

§ 39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:

Airbus: Docket 2002–NM–49–AD.

Applicability: Model A319, A320, and A321 series airplanes; certificated in any category; except those airplanes on which Airbus Modification 30648 has been installed.

Compliance: Required as indicated, unless accomplished previously.

To prevent failure of the MLG side-stay cuff lugs or down-lock spring attachments, which could result in improper down-lock of the MLG during a freefall extension, and possible collapse of the MLG, accomplish the following:

Inspection

(a) Do a detailed inspection of the left- and right-side main landing gear (MLG) side-stay cuff lugs and down-lock spring attachments to detect failures (cracked or fractured lugs), in accordance with the Accomplishment Instructions of Airbus Service Bulletin A320–32A1224, dated January 18, 2001, at the later of the times specified in paragraphs (a)(1) and (a)(2) of this AD.

(1) Within 60 months from the first entry into service of the MLG, or before the accumulation of 9,000 total flight hours on the MLG, whichever occurs first.

(2) Within 500 flight hours on the MLG after the effective date of this AD.

Note 1: For the purposes of this AD, a detailed inspection is defined as: “An intensive visual examination of a specific structural area, system, installation, or assembly to detect damage, failure, or irregularity. Available lighting is normally supplemented with a direct source of good lighting at intensity deemed appropriate by the inspector. Inspection aids such as mirror, magnifying lenses, etc., may be used. Surface cleaning and elaborate access procedures may be required.”

(b) If during any inspection required by paragraph (a) of this AD no crack or fracture is detected: Repeat the inspection required by paragraph (a) of this AD thereafter at intervals not to exceed 500 flight cycles until the actions specified in paragraph (e) of this AD are accomplished.

(c) If during any inspection required by paragraph (a) of this AD any crack or fracture is detected: Before further flight, replace any discrepant part with a new part of the same type in accordance with the Accomplishment Instructions of Airbus Service Bulletin A320–32A1224, dated January 18, 2001. Repeat the inspection required by paragraph (a) of this AD thereafter at intervals not to exceed 500 flight cycles until the actions specified in paragraph (e) of this AD are accomplished.

Credit for Actions Done per the Maintenance Planning Document

(d) Compliance with task number 321119.01.1, “Visual Check of Main Landing Gear Downlocking Springs,” of the Airbus A319/A320/A321 Maintenance Planning Document, Revision 25, dated October 2001, is considered acceptable for compliance with the inspection requirements of paragraph (a)

of this AD. Operators should note that this task requires repetitive inspections at 8-day intervals, instead of intervals not to exceed 500 flight cycles.

Optional Terminating Action

(e) Replacement of the MLG side-stay lugs and links on the left and right sides of the airplane with lugs and links made of new, improved material, in accordance with Airbus Service Bulletin A320–32–1223, dated March 5, 2001, terminates the repetitive inspections required by paragraphs (b) and (c) of this AD.

Alternative Methods of Compliance

(f) In accordance with 14 CFR 39.19, the Manager, International Branch, ANM–116, FAA, Transport Airplane Directorate, is authorized to approve alternative methods of compliance for this AD.

Note 2: The subject of this AD is addressed in French airworthiness directive 2002–075(B), dated January 23, 2002.

Issued in Renton, Washington, on October 7, 2003.

Ali Bahrami,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 03–25978 Filed 10–14–03; 8:45 am]

BILLING CODE 4910–13–P

DEPARTMENT OF TRANSPORTATION**Federal Aviation Administration****14 CFR Part 39**

[Docket No. 2001–NM–362–AD]

RIN 2120–AA64

Airworthiness Directives; McDonnell Douglas Model DC–10–10, DC–10–10F, DC–10–15, DC–10–30, DC–10–30F (KC–10A and KDC–10), DC–10–40, DC–10–40F, MD–10–10F, MD–10–30F, MD–11, and MD–11F Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: This document proposes the adoption of a new airworthiness directive (AD) that is applicable to certain McDonnell Douglas airplanes as listed above. This proposal would require modification of the installation wiring for the electric motor operated auxiliary hydraulic pumps in the right wheel well area of the main landing gear, and repetitive inspections of the number 1 and 2 electric motors of the auxiliary hydraulic pumps for electrical resistance, continuity, mechanical rotation, and associated airplane wiring resistance/voltage; and corrective actions if necessary. This action is necessary to prevent failure of the

electric motors of the hydraulic pump and associated wiring, which could result in fire at the auxiliary hydraulic pump and consequent damage to the adjacent electrical equipment and/or structure. This action is intended to address the identified unsafe condition.

DATES: Comments must be received by December 1, 2003.

ADDRESSES: Submit comments in triplicate to the Federal Aviation Administration (FAA), Transport Airplane Directorate, ANM–114, Attention: Rules Docket No. 2001–NM–362–AD, 1601 Lind Avenue, SW., Renton, Washington 98055–4056. Comments may be inspected at this location between 9 a.m. and 3 p.m., Monday through Friday, except Federal holidays. Comments may be submitted via fax to (425) 227–1232. Comments may also be sent via the Internet using the following address: *9-anm-nprmcomment@faa.gov*. Comments sent via fax or the Internet must contain “Docket No. 2001–NM–362–AD” in the subject line and need not be submitted in triplicate. Comments sent via the Internet as attached electronic files must be formatted in Microsoft Word 97 for Windows or ASCII text.

The service information referenced in the proposed rule may be obtained from Boeing Commercial Aircraft Group, Long Beach Division, 3855 Lakewood Boulevard, Long Beach, California 90846, Attention: Technical Publications Business Administration, Dept. C1–L51 (2–60). This information may be examined at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the FAA, Los Angeles Aircraft Certification Office, 3960 Paramount Boulevard, Lakewood, California.

FOR FURTHER INFORMATION CONTACT: Ken Sujishi, Aerospace Engineer, Systems and Equipment Branch, ANM–130L, FAA, Los Angeles Aircraft Certification Office, 3960 Paramount Boulevard, Lakewood, California 90712; telephone (562) 627–5353; fax (562) 627–5210.

SUPPLEMENTARY INFORMATION:**Comments Invited**

Interested persons are invited to participate in the making of the proposed rule by submitting such written data, views, or arguments as they may desire. Communications shall identify the Rules Docket number and be submitted in triplicate to the address specified above. All communications received on or before the closing date for comments, specified above, will be considered before taking action on the proposed rule.

The proposals contained in this action may be changed in light of the comments received.

Submit comments using the following format:

- Organize comments issue-by-issue. For example, discuss a request to change the compliance time and a request to change the service bulletin reference as two separate issues.
- For each issue, state what specific change to the proposed AD is being requested.
- Include justification (*e.g.*, reasons or data) for each request.

Comments are specifically invited on the overall regulatory, economic, environmental, and energy aspects of the proposed rule. All comments submitted will be available, both before and after the closing date for comments, in the Rules Docket for examination by interested persons. A report summarizing each FAA-public contact concerned with the substance of this proposal will be filed in the Rules Docket.

Commenters wishing the FAA to acknowledge receipt of their comments submitted in response to this action must submit a self-addressed, stamped postcard on which the following statement is made: "Comments to Docket Number 2001-NM-362-AD." The postcard will be date stamped and returned to the commenter.

Availability of NPRMs

Any person may obtain a copy of this NPRM by submitting a request to the FAA, Transport Airplane Directorate, ANM-114, Attention: Rules Docket No. 2001-NM-362-AD, 1601 Lind Avenue, SW., Renton, Washington 98055-4056.

Discussion

The FAA has received several reports of failure of the auxiliary hydraulic pump systems on Model DC-10 airplanes. Some failures resulted in burnt electrical wiring leading to the electric motor of the auxiliary hydraulic pump, within the right wheel well area of the main landing gear (MLG). Damage also was found on the adjacent structure, control cables, hydraulic pipes, and hoses. These failures occurred during ground operations, or when powered in-flight by the air-driven generator. The failures were due to an electrical short within the electric motor which resulted in arcing damage. These failures consisted of a seized or difficult-to-turn rotor on the pump assembly, burnt and shorted motor feeder cables, and/or uncontained internal electric arcing failures within the electric motor. Investigation revealed that these conditions occurred

on airplanes that had been in service several years and/or had the auxiliary hydraulic pump previously overhauled. These discrepancies can be caused by hydraulic fluid contamination to the electric motor portion of the pump assembly which causes failed rotor bearing, and/or degradation of the stator's encapsulate material. Such discrepancies, if not corrected, could result in fire at the auxiliary hydraulic pump and consequent damage to the adjacent electrical equipment and/or structure.

Explanation of Relevant Service Information

We have reviewed and approved Boeing Alert Service Bulletin DC10-29A144, Revision 2, dated August 1, 2003 (for Model DC-10-10, DC-10-10F, DC-10-15, DC-10-30, DC-10-30F (KC-10A and KDC-10), DC-10-40, DC-10-40F, MD-10-10F, and MD-10-30F airplanes), which describes procedures for modification of the electrical wiring of the auxiliary hydraulic pump installation in the right wheel well area of the MLG. The modification includes, but is not limited to, removing existing clamps, ground wires (if required), and sleeving from the wire assemblies; inspecting for cracks and chafing; installing new support brackets, clips, and bracket assemblies, as applicable; installing sleeving; re-routing the wire assemblies using new clamps and attachments, installing an additional routing clip on the lower bracket of the fuel motor control valve, if applicable, and doing a voltage check and a functional test.

Service Bulletin DC10-29A144 recommends prior or concurrent accomplishment of Boeing Alert Service Bulletin DC10-29A142, Revision 02, dated April 17, 2003, which describes procedures for repetitive inspections (checks) of the number 1 and 2 electric motors of the auxiliary hydraulic pumps for electrical resistance, continuity, mechanical rotation, and associated airplane wiring resistance/voltage, and corrective actions if necessary. The corrective actions include replacing the auxiliary hydraulic pump with a serviceable pump, and repairing the wiring.

We also have reviewed and approved Boeing Alert Service Bulletin MD11-29A059, Revision 2, dated August 1, 2003 (for Model MD-11 and MD-11F airplanes), which describes procedures for modification of the wiring of the electric motors of the auxiliary hydraulic pump in the right wheel well area of the MLG. The modification includes, but is not limited to, removing and retaining wire assembly clamps, if

applicable; retaining the existing ground wire assemblies; retaining or replacing all other wire assemblies for both connectors; installing spiral wrap and sleeving; wrapping upper ends of individual wires with tape; installing new support bracket assemblies, if applicable; re-routing and attaching wire assemblies using new clamps and attachments, if applicable; and doing a voltage check and a functional test.

Service Bulletin MD11-29A059 recommends prior or concurrent accomplishment of Boeing Alert Service Bulletin MD11-29A057, Revision 02, dated April 17, 2003, which describes procedures for repetitive inspections (checks) of the number 1 and 2 electric motors of the auxiliary hydraulic pumps for electrical resistance, continuity, mechanical rotation, associated airplane wiring resistance/voltage, and corrective actions if necessary. The corrective actions include replacing the auxiliary hydraulic pump with a serviceable pump and repairing the wiring.

Accomplishment of the actions specified in the service bulletins is intended to adequately address the identified unsafe condition.

Related Rulemaking

On July 2, 2001, we issued AD 2001-14-08, amendment 39-12319 (66 FR 36441, July 12, 2001), for certain McDonnell Douglas Model DC-10 series airplanes, Model MD-10 series airplanes, and Model MD-11 series airplanes. That AD requires repetitive inspections of the number 1 and 2 electric motors of the auxiliary hydraulic pump for electrical resistance, continuity, mechanical rotation, and associated wiring resistance/voltage; and corrective actions if necessary. The actions specified by that AD are intended to prevent various failures of electric motors of the auxiliary hydraulic pump and associated wiring, which could result in fire at the auxiliary hydraulic pump and consequent damage to the adjacent electrical equipment and/or structure. That AD is being superseded by a separate action to reduce the repetitive inspection intervals currently required.

Explanation of Requirements of Proposed Rule

Since an unsafe condition has been identified that is likely to exist or develop on other products of this same type design, the proposed AD would require accomplishment of the actions specified in the service bulletins described previously, except that the proposed AD does not require completing the Evaluation Forms.

Cost Impact

There are approximately 409 Model DC-10 airplanes of the affected design in the worldwide fleet. The FAA estimates that 322 airplanes of U.S. registry would be affected by this proposed AD.

It would take approximately 9 work hours per airplane to do the modification specified in Service Bulletin DC10-29A144, at an average labor rate of \$65 per work hour. Required parts would cost would be between \$4,886 and \$7,920 per airplane. Based on these figures, the cost impact of the proposed modification is estimated to be between \$5,471 and \$8,505 per airplane.

It would take approximately 1 work hour per airplane to do the inspection specified in Service Bulletin DC10-29A142, at an average labor rate of \$65 per work hour. Based on these figures, the cost impact of the proposed inspection is estimated to be \$65 per airplane, per inspection cycle.

There are approximately 195 Model MD-11 airplanes of the affected design in the worldwide fleet. The FAA estimates that 74 airplanes of U.S. registry would be affected by this proposed AD.

It would take approximately 13 work hours per airplane to do the modification specified in Service Bulletin MD11-29A059, at an average labor rate of \$65 per work hour. Required parts would cost between \$5,183 and \$9,182 per airplane. Based on these figures, the cost impact of the proposed modification is estimated to be between \$6,028 and \$10,027 per airplane.

It would take approximately 1 work hour per airplane to do the inspection specified in Service Bulletin MD11-29A057, at an average labor rate of \$65 per work hour. Based on these figures, the cost impact of the proposed inspection is estimated to be \$65 per airplane, per inspection cycle.

The cost impact figures discussed above are based on assumptions that no operator has yet accomplished any of the proposed requirements of this AD action, and that no operator would accomplish those actions in the future if this proposed AD were not adopted. The cost impact figures discussed in AD rulemaking actions represent only the time necessary to perform the specific actions actually required by the AD. These figures typically do not include incidental costs, such as the time required to gain access and close up, planning time, or time necessitated by other administrative actions.

Regulatory Impact

The regulations proposed herein would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, it is determined that this proposal would not have federalism implications under Executive Order 13132.

For the reasons discussed above, I certify that this proposed regulation: (1) Is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) if promulgated, will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A copy of the draft regulatory evaluation prepared for this action is contained in the Rules Docket. A copy of it may be obtained by contacting the Rules Docket at the location provided under the caption **ADDRESSES**.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

The Proposed Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration proposes to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:

McDonnell Douglas: Docket 2001-NM-362-AD.

Applicability: Model DC-10-10, DC-10-10F, DC-10-15, DC-10-30, DC-10-30F (KC-10A and KDC-10), DC-10-40, DC-10-40F, MD-10-10F, MD-10-30F, MD-11, and MD-11F airplanes; certificated in any category.

Compliance: Required as indicated, unless accomplished previously.

To prevent failure of the electric motors of the hydraulic pump and associated wiring, which could result in fire at the auxiliary hydraulic pump and consequent damage to the adjacent electrical equipment and/or structure, accomplish the following:

Modification/Prior or Concurrent Actions

(a) For airplanes listed in Boeing Alert Service Bulletin DC10-29A144, Revision 2, dated August 1, 2003: Within 18 months after the effective date of this AD, do the actions specified in paragraphs (a)(1) and (a)(2) of this AD.

(1) Modify the installation wiring of the electric motor operated auxiliary hydraulic pumps in the right wheel well area of the main landing gear (MLG) (including removing existing clamps, ground wires, if required, and sleeving from the wire assemblies; inspecting for cracks and chafing, installing new support bracket, clips, and bracket assemblies, as applicable; installing sleeving; re-routing and attaching wire assemblies using new clamps and attachments; installing an additional routing clip on lower bracket of fuel motor control valve, if applicable; and doing a voltage check and a functional test), per the Accomplishment Instructions of Boeing Alert Service Bulletin DC10-29A144, Revision 2, dated August 1, 2003.

(2) Prior to or concurrent with accomplishment of paragraph (a)(1) of this AD: Do the actions specified in Boeing Alert Service Bulletin DC10-29A142, Revision 02, dated April 17, 2003 (including inspecting the number 1 and 2 electric motors of the auxiliary hydraulic pumps for electrical resistance, continuity, mechanical rotation, and associated airplane wiring resistance/voltage; and replacing the auxiliary hydraulic pump with a serviceable pump and repairing the wiring if necessary), per the Accomplishment Instructions of the service bulletin. Repeat the actions after that at intervals not to exceed 2,500 flight hours.

(b) For airplanes listed in Boeing Alert Service Bulletin MD11-29A059, Revision 2, dated August 1, 2003: Within 18 months after the effective date of this AD, do the actions specified in paragraphs (b)(1) and (b)(2) of this AD.

(1) Modify the installation wiring of the electric motor auxiliary hydraulic pumps in the wheel well area of the right MLG (including removing and retaining wire assembly clamps, if applicable; retaining the existing ground wire assemblies; retaining or replacing all other wire assemblies for both connectors; installing spiral wrap and sleeving; wrapping upper ends of individual wires with tape; installing new support bracket assemblies, if applicable; re-routing and attaching wire assemblies using new clamps and attachments, if applicable; and doing a voltage check and a functional test), per the Accomplishment Instructions of Boeing Alert Service Bulletin MD11-29A059, Revision 2, dated August 1, 2003.

(2) Prior to or concurrent with accomplishment of paragraph (b)(1) of this AD: Do the actions specified in Boeing Alert Service Bulletin MD11-29A057, Revision 02, dated April 17, 2003 (including inspecting the number 1 and 2 electric motors of the auxiliary hydraulic pumps for electrical resistance, continuity, mechanical rotation, and associated airplane wiring resistance/voltage; and replacing the auxiliary hydraulic pump with a serviceable pump and repairing the wiring if necessary), per the Accomplishment Instructions of the service

bulletin. Repeat the actions after that at intervals not to exceed 2,500 flight hours.

Alternative Methods of Compliance

(c) In accordance with 14 CFR 39.19, the Manager, Los Angeles Aircraft Certification Office, FAA, is authorized to approve alternative methods of compliance (AMOCs) for this AD.

Issued in Renton, Washington, on October 7, 2003.

Ali Bahrami,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 03-25979 Filed 10-14-03; 8:45 am]

BILLING CODE 4910-13-P

DEPARTMENT OF THE INTERIOR

Office of Surface Mining Reclamation and Enforcement

30 CFR Part 914

[IN-153-FOR, State Program Amendment No. 02-034R]

Indiana Regulatory Program

AGENCY: Office of Surface Mining Reclamation and Enforcement, Interior.

ACTION: Proposed rule; public comment period and opportunity for public hearing on proposed amendment.

SUMMARY: We, the Office of Surface Mining Reclamation and Enforcement (OSM), are announcing receipt of a proposed amendment to the Indiana regulatory program (Indiana program) under the Surface Mining Control and Reclamation Act of 1977 (SMCRA or the Act). Indiana proposes revisions to and additions of rules concerning protection of ground water quality. Indiana intends to revise its program to provide additional safeguards for ground water.

This document gives the times and locations that the Indiana program and proposed amendment to that program are available for your inspection, the comment period during which you may submit written comments on the amendment, and the procedures that we will follow for the public hearing, if one is requested.

DATES: We will accept written comments on this amendment until 4 p.m., e.s.t., November 14, 2003. If requested, we will hold a public hearing on the amendment on November 10, 2003. We will accept requests to speak at a hearing until 4 p.m., e.s.t. on October 30, 2003.

ADDRESSES: You should mail or hand deliver written comments and requests to speak at the hearing to Andrew R. Gilmore, Director, Indianapolis Field Office, at the address listed below.

You may review copies of the Indiana program, this amendment, a listing of any scheduled public hearings, and all written comments received in response to this document at the addresses listed below during normal business hours, Monday through Friday, excluding holidays. You may receive one free copy of the amendment by contacting OSM's Indianapolis Field Office.

Andrew R. Gilmore, Director,
Indianapolis Field Office, Office of
Surface Mining Reclamation and
Enforcement, Minton-Capehart
Federal Building, 575 North
Pennsylvania Street, Room 301,
Indianapolis, Indiana 46204,
Telephone: (317) 226-6700, Internet
address: IFOMAIL@osmre.gov.

Indiana Department of Natural
Resources, Bureau of Mine
Reclamation, 402 West Washington
Street, Room W-295, Indianapolis,
Indiana 46204, Telephone: (317) 232-
1291.

FOR FURTHER INFORMATION CONTACT:

Andrew R. Gilmore, Director,
Indianapolis Field Office. Telephone:
(317) 226-6700. Internet address:
IFOMAIL@osmre.gov.

SUPPLEMENTARY INFORMATION:

- I. Background on the Indiana Program
- II. Description of the Proposed Amendment
- III. Public Comment Procedures
- IV. Procedural Determinations

I. Background on the Indiana Program

Section 503(a) of the Act permits a State to assume primacy for the regulation of surface coal mining and reclamation operations on non-Federal and non-Indian lands within its borders by demonstrating that its program includes, among other things, "a State law which provides for the regulation of surface coal mining and reclamation operations in accordance with the requirements of this Act * * *; and rules and regulations consistent with regulations issued by the Secretary pursuant to this Act." See 30 U.S.C. 1253(a)(1) and (7). On the basis of these criteria, the Secretary of the Interior conditionally approved the Indiana program effective July 29, 1982. You can find background information on the Indiana program, including the Secretary's findings, the disposition of comments, and the conditions of approval of the Indiana program in the July 26, 1982, **Federal Register** (47 FR 32071). You can also find later actions concerning the Indiana program and program amendments at 30 CFR 914.10, 914.15, 914.16, and 914.17.

II. Description of the Proposed Amendment

By letter dated September 3, 2003 (Administrative Record No. IND-1719), Indiana sent us an amendment to its program under SMCRA (30 U.S.C. 1201 *et seq.*). Indiana sent the amendment at its own initiative. Below is a summary of the changes proposed by Indiana. The full text of the program amendment is available for you to read at the locations listed above under **ADDRESSES**.

A. Definitions

1. At 312 IAC 25-1-45.5, Indiana is adding the following definition for "Drinking water well."

"Drinking water well," for the purposes of 312 IAC 25-6-12.5 and 312 IAC 25-6-76.5, means a bored, drilled, or driven shaft or a dug hole that meets each of the following:

- (1) Supplies ground water for human consumption.
- (2) Has a depth greater than its largest surface dimension.
- (3) Is not permanently abandoned under 312 IAC 13-10-2.

2. At 312 IAC 25-1-60.5, Indiana is adding the following definition for "Ground water management zone."

"Ground water management zone" means a three (3) dimensional region of ground water around a potential or existing contaminant source where a contaminant is or was managed to prevent or mitigate deterioration of ground water quality such that the criteria established in 312 IAC 25-6-12.5(a) or 312 IAC 25-6-76.5(a) are met at and beyond the boundary of the region.

3. At 312 IAC 25-1-109.5, Indiana is adding the following definition for "Property boundary."

"Property boundary," for the purposes of 312 IAC 25-6-12.5 and 312 IAC 25-6-76.5, means the edge of a contiguous parcel of land owned by or leased to the permittee. Contiguous land shall include land separated by a public right-of-way, if that land would otherwise be contiguous.

B. Surface Mining Permit Applications

1. At 312 IAC 25-4-43, Indiana is adding subdivision (4). This new subdivision requires the maps and plans of the proposed permit and adjacent areas to include all monitoring locations used to demonstrate compliance with 312 IAC 25-6-12.5.

2. At 312 IAC 25-4-47(b), protection of hydrologic balance, Indiana is adding subdivision (9). This new subdivision requires the reclamation plan to contain a description, with appropriate maps and cross section drawings, of a plan to demonstrate compliance with 312 IAC 25-6-12.5.