1 work hour per airplane, at an average labor rate of \$80 per work hour. Based on these figures, the estimated cost of the proposed AD for U.S. operators is \$13,680, or \$80 per airplane.

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, Section 106, describes the authority of the FAA Administrator. Subtitle VII, Aviation Programs, describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section 44701, "General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

We have determined that this proposed AD would not have federalism implications under Executive Order 13132. This proposed AD would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or

on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that the proposed regulation:

1. Is not a "significant regulatory action" under Executive Order 12866;
2. Is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and
3. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared a regulatory evaluation of the estimated costs to comply with this proposed AD and placed it in the AD docket. See the **ADDRESSES** section for a location to examine the regulatory evaluation.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

The Proposed Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA proposes to amend 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. The Federal Aviation Administration (FAA) amends § 39.13 by adding the following new airworthiness directive (AD):

Learjet: Docket No. FAA-2006-25174; Directorate Identifier 2005-NM-007-AD.

Comments Due Date

- (a) The FAA must receive comments on this AD action by August 10, 2006.

Affected ADs

- (b) None.

Applicability

- (c) This AD applies to Learjet Model 45 airplanes, certificated in any category; serial numbers (S/Ns) 45-002 through 45-233 inclusive, and S/Ns 45-2001 through 45-2031 inclusive.

Unsafe Condition

- (d) This AD results from new and more restrictive life limits and inspection intervals for certain principal structural elements (PSEs). We are issuing this AD to ensure that fatigue cracking of various PSEs is detected and corrected; such fatigue cracking could adversely affect the structural integrity of these airplanes.

Compliance

(e) You are responsible for having the actions required by this AD performed within the compliance times specified, unless the actions have already been done.

Note 1: This AD requires revisions to certain operator maintenance documents to include new inspections. Compliance with these inspections is required by 14 CFR 91.403(c). For airplanes that have been previously modified, altered, or repaired in the areas addressed by these inspections, the operator may not be able to accomplish the inspections described in the revisions. In this situation, to comply with 14 CFR 91.403(c), the operator must request approval for an alternative method of compliance (AMOC) according to paragraph (g) of this AD. The request should include a description of changes to the required inspections that will ensure the continued damage tolerance of the affected structure. The FAA has provided guidance for this determination in Advisory Circular (AC) 25–1529.

Revise the Airworthiness Limitations Section (ALS)

(f) Within 30 days after the effective date of this AD, revise the ALS of the airplane maintenance manual (AMM) to include new life limits and inspection intervals according to a method approved by the Manager, Wichita Aircraft Certification Office (ACO), FAA. Incorporating the applicable chapters in paragraph (f)(1) or (f)(2) of this AD in the AMM is one approved method for doing the revision. Thereafter, except as provided in paragraph (g) of this AD, no alternative life limits or inspection intervals may be approved for the affected PSEs.

(1) For Learjet Model 45 airplanes, S/Ns 45–002 through 45–233 inclusive: Chapter 4 of the Learjet 45 Maintenance Manual, Revision 38, dated April 24, 2006.

(2) For Learjet Model 45 airplanes, S/Ns 45–2001 through 45–2031 inclusive: Chapter 4 of the Learjet 40 Maintenance Manual, Revision 6, dated April 24, 2006.

AMOCs

(g)(1) The Manager, Wichita ACO, FAA, has the authority to approve AMOCs for this AD, if requested in accordance with the procedures found in 14 CFR 39.19.

(2) Before using any AMOC approved in accordance with § 39.19 on any airplane to which the AMOC applies, notify the appropriate principal inspector in the FAA Flight Standards Certificate Holding District Office.

Issued in Renton, Washington, on June 14, 2006.

Kalene C. Yanamura,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. E6–10004 Filed 6–23–06; 8:45 am]

BILLING CODE 4910–13–P

DEPARTMENT OF TRANSPORTATION**Federal Aviation Administration****14 CFR Part 71**

[Docket No. FAA–2006–25059; Airspace Docket No. 06–ACE–8]

Proposed Establishment of Class E5 Airspace; Higginsville, MO

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Notice of proposed rulemaking.

SUMMARY: This notice proposes to amend part 71 of the Federal Aviation Regulations (14 CFR part 71) by establishing a Class E airspace area extending upward from 700 feet above the surface at Higginsville Industrial Municipal Airport, MO.

DATES: Comments for inclusion in the Rules Docket must be received on or before August 1, 2006.

ADDRESSES: Send comments on this proposal to the Docket Management System, U.S. Department of Transportation, Room Plaza 401, 400 Seventh Street, SW., Washington, DC 20590–0001. You must identify the docket number FAA–2006–25059/ Airspace Docket No. 06–ACE–8, at the beginning of your comments. You may also submit comments on the Internet at <http://dms.dot.gov>. You may review the public docket containing the proposal, any comments received, and any final disposition in person the Dockets Office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The Docket Office (telephone 1–800–647–5527) is on the plaza level of the Department of Transportation NASSIF Building at the above address.

FOR FURTHER INFORMATION CONTACT: Brenda Mumper, Air Traffic Division, Airspace Branch, ACE–520A, DOT Regional Headquarters Building, Federal Aviation Administration, 901 Locust, Kansas City, MO 64106; telephone: (816) 329–2524.

SUPPLEMENTARY INFORMATION:**Comments Invited**

Interested parties are invited to participate in this proposed rulemaking by submitting such written data, views, or arguments, as they may desire. Comments that provide the factual basis supporting the views and suggestions presented are particularly helpful in developing reasoned regulatory decisions on the proposal. Comments are specifically invited on the overall regulatory, aeronautical, economic, environmental, and energy-related aspects of the proposal.

Communications should identify both docket numbers and be submitted in triplicate to the address listed above. Commenters wishing the FAA to acknowledge receipt of their comments on this notice must submit with those comments a self-addressed, stamped postcard on which the following statement is made: “Comments to Docket No. FAA–2006–25059/Airspace Docket No. 06–ACE–8.” The postcard will be date/time stamped and returned to the commenter.

Availability of NPRM's

An electronic copy of this document may be downloaded through the Internet at <http://dms.dot.gov>. Recently published rulemaking documents can also be accessed through the FAA's Web page at <http://www.faa.gov> or the Superintendent of Document's Web page at <http://www.access.gpo.gov/nara>.

Additionally, any person may obtain a copy of this notice by submitting a request to the Federal Aviation Administration (FAA), Office of Air Traffic Airspace Management, ATA–400, 800 Independence Avenue, SW., Washington, DC 20591, or by calling (202) 267–8783. Communications must identify both docket number for this notice. Persons interested in being placed on a mailing list for future NPRM's should contact the FAA's Office of Rulemaking (202) 267–9677, to request a copy of Advisory Circular No. 11–2A, Notice of Proposed Rulemaking Distribution System, which describes the application procedure.

The Proposal

This notice proposes to amend part 71 of the Federal Aviation Regulations (14 CFR part 71) by establishing a Class E airspace area extending upward from 700 feet above the surface at Higginsville Industrial Municipal Airport, MO. The establishment of Area Navigation (RNAV) Global Positioning System (GPS) Instrument Approach Procedures (IAP) to Runways 16 and 34 have made this action necessary. The intended effect of this proposal is to provide adequate controlled airspace for Instrument Flight Rules operations at Higginsville Industrial Municipal Airport, MO. The area would be depicted on appropriate aeronautical charts.

Class E airspace areas extending upward from 700 feet or more above the surface of the earth are published in Paragraph 6005 of FAA Order 7400.9N, dated September 1, 2005, and effective September 16, 2005, which is incorporated by reference in 14 CFR 71.1. The Class E airspace designations