

(2) Is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and

(3) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared a summary of the costs to comply with this AD and placed it in the AD Docket. You may get a copy of this summary at the address listed under ADDRESSES.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

Accordingly, under the authority delegated to me by the Administrator, the Federal Aviation Administration amends 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. The FAA amends § 39.13 by adding the following new airworthiness directive:

2006-08-03 Sicma Aero Seat (formerly Farner): Amendment 39-14557. Docket

No. FAA-2005-22109; Directorate Identifier 2005-NE-32-AD.

Effective Date

(a) This airworthiness directive (AD) becomes effective May 18, 2006.

Affected ADs

(b) None.

Applicability

(c) This AD applies to Sicma Aero Seat (formerly Farner) cabin attendant seats series 150 type FN and 151 type WN, all part and serial numbers. These attendant seats are installed on, but not limited to, Airbus A319, A320, and A321 series airplanes.

Unsafe Condition

(d) This AD results from a child catching its fingers in the folding mechanism of the bottom of the attendant seat. We are issuing this AD to prevent injury resulting from contact with the bottom folding mechanism.

Compliance

(e) You are responsible for having the actions required by this AD performed within the compliance times specified unless the actions have already been done.

Installing Protective Fairings

(f) Within 90 days after the effective date of this AD, install two protective fairings, part number (P/N) 160100-49, on each affected cabin attendant seat. Use the instructions in paragraph 2 of Sicma Aero Seat Service Bulletin 150-25-036, Issue 1, dated February 10, 1999, and Sicma Aero Seat Service Bulletin 151-25-037, Issue 1, dated February 10, 1999.

(g) After installing the fairings, add or complete a modification placard, part number (P/N) 00-5179, indicating that the

service bulletin has been completed. Use the instructions in paragraph 3 of Sicma Aero Seat Service Bulletin 150-25-036, Issue 1, dated February 10, 1999, and Sicma Aero Seat Service Bulletin 151-25-037, Issue 1, dated February 10, 1999.

Alternative Methods of Compliance

(h) The Manager, Boston Aircraft Certification Office, has the authority to approve alternative methods of compliance for this AD if requested using the procedures found in 14 CFR 39.19.

Related Information

(i) Direction Generale de L'Aviation Civile airworthiness directive 1999-004 (AB), dated January 13, 1999, also addresses the subject of this AD.

Material Incorporated by Reference

(j) You must use the Sicma Aero Seat Service Bulletins specified in Table 1 to perform the installations required by this AD. The Director of the Federal Register approved the incorporation by reference of the documents listed in Table 1 of this AD in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Contact Sicma Aero Seat, 7 Rue Lucien Coupet, 36100 Issoudun, France; telephone 33 (0) 2 54 03 39 39, fax 33 (0) 2 54 03 15 16 for a copy of this service information. You may review copies at the Docket Management Facility; U.S. Department of Transportation, 400 Seventh Street, SW., Nassif Building, Room PL-401, Washington, DC 20590-0001, on the internet at <http://dms.dot.gov>, or at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call 202-741-6030, or go to: <http://www.archives.gov/federal-register/cfr/ibr-locations.html>.

TABLE 1.—INCORPORATION BY REFERENCE

Service bulletin No.	Page	Issue	Date
150-25-036 Total Pages: 8	ALL	1	February 10, 1999.
151-25-037 Total Pages: 8	ALL	1	February 10, 1999.

Issued in Burlington, Massachusetts, on April 5, 2006.

Peter A. White,

Acting Manager, Engine and Propeller Directorate, Aircraft Certification Service.

[FR Doc. 06-3479 Filed 4-12-06; 8:45 am]

BILLING CODE 4910-13-P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA-2006-24429; Directorate Identifier 2006-NM-003-AD; Amendment 39-14559; AD 2006-08-05]

RIN 2120-AA64

Airworthiness Directives; Fokker Model F.28 Mark 0100 Airplanes

AGENCY: Federal Aviation Administration (FAA), Department of Transportation (DOT).

ACTION: Final rule; request for comments.

SUMMARY: The FAA is adopting a new airworthiness directive (AD) for certain Fokker Model F.28 Mark 0100 airplanes. This AD requires reinforcing the fuselage at the emergency locator transmitter system's antenna area. This AD results from a report that certain structural provisions for an optional antenna are of insufficient strength. We are issuing this AD to prevent reduced structural integrity of the airplane.

DATES: This AD becomes effective April 28, 2006.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in the AD as of April 28, 2006.

We must receive comments on this AD by June 12, 2006.

ADDRESSES: Use one of the following addresses to submit comments on this AD.

- DOT Docket Web site: Go to <http://dms.dot.gov> and follow the instructions for sending your comments electronically.

- Government-wide rulemaking Web site: Go to <http://www.regulations.gov> and follow the instructions for sending your comments electronically.

- Mail: Docket Management Facility; U.S. Department of Transportation, 400 Seventh Street, SW., Nassif Building, Room PL-401, Washington, DC 20590.

- Fax: (202) 493-2251.

- Hand Delivery: Room PL-401 on the plaza level of the Nassif Building, 400 Seventh Street, SW., Washington, DC, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

Contact Fokker Services B.V., P.O. Box 231, 2150 AE Nieuw-Vennep, the Netherlands, for service information identified in this AD.

FOR FURTHER INFORMATION CONTACT: Tom Rodriguez, Aerospace Engineer, International Branch, ANM-116, Transport Airplane Directorate, FAA, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 227-1137; fax (425) 227-1149.

SUPPLEMENTARY INFORMATION:

Discussion

The Civil Aviation Authority—The Netherlands (CAA-NL), which is the airworthiness authority for the Netherlands, notified us that an unsafe condition may exist on certain Fokker Model F.28 Mark 0100 airplanes. The CAA-NL advises that in 2003 and 2004 the manufacturer developed two optional modifications for airplanes with an emergency locator transmitter (ELT) system installed in production. These modifications included removing existing structural provisions in the rear fuselage, and installing new structural provisions for a new antenna and tuner. Recent investigations showed that the new structural provisions for the new antenna are of insufficient strength. This condition, if not corrected, could result in reduced structural integrity of the airplane.

Relevant Service Information

Fokker Services B.V. has issued Service Bulletin SBF100-53-100, dated February 28, 2005, including Drawing W98488, Sheets 07 and 08, Issue F, dated February 9, 2004, and Sheets 09 and 10, Issue F, dated February 10, 2004. The service bulletin describes procedures for reinforcing the fuselage at the ELT system's antenna area. The reinforcement involves installing new,

improved angles at the existing attachment holes. Accomplishing the actions specified in the service information is intended to adequately address the unsafe condition. The CAA-NL mandated the service information and issued Dutch airworthiness directive NL-2005-004, dated April 29, 2005, to ensure the continued airworthiness of these airplanes in the Netherlands.

FAA's Determination and Requirements of This AD

This airplane model is manufactured in the Netherlands and is type certificated for operation in the United States under the provisions of section 21.29 of the Federal Aviation Regulations (14 CFR 21.29) and the applicable bilateral airworthiness agreement. Pursuant to this bilateral airworthiness agreement, the CAA-NL has kept the FAA informed of the situation described above. We have examined the CAA-NL's findings, evaluated all pertinent information, and determined that we need to issue an AD for products of this type design that are certificated for operation in the United States.

Therefore, we are issuing this AD to prevent reduced structural integrity of the airplane. This AD requires accomplishing the actions specified in the service information described previously.

Costs of Compliance

None of the airplanes affected by this action are on the U.S. Register. All airplanes affected by this AD are currently operated by non-U.S. operators under foreign registry; therefore, they are not directly affected by this AD action. However, we consider this AD necessary to ensure that the unsafe condition is addressed if any affected airplane is imported and placed on the U.S. Register in the future.

If an affected airplane is imported and placed on the U.S. Register in the future, the required actions would take about 3 work hours per airplane, at an average labor rate of \$80 per work hour. Required parts are free of charge. Based on these figures, the estimated cost of the AD would be \$240 per airplane.

FAA's Determination of the Effective Date

No airplane affected by this AD is currently on the U.S. Register. Therefore, providing notice and opportunity for public comment is unnecessary before this AD is issued, and this AD may be made effective in less than 30 days after it is published in the **Federal Register**.

Comments Invited

This AD is a final rule that involves requirements that affect flight safety and was not preceded by notice and an opportunity for public comment; however, we invite you to submit any relevant written data, views, or arguments regarding this AD. Send your comments to an address listed in the **ADDRESSES** section. Include "Docket No. FAA-2006-24429; Directorate Identifier 2006-NM-003-AD" at the beginning of your comments. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of the AD that might suggest a need to modify it.

We will post all comments we receive, without change, to <http://dms.dot.gov>, including any personal information you provide. We will also post a report summarizing each substantive verbal contact with FAA personnel concerning this AD. Using the search function of that web site, anyone can find and read the comments in any of our dockets, including the name of the individual who sent the comment (or signed the comment on behalf of an association, business, labor union, etc.). You may review the DOT's complete Privacy Act Statement in the **Federal Register** published on April 11, 2000 (65 FR 19477-78), or you may visit <http://dms.dot.gov>.

Examining the Docket

You may examine the AD docket on the Internet at <http://dms.dot.gov>, or in person at the Docket Management Facility office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The Docket Management Facility office (telephone (800) 647-5227) is located on the plaza level of the Nassif Building at the DOT street address stated in the **ADDRESSES** section. Comments will be available in the AD docket shortly after the Docket Management System receives them.

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. Subtitle VII, Aviation Programs, describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in subtitle VII, part A, subpart III, section 44701, "General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures

the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

We have determined that this AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that the regulation:

1. Is not a "significant regulatory action" under Executive Order 12866;
2. Is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and
3. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared a regulatory evaluation of the estimated costs to comply with this AD and placed it in the AD docket. See the ADDRESSES section for a location to examine the regulatory evaluation.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. The Federal Aviation Administration (FAA) amends § 39.13 by adding the following new airworthiness directive (AD):

2006-08-05 Fokker Services B.V.:
Amendment 39-14559. Docket No. FAA-2006-24429; Directorate Identifier 2006-NM-003-AD.

Effective Date

(a) This AD becomes effective April 28, 2006.

Affected ADs

(b) None.

Applicability

(c) This AD applies to Fokker Model F.28 Mark 0100 airplanes, certificated in any category; serial numbers 11290, 11296, 11323, 11329, and 11330, if modified in accordance with Fokker Service Bulletin SBF100-25-038 or SBF100-25-096.

Unsafe Condition

(d) This AD results from a report that certain structural provisions for an optional antenna are of insufficient strength. We are issuing this AD to prevent reduced structural integrity of the airplane.

Compliance

(e) You are responsible for having the actions required by this AD performed within the compliance times specified, unless the actions have already been done.

Installation

(f) Within 24 months or 3,000 flight cycles after the effective date of this AD, whichever occurs first: Reinforce the fuselage at the emergency locator transmitter system's antenna area in accordance with Part 1 or Part 2 of the Accomplishment Instructions of Fokker Service Bulletin SBF100-53-100, dated February 28, 2005, including Drawing W98488, Sheets 07 and 08, Issue F, dated February 9, 2004, and Sheets 09 and 10, Issue F, dated February 10, 2004.

Alternative Methods of Compliance (AMOCs)

(g)(1) The Manager, International Branch, ANM-116, Transport Airplane Directorate, FAA, has the authority to approve AMOCs for this AD, if requested in accordance with the procedures found in 14 CFR 39.19.

(2) Before using any AMOC approved in accordance with § 39.19 on any airplane to which the AMOC applies, notify the appropriate principal inspector in the FAA Flight Standards Certificate Holding District Office.

Related Information

(h) Dutch airworthiness directive NL-2005-004, dated April 29, 2005, also addresses the subject of this AD.

Material Incorporated by Reference

(i) You must use Fokker Service Bulletin SBF100-53-100, dated February 28, 2005, including Drawing W98488, Sheets 07 and 08, Issue F, dated February 9, 2004, and Sheets 09 and 10, Issue F, dated February 10, 2004, to perform the actions that are required by this AD, unless the AD specifies otherwise. Fokker Service Bulletin SBF100-53-100 contains the following effective pages:

Page No.	Revision level shown on page	Date shown on page
1-9	Original	February 28, 2005.
Drawing W98488		
07, 08	F	February 9, 2004.

Page No.	Revision level shown on page	Date shown on page
09, 10	F	February 10, 2004.

The Director of the Federal Register approved the incorporation by reference of this document in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Contact Fokker Services B.V., P.O. Box 231, 2150 AE Nieuw-Vennep, the Netherlands, for a copy of this service information. You may review copies at the Docket Management Facility, U.S. Department of Transportation, 400 Seventh Street, SW., Room PL-401, Nassif Building, Washington, DC; on the Internet at <http://dms.dot.gov>; or at the National Archives and Records Administration (NARA). For information on the availability of this material at the NARA, call (202) 741-6030, or go to http://www.archives.gov/federal_register/code_of_federal_regulations/ibr_locations.html.

Issued in Renton, Washington, on April 4, 2006.

Kevin M. Mullin,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 06-3480 Filed 4-12-06; 8:45 am]

BILLING CODE 4910-13-P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA-2006-24446; Directorate Identifier 2005-SW-15-AD; Amendment 39-14561; AD 2006-08-06]

RIN 2120-AA64

Airworthiness Directives; Eurocopter France Model SA-360C, SA-365C, SA-365C1, and SA-365C2 Helicopters

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule; request for comments.

SUMMARY: This amendment adopts a new airworthiness directive (AD) for Eurocopter France (Eurocopter) Model SA-360C, SA-365C, SA-365C1, and SA-365C2 helicopters. This action requires inspecting the main gearbox (MGB) base plate for a crack and replacing the MGB if a crack is found. This amendment is prompted by the discovery of a crack in a MGB base plate. The actions specified in this AD are intended to detect a crack in a MGB base plate and prevent failure of one of the MGB attachment points to the frame, which could result in severe vibration and subsequent loss of control of the helicopter.

DATES: Effective April 28, 2006.