

Proposed Rules

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This section of the FEDERAL REGISTER contains notices to the public of the proposed issuance of rules and regulations. The purpose of these notices is to give interested persons an opportunity to participate in the rule making prior to the adoption of the final rules.

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2003-NE-25-AD]

RIN 2120-AA64

Airworthiness Directives; Pratt & Whitney Canada PW206A and PW206E Turboshaft Engines

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: The FAA proposes to supersede an existing airworthiness directive (AD) for Pratt & Whitney Canada (PWC) PW206A and PW206E turboshaft engines. That AD currently requires initial and repetitive borescope inspections of compressor turbine and power turbine blades for blade axial shift, and replacement of blade retaining rivets and certain rotor air seals as terminating action for the repetitive borescope inspections.

This proposed AD would require the same actions but needs to clarify the extent of engine disassembly that triggers the required part replacements. This proposed AD is prompted by reports of engine shutdowns and emergency landings due to severe vibration, resulting in exhaust gases escaping from the engine-to-exhaust nozzle interface, thereby triggering in-flight engine fire warnings. We are proposing this AD to prevent turbine blade axial shift, which could cause high levels of vibration, loss of engine torque, in-flight engine shutdown, and loss of the airframe exhaust duct.

DATES: We must receive any comments on this proposed AD by April 20, 2004.

ADDRESSES: Use one of the following addresses to submit comments on this proposed AD:

- By mail: Federal Aviation Administration (FAA), New England Region, Office of the Regional Counsel, Attention: Rules Docket No. 2003-NE-

25-AD, 12 New England Executive Park, Burlington, MA 01803-5299.

- By fax: (781) 238-7055.
- By e-mail: 9-ane-adcomment@faa.gov

You can get the service information identified in this proposed AD from Pratt & Whitney Canada, 1000 Marie-Victorin, Longueuil, Quebec, Canada J4G1A1.

You may examine the AD docket, by appointment, at the FAA, New England Region, Office of the Regional Counsel, 12 New England Executive Park, Burlington, MA.

FOR FURTHER INFORMATION CONTACT: Ian Dargin, Aerospace Engineer, Engine Certification Office, FAA, Engine and Propeller Directorate, 12 New England Executive Park, Burlington, MA 01803-5299; telephone (781) 238-7178; fax (781) 238-7199.

SUPPLEMENTARY INFORMATION:

Comments Invited

We invite you to submit any written relevant data, views, or arguments regarding this proposal. Send your comments to an address listed under **ADDRESSES**. Include "AD Docket No. 2003-NE-25-AD" in the subject line of your comments. If you want us to acknowledge receipt of your mailed comments, send us a self-addressed, stamped postcard with the docket number written on it; we will date-stamp your postcard and mail it back to you. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of the proposed AD. If a person contacts us through a nonwritten communication, and that contact relates to a substantive part of this proposed AD, we will summarize the contact and place the summary in the docket. We will consider all comments received by the closing date and may amend the proposed AD in light of those comments.

We are reviewing the writing style we currently use in regulatory documents. We are interested in your comments on whether the style of this document is clear, and your suggestions to improve the clarity of our communications that affect you. You may get more information about plain language at <http://www.faa.gov/language> and <http://www.plainlanguage.gov>.

Examining the AD Docket

You may examine the AD Docket (including any comments and service information), by appointment, between 8 a.m. and 4:30 p.m., Monday through Friday, except Federal holidays. See **ADDRESSES** for the location.

Discussion

On August 4, 2003, the FAA issued airworthiness directive (AD) 2003-16-10, Amendment 39-13263 (68 FR 48544, August 14, 2003), to require initial and repetitive borescope inspections of compressor turbine and power turbine blades for blade axial shift. That AD also required replacement of blade retaining rivets and certain rotor air seals as terminating action for the repetitive borescope inspections. That action was prompted by reports of engine shutdowns and emergency landings due to severe vibration and drops in engine torque, and an increase in internal engine temperature, triggering in-flight engine fire warnings. That condition, if not corrected, could result in turbine blade axial shift, which could cause high levels of vibration, loss of engine torque, in-flight engine shutdown, and loss of the airframe exhaust duct.

Comments Received Since AD 2003-16-10 Was Issued

Since that final rule; request for comments was issued, we received two comments on that AD. We have considered those comments.

Request To Clarify the Extent of Engine Disassembly Required

One commenter requests clarification in the AD of the extent of engine disassembly that would trigger the part replacements and clarification of the rework specified in the terminating action. The commenter states that more extensive disassembly is required to do the part replacement specified in Part B of PWC SB No. 200-72-28069, Revision 5, dated February 10, 2003, than to do the part rework specified in Part A of that SB. The commenter also states that the triggering event of a shop visit for any reason is too restrictive.

The FAA agrees. We have rewritten the terminating action to be done at the next engine shop visit when access is available to subassemblies, such as modules, accessories, and components, or at the next engine overhaul,

whichever occurs first, but before accumulating 1,800 flight hours from the effective date of this AD or before December 31, 2009, whichever occurs first.

Request To Clarify the Preamble

One commenter requests clarification of what prompted the AD. The commenter suggests that the words describing the actions prompting this AD be changed to state that the AD is being issued to prevent turbine blade axial shift, leading to high levels of vibration and possible in-flight engine shutdown.

The FAA agrees to the suggested changes to the preamble, which have been incorporated into this document. There have been six emergency landings due to high vibration levels and in-flight engine fire warnings, one incident of the loss of the airframe exhaust duct, one in-flight shutdown, one pilot report of high oil consumption, and one pilot report of loss of torque. To date, there have been no failures that have resulted in uncontained engine failures. The aircraft warning and detection system should preclude uncontained engine failures from occurring. We agree that the loss of the airframe exhaust duct should have been referenced in the preamble to the current AD. Therefore, the preamble of this proposal is written to reference the loss of the airframe exhaust duct.

Correction To Include No. 4 Bearing Rear Rotor Air Seal

The reference to replacing the No. 4 bearing rear rotor air seal was inadvertently omitted from the compliance section of the AD. We have rewritten paragraph (i) of the AD to include replacing of the No. 4 bearing rear rotor air seal.

Relevant Service Information

We have reviewed and approved the technical contents of the following Pratt & Whitney Canada service documents:

- Alert Service Bulletin (ASB) No. PW200-72-A28242, Revision 1, dated October 2, 2002, that describes procedures for borescope inspecting of compressor turbine blades and power turbine blades for axial shift within the disks.
- Service Bulletin (SB) No. PW200-72-28069, Revision 5, dated February 10, 2003, that describes procedures for replacing compressor turbine blade retaining rivets, the No. 3 bearing rotor air seal, and the No. 4 bearing front rotor air seal.
- SB No. PW200-72-28239, Revision 2, dated February 10, 2003, that describes procedures for replacing

power turbine blade retaining rivets. Transport Canada, which is the airworthiness authority for Canada, classified these service bulletins as mandatory and issued AD CF-2003-06, dated February 4, 2003, in order to ensure the airworthiness of these PWC PW206A and PW206E turboshaft engines in Canada.

Bilateral Agreement Information

This engine model is manufactured in Canada and is type certificated for operation in the United States under the provisions of section 21.29 of the Federal Aviation Regulations (14 CFR 21.29) and the applicable bilateral airworthiness agreement. Under this bilateral airworthiness agreement, Transport Canada has kept the FAA informed of the situation described above. We have examined the findings of Transport Canada, reviewed all available information, and determined that AD action is necessary for products of this type design that are certificated for operation in the United States.

FAA's Determination and Requirements of the Proposed AD

We have evaluated all pertinent information and identified an unsafe condition that is likely to exist or develop on other products of this same type design. Therefore, we are proposing this AD, which would require:

- Initial and repetitive borescope inspections of compressor turbine blades and power turbine blades for blade axial shift within the turbine disks; and
- Replacement of blade retaining rivets, the No. 3 bearing rotor air seal, and the No. 4 bearing front rotor air seal as mandatory terminating action for the repetitive borescope inspections.

You must use the service information described previously to perform the actions required by this AD.

Changes to 14 CFR Part 39—Effect on the Proposed AD

On July 10, 2002, we issued a new version of 14 CFR part 39 (67 FR 47998, July 22, 2002), which governs the FAA's AD system. This regulation now includes material that relates to altered products, special flight permits, and alternative methods of compliance. This material previously was included in each individual AD. Since this material is included in 14 CFR part 39, we will not include it in future AD actions.

Costs of Compliance

There are about 130 PWC PW206A and PW206E turboshaft engines of the affected design in the worldwide fleet.

We estimate that 15 engines installed on airplanes of U.S. registry would be affected by this proposed AD. We also estimate that it would take about 0.5 work hours per engine to perform the proposed actions, and that the average labor rate is \$65 per work hour. Required parts would cost about \$9,077 per engine. Based on these figures, we estimate the total cost of the proposed AD to U.S. operators to be \$136,656. The manufacturer has stated that it may provide replacement parts at no cost to operators.

Regulatory Findings

We have determined that this proposed AD would not have federalism implications under Executive Order 13132. This proposed AD would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that the proposed regulation:

1. Is not a "significant regulatory action" under Executive Order 12866;
2. Is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and
3. Would not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared a summary of the costs to comply with this proposal and placed it in the AD Docket. You may get a copy of this summary by sending a request to us at the address listed under **ADDRESSES**. Include "AD Docket No. 2003-NE-25-AD" in your request.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

The Proposed Amendment

Accordingly, under the authority delegated to me by the Administrator, the Federal Aviation Administration proposes to amend 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. The FAA amends § 39.13 by removing Amendment 39-13263 (68 FR 48544, August 14, 2003) and by adding a new airworthiness directive, to read as follows:

Pratt & Whitney Canada: Docket No. 2003–NE–25–AD. Supersedes AD 2003–16–10, Amendment 39–13263.

Comments Due Date

(a) The Federal Aviation Administration (FAA) must receive comments on this airworthiness directive (AD) action by April 20, 2004.

Affected ADs

(b) This AD supersedes AD 2003–16–10, Amendment 39–13263.

Applicability

(c) This AD applies to Pratt & Whitney Canada (PWC) PW206A and PW206E turboshaft engines. These PWC engines are installed on, but not limited to, MD Helicopters Inc. Model MD–900 helicopters.

Unsafe Condition

(d) This AD is prompted by the need to clarify the extent of engine disassembly that triggers the required part replacements. This AD is also prompted by reports of engine shutdowns and emergency landings due to severe vibration, resulting in exhaust gases escaping from the engine-to-exhaust nozzle interface, thereby triggering in-flight engine fire warnings. The actions specified in this AD are intended to prevent turbine blade axial shift, leading to high levels of vibration, in-flight engine shutdowns and loss of the airframe exhaust duct.

Compliance

(e) You are responsible for having the actions required by this AD performed within the compliance times specified unless the actions have already been done.

Initial Sequence of Borescope Inspections

(f) Perform an initial sequence of borescope inspections of compressor turbine blades and power turbine blades for blade axial shift within the turbine disks. Use paragraph 3. of Accomplishment Instructions of PWC Alert Service Bulletin (ASB) No. PW200–72–A28242, Revision 1, dated October 2, 2002, for the borescope inspection and determination of blade shift. Do the inspections at the following times:

(1) Within 25 flight hours accumulated, or 30 days after the effective date of this AD, whichever occurs earlier.

(2) After 30 flight hours, but before 50 flight hours accumulated since inspection of paragraph (f)(1) of this AD.

(3) After 80 flight hours, but before 100 flight hours accumulated since inspection of paragraph (f)(1) of this AD.

(4) After 180 flight hours, but before 200 flight hours accumulated since inspection of paragraph (f)(1) of this AD.

Repetitive Borescope Inspections

(g) Thereafter, perform repetitive borescope inspections at intervals of not less than 280 nor more than 300 flight hours since-last-inspection. Use paragraph 3. of Accomplishment Instructions of PWC ASB No. PW200–72–A28242, Revision 1, dated October 2, 2002, for the borescope inspections and determination of blade shift.

Disposition

(h) If you find any blade shift, remove engine from service before further flight and perform rivet and rotor air seal replacements, as specified in paragraphs (i)(1) through (i)(3) of this AD, to return the engine to service.

Terminating Action

(i) At the next engine shop visit when access is available to subassemblies, such as modules, accessories, and components, or at the next engine overhaul, whichever occurs first, but before accumulating 1,800 flight hours from the effective date of this AD or before December 31, 2009, whichever occurs first, do the following:

(1) Replace the compressor turbine blade retaining rivets with new P/N retaining rivets, and the No. 4 bearing rear rotor air seal with the new P/N No. 4 bearing rear rotor air seal. Use paragraph 3., Part A, of Accomplishment Instructions of SB No. PW200–72–28069, Revision 5, dated February 10, 2003.

(2) Replace the No. 3 bearing rotating air seal with the new P/N air seal, and the No. 4 bearing front rotor air seal with the new P/N No. 4 bearing front rotor air seal. Use paragraph 3., Part B, of Accomplishment Instructions of SB No. PW200–72–28069, Revision 5, dated February 10, 2003.

(3) Replace the power turbine blade retaining rivets with new P/N power turbine blade retaining rivets. Use paragraph 3. of Accomplishment Instructions of SB No. PW200–72–28239, Revision 2, dated February 10, 2003.

(j) Completing the actions in paragraphs (i)(1) through (i)(3) of this AD terminate all inspection requirements of this AD.

Previous Credit

(k) Previous credit is allowed:

(1) For performing the initial sequence for borescope inspections in paragraph (f) of this AD, that were done using AD 2003–16–10.

(2) For terminating action in paragraphs (i)(1) through (i)(3) of this AD that was done using Accomplishment Instructions of SB No. PW200–72–28069, Revision 4, dated December 27, 2000, and Accomplishment Instructions of SB No. PW200–72–28239, dated September 5, 2002, or Revision 1, dated December 5, 2002, before the effective date of this AD.

Alternative Methods of Compliance

(l) The Manager, Engine Certification Office, has the authority to approve alternative methods of compliance for this AD if requested using the procedures found in 14 CFR 39.19.

Material Incorporated by Reference

(m) You must use the following Pratt & Whitney Canada Service Bulletins and Alert Service Bulletin to perform the inspections and replacement actions required by this AD. The Director of the Federal Register approved the incorporation by reference of the documents listed in Table 1 of this AD as of August 29, 2003 (68 FR 48544, August 14, 2003), in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. You may get a copy from Pratt & Whitney Canada, 1000 Marie-Victorin, Longueuil, Quebec, Canada J4G1A1. You may review copies at Federal Aviation Administration (FAA), New England Region, Office of the Regional Counsel, Attention: Rules Docket No. 2003–NE–25–AD, 12 New England Executive Park, Burlington, MA 01803–5299; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC. Table 1 follows:

TABLE 1.—INCORPORATION BY REFERENCE

Service bulletin	Page number(s)	Revision	Date
PW200–72–A28242	All	1	October 2, 2002.
Total Pages—7.			
PW200–72–28069	All	5	February 10, 2003.
Total Pages—17.			
PW200–72–28239	All	2	February 10, 2003.
Total Pages—20.			

Related Information

(n) Transport Canada issued airworthiness directive CF-2003-06, dated February 4, 2003, which pertains to the subject of this AD, in order to assure the airworthiness of these PWC PW206A and PW206E turboshaft engines in Canada.

Issued in Burlington, Massachusetts, on February 13, 2004.

Peter A. White,

Acting Manager, Engine and Propeller Directorate, Aircraft Certification Service.

[FR Doc. 04-3682 Filed 2-19-04; 8:45 am]

BILLING CODE 4910-13-P

DEPARTMENT OF LABOR**Mine Safety and Health Administration****30 CFR Part 57**

RIN 1219-AB29

Diesel Particulate Matter Exposure of Underground Metal and Nonmetal Miners

AGENCY: Mine Safety and Health Administration (MSHA), Labor.

ACTION: Proposed rule; limited reopening of comment period.

SUMMARY: This document announces a limited reopening of the comment period on the notice of proposed rulemaking, published in the **Federal Register** on August 14, 2003, to obtain public comment on three new documents related to this rulemaking. We will consider these comments as we develop the final rule.

DATES: Comments must be received by April 5, 2004.

ADDRESSES: You may submit comments, identified by RIN 1219-AB29, by any of the following methods:

- Federal eRulemaking Portal: <http://www.regulations.gov>.

- E-mail: comments@msha.gov.

Include "RIN 1219-AB29" in the subject line of the message.

- Fax: (202) 693-9441.

- Mail, Hand Delivery, or Courier: MSHA, 1100 Wilson Blvd, Room 2350, Arlington, Virginia 22209.

Instructions: All comments, including any personal information contained therein, will be posted without change to <http://www.msha.gov/currentcomments.htm>.

Docket: The entire rulemaking record may be viewed in MSHA's public reading room at 1100 Wilson Boulevard, Room 2349, Arlington, Virginia.

FOR FURTHER INFORMATION CONTACT: Marvin W. Nichols, Jr., Director, Office of Standards, Regulations, and Variances, MSHA, 1100 Wilson Blvd.,

Room 2350, Arlington, Virginia 22209-3939, Nichols.Marvin@dol.gov, (202) 693-9440 (telephone), or (202) 693-9441 (facsimile).

SUPPLEMENTARY INFORMATION:**Background**

On January 19, 2001, we published a rule at 66 FR 5706 that established new health standards for underground metal and nonmetal miners by requiring use of approved equipment and low sulfur fuel, and by setting an interim and final concentration limit for diesel particulate matter (DPM) in the underground mining environment. Under a settlement agreement reached in response to legal challenges to the 2001 rule, we amended portions of the rule on February 27, 2002 (67 FR 9180), and initiated this rulemaking. We published the advance notice of proposed rulemaking (ANPRM) on September 25, 2002 (67 FR 60199), and published the proposed rule on August 14, 2003 (68 FR 48668). The proposed rule would revise the interim concentration limit; designate elemental carbon as the surrogate for measuring DPM for the interim limit; allow an extension of time in which to achieve compliance with the interim limit; apply our longstanding hierarchy of controls used for other exposure-based health standards for metal and nonmetal mines, including engineering and administrative controls supplemented by respiratory protection, but prohibit rotation of miners; and revise the requirements for the DPM control plan. Four public hearings were held on the proposed rule between September 16, 2003 and October 7, 2003. The comment period closed on October 14, 2003. The legal challenge is stayed pending completion of additional rulemaking actions.

Limited Reopening of Comment Period

We recently received new information related to this rulemaking, and concluded that it is in the public interest to obtain comments on this information. Therefore, the comment period is reopened for the limited purpose of obtaining public comment on:

- U.S. Department of Health and Human Services, Center for Disease Control, National Institute of Occupational Safety and Health, "The Effectiveness of Selected Technologies in Controlling Diesel Emissions in an Underground Mine—Isolated Zone Study at Stillwater Mining Company's Nye Mine," January 5, 2004.

In addition, two other documents have come to the Agency's attention and MSHA is also seeking comments on:

- U.S. Department of Labor, Bureau of Labor Statistics, and U.S. Department of Health and Human Services, Center for Disease Control, National Institute of Occupational Safety and Health, "Respirator Usage in Private Sector Firms, 2001," September, 2003.

- Chase, Gerald, "Characterizations of Lung Cancer in Cohort Studies and a NIOSH Study on Health Effects of Diesel Exhaust in Miners," undated, received January 5, 2004.

These documents can be accessed at <http://www.msha.gov>. We invite public comment on the findings of these documents and their impact on this rulemaking. We will disregard any comments that are outside the scope of these documents.

Dated: February 13, 2004.

Dave D. Lauriski,

Assistant Secretary of Labor for Mine Safety and Health.

[FR Doc. 04-3656 Filed 2-19-04; 8:45 am]

BILLING CODE 4510-43-P

NATIONAL ARCHIVES AND RECORDS ADMINISTRATION**36 CFR Part 1200**

RIN 3095-AB19

Official Seals and Logos

AGENCY: National Archives and Records Administration (NARA).

ACTION: Proposed rule.

SUMMARY: The National Archives and Records Administration (NARA) is proposing to modify its regulations on the use of official NARA seals by the public and other Federal agencies by extending the regulations to apply to the use of official NARA logos. This part applies to the public and other Federal agencies.

DATES: Comments are due by April 20, 2004.

ADDRESSES: NARA invites interested persons to submit comments on this proposed rule. Comments may be submitted by any of the following methods:

- Mail: Send comments to: Regulation Comments Desk (NPOL), Room 4100, Policy and Communications Staff, National Archives and Records Administration, 8601 Adelphi Road, College Park, MD 20740-6001.

- Fax: Submit comments by facsimile transmission to: 301-837-0319.

- E-mail: Send comments to <http://www.regulations.gov>. You may also comment via e-mail to

comments@nara.gov. See the

SUPPLEMENTARY INFORMATION for details.