

appropriate principal inspector in the FAA Flight Standards Certificate Holding District Office.

#### Related Information

(i) Canadian airworthiness directive CF-2004-23R1, dated July 18, 2005, also addresses the subject of this AD.

#### Material Incorporated by Reference

(j) You must use Bombardier Service Bulletin 601R-31-030, Revision F, dated September 1, 2005, to perform the actions that are required by this AD, unless the AD specifies otherwise. The Director of the Federal Register approved the incorporation by reference of this document in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Contact Bombardier, Inc., Canadair, Aerospace Group, P.O. Box 6087, Station Centre-ville, Montreal, Quebec H3C 3G9, Canada, for a copy of this service information. You may review copies at the Docket Management Facility, U.S. Department of Transportation, 400 Seventh Street SW., room PL-401, Nassif Building, Washington, DC; on the Internet at <http://dms.dot.gov>; or at the National Archives and Records Administration (NARA). For information on the availability of this material at the NARA, call (202) 741-6030, or go to [http://www.archives.gov/federal\\_register/code\\_of\\_federal\\_regulations/ibr\\_locations.html](http://www.archives.gov/federal_register/code_of_federal_regulations/ibr_locations.html).

Issued in Renton, Washington, on January 10, 2006.

**Ali Bahrami,**

Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 06-466 Filed 1-19-06; 8:45 am]

BILLING CODE 4910-13-P

## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 39

[Docket No. FAA-2005-22917; Directorate Identifier 2005-NM-157-AD; Amendment 39-14456; AD 2006-02-04]

RIN 2120-AA64

#### Airworthiness Directives; Bombardier Model CL-600-1A11 (CL-600), CL-600-2A12 (CL-601), and CL-600-2B16 (CL-601-3A, CL-601-3R, and CL-604) Airplanes

**AGENCY:** Federal Aviation Administration (FAA), Department of Transportation (DOT).

**ACTION:** Final rule.

**SUMMARY:** The FAA is adopting a new airworthiness directive (AD) for certain Bombardier Model CL-600-1A11 (CL-600), CL-600-2A12 (CL-601), and CL-600-2B16 (CL-601-3A, CL-601-3R, & CL-604) airplanes. This AD requires modifying the rudder balance spring assembly by installing a new adjustable

balance spring, and rigging the assembly to suit the rudder of each airplane. This AD results from production inspections that showed that the spring assembly that controls rudder balance may not have the correct pre-load on some airplanes. We are issuing this AD to prevent uncommanded yaw movements and consequent reduced controllability of the airplane.

**DATES:** This AD becomes effective February 24, 2006.

The Director of the Federal Register approved the incorporation by reference of certain publications listed in the AD as of February 24, 2006.

**ADDRESSES:** You may examine the AD docket on the Internet at <http://dms.dot.gov> or in person at the Docket Management Facility, U.S. Department of Transportation, 400 Seventh Street, SW., Nassif Building, room PL-401, Washington, DC.

Contact Bombardier, Inc., Canadair, Aerospace Group, P.O. Box 6087, Station Centre-ville, Montreal, Quebec H3C 3G9, Canada, for service information identified in this AD.

#### FOR FURTHER INFORMATION CONTACT:

Daniel Parrillo, Aerospace Engineer, Systems and Flight Test Branch, ANE-172, FAA, New York Aircraft Certification Office, 1600 Stewart Avenue, Suite 410, Westbury, New York 11590; telephone (516) 228-7305; fax (516) 794-5531.

#### SUPPLEMENTARY INFORMATION:

##### Examining the Docket

You may examine the airworthiness directive (AD) docket on the Internet at <http://dms.dot.gov> or in person at the Docket Management Facility office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The Docket Management Facility office (telephone (800) 647-5227) is located on the plaza level of the Nassif Building at the street address stated in the **ADDRESSES** section.

##### Discussion

The FAA issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 to include an AD that would apply to certain Bombardier Model CL-600-1A11 (CL-600), CL-600-2A12 (CL-601), and CL-600-2B16 (CL-601-3A, CL-601-3R, and CL-604) airplanes. That NPRM was published in the **Federal Register** on November 10, 2005 (70 FR 68377). That NPRM proposed to require modifying the rudder balance spring assembly by installing a new adjustable balance spring, and rigging the assembly to suit the rudder of each airplane.

#### Comments

We provided the public the opportunity to participate in the development of this AD. We received no comments on the NPRM or on the determination of the cost to the public.

#### Explanation of Change to Related Information

We have revised paragraph (k) of this AD to include reference to Canadian airworthiness directive CF-2005-21R1, dated November 23, 2005, which was issued to revise the applicability of the affected airplanes. This change was reflected in the applicability of the proposed AD.

#### Conclusion

We have carefully reviewed the available data and determined that air safety and the public interest require adopting the AD with the change described previously. We have determined that this change will neither increase the economic burden on any operator nor increase the scope of the AD.

#### Costs of Compliance

This AD will affect about 501 airplanes of U.S. registry. The required actions will take about 12 work hours per airplane, at an average labor rate of \$65 per work hour. Required parts will cost about \$1,749 per airplane. Based on these figures, the estimated cost of the AD for U.S. operators is \$1,267,029, or \$2,529 per airplane.

#### Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. Subtitle VII, Aviation Programs, describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in subtitle VII, part A, subpart III, section 44701, "General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

#### Regulatory Findings

We have determined that this AD will not have federalism implications under Executive Order 13132. This AD will

not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that this AD:

- (1) Is not a "significant regulatory action" under Executive Order 12866;
- (2) Is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and
- (3) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared a regulatory evaluation of the estimated costs to comply with this AD and placed it in the AD docket. See the **ADDRESSES** section for a location to examine the regulatory evaluation.

**List of Subjects in 14 CFR Part 39**

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

**Adoption of the Amendment**

Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

**PART 39—AIRWORTHINESS DIRECTIVES**

1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

**§ 39.13 [Amended]**

2. The Federal Aviation Administration (FAA) amends § 39.13

by adding the following new airworthiness directive (AD):

**2006-02-04 Bombardier, Inc. (Formerly Canadair):** Amendment 39-14456. Docket No. FAA-2005-22917; Directorate Identifier 2005-NM-157-AD.

**Effective Date**

(a) This AD becomes effective February 24, 2006.

**Affected ADs**

(b) None.

**Applicability**

(c) This AD applies to Bombardier Model CL-600-1A11 (CL-600), CL-600-2A12 (CL-601), and CL-600-2B16 (CL-601-3A, CL-601-3R, and CL-604) airplanes, certificated in any category; as identified in Table 1 of this AD.

TABLE 1.—AFFECTED AIRPLANES BY SERIAL NUMBER

Bombardier airplane model	Affected serial numbers
CL-600-1A11 (CL-600) .....	1004 through 1085 inclusive.
CL-600-2A12 (CL-601) .....	3001 through 3066 inclusive.
CL-600-2B16 (CL-601-3A and CL-601-3R) .....	5001 through 5194 inclusive.
CL-600-2B16 (CL-604) .....	5301 through 5564 inclusive.

**Unsafe Condition**

(d) This AD results from production inspections that showed that the spring assembly that controls rudder balance may not have the correct pre-load on some airplanes. We are issuing this AD to prevent uncommanded yaw movements and

consequent reduced controllability of the airplane.

**Compliance**

(e) You are responsible for having the actions required by this AD performed within the compliance times specified, unless the actions have already been done.

**Service Bulletin Reference**

(f) The term "service bulletin," as used in this AD, means the Accomplishment Instructions of the applicable service bulletin in Table 2 of this AD.

TABLE 2.—SERVICE BULLETINS

Bombardier airplane model	Bombardier service bulletin
CL-600-1A11 (CL-600) .....	600-0714, including Appendix 1 and excluding Appendix 2, dated April 4, 2003.
CL-600-2A12 (CL-601), and CL-600-2B16 (CL-601-3A and CL-601-3R) .....	(601-0549, including Appendix 1 and excluding Appendix 2, dated April 4, 2003.
CL-600-2B16 (CL-604) .....	604-27-013, including Appendix 1 and excluding Appendix 2, dated April 4, 2003.

**Modification and Rigging**

(g) Within 12 months after the effective date of this AD: Modify and rig the adjustable rudder balance spring assembly for the rudder control surface, in accordance with the Accomplishment Instructions of the applicable service bulletin specified in Table 2 of this AD. Where the service bulletin specifies contacting Bombardier for instructions on making certain adjustments: Before further flight, adjust according to a method approved by the Manager, New York Aircraft Certification Office (ACO), FAA; or Transport Canada Civil Aviation (TCCA) (or its delegated agent).

**No Reporting Required**

(h) Although the service bulletins referenced in this AD specify to submit certain information to the manufacturer, this AD does not include that requirement.

**Parts Installation**

(i) After the effective date of this AD, no person may install on any airplane a rudder balance spring assembly unless it has been modified and rigged in accordance with paragraph (g) of this AD.

**Alternative Methods of Compliance (AMOCs)**

(j)(1) The Manager, New York ACO, has the authority to approve AMOCs for this AD, if

requested in accordance with the procedures found in 14 CFR 39.19.

(2) Before using any AMOC approved in accordance with § 39.19 on any airplane to which the AMOC applies, notify the appropriate principal inspector in the FAA Flight Standards Certificate Holding District Office.

**Related Information**

(k) Canadian airworthiness directives CF-2005-21, dated June 23, 2005, and CF-2005-21R1, dated November 23, 2005, also address the subject of this AD.

**Material Incorporated by Reference**

(1) You must use the service documents identified in Table 3 of this AD to perform the actions that are required by this AD, unless the AD specifies otherwise. The Director of the Federal Register approved the incorporation by reference of these documents in accordance with 5 U.S.C.

552(a) and 1 CFR part 51. Contact Bombardier, Inc., Canadair, Aerospace Group, P.O. Box 6087, Station Centre-ville, Montreal, Quebec H3C 3G9, Canada, for a copy of this service information. You may review copies at the Docket Management Facility, U.S. Department of Transportation, 400 Seventh Street, SW., room PL-401,

Nassif Building, Washington, DC; on the Internet at <http://dms.dot.gov>; or at the National Archives and Records Administration (NARA). For information on the availability of this material at the NARA, call (202) 741-6030, or go to [http://www.archives.gov/federal\\_register/code\\_of\\_federal\\_regulations/ibr\\_locations.html](http://www.archives.gov/federal_register/code_of_federal_regulations/ibr_locations.html).

TABLE 3.—MATERIAL INCORPORATED BY REFERENCE

Bombardier service bulletin	Date
600-0714, including Appendix 1 and excluding Appendix 2	April 4, 2003.
601-0549, including Appendix 1 and excluding Appendix 2	April 4, 2003.
604-27-013, including Appendix 1 and excluding Appendix 2	April 4, 2003.

Issued in Renton, Washington, on January 10, 2006.

**Ali Bahrami,**

Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 06-467 Filed 1-19-06; 8:45 am]

BILLING CODE 4910-13-P

**DEPARTMENT OF TRANSPORTATION**

**Federal Aviation Administration**

**14 CFR Part 39**

[Docket No. FAA-2005-22810; Directorate Identifier 2005-NM-143-AD; Amendment 39-14458; AD 2006-02-06]

RIN 2120-AA64

**Airworthiness Directives; Airbus Model A310-203, -204, and -222 Airplanes, and Model A310-300 Series Airplanes**

**AGENCY:** Federal Aviation Administration (FAA), Department of Transportation (DOT).

**ACTION:** Final rule.

**SUMMARY:** The FAA is adopting a new airworthiness directive (AD) for certain Airbus Model A310-203, -204, and -222 airplanes, and Model A310-300 series airplanes. This AD requires a one-time rototest inspection for cracking of the frame foot and adjacent frames and skin in the area surrounding the frame foot run-outs from fuselage frames (FR) 43 through FR 46, and repair if necessary. This AD also requires modification of certain fastener holes. This AD results from a structural evaluation of Model A310 airplanes for widespread fatigue damage of the frame foot run-outs from FR 43 through FR 46. The evaluation revealed that, on in-service airplanes, undetected cracking in this area can lead to the rupture of the frame foot and subsequent cracking of the adjacent frames and fuselage skin. We are issuing this AD to prevent fatigue cracking of the frame foot run-outs, which could lead to rupture of the

frame foot and cracking in adjacent frames and skin, and result in reduced structural integrity of the fuselage.

**DATES:** This AD becomes effective February 24, 2006.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in the AD as of February 24, 2006.

**ADDRESSES:** You may examine the AD docket on the Internet at <http://dms.dot.gov> or in person at the Docket Management Facility, U.S. Department of Transportation, 400 Seventh Street, SW., Nassif Building, Room PL-401, Washington, DC.

Contact Airbus, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France, for service information identified in this AD.

**FOR FURTHER INFORMATION CONTACT:** Dan Rodina, Aerospace Engineer, International Branch, ANM-116, Transport Airplane Directorate, FAA, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 227-2125; fax (425) 227-1149.

**SUPPLEMENTARY INFORMATION:**

**Examining the Docket**

You may examine the airworthiness directive (AD) docket on the Internet at <http://dms.dot.gov> or in person at the Docket Management Facility office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The Docket Management Facility office (telephone (800) 647-5227) is located on the plaza level of the Nassif Building at the street address stated in the **ADDRESSES** section.

**ADDRESSES** section.

**Discussion**

The FAA issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 to include an AD that would apply to certain Airbus Model A310-203, -204, and -222 airplanes, and Model A310-300 series airplanes. That NPRM was published in the **Federal Register** on October 27, 2005 (70 FR

61924). That NPRM proposed to require a one-time rototest inspection for cracking of the frame foot and adjacent frames and skin in the area surrounding the frame foot run-outs from fuselage frames (FR) 43 through FR 46, and repair if necessary. The NPRM also proposed to require modification of certain fastener holes.

**Comments**

We provided the public the opportunity to participate in the development of this AD. We have considered the single comment received.

**Request for Clarification of Service Bulletin Reference**

The commenter states that a reference to Airbus Service Bulletin A310-53-2124, dated April 4, 2005, was omitted in paragraph (f) of the NPRM. We agreed with the commenter. We inadvertently omitted the service bulletin number in paragraph (f); that error has been corrected in this final rule.

**Conclusion**

We have carefully reviewed the available data, including the comment received, and determined that air safety and the public interest require adopting the AD with the change described previously. This change will neither increase the economic burden on any operator nor increase the scope of the AD.

**Costs of Compliance**

This AD will affect about 59 airplanes of U.S. registry. The actions will take about 31 work hours per airplane, at an average labor rate of \$65 per work hour. Required parts will cost about \$1,730 per kit (two kits per airplane). Based on these figures, the estimated cost of the AD for U.S. operators is \$323,025, or \$5,475 per airplane.