

## Adoption of the Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

### PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

#### § 39.13 [Amended]

2. Section 39.13 is amended by adding a new airworthiness directive to read as follows:

**2003-04-23 Hartzell Propeller Inc.:**  
Amendment 39-13072. Docket No. 2001-NE-44-AD.

**Applicability:** This airworthiness directive (AD) is applicable to Hartzell Propeller Inc. model HC-B3TN-5( ) propellers, with part numbers (P/N's) T10176H(B)-5, T10176H(K)-5, T10176H-5, T10178H-11, T10178H-11R, T10178H(B)-11, and T10178H(B)-11R blades that are installed on Mitsubishi Heavy Industries, Ltd, MU-2 series airplanes.

**Note 1:** The parentheses indicate the presence or absence of an additional letter(s) which vary the basic propeller blade model designation. This AD still applies regardless of whether these letters are present or absent on the propeller blade model designation.

**Note 2:** This AD applies to each propeller identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For propellers that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (c) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

**Compliance:** Compliance with this AD is required within 200 flight hours or 1 year from the effective date of this AD, whichever occurs first, unless already done.

To prevent propeller blade separation, damage to the airplane, and possible loss of the airplane, do the following:

(a) Remove and replace propeller blades in accordance with paragraphs 3.A. through 3.C.(3) of the Accomplishment Instructions of Hartzell Propeller Inc. Service Bulletin (SB) HC-SB-61-250, Revision 1, dated April 8, 2002.

(b) After the effective date of this AD, do not install any propeller blade removed in accordance with paragraph (a) of this AD, on any airplane.

## Alternative Methods of Compliance

(c) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, Chicago Aircraft Certification Office (ACO). Operators must submit their request through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, Chicago ACO.

**Note 3:** Information concerning the existence of approved alternative methods of compliance with this airworthiness directive, if any, may be obtained from the Chicago ACO.

## Special Flight Permits

(d) Special flight permits may be issued in accordance with §§ 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the aircraft to a location where the requirements of this AD can be done.

## Documents That Have Been Incorporated by Reference

(e) The blade removal and replacement must be done in accordance with Hartzell Propeller Inc. Service Bulletin (SB) HC-SB-61-250, Revision 1, dated April 8, 2002. This incorporation by reference was approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Copies may be obtained from Hartzell Propeller Inc. Technical Publications Department, One Propeller Place, Piqua, OH 45356; telephone (937) 778-4200, fax (937) 778-4391. Copies may be inspected at the FAA, New England Region, Office of the Regional Counsel, 12 New England Executive Park, Burlington, MA; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

## Effective Date

(f) This amendment becomes effective on April 4, 2003.

Issued in Burlington, Massachusetts, on February 19, 2003.

**Jay J. Pardee,**

*Manager, Engine and Propeller Directorate, Aircraft Certification Service.*

[FR Doc. 03-4484 Filed 2-27-03; 8:45 am]

**BILLING CODE 4910-13-P**

## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 39

[Docket No. 2002-NM-100-AD; Amendment 39-13070; AD 2003-04-21]

**RIN 2120-AA64**

### Airworthiness Directives; Bombardier Model CL-600-2B19 (Regional Jet Series 440) Series Airplanes

**AGENCY:** Federal Aviation Administration, DOT.

**ACTION:** Final rule.

**SUMMARY:** This amendment adopts a new airworthiness directive (AD), applicable to certain Bombardier Model CL-600-2B19 (Regional Jet Series 440) series airplanes, that requires replacement of the overwing emergency exit placards, door weight placards, and no baggage placards with new placards. This action is necessary to prevent the inability of a passenger to open and dispose of the overwing emergency exit door during an emergency evacuation due to incorrect placards. This action is intended to address the identified unsafe condition.

**DATES:** Effective April 4, 2003.

The incorporation by reference of certain publications listed in the regulations is approved by the Director of the Federal Register as of April 4, 2003.

**ADDRESSES:** The service information referenced in this AD may be obtained from Bombardier, Inc., Canadair, Aerospace Group, P.O. Box 6087, Station Centre-ville, Montreal, Quebec H3C 3G9, Canada. This information may be examined at the Federal Aviation Administration (FAA), Transport Airplane Directorate, Rules Docket, 1601 Lind Avenue, SW., Renton, Washington; or at the FAA, New York Aircraft Certification Office, 10 Fifth Street, Third Floor, Valley Stream, New York; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

**FOR FURTHER INFORMATION CONTACT:** Dan Parrillo, Aerospace Engineer, Systems and Flight Test Branch, ANE-172, FAA, New York Aircraft Certification Office, 10 Fifth Street, Third Floor, Valley Stream, New York 11581; telephone (516) 256-7505; fax (516) 568-2716.

**SUPPLEMENTARY INFORMATION:** A proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) to include an airworthiness directive (AD) that is applicable to certain Bombardier Model CL-600-2B19 (Regional Jet Series 440) series airplanes was published in the **Federal Register** on August 7, 2002 (67 FR 51147). That action proposed to require replacement of the overwing emergency exit placards, door weight placards, and no baggage placards with new placards.

## Comments

Interested persons have been afforded an opportunity to participate in the making of this amendment. Due consideration has been given to the comments received.

### Requests That Credit Be Given for Previous Replacement

Two commenters request that the FAA revise the notice of proposed rulemaking (NPRM) to specify that replacement accomplished before the effective date of this AD per Bombardier Alert Service Bulletin A601R-11-077, dated July 12, 2001, is considered acceptable for compliance with the replacement specified in paragraph (a) of the NPRM. One commenter states that, with the exception of the pre-modification location of placards for certain airplanes, the changes embodied in Revision "A" of the service bulletin are not significant for the proper execution of the bulletin and, therefore, result in the same level of safety. Another commenter states that Canadian airworthiness directive CF-2002-12, dated February 4, 2002, allows credit for airplanes modified per the original service bulletin.

We agree. We have added a new paragraph (b) in the final rule to provide operators with credit for previously accomplishing Bombardier Alert Service Bulletin A601R-11-077, dated July 12, 2001, and have redesignated the subsequent paragraphs.

### Request To Clarify Reference to Attachments

One commenter requests clarification about the attachments specified in paragraph (a) of the NPRM. The commenter states that Bombardier Alert Service Bulletin A601R-11-077, Revision "A," does not reference "Attachments" within its text. However, the commenter notes that, following page 46 of 46 of the service bulletin, there are two pages—a comment sheet and a compliance sheet. The commenter recommends calling those pages by their given names.

We agree and have revised the names of those sheets in the final rule accordingly. In addition, for clarification purposes, we have changed the service bulletin citation throughout this final rule to exclude those sheets. As stated in the NPRM, this AD does not require operators to complete the comment and compliance sheets.

### Explanation of Change to Applicability

We have revised the applicability of the final rule to identify model designations as published in the most recent type certificate data sheet for the affected models.

### Conclusion

After careful review of the available data, including the comments noted above, we have determined that air safety and the public interest require the

adoption of the rule with the changes described previously. We have determined that these changes will neither increase the economic burden on any operator nor increase the scope of the AD.

### Cost Impact

We estimate that 284 Model CL-600-2B19 (Regional Jet Series 440) series airplanes of U.S. registry will be affected by this AD, that it will take approximately between 1 and 2 hours per airplane depending on the airplane configuration to accomplish the required actions, and that the average labor rate is \$60 per work hour. Required parts will cost approximately between \$47 and \$195 per airplane depending on the configuration of the airplane. Based on these figures, the cost impact of the AD on U.S. operators is estimated to be between \$30,388 and \$89,460, or \$107 and \$315 per airplane depending on the configuration of the airplane.

The cost impact figures discussed above are based on assumptions that no operator has yet accomplished any of the requirements of this AD action, and that no operator would accomplish those actions in the future if this AD were not adopted. The cost impact figures discussed in AD rulemaking actions represent only the time necessary to perform the specific actions actually required by the AD. These figures typically do not include incidental costs, such as the time required to gain access and close up, planning time, or time necessitated by other administrative actions. However, for affected airplanes within the period under the warranty agreement, we have been advised that the manufacturer has committed previously to its customers that it will bear the cost of the placard kits.

### Regulatory Impact

The regulations adopted herein will not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, it is determined that this final rule does not have federalism implications under Executive Order 13132.

For the reasons discussed above, I certify that this action (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) will not have a significant economic impact, positive or negative, on a

substantial number of small entities under the criteria of the Regulatory Flexibility Act. A final evaluation has been prepared for this action and it is contained in the Rules Docket. A copy of it may be obtained from the Rules Docket at the location provided under the caption **ADDRESSES**.

### List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

### Adoption of the Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

### PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

#### § 39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:

**2003-04-21 Bombardier, Inc. (Formerly Canadair):** Amendment 39-13070. Docket 2002-NM-100-AD.

*Applicability:* Model CL-600-2B19 (Regional Jet Series 440) series airplanes, certificated in any category, having the serial numbers listed in the following table:

TABLE—SERIAL NUMBERS

Serial Nos.
7003 through 7434 inclusive.
7436 through 7442 inclusive.
7444 through 7452 inclusive.
7454 through 7458 inclusive.
7460 through 7497 inclusive.
7499 through 7504 inclusive.

**Note 1:** This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (c) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not

been eliminated, the request should include specific proposed actions to address it.

**Compliance:** Required as indicated, unless accomplished previously.

To prevent the inability of a passenger to open and dispose of the overwing emergency exit door during an emergency evacuation due to incorrect placards, accomplish the following:

#### Replacement of Placards

(a) Within 12 months after the effective date of this AD, replace the overwing emergency exit placards, door weight placards, and no baggage placards with new placards (including cleaning of the applicable surface), as applicable, per Bombardier Alert Service Bulletin A601R-11-077, Revision "A," dated December 11, 2001, excluding Service Bulletin Comment Sheet-Facsimile Reply Sheet and CRJ 100/200 Service Bulletin Compliance Facsimile Reply Sheet.

(b) Replacement accomplished before the effective date of this AD per Bombardier Alert Service Bulletin A601R-11-077, dated July 12, 2001, is considered acceptable for compliance with the replacement specified in paragraph (a) of this AD.

#### Alternative Methods of Compliance

(c) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, New York Aircraft Certification Office (ACO), FAA. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, New York ACO.

**Note 2:** Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the New York ACO.

#### Special Flight Permits

(d) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

#### Incorporation by Reference

(e) Unless otherwise specified in this AD, the actions shall be done in accordance with Bombardier Alert Service Bulletin A601R-11-077, Revision "A," dated December 11, 2001, excluding Service Bulletin Comment Sheet-Facsimile Reply Sheet and CRJ 100/200 Service Bulletin Compliance Facsimile Reply Sheet. This incorporation by reference was approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Copies may be obtained from Bombardier, Inc., Canadair, Aerospace Group, P.O. Box 6087, Station A, Montreal, Quebec H3C 3G9, Canada. Copies may be inspected at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the FAA, New York Aircraft Certification Office, 10 Fifth Street, Third Floor, Valley Stream, New York; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

**Note 3:** The subject of this AD is addressed in Canadian airworthiness directive CF-2002-12, dated February 4, 2002.

#### Effective Date

(f) This amendment becomes effective on April 4, 2003.

Issued in Renton, Washington, on February 19, 2003.

**Ali Bahrami,**

*Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.*

[FR Doc. 03-4347 Filed 2-27-03; 8:45 am]

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## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 39

[Docket No. 2001-NM-289-AD; Amendment 39-13068; AD 2003-04-19]

**RIN 2120-AA64**

#### Airworthiness Directives; Fokker Model F.28 Mark 0070 and 0100 Series Airplanes

**AGENCY:** Federal Aviation Administration, DOT.

**ACTION:** Final rule.

**SUMMARY:** This amendment adopts a new airworthiness directive (AD), applicable to all Fokker Model F.28 Mark 0070 and 0100 series airplanes, that requires a one-time general visual inspection to detect any missing attachment bolts in the replaceable frame struts, and corrective actions, if necessary. This action is necessary to prevent excessive deformation of the floor structure in the event of rapid decompression in the lower cargo hold due to missing attachment bolts in the replaceable frame struts. Such deformation may result in the flight and engine control cables becoming jammed, and consequent reduced controllability of the airplane. This action is intended to address the identified unsafe condition.

**DATES:** Effective April 4, 2003.

The incorporation by reference of certain publications listed in the regulations is approved by the Director of the Federal Register as of April 4, 2003.

**ADDRESSES:** The service information referenced in this AD may be obtained from Fokker Services B.V., P.O. Box 231, 2150 AE Nieuw-Vennep, the Netherlands. This information may be examined at the Federal Aviation Administration (FAA), Transport Airplane Directorate, Rules Docket, 1601 Lind Avenue, SW., Renton,

Washington; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

**FOR FURTHER INFORMATION CONTACT:** Tom Rodriguez, Aerospace Engineer, International Branch, ANM-116, FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 227-1137; fax (425) 227-1149.

**SUPPLEMENTARY INFORMATION:** A proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) to include an airworthiness directive (AD) that is applicable to all Fokker Model F.28 Mark 0070 and 0100 series airplanes was published in the **Federal Register** on March 28, 2002 (67 FR 14891). That action proposed to require a one-time general visual inspection to detect any missing attachment bolts in the replaceable frame struts, and corrective actions, if necessary.

#### Comments

Interested persons have been afforded an opportunity to participate in the making of this amendment. Due consideration has been given to the comments received.

#### Support for the Proposed Rule

One commenter supports the proposed rule.

#### Request To Reference Latest Service Information

One commenter requests that Revision 1 to Fokker Service Bulletin SBF100-53-096 be cited as an acceptable source of service information for compliance with the proposed AD. That revision adds two figures to the service bulletin that are applicable to the operator's fleet.

The commenter also states that it notified the manufacturer of typographical errors in Figures 2 and 17 of the service bulletin. The manufacturer informed the commenter that a Service Bulletin Change Notification (SBCN) to correct the typographical errors would be issued. The commenter requests that a statement allowing the use of future service bulletin revisions and SBCNs be included in the proposed AD.

The FAA concurs with the commenter's request to reference Revision 1 of the service bulletin and the applicable SBCN. Since the issuance of the proposed AD, the manufacturer has issued Fokker Service Bulletin SBF100-53-096, Revision 1, dated November 22, 2001; and Fokker SBCN SBF100-53-096/02, dated January 28, 2002. The proposed rule references the original issue of the service bulletin, dated April 11, 2001, as the appropriate