

**Compliance:** Required as indicated, unless accomplished previously.

To prevent electrical arcing between the integrated drive generator (IDG) cables and the firewall separators due to IDG cable chafing, which could result in an in-flight fire and/or loss of electrical power, accomplish the following:

#### Part Number Identification

(a) For airplanes that have been repaired or modified before the effective date of this AD in accordance with Bombardier Alert Service Bulletin A601R-24-091, dated March 9, 2000; or Revision "A," dated May 10, 2000: Within 550 flight hours or 2 months after the effective date of this AD, whichever occurs first, determine the part numbers (P/Ns) of the clamps that hold the IDG cables on the left and right pylons.

(1) If the P/N of all clamps is TA121010R14-04: No further action is required by this paragraph.

(2) If the P/N of any clamp is NOT TA121010R14-04: Before further flight, replace the discrepant clamp with a clamp having P/N TA121010R14-04, in accordance with Bombardier Alert Service Bulletin A601R-24-091, Revision 'C,' dated February 1, 2001.

#### Inspection

(b) For airplanes not identified in paragraph (a) of this AD: Within 550 flight hours or 2 months after the effective date of this AD, whichever occurs first, perform a one-time general visual inspection to detect chafing and other damage of the IDG cables and the firewall separators of the pylon, in accordance with Bombardier Alert Service Bulletin A601R-24-091, Revision 'C,' dated February 1, 2001. Prior to further flight thereafter, perform all applicable corrective actions and install a clamp, a conduit, and Teflon strips, in accordance with the alert service bulletin. If a temporary repair is performed, replace the harnesses with new parts within 4,000 flight hours after the repair, in accordance with the alert service bulletin.

(c) Accomplishment of an inspection and applicable corrective actions before the effective date of this AD in accordance with Bombardier Alert Service Bulletin A601R-24-091, Revision 'B,' dated September 14, 2000, is acceptable for compliance with the requirements of paragraph (b) of this AD.

**Note 2:** For the purposes of this AD, a general visual inspection is defined as: "A visual examination of an interior or exterior area, installation, or assembly to detect obvious damage, failure, or irregularity. This level of inspection is made under normally available lighting conditions such as daylight, hangar lighting, flashlight, or drop-light, and may require removal or opening of access panels or doors. Stands, ladders, or platforms may be required to gain proximity to the area being checked."

#### Part Installation

(d) As of the effective date of this AD, no person may install an IDG cable clamp, P/N TA121010L14-04, during incorporation of Revision 'C' of Bombardier Alert Service Bulletin A601R-24-091.

#### Alternative Methods of Compliance

(e) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, New York Aircraft Certification Office (ACO), FAA. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, New York ACO.

**Note 3:** Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the New York ACO.

#### Special Flight Permits

(f) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

#### Incorporation by Reference

(g) Unless otherwise specified in this AD, the actions must be done in accordance with Bombardier Alert Service Bulletin A601R-24-091, Revision 'C,' dated February 1, 2001. This incorporation by reference was approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Copies may be obtained from Bombardier, Inc., Canadair, Aerospace Group, P.O. Box 6087, Station Centreville, Montreal, Quebec H3C 3G9, Canada. Copies may be inspected at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

**Note 4:** The subject of this AD is addressed in Canadian airworthiness directive CF-2000-17R1, dated October 30, 2000.

#### Effective Date

(h) This amendment becomes effective on October 27, 2003.

Issued in Renton, Washington, on September 11, 2003.

**Vi L. Lipski,**

*Manager, Transport Airplane Directorate, Aircraft Certification Service.*

[FR Doc. 03-23671 Filed 9-18-03; 12:01 pm]

**BILLING CODE 4910-13-U**

## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 39

[Docket No. 2002-NM-60-AD; Amendment 39-13306; AD 2003-19-03]

RIN 2120-AA64

### Airworthiness Directives; Dornier Model 328-100 and -300 Series Airplanes

**AGENCY:** Federal Aviation Administration, DOT.

**ACTION:** Final rule.

**SUMMARY:** This amendment adopts a new airworthiness directive (AD), applicable to certain Dornier Model 328-100 and -300 series airplanes, that requires inspection of the nose landing gear (NLG) and main landing gear (MLG) to ensure that certain bolts are in place; repetitive inspections of the bolts and bolt areas for evidence of corrosion; and corrective action, if necessary. This action is necessary to prevent failure of the NLG or MLG due to corroded or missing bolts, which could cause loss of connection pins, and consequent collapse of the landing gear during ground maneuvers or upon landing. This action is intended to address the identified unsafe condition.

**DATES:** Effective October 27, 2003.

The incorporation by reference of certain publications listed in the regulations is approved by the Director of the Federal Register as of October 27, 2003.

**ADDRESSES:** The service information referenced in this AD may be obtained from AvCraft Aerospace GmbH, P.O. Box 1103, D-82230 Wessling, Germany. This information may be examined at the Federal Aviation Administration (FAA), Transport Airplane Directorate, Rules Docket, 1601 Lind Avenue, SW., Renton, Washington; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

**FOR FURTHER INFORMATION CONTACT:** Tom Groves, Aerospace Engineer, International Branch, ANM-116, FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 227-1503; fax (425) 227-1149.

**SUPPLEMENTARY INFORMATION:** A proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) to include an airworthiness directive (AD) that is applicable to certain Dornier Model 328-100 and -300 series airplanes was published in the **Federal Register** on July 9, 2003 (68 FR 40831). That action proposed to require inspection of the nose landing gear (NLG) and main landing gear (MLG) to ensure that certain bolts are in place; repetitive inspections of the bolts and bolt areas for evidence of corrosion; and corrective action, if necessary.

#### Comments

Interested persons have been afforded an opportunity to participate in the making of this amendment. No comments were submitted in response to the proposal or the FAA's determination of the cost to the public.

## Conclusion

We have determined that air safety and the public interest require the adoption of the rule as proposed.

## Changes to 14 CFR Part 39/Effect on the AD

On July 10, 2002, we issued a new version of 14 CFR part 39 (67 FR 47997, July 22, 2002), which governs the FAA's airworthiness directives system. The regulation now includes material that relates to altered products, special flight permits, and alternative methods of compliance. However, for clarity and consistency in this final rule, we have retained the language of the NPRM regarding that material.

## Change to Labor Rate Estimate

We have reviewed the figures we have used over the past several years to calculate AD costs to operators. To account for various inflationary costs in the airline industry, we find it necessary to increase the labor rate used in these calculations from \$60 per work hour to \$65 per work hour. The cost impact information, below, reflects this increase in the specified hourly labor rate.

## Cost Impact

We estimate that 53 Model 328-100 series airplanes and 39 Model 328-300 series airplanes of U.S. registry will be affected by this AD, that it will take approximately 1 work hour per airplane to accomplish the required inspection for bolt placement, and that the average labor rate is \$65 per work hour. Based on these figures, the cost impact of the AD on U.S. operators is estimated to be \$5,980, or \$65 per airplane.

We estimate that it will take approximately 5 work hours per airplane to accomplish the required inspection for corrosion, and that the average labor rate is \$65 per work hour. Based on these figures, the cost impact on U.S. operators for the required inspection for corrosion is estimated to be \$29,900, or \$325 per airplane.

The cost impact figures discussed above are based on assumptions that no operator has yet accomplished any of the requirements of this AD action, and that no operator would accomplish those actions in the future if this AD were not adopted. The cost impact figures discussed in AD rulemaking actions represent only the time necessary to perform the specific actions actually required by the AD. These figures typically do not include incidental costs, such as the time required to gain access and close up, planning time, or time necessitated by other administrative actions.

## Regulatory Impact

The regulations adopted herein will not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, it is determined that this final rule does not have federalism implications under Executive Order 13132.

For the reasons discussed above, I certify that this action (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A final evaluation has been prepared for this action and it is contained in the Rules Docket. A copy of it may be obtained from the Rules Docket at the location provided under the caption **ADDRESSES**.

## List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

## Adoption of the Amendment

■ Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

## PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

### § 39.13 [Amended]

■ 2. Section 39.13 is amended by adding the following new airworthiness directive:

**2003-19-03 Fairchild Dornier GMBH** (Formerly Dornier Luftfahrt GmbH): Amendment 39-13306. Docket 2002-NM-60-AD.

**Applicability:** Model 328-100 series airplanes having serial numbers 3005 through 3119 inclusive, and Model 328-300 series airplanes having serial numbers 3105 through 3200 inclusive; certificated in any category.

**Compliance:** Required as indicated, unless accomplished previously.

To prevent failure of the nose landing gear (NLG) or main landing gear (MLG) due to corroded or missing bolts, which could cause loss of connection pins, and consequent collapse of the landing gear during ground

maneuvers or upon landing, accomplish the following:

## Service Bulletin Reference

(a) The term "service bulletin," as used in this AD, means the Accomplishment Instructions of the following service bulletins, as applicable:

(1) For Model 328-100 series airplanes: Dornier Service Bulletin SB-328-32-414, dated December 3, 2001.

(2) For Model 328-300 series airplanes: Dornier Service Bulletin SB-328J-32-147, dated December 3, 2001.

## Inspection of Bolt Placement

(b) Perform a one-time general visual inspection of the NLG and MLG to ensure that the bolts are in place, per paragraph 2.B1) of the applicable service bulletin. Do the inspection at the later of the times specified in paragraphs (b)(1) and (b)(2) of this AD. If all bolts are in place, no further action is required by this paragraph.

**Note 1:** For the purposes of this AD, a general visual inspection is defined as: "A visual examination of an interior or exterior area, installation, or assembly to detect obvious damage, failure, or irregularity. This level of inspection is made from within touching distance unless otherwise specified. A mirror may be necessary to enhance visual access to all exposed surfaces in the inspection area. This level of inspection is made under normally available lighting conditions such as daylight, hangar lighting, flashlight, or droplight and may require removal or opening of access panels or doors. Stands, ladders, or platforms may be required to gain proximity to the area being checked."

(1) Within 4,000 total flight hours, or within 24 months since the date of issuance of the original Airworthiness Certificate, or within 24 months since the date of issuance of the Export Certificate of Airworthiness, whichever occurs first.

(2) Within 6 days after the effective date of this AD.

## Corrective Action

(c) During the inspection required by paragraph (b) of this AD, if any bolt is missing or is not in position: Prior to further flight, replace the bolt with a bolt having the same part number, per the applicable service bulletin.

## Inspections for Corrosion

(d) Within 400 flight hours or 6 months after accomplishing the inspection required by paragraph (b) of this AD, whichever occurs first: Remove the nuts, bolts, and washers of the NLG and MLG, and perform a detailed inspection for evidence of corrosion. Do the inspection per the applicable service bulletin. Repeat the inspection thereafter at intervals not to exceed 4,000 flight hours or 24 months, whichever occurs first.

**Note 2:** For the purposes of this AD, a detailed inspection is defined as: "An intensive visual examination of a specific structural area, system, installation, or assembly to detect damage, failure, or irregularity. Available lighting is normally supplemented with a direct source of good

lighting at intensity deemed appropriate by the inspector. Inspection aids such as mirror, magnifying lenses, etc., may be used. Surface cleaning and elaborate access procedures may be required."

(1) If no evidence of corrosion is found on any part, or if a new bolt is installed: Prior to further flight, apply corrosion prevention compound to the bolt shaft and install the bolt, per the applicable service bulletin.

(2) If any evidence of corrosion is found: Prior to further flight, replace the bolt with a part having the same part number and apply corrosion prevention compound to the bolt shaft and install the bolt, per the applicable service bulletin.

#### Alternative Methods of Compliance

(e) In accordance with 14 CFR 39.19, the Manager, International Branch, ANM-116, FAA, is authorized to approve alternative methods of compliance for this AD.

#### Incorporation by Reference

(f) The actions shall be done in accordance with Dornier Service Bulletin SB-328-32-414, dated December 3, 2001; or Dornier Service Bulletin SB-328J-32-147, dated December 3, 2001; as applicable. This incorporation by reference was approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Copies may be obtained from AvCraft Aerospace GmbH, P.O. Box 1103, D-82230 Wessling, Germany. Copies may be inspected at the FAA, Transport Airplane Directorate, 1601 Lind Avenue SW, Renton, Washington; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

**Note 3:** The subject of this AD is addressed in German airworthiness directives 2002-014/2 and 2002-015/2, both dated March 7, 2002.

#### Effective Date

(g) This amendment becomes effective on October 27, 2003.

Issued in Renton, Washington, on September 11, 2003.

**Vi L. Lipski,**

*Manager, Transport Airplane Directorate,  
Aircraft Certification Service.*

[FR Doc. 03-23672 Filed 9-18-03; 12:01 pm]

**BILLING CODE 4910-13-U**

## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 97

[Docket No. 30387 ; Amdt. No. 3075 ]

#### Standard Instrument Approach Procedures; Miscellaneous Amendments

**AGENCY:** Federal Aviation Administration (FAA), DOT.

**ACTION:** Final rule.

**SUMMARY:** This amendment establishes, amends, suspends, or revokes Standard

Instrument Approach Procedures (SIAPs) for operations at certain airports. These regulatory actions are needed because of the adoption of new or revised criteria, or because of changes occurring in the National Airspace System, such as the commissioning of new navigational facilities, addition of new obstacles, or changes in air traffic requirements. These changes are designed to provide safe and efficient use of the navigable airspace and to promote safe flight operations under instrument flight rules at the affected airports.

**DATES:** This rule is effective September 22, 2003. The compliance date for each SIAP is specified in the amendatory provisions.

The incorporation by reference of certain publications listed in the regulations is approved by the Director of the Federal Register as of September 22, 2003.

**ADDRESSES:** Availability of matters incorporated by reference in the amendment is as follows:

*For Examination—*

1. FAA Rules Docket, FAA Headquarters Building, 800 Independence Avenue, SW., Washington, DC 20591;
2. The FAA Regional Office of the region in which the affected airport is located;
3. The Flight Inspection Area Office which originated the SIAP; or,
4. The Office of Federal Register, 800 North Capitol Street, NW., Suite 700, Washington, DC.

*For Purchase—*Individual SIAP copies may be obtained from:

1. FAA Public Inquiry Center (APA-200), FAA Headquarters Building, 800 Independence Avenue, SW., Washington, DC 20591; or
2. The FAA Regional Office of the region in which the affected airport is located.

*By Subscription—*Copies of all SIAPs, mailed once every 2 weeks, are for sale by the Superintendent of Documents, U.S. Government Printing Office, Washington, DC 20402.

**FOR FURTHER INFORMATION CONTACT:**

Donald P. Pate, Flight Procedure Standards Branch (AMCAFS-420), Flight Technologies and Programs Division, Flight Standards Service, Federal Aviation Administration, Mike Monroney Aeronautical Center, 6500 South MacArthur Blvd., Oklahoma City, OK 73169 (Mail Address: P.O. Box 25082, Oklahoma City, OK 73125) telephone: (405) 954-4164.

**SUPPLEMENTARY INFORMATION:** This amendment to part 97 of the Federal Aviation Regulations (14 CFR part 97)

establishes, amends, suspends, or revokes Standard Instrument Approach Procedures (SIAPs). The complete regulatory description of each SIAP is contained in official FAA form documents which are incorporated by reference in this amendment under 5 U.S.C. 552(a), 1 CFR part 51, and § 97.20 of the Federal Aviation Regulations (FAR). The applicable FAA Forms are identified as FAA Forms 8260-3, 8260-4, and 8260-5. Materials incorporated by reference are available for examination or purchase as stated above.

The large number of SIAPs, their complex nature, and the need for a special format make their verbatim publication in the **Federal Register** expensive and impractical. Further, airmen do not use the regulatory text of the SIAPs, but refer to their graphic depiction on charts printed by publishers of aeronautical materials. Thus, the advantages of incorporation by reference are realized and publication of the complete description of each SIAP contained in FAA form documents is unnecessary. The provisions of this amendment state the affected CFR (and FAR) sections, with the types and effective dates of the SIAPs. This amendment also identifies the airport, its location, the procedure identification and the amendment number.

#### The Rule

This amendment to part 97 is effective upon publication of each separate SIAP as contained in the transmittal. Some SIAP amendments may have been previously issued by the FAA in a National Flight Data Center (NFDC) Notice to Airmen (NOTAM) as an emergency action of immediate flight safety relating directly to published aeronautical charts. The circumstances which created the need for some SIAP amendments may require making them effective in less than 30 days. For the remaining SIAPs, an effective date at least 30 days after publication is provided.

Further, the SIAPs contained in this amendment are based on the criteria contained in the U.S. Standard for Terminal Instrument Procedures (TERPS). In developing these SIAPs, the TERPS criteria were applied to the conditions existing or anticipated at the affected airports. Because of the close and immediate relationship between these SIAPs and safety in air commerce, I find that notice and public procedure before adopting these SIAPs are impracticable and contrary to the public interest and, where applicable, that