

For the reasons discussed above, I certify that the proposed regulation:

1. Is not a "significant regulatory action" under Executive Order 12866;
2. Is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and
3. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared a regulatory evaluation of the estimated costs to comply with this proposed AD. See the **ADDRESSES** section for a location to examine the regulatory evaluation.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

The Proposed Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA proposes to amend 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. The FAA amends § 39.13 by adding the following new airworthiness directive (AD):

McDonnell Douglas: Docket No. FAA-2005-20882; Directorate Identifier 2004-NM-241-AD.

Comments Due Date

(a) The Federal Aviation Administration (FAA) must receive comments on this AD action by May 23, 2005.

Affected ADs

- (b) None.

Applicability

(c) This AD applies to the airplanes identified in Table 1 of this AD; certificated in any category.

TABLE 1.—APPLICABILITY

Airplane model	Applicable service bulletin
DC-10-10, DC-10-10F, DC-10-15, DC-10-30, DC-10-30F airplanes (KC-10A and KDC-10), DC-10-40, DC-10-40F, MD-10-10F, MD-10-30F airplanes.	McDonnell Douglas Service Bulletin DC10-52-219, Revision 1, dated September 3, 2004.
MD-11 and MD-11F airplanes	McDonnell Douglas Service Bulletin MD11-52-044, Revision 1, dated September 3, 2004.

Unsafe Condition

(d) This AD was prompted by a report indicating that the exterior emergency control mechanism handles of the mid, overwing and aft passenger doors were inoperative. We are issuing this AD to prevent failure of the passenger doors to operate properly in an emergency condition, which could delay an emergency evacuation and possibly result in injury to passengers and flightcrew.

Compliance

(e) You are responsible for having the actions required by this AD performed within the compliance times specified, unless the actions have already been done.

Service Bulletin Reference

(f) The term "service bulletin," as used in this AD, means the Accomplishment Instructions of McDonnell Douglas Service Bulletin MD11-52-044, Revision 1, and Service Bulletin DC10-52-219, Revision 1; both dated September 3, 2004; as applicable.

Functional Test

(g) Within 6,000 flight hours or 18 months after the effective date of this AD, whichever occurs later, perform a functional test of the exterior emergency control handle assemblies of the mid, overwing, and aft passenger doors; by doing all actions specified in the applicable service bulletin, except as provided by paragraph (i) of this AD.

(1) If the functional test reveals no noisy operation or binding: Repeat the functional test at intervals not to exceed 6,000 flight hours or 18 months, whichever occurs later, until the terminating action of paragraph (h) of this AD has been accomplished.

(2) If any functional test required by this AD reveals noisy operation or binding: Prior

to further flight, replace the steel bearings with bearings made from corrosion-resistant material in accordance with the applicable service bulletin.

Optional Terminating Action

(h) Accomplishment of the actions required by paragraph (g)(2) of this AD constitutes terminating action for the repetitive tests required by paragraph (g)(1) of this AD only for the modified doors.

Inoperable Doors

(i) Any mid, overwing, or aft passenger door that has been fastened shut and rendered inoperable according to some approved airplane freighter configuration is not subject to the requirements of this AD.

Alternative Methods of Compliance (AMOCs)

(j) The Manager, Los Angeles Aircraft Certification Office (ACO), FAA, has the authority to approve AMOCs for this AD, if requested in accordance with the procedures found in 14 CFR 39.19.

Issued in Renton, Washington, on March 31, 2005.

Kalene C. Yanamura,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.
[FR Doc. 05-6910 Filed 4-6-05; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA-2004-19563; Directorate Identifier 2003-NM-10-AD]

RIN 2120-AA64

Airworthiness Directives; Bombardier Model CL-600-2B16 (CL-604) Series Airplanes

AGENCY: Federal Aviation Administration (FAA), Department of Transportation (DOT).

ACTION: Proposed rule; withdrawal.

SUMMARY: The FAA withdraws a notice of proposed rulemaking (NPRM) that proposed a new airworthiness directive (AD) for certain Bombardier Model CL-600-2B16 (CL-604) series airplanes. The proposed AD would have required replacing the side-brace fitting shafts of the main landing gear (MLG) with new, improved side-brace fitting shafts; inspecting for corrosion of the MLG side-brace fitting shafts; and replacing the nut, washer, and cotter pin of the MLG side-brace fitting shafts with new parts; as applicable. Since the proposed AD was issued, we have received new data that the actions that would have been required by the proposed AD have already been accomplished on all of the affected airplanes. Accordingly, the proposed AD is withdrawn.

ADDRESSES: You can examine the contents of this AD docket on the Internet at <http://dms.dot.gov>, or in person at the Docket Management Facility, U.S. Department of Transportation, 400 Seventh Street, SW., room PL-401, on the plaza level of the Nassif Building, Washington, DC. This docket number is FAA-2004-19563; the directorate identifier for this docket is 2003-NM-10-AD.

FOR FURTHER INFORMATION CONTACT: Serge Napoleon, Aerospace Engineer, Airframe and Propulsion Branch, ANE-171, FAA, New York Aircraft Certification Office, 1600 Stewart Avenue, suite 410, Westbury, New York 11590; telephone (516) 228-7312; fax (516) 794-5531.

SUPPLEMENTARY INFORMATION:

Discussion

We proposed to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) with a notice of proposed rulemaking (NPRM) for a new AD for certain Bombardier Model CL-600-2B16 (CL-604) series airplanes. That NPRM was published in the **Federal Register** on November 9, 2004 (69 FR 64869). The NPRM would have required replacing the side-brace fitting shafts of the main landing gear (MLG) with new, improved side-brace fitting shafts; inspecting for corrosion of the MLG side-brace fitting shafts; and replacing the nut, washer, and cotter pin of the MLG side-brace fitting shafts with new parts; as applicable. The NPRM was prompted by the discovery of fractures of the MLG side-brace fitting shafts caused by corrosion on the forward side of the side-brace fitting shafts. The proposed actions were intended to prevent fracture of the MLG side-brace fitting shafts, which could result in collapse of the MLG.

Actions Since NPRM Was Issued

Since we issued the NPRM, the airplane manufacturer has informed us that all airplanes identified in the applicability section of the NPRM have already accomplished the actions specified in Bombardier Alert Service Bulletin A604-32-018, Revision 01, dated February 22, 2002, which would have been required by the proposed AD. Transport Canada Civil Aviation (TCCA), which is the airworthiness authority for Canada, has also confirmed that the proposed requirements have already been accomplished on all affected airplanes.

FAA's Conclusions

Upon further consideration, we have determined that it is unnecessary to

issue the proposed AD. Accordingly, the NPRM is withdrawn.

Withdrawal of the NPRM does not preclude the FAA from issuing another related action or commit the FAA to any course of action in the future.

Regulatory Impact

Since this action only withdraws an NPRM, it is neither a proposed nor a final rule and therefore is not covered under Executive Order 12866, the Regulatory Flexibility Act, or DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979).

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

The Withdrawal

Accordingly, we withdraw the NPRM, Docket No. FAA-2004-19563, Directorate Identifier 2003-NM-10-AD, which was published in the **Federal Register** on November 9, 2004 (69 FR 64869).

Issued in Renton, Washington, on March 31, 2005.

Kalene C. Yanamura,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA-2005-20863; Directorate Identifier 2004-SW-36-AD]

RIN 2120-AA64

Airworthiness Directives; Eurocopter France Model AS350B, BA, B1, B2, B3, D, and AS355E Helicopters

AGENCY: Federal Aviation Administration, DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: This document proposes adopting a new airworthiness directive (AD) for the specified Eurocopter France (ECF) model helicopters. This proposal would require replacing the hydraulic fluid at a specified time interval when operating in cold weather. This proposal is prompted by reports of ice forming due to condensation in some parts of the hydraulic system during cold weather operation. The actions specified by this proposed AD are intended to prevent ice from forming in the hydraulic system resulting in an unintended movement of

the flight controls and subsequent loss of control of the helicopter.

DATES: Comments must be received on or before June 6, 2005.

ADDRESSES: Use one of the following addresses to submit comments on this proposed AD:

- DOT Docket Web site: Go to <http://dms.dot.gov> and follow the instructions for sending your comments electronically;
- Government-wide rulemaking Web site: Go to <http://www.regulations.gov> and follow the instructions for sending your comments electronically;
- Mail: Docket Management Facility; U.S. Department of Transportation, 400 Seventh Street, SW., Nassif Building, Room PL-401, Washington, DC 20590;
- Fax: 202-493-2251; or
- Hand Delivery: Room PL-401 on the plaza level of the Nassif Building, 400 Seventh Street, SW., Washington, DC, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

You may get the service information identified in this proposed AD from American Eurocopter Corporation, 2701 Forum Drive, Grand Prairie, Texas 75053-4005, telephone (972) 641-3460, fax (972) 641-3527.

You may examine the comments to this proposed AD in the AD docket on the Internet at <http://dms.dot.gov>.

FOR FURTHER INFORMATION CONTACT: Ed Cuevas, Aviation Safety Engineer, FAA, Rotorcraft Directorate, Safety Management Group, Fort Worth, Texas 76193-0111, telephone (817) 222-5355, fax (817) 222-5961.

SUPPLEMENTARY INFORMATION:

Comments Invited

We invite you to submit any written data, views, or arguments regarding this proposed AD. Send your comments to the address listed under the caption **ADDRESSES**. Include the docket number "FAA-2005-20863, Directorate Identifier 2004-SW-36-AD" at the beginning of your comments. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of the proposed AD. We will consider all comments received by the closing date and may amend the proposed AD in light of those comments.

We will post all comments we receive, without change, to <http://dms.dot.gov>, including any personal information you provide. We will also post a report summarizing each substantive verbal contact with FAA personnel concerning this proposed rulemaking. Using the search function of our docket Web site, you can find and read the comments to any of our