

Dated: August 28, 2003.

A.J. Yates,

Administrator, Agricultural Marketing Service.

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2003-NE-32-AD; Amendment 39-13285; AD 2003-17-10]

RIN 2120-AA64

Airworthiness Directives; McCauley Propeller Systems, Inc. Propeller Hub Models B5JFR36C1101, C5JFR36C1102, B5JFR36C1103, and C5JFR36C1104; Correction

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule; correction.

SUMMARY: This document makes a correction to Airworthiness Directive (AD) 2003-17-10. That AD applies to McCauley Propeller Systems, Inc. Propeller Hub Models B5JFR36C1101, C5JFR36C1102, B5JFR36C1103, and C5JFR36C1104 propellers. AD 2003-17-10 was published in the **Federal Register** on August 21, 2003 (68 FR 50462). Paragraph (o) incorrectly references Table 3 and should reference Table 2. This document corrects that reference. In all other respects, the original document remains the same. **EFFECTIVE DATE:** Effective September 3, 2003.

FOR FURTHER INFORMATION CONTACT: Timothy Smyth, Aerospace Engineer, Chicago Aircraft Certification Office, FAA, Small Airplane Directorate, 2300 East Devon Avenue, Room 107, Des Plaines, IL 60018; telephone: (847) 294-7132; fax: (847) 294-7834.

SUPPLEMENTARY INFORMATION: A final rule; request for comments to supersede an existing AD, FR Doc. 03-21519 that applies to McCauley Propeller Systems, Inc. Propeller Hub Models B5JFR36C1101, C5JFR36C1102, B5JFR36C1103, and C5JFR36C1104 propellers, was published in the **Federal Register** on August 21, 2003 (68 FR 50462). The following correction is needed:

§ 39.13 [Corrected]

■ On page 50464, in the third column, in the paragraph entitled Material Incorporated by Reference, paragraph (o), in the sixth line, "listed in Table 3

of this AD" is corrected to read "listed in Table 2 of this AD".

Issued in Burlington, MA, on August 27, 2003.

Francis A. Favara,

Assistant Manager, Engine and Propeller Directorate, Aircraft Certification Service.

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2003-NM-164-AD; Amendment 39-13292; AD 2003-18-01]

RIN 2120-AA64

Airworthiness Directives; General Dynamics (Convair) Model P4Y-2 Airplanes, General Dynamics (Consolidated-Vultee) (Army) Model LB-30 Airplanes, and General Dynamics (Consolidated) (Army) Model C-87A Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule; request for comments.

SUMMARY: This amendment supersedes an existing airworthiness directive (AD), applicable to various surplus military airplanes manufactured by Consolidated, Consolidated Vultee, and Convair, that currently requires repetitive inspections to find fatigue cracks in the lower rear cap of the wing front spar, front spar web, and lower skin of the wings; repair or replacement of any cracked part with a new part; and follow-on inspections at new intervals. This amendment continues to require those actions and revises and clarifies the applicability of the existing AD. The actions specified in this AD are intended to find and fix fatigue cracking, which could result in structural failure of the wings and consequent loss of control of the airplane. This action is intended to address the identified unsafe condition. **DATES:** Effective September 18, 2003. Comments for inclusion in the Rules Docket must be received on or before November 3, 2003. **ADDRESSES:** Submit comments in triplicate to the Federal Aviation Administration (FAA), Transport Airplane Directorate, ANM-114, Attention: Rules Docket No. 2003-NM-164-AD, 1601 Lind Avenue, SW., Renton, Washington 98055-4056. Comments may be inspected at this location between 9 a.m. and 3 p.m.,

Monday through Friday, except Federal holidays. Comments may be submitted via fax to (425) 227-1232. Comments may also be sent via the Internet using the following address: *9-anm-iarcomment@faa.gov*. Comments sent via fax or the Internet must contain "Docket No. 2003-NM-164-AD" in the subject line and need not be submitted in triplicate. Comments sent via the Internet as attached electronic files must be formatted in Microsoft Word 97 for Windows or ASCII text.

Information pertaining to this AD may be examined at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the FAA, Los Angeles Aircraft Certification Office, 3960 Paramount Boulevard, Lakewood, California.

FOR FURTHER INFORMATION CONTACT: John Cecil, Aerospace Engineer, Airframe Branch, ANM-120L, FAA, Los Angeles Aircraft Certification Office, 3960 Paramount Boulevard, Lakewood, California 90712-4137; telephone (562) 627-5228; fax (562) 627-5210.

SUPPLEMENTARY INFORMATION: On April 16, 2003, the FAA issued AD 2003-08-13, amendment 39-13126 (68 FR 19728, April 22, 2003), applicable to various surplus military airplanes manufactured by Consolidated, Consolidated Vultee, and Convair, to require repetitive inspections to find fatigue cracks in the lower rear cap of the wing front spar, front spar web, and lower skin of the wings; repair or replacement of any cracked part with a new part; and follow-on inspections at new intervals. That action was prompted by an accident resulting from the structural failure of the center wing of a United States Department of Agriculture (USDA) Forest Service Model P4Y-2 airplane, and results of an investigation, which revealed fatigue cracking of the lower rear cap of the wing front spar, front spar web, and lower skin of the wings. Such fatigue cracking, if not found and fixed in a timely manner, could result in structural failure of the wings and consequent loss of control of the airplane.

Actions Since Issuance of Previous Rule

Since the issuance of that AD, the FAA has received inquiries concerning the applicability of the AD. The commenters indicate that the applicability of the AD, as published, contains a phrase that could lead the reader to believe that the AD applies to all former military surplus aircraft, rather than just those airplanes specifically called out by model in the AD.