

olive industry and all interested persons were invited to attend the meeting and participate in committee deliberations on all issues. Like all committee meetings, the December 11, 2002, meeting was a public meeting and all entities, both large and small, were able to express views on this issue. Finally, interested persons are invited to submit information on the regulatory and informational impacts of this action on small businesses.

This proposed rule would impose no additional reporting or recordkeeping requirements on California olive handlers. As with all Federal marketing order programs, reports and forms are periodically reviewed to reduce information requirements and duplication by industry and public sector agencies.

USDA has not identified any relevant Federal rules that duplicate, overlap, or conflict with this rule.

A small business guide on complying with fruit, vegetable, and specialty crop marketing agreements and orders may be viewed at: <http://www.ams.usda.gov/fv/moab.html>. Any questions about the compliance guide should be sent to Jay Guerber at the previously mentioned address in the **FOR FURTHER INFORMATION CONTACT** section.

A 30-day comment period is provided to allow interested persons to respond to this proposed rule. Thirty days is deemed appropriate because: (1) The 2003 fiscal year began on January 1, 2003, and the marketing order requires that the rate of assessment for each fiscal year apply to all assessable olives handled during such fiscal year; (2) the committee needs to have sufficient funds to pay its expenses which are incurred on a continuous basis; and (3) handlers are aware of this action which was unanimously recommended by the committee at a public meeting and is similar to other assessment rate actions issued in past years.

List of Subjects in 7 CFR Part 932

Marketing agreements, Olives, Reporting and recordkeeping requirements.

For the reasons set forth in the preamble, 7 CFR part 932 is proposed to be amended as follows:

PART 932—OLIVES GROWN IN CALIFORNIA

1. The authority citation for 7 CFR part 932 continues to read as follows:

Authority: 7 U.S.C. 601–674.

2. Section 932.230 is revised to read as follows:

§ 932.230 Assessment rate.

On and after January 1, 2003, an assessment rate of \$13.89 per ton is established for California olives.

Dated: March 4, 2003.

A.J. Yates,
Administrator, Agricultural Marketing Service.

[FR Doc. 03–5561 Filed 3–7–03; 8:45 am]

BILLING CODE 3410–02–P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 94–ANE–08–AD]

RIN 2120–AA64

Airworthiness Directives; Turbomeca Arriel 1 Series Turboshaft Engines

AGENCY: Federal Aviation Administration, DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: The Federal Aviation Administration (FAA) proposes to supersede an existing airworthiness directive (AD), applicable to Turbomeca Arriel 1 A, 1 A1, 1 A2, 1 B, 1 C, 1 C1, 1 C2, 1 D, and 1 D1 turboshaft engines. That AD currently requires repetitive checks for engine rubbing noise during gas generator rundown following engine shutdown, and for free rotation of the gas generator by rotating the compressor manually after the last flight of the day. In addition, the AD 95–11–01 requires installation of modification TU 202 or TU 197 as terminating action to the repetitive checks. This proposal would add additional engine models to the applicability section, would eliminate the installation of modification TU 197 as a terminating action to the repetitive checks, would require additional inspections for engines that have modification TU 197 installed, and would require the replacement of modifications TU 76 and TU 197 with modification TU 202, as a terminating action to the repetitive checks and inspections. This proposal is prompted by a report of an in-flight engine shutdown on an engine that had modification TU 197 installed, and the need to update the modification standard on certain engine models. The actions specified by the proposed AD are intended to prevent engine failure due to rubbing of the 2nd stage turbine disk on the 2nd stage turbine nozzle guide vanes, which could result in complete engine failure and damage to the helicopter.

DATES: Comments must be received by May 9, 2003.

ADDRESSES: Submit comments in triplicate to the Federal Aviation Administration (FAA), New England Region, Office of the Regional Counsel, Attention: Rules Docket No. 94–ANE–08–AD, 12 New England Executive Park, Burlington, MA 01803–5299. Comments may be inspected at this location, by appointment, between 8 a.m. and 4:30 p.m., Monday through Friday, except Federal holidays. Comments may also be sent via the Internet using the following address: 9-ane-adcomment@faa.gov. Comments sent via the Internet must contain the docket number in the subject line.

The service information referenced in the proposed rule may be obtained from 40220 Tarnos, France; telephone (33) 05 59 64 40 00, fax (33) 05 59 64 60 80. This information may be examined, by appointment, at the FAA, New England Region, Office of the Regional Counsel, 12 New England Executive Park, Burlington, MA.

FOR FURTHER INFORMATION CONTACT:

Antonio Cancelliere, Aerospace Engineer, Engine Certification Office, FAA, Engine and Propeller Directorate, 12 New England Executive Park, Burlington, MA 01803–5299; telephone (781) 238–7751; fax (781) 238–7199.

SUPPLEMENTARY INFORMATION:

Comments Invited

Interested persons are invited to participate in the making of the proposed rule by submitting such written data, views, or arguments as they may desire. Communications should identify the Rules Docket number and be submitted in triplicate to the address specified above. All communications received on or before the closing date for comments, specified above, will be considered before taking action on the proposed rule. The proposals contained in this action may be changed in light of the comments received.

Comments are specifically invited on the overall regulatory, economic, environmental, and energy aspects of the proposed rule. All comments submitted will be available, both before and after the closing date for comments, in the Rules Docket for examination by interested persons. A report summarizing each FAA-public contact concerned with the substance of this proposal will be filed in the Rules Docket.

Commenters wishing the FAA to acknowledge receipt of their comments submitted in response to this action must submit a self-addressed, stamped

postcard on which the following statement is made: "Comments to Docket Number 94-ANE-08-AD." The postcard will be date stamped and returned to the commenter.

Availability of NPRM's

Any person may obtain a copy of this NPRM by submitting a request to the FAA, New England Region, Office of the Regional Counsel, Attention: Rules Docket No. 94-ANE-08-AD, 12 New England Executive Park, Burlington, MA 01803-5299.

Discussion

On May 15, 1995, the FAA issued airworthiness directive (AD) 95-11-01, Amendment 39-9235 (60 FR 27023, May 22, 1995), applicable to Turbomeca Arriel 1 series turboshaft engines, to require repetitive checks for engine rubbing noise during gas generator rundown following engine shutdown, and for free rotation of the gas generator by rotating the compressor manually at a daily interval until installation of improved 2nd stage turbine nozzle guide vanes. That action was prompted by comments submitted by operators of the affected engines in response to a previous AD and the availability of improved design 2nd stage turbine nozzle guide vanes. That condition, if not corrected, could result in engine failure due to rubbing of the 2nd stage turbine disk on the 2nd stage turbine nozzle guide vanes, which could result in complete engine failure and damage to the helicopter.

The Direction Generale de L'Aviation Civile (DGAC), which is the airworthiness authority for France, notified the FAA that an unsafe condition may exist on Turbomeca Arriel 1 B, 1 C, 1 C1, 1 C2, 1 D, and 1 D1 turboshaft engines.

Since AD 95-11-01 was issued, the DGAC advises that it has received a report of an in-flight engine shutdown on an engine that had improved 2nd stage turbine nozzle guide vanes, modification TU 197, installed. In this particular event, a crack initiated in a machined slot located between the vanes on the inner ring. The crack propagated and resulted in separation of the inner ring. This failure mode is different than that experienced on engines that have premodification TU 197 2nd stage turbine nozzle guide vanes installed. Installation of modification TU 197 is identified as a terminating action to the repetitive checks for the current AD. As a result of this event, the proposed AD requires additional borescope inspections for engines that have modification TU 197 installed.

The DGAC has also determined that modification TU 76 should be replaced with modification TU 202 on Arriel 1 B, 1 D, and 1 D1 engines. The manufacturer has informed the FAA that modification TU 202 may have already been installed on all Arriel 1 B, 1 D, and 1 D1 engines installed on helicopters of U.S. registry. However, for completeness, the proposed AD requires the removal of modification TU 76 or TU 197 and replacement with modification TU 202 before further flight after the effective date of this AD, to cover any potential engines that may not have been modified already.

The proposed AD will also require replacement of the 2nd stage nozzle guide vanes, having modification TU 197, with modification TU 202 on all Arriel 1 A, 1 A1, 1 A2, 1 C, 1 C1, 1 C2, 1 E2, 1 K, 1 K1, 1 S, and 1 S1 engines at next engine shop visit after the effective date of the proposed AD, but no later than December 31, 2006. Installation of modification TU 202 constitutes terminating action to the repetitive checks and inspections.

The Arriel 1 E2, 1 K, 1 K1, 1 S, and 1 S1 engine models have also been added to the applicability section of the proposed AD since they are susceptible to the same problem.

Manufacturer's Service Information

Turbomeca has issued the following Arriel 1 service bulletins (SBs) and alert service bulletins (ASBs):

- SB No. 292 72 0181, Update 3, dated September 15, 1995, that describes procedures for checking for unusual noise during gas generator rundown on engine shutdown and after the last flight of the day.
- ASB No. A292 72 0212, Update 5, dated August 8, 2001, that describes procedures for post Module TU 197 initial and repetitive borescope inspections of the nozzle guide vanes for cracks.
- ASB No. A292 72 0150, Update 6, dated September 4, 2000, that describes procedures for replacing modifications TU 76 and TU 197 with modification TU 202.

The DGAC has classified ASB No. A292 72 0212, Update 5, dated August 8, 2001, as mandatory and issued AD DGAC 98-311 (A) R1, dated October 7, 1998, in order to assure the airworthiness of these Turbomeca engines in France.

Differences Between the Manufacturer's Service Information and This Proposed AD

Turbomeca SB No. 292 72 0181 allows 50 flight hours between checks for unusual noise during gas generator

rundown on Arriel 1 A, 1 A1, and 1 A2 engines with modification TU 76; and 1 C, 1 C1, 1 C2, 1 K, 1 K1, and 1 S engines before modification to TU 197 or TU 202. This proposed AD would require that the checks be performed during engine shutdown after the last flight of the day or after a 5 second ventilation.

Bilateral Agreement Information

This engine model is manufactured in France and is type certificated for operation in the United States under the provisions of section 21.29 of the Federal Aviation Regulations (14 CFR 21.29) and the applicable bilateral airworthiness agreement. Pursuant to this bilateral airworthiness agreement, the DGAC has kept the FAA informed of the situation described above. The FAA has examined the findings of the DGAC, reviewed all available information, and determined that AD action is necessary for products of this type design that are certificated for operation in the United States.

Proposed Requirements of This AD

Since an unsafe condition has been identified that is likely to exist or develop on other Turbomeca Arriel 1 A, 1 A1, 1 A2, 1 B, 1 C, 1 C1, 1 C2, 1 D, 1 D1, 1 E2, 1 K, 1 K1, 1 S, and 1 S1 turboshaft engines of the same type design that are installed on helicopters registered in U.S., the proposed AD would supersede AD 95-11-01 to:

- Add additional engine models to the applicability section, and
- Eliminate the installation of modification TU 197 as a terminating action to the repetitive checks, and
- Require additional inspections for engines that have modification TU 197 installed, and
- Require the replacement of modifications TU 76 and TU 197 with modification TU 202 on Arriel 1 B, 1 D, and 1 D1 engines before further flight, and
- Replacement of TU 197 with modification TU 202 as a terminating action to the repetitive checks and inspections.

Economic Analysis

There are approximately 487 engines of the affected design in the worldwide fleet. The FAA estimates that 47 engines installed on aircraft of U.S. registry would be affected by this proposed AD, that it would take approximately 4 work hours per engine to do the proposed inspections, including removal and installation of the gas generator module, and that the average labor rate is \$60 per work hour. Required parts would cost approximately \$18,600 per engine. Based on these figures, the cost per

inspection to U.S. operators is estimated to be \$885,480. The manufacturer has advised the DGAC that it may provide modification TU 202 at no cost to the operator, thereby substantially reducing the cost of this proposed rule.

Regulatory Analysis

This proposed rule does not have federalism implications, as defined in Executive Order 13132, because it would not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government. Accordingly, the FAA has not consulted with state authorities prior to publication of this proposed rule.

For the reasons discussed above, I certify that this proposed regulation (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) if promulgated, will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A copy of the draft regulatory evaluation prepared for this action is contained in the Rules Docket. A copy of it may be obtained by contacting the Rules Docket at the location provided under the caption **ADDRESSES**.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

The Proposed Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration proposes to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by removing Amendment 39-9235, (60 FR 27023, May 22, 1995), and by adding a new airworthiness directive:

Turbomeca: Docket No. 94-ANE-08-AD. Supersedes AD 95-11-01, Amendment 39-9235.

Applicability: This airworthiness directive (AD) is applicable to Turbomeca turboshaft engine models Arriel 1 A, 1 A1, 1 A2, 1 B, 1 C, 1 C2, 1 D, 1 D1, 1 E2, 1 K, 1 K1, 1 S, and 1 S1 that have not incorporated modification TU 202. These engines are installed on but not limited to Eurocopter AS-350 B, B1, and B2; SA-365 C, C2, N, N1, and N2; MBB-BK 117 C-1 and C-2, Sikorsky S-76 C, and Agusta A109 K2 helicopters.

Note 1: This airworthiness directive (AD) applies to each engine identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For engines that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (k) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Compliance with this AD is required as indicated, unless already done. To prevent engine failure due to rubbing of the 2nd stage turbine disk on the 2nd stage nozzle guide vanes, which could result in complete engine failure and damage to the helicopter, do the following:

(a) For Turbomeca Arriel 1 A, 1 A1, 1 A2, 1 B, 1 C, 1 C1, 1 C2, 1 D, 1 D1, 1 E2, 1 K, 1 K1, 1 S, and 1 S1 turboshaft engines that have incorporated modification TU 202, no further action is required.

(b) For Turbomeca Arriel turboshaft engines Models 1 B, 1 D, or 1 D1 that have modification TU 76 or TU 197 installed, before further flight after the effective date of this AD, replace modification TU 76 or TU 197 with modification TU 202 in accordance with 2.B.(1) through 2.C.(2) of Arriel 1 Alert Service Bulletin (ASB) No. A292 72 0150, Update 6, dated September 4, 2000.

Daily Inspection for Engine Rubbing and Free Rotation

(c) For Arriel 1 A, 1 A1, 1 A2, 1 C, 1 C1, 1 C2, 1 E2, 1 K, 1 K1, 1 S, and 1 S1 engines

with modification TU 197 installed, perform the following daily checks:

(1) After the last flight of the day or after a ventilation (maximum of 5 seconds), immediately after engine stopping, listen for unusual engine rubbing noise during the gas generator rundown, and

(2) During the check after the last flight of the day, when the T4 temperature is below 150°C (302°F), perform a ventilation (5 seconds maximum) during gas generator rundown or check for free rotation of the gas generator and unusual noise by turning the compressor by hand.

(3) If any rubbing noise is heard and the source of the noise cannot be identified, replace module M03.

Initial Borescope Inspection

(d) For Arriel 1 A, 1 A1, 1 A2, 1 C, 1 C1, 1 C2, 1 E2, 1 K, 1 K1, 1 S, and 1 S1 engines with modification TU 197 installed, do the following:

(1) Perform initial borescope inspections for cracks of the second stage nozzle guide vanes (NGV2) in accordance with 2.B.(a) through 2.B.(c)(2) of Turbomeca ASB No. A292 72 0212, Update 5, dated August 8, 2001, and the schedules specified in the following Table 1:

TABLE 1.—INITIAL BORESCOPE INSPECTION

Number of cycles-since-new or overhaul (CSN) on the effective date of this AD.	Initial inspection
(1) Modules M03 with fewer than 1,000 CSN.	Before accumulating 1,100 CSN.
(2) Modules M03 with 1,000 CSN or greater.	Within 100 additional cycles-in-service (CIS) after the effective date of this AD.

(2) If the 2nd stage nozzle guide vanes do not meet the acceptance criteria specified in 2.B.(c)(2) of ASB A292 72 0212, Update 5, dated August 8, 2001, replace module M03.

First Repetitive Borescope Inspection

(e) Thereafter, for Arriel 1 A, 1 A1, 1 A2, 1 C, 1 C1, 1 C2, 1 E2, 1 K, 1 K1, 1 S, and 1 S1 engines with modification TU 197 installed, do the following:

(1) Perform the first repetitive borescope inspection for cracks of the NGV2 in accordance with 2.B.(a) through 2.(c)(2) of Turbomeca ASB No. A292 72 0212, Revision 5, dated August 8, 2001, and the schedules specified in the following Table 2:

TABLE 2.—REPETITIVE BORESCOPE INSPECTIONS

If module M03 has already been checked	Then repeat inspection
(i) Once, before 900 CSN	Before 1,100 CSN and then between 1,900 and 2,100 CSN.
(ii) Twice, before 900 CSN without propagation of cracks recorded between the first and second check.	Before 1,500 CSN.
(iii) Twice, before 900 CSN with propagation of cracks recorded between the first and second check.	Before 1,100 CSN and then between 1,900 and 2,100 CSN.
(iv) Once, after 900 CSN	Between 1,900 and 2,100 CSN.

(2) If the 2nd stage nozzle guide vanes do not meet the acceptance criteria specified in 2.B.(c)(2) of ASB A292 72 0212, Update 5, dated August 8, 2001, replace module M03.

Subsequent Repetitive Borescope Inspection

(f) Thereafter, for Arriel 1 A, 1 A1, 1 A2, 1 C, 1 C1, 1 C2, 1 E2, 1 K, 1 K1, 1 S, and 1 S1 engines with modification TU 197 installed, do the following:

(1) Repeat the borescope inspection of the NGV2 in accordance with 2.B.(a) through 2.B.(c)(2) of Turbomeca ASB No. A292 72 0212, Update 5, dated August 8, 2001, at intervals not to exceed 2,100 cycles-since-last-inspection (CSLI).

(2) If the 2nd stage nozzle guide vanes do not meet the acceptance criteria specified in 2.B.(c)(2) of ASB A292 72 0212, Update 5, dated August 8, 2001, replace module M03.

Replacement of Modification TU 197

(g) For 1 A, 1 A1, 1 A2, 1 C, 1 C1, 1 C2, 1 E2, 1K, 1 K1, 1 S, and 1 S1 engines that have modification TU 197 installed, install the improved 2nd stage nozzle guide vanes, modification TU 202 at next shop visit after the effective date of this AD, but not later than December 31, 2006, in accordance with 2.B. through 2.C. of Arriel 1 ASB No. A292 72 0150, Update No. 6, dated September 4, 2000.

Terminating Action

(h) Installation of the improved 2nd stage nozzle guide vane, modification TU 202, constitutes terminating action to the checks and inspections required by paragraphs (c)(1), (c)(2), and (d)(1) through (d)(3) of this AD.

(i) The checks required by paragraph (c)(1) and (c)(2) of this AD may be performed by the pilot holding at least a private pilot certificate as an exception to the requirements of part 43 of the Federal Aviation Regulations (14 CFR part 43). The checks must be recorded in accordance with §§ 43.9 and 91.417(a)(2)(v) of the Federal Aviation Regulations (14 CFR 43.9 and 14 CFR 91.417(a)(2)(v)), and the records must be maintained as required by the applicable Federal Aviation Regulation.

Alternative Methods of Compliance

(j) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, Engine Certification Office (ECO). Operators must submit their request through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, ECO.

Note 2: Information concerning the existence of approved alternative methods of compliance with this airworthiness directive, if any, may be obtained from the ECO.

Special Flight Permits

(k) Special flight permits may be issued in accordance with §§ 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the helicopter to a location where the requirements of this AD can be done.

Note 3: The subject of this AD is addressed in DGAC airworthiness directive DGAC 98-311 (A) R1, dated October 7, 1998.

Issued in Burlington, Massachusetts, on March 3, 2003.

Jay J. Pardee,

Manager, Engine and Propeller Directorate, Aircraft Certification Service.

[FR Doc. 03-5577 Filed 3-7-03; 8:45 am]

BILLING CODE 4910-13-P

DEPARTMENT OF THE INTERIOR

Bureau of Land Management

43 CFR Part 4100

[WO-220-1020-24 1A]

RIN 1004-AD42

Grazing Administration—Exclusive of Alaska

AGENCY: Bureau of Land Management, Interior.

ACTION: Advance notice of proposed rulemaking for proposed amendments to the BLM's Grazing Administration Regulations and announcement of public meetings; correction.

SUMMARY: The Bureau of Land Management (BLM) in this document corrects one internet address and removes reference to another internet address to which the public cannot get access that appear in the advance notice of proposed rulemaking regarding proposed amendments to BLM's Grazing Administration Regulations, published in the **Federal Register** of March 3, 2003.

FOR FURTHER INFORMATION CONTACT: Ted Hudson, 202-452-5042.

Correction

In proposed rule FR Doc. 03-4933, beginning on page 9964 in the issue of March 3, 2003, make the following corrections:

1. In the Addresses section, on page 9964 in the 3rd column, correct the internet address immediately following the subheading "Direct Internet response" to read: "<http://www.blm.gov/nhp/news/regulatory/index.htm>".

2. In the Supplementary Information section, on page 9966, in the 2nd column, correct the final paragraph of the advance notice of proposed rulemaking by revising it to read:

"Additional information about BLM's Rangeland, Soils, Water, and Air Program is available at any State Office or field office of the Bureau of Land Management."

Dated: March 5, 2003.

Jim Hughes,

Deputy Director, Bureau of Land Management.

[FR Doc. 03-5718 Filed 3-7-03; 8:45 am]

BILLING CODE 4310-84-P

FEDERAL COMMUNICATIONS COMMISSION

47 CFR Part 73

[DA 03-437; MB Docket No. 03-47, RM-10592]

Radio Broadcasting Services; Midlothian, Orange and South Hill, VA, and Reidsville, NC

AGENCY: Federal Communications Commission.

ACTION: Proposed rule.

SUMMARY: This document sets forth a proposal to amend the FM Table of Allotment of the Commission's rules. The Commission requests comment on a petition filed by Piedmont Communications, Inc. and Old Belt Broadcasting Corporation (together, "Joint Petitioners") pursuant to section 1.420(i) of the Commission's rules. Joint Petitioners propose to change the community of allotment and the corresponding channel allotment for Channel 255A at Orange, Virginia, to Channel 255B1 at Midlothian, Virginia, and to modify the license of WJMA-FM accordingly. In order to facilitate those changes, Joint Petitioners further propose to substitute Channel 270A for Channel 255C3 at South Hill, Virginia, and to modify the WKSK-FM license to specify operation on Channel 270A. To accommodate this proposal, Joint Petitioners also request substitution of Channel 271C0 for Channel 271C at Reidsville, North Carolina. Channel 255B1 can be allotted to Midlothian in compliance with the Commission's minimum distance separation requirements with a site restriction of 12.7 km (7.9 miles) northwest of Midlothian. The coordinates for Channel 255B1 at Midlothian are 37-35-23 North Latitude and 77-44-49 West Longitude. Channel 270A can be allotted to South Hill in compliance with the Commission's minimum distance separation requirements with a site restriction of 12.4 km (7.7 miles) northwest of South Hill. The coordinates for Channel 270A at South Hill are 36-46-48 North Latitude and 78-15-04 West Longitude. Channel 271C0 can be allotted at Reidsville, North Carolina, at the current coordinates for Channel 271C. Because Midlothian is not listed in the United