

TABLE 2.—UNIT RATES—Continued

Service ^{1,3}	Rough rice	Brown rice for processing	Milled rice
(a) Milling yield (per sample)	27.50	27.50
(b) All other factors (per factor)	13.20	13.20	13.20
Total oil and free fatty acid	43.00	43.00
Interpretive line samples: ²			
(a) Milling degree (per set)	94.00
(b) Parboiled light (per sample)	23.00
Extra copies of certificates (per copy)	3.00	3.00	3.00

¹ Fees apply to determinations (original or appeals) for kind, class, grade, factor analysis, equal to type, milling yield, or any other quality designation as defined in the U.S. Standards for Rice or applicable instructions, whether performed singly or in combination at other than at the applicant's facility.

² Interpretive line samples may be purchased from the U.S. Department of Agriculture, GIPSA, FGIS, Technical Services Division, 10383 North Ambassador Drive, Kansas City, Missouri 64153-1394. Interpretive line samples also are available for examination at selected FGIS field offices. A list of field offices may be obtained from the Director, Field Management Division, USDA, GIPSA, FGIS, 1400 Independence Avenue, SW, STOP 3630, Washington, D.C. 20250-3630. The interpretive line samples illustrate the lower limit for milling degrees only and the color limit for the factor "Parboiled Light" rice.

³ Fees for other services not referenced in Table 2 will be based on the noncontract hourly rate listed in § 868.90, Table 1.

Dated: May 6, 2003.

Donna Reifschneider,

Administrator, Grain Inspection, Packers and Stockyards Administration.

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2000-NE-48-AD; Amendment 39-13107; AD 2003-07-11]

RIN 2120-AA64

Airworthiness Directives; Rolls-Royce Deutschland Ltd & Co KG Models BR700-710A1-10 and BR700-710A2-20 Turbofan Engines; Correction

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule; request for comments, correction.

SUMMARY: This document makes a correction to Airworthiness Directive (AD) 2003-07-11, applicable to Rolls-Royce Deutschland Ltd & Co KG (formerly Rolls-Royce Deutschland GmbH, formerly BMW Rolls-Royce GmbH), models BR700-710A1-10 and BR700-710A2-20 turbofan engines. AD 2003-07-11 was published in the **Federal Register** on April 11, 2003 (68 FR 17727). Subsequently, a correction document was published in the **Federal Register** on April 23, 2003 (68 FR 19944) that made corrections to the compliance section starting at paragraph (g). This document corrects paragraph (f) of the compliance section that incorrectly references cycles instead of hours. This document changes cycles to hours in paragraph (f). In all other

respects, the original document, with the corrections published on April 23, 2003, remains the same.

EFFECTIVE DATE: April 28, 2003.

FOR FURTHER INFORMATION CONTACT:

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SUPPLEMENTARY INFORMATION: A final rule; request for comments airworthiness directive FR DOC. 03-8327, applicable to Rolls-Royce Deutschland Ltd & Co KG models BR700-710A1-10 and BR700-710A2-20 turbofan engines, was published in the **Federal Register** on April 11, 2003 (68 FR 17727). The following correction is needed:

■ On page 17729, in the second column, under Repetitive Inspections heading, paragraph (f), third line, which reads "500 CSN, in accordance with paragraphs" is corrected to read "500 hours-since-new, in accordance with paragraphs".

Issued in Burlington, MA, on May 2, 2003.

Francis A. Favara,

Acting Manager, Engine and Propeller Directorate, Aircraft Certification Service.

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2002-CE-44-AD; Amendment 39-13142; AD 2003-09-13]

RIN 2120-AA64

Airworthiness Directives; the New Piper Aircraft, Inc. Models PA-23, PA-23-160, PA-23-235, PA-23-250, and PA-E23-250 Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule.

SUMMARY: This amendment adopts a new airworthiness directive (AD) that applies to certain The New Piper Aircraft, Inc. (Piper) Models PA-23, PA-23-160, PA-23-235, PA-23-250, and PA-E23-250 airplanes that do not incorporate a certain design flap control torque tube or torque tube assembly. This AD requires you to repetitively inspect the flap control torque tube for cracks, corrosion, wear, or elongation of the attachment bolt holes (referred to as damage); and requires you to replace any damaged torque tube with either an improved design flap control torque tube or flap control torque tube assembly. The repetitive inspections will no longer be necessary when the improved design torque tube or torque tube assembly is installed. This AD is the result of several reports of damage found in the flap control torque tube on the affected airplanes. The actions specified by this AD are intended to detect and correct damage to the flap control torque tube, which could result in failure of the flap operating system. If such failure occurred during landing or takeoff, then a split flap condition