

appropriate restrictions to the uses of the property; or

(2) Withhold the property from conveyance.

(b) Before completing any transaction that DOE guarantees, approves, regulates, or insures that is related to an area located in a floodplain, DOE shall inform any private party participating in the transaction of the hazards associated with locating facilities or structures in the floodplain.

§ 1022.22 Requests for authorizations or appropriations.

It is DOE policy to indicate in any requests for new authorizations or appropriations transmitted to the White House Office of Management and Budget, if a proposed action is located in a floodplain or wetland and whether the proposed action is in accord with the requirements of E.O. 11988 and E.O. 11990 and this part.

§ 1022.23 Applicant responsibilities.

DOE may require applicants for any use of real property (*e.g.*, license, easement, lease, transfer, or disposal), permits, certificates, loans, grants, contract awards, allocations, or other forms of assistance or other entitlement related to activities in a floodplain or wetland of the requirements of this part to provide information necessary for DOE to comply with this part.

§ 1022.24 Interagency cooperation.

If DOE and one or more agencies are directly involved in a proposed floodplain or wetland action, in accordance with DOE's NEPA or CERCLA procedures, DOE shall consult with such other agencies to determine if a floodplain or wetland assessment is required by Subpart B of this part, identify the appropriate lead or joint agency responsibilities, identify the applicable regulations, and establish procedures for interagency coordination during the environmental review process.

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2002-NM-200-AD]

RIN 2120-AA64

Airworthiness Directives; Saab Model SAAB 340B Series Airplanes Equipped With Hamilton Sundstrand Propellers

AGENCY: Federal Aviation Administration, DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: This document proposes the adoption of a new airworthiness directive (AD) that is applicable to certain Saab Model SAAB 340B series airplanes equipped with Hamilton Sundstrand propellers. This proposal would require a one-time inspection of two remote controlled circuit breakers (RCCB), located in specific electrical compartments, to identify the part number, and replacement of the RCCBs with new RCCBs having a different part number if necessary. This action is necessary to ensure removal of 35-ampere (amp) RCCBs on a 50-amp electrical circuit. Incorrect RCCBs on an electrical circuit could result in erroneous tripping of the RCCBs (even though an overload condition does not exist), premature failure of the RCCBs, loss of power to the feather pump system, and consequent reduced controllability of the airplane. This action is intended to address the identified unsafe condition.

DATES: Comments must be received by December 18, 2002.

ADDRESSES: Submit comments in triplicate to the Federal Aviation Administration (FAA), Transport Airplane Directorate, ANM-114, Attention: Rules Docket No. 2002-NM-200-AD, 1601 Lind Avenue, SW., Renton, Washington 98055-4056. Comments may be inspected at this location between 9 a.m. and 3 p.m., Monday through Friday, except Federal holidays. Comments may be submitted via fax to (425) 227-1232. Comments may also be sent via the Internet using the following address: 9-anm-nprmcomment@faa.gov. Comments sent via fax or the Internet must contain "Docket No. 2002-NM-200-AD" in the subject line and need not be submitted in triplicate. Comments sent via the Internet as attached electronic files must be formatted in Microsoft Word 97 for Windows or ASCII text.

The service information referenced in the proposed rule may be obtained from

Saab Aircraft AB, SAAB Aircraft Product Support, S-581.88, Linköping, Sweden. This information may be examined at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington.

FOR FURTHER INFORMATION CONTACT:

Rosanne Ryburn, Aerospace Engineer, International Branch, ANM-116, FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 227-2139; fax (425) 227-1149.

SUPPLEMENTARY INFORMATION:

Comments Invited

Interested persons are invited to participate in the making of the proposed rule by submitting such written data, views, or arguments as they may desire. Communications shall identify the Rules Docket number and be submitted in triplicate to the address specified above. All communications received on or before the closing date for comments, specified above, will be considered before taking action on the proposed rule. The proposals contained in this action may be changed in light of the comments received.

Submit comments using the following format:

- Organize comments issue-by-issue. For example, discuss a request to change the compliance time and a request to change the service bulletin reference as two separate issues.
- For each issue, state what specific change to the proposed AD is being requested.
- Include justification (*e.g.*, reasons or data) for each request.

Comments are specifically invited on the overall regulatory, economic, environmental, and energy aspects of the proposed rule. All comments submitted will be available, both before and after the closing date for comments, in the Rules Docket for examination by interested persons. A report summarizing each FAA-public contact concerned with the substance of this proposal will be filed in the Rules Docket.

Commenters wishing the FAA to acknowledge receipt of their comments submitted in response to this action must submit a self-addressed, stamped postcard on which the following statement is made: "Comments to Docket Number 2002-NM-200-AD." The postcard will be date stamped and returned to the commenter.

Availability of NPRMs

Any person may obtain a copy of this NPRM by submitting a request to the FAA, Transport Airplane Directorate,

ANM-114, Attention: Rules Docket No. 2002-NM-200-AD, 1601 Lind Avenue, SW., Renton, Washington 98055-4056.

Discussion

The Luftfartsverket (LFV), which is the airworthiness authority for Sweden, notified the FAA that an unsafe condition may exist on certain Saab Model SAAB 340B series airplanes. The LFV advises that these airplanes were manufactured with 50-ampere (amp) remote controlled circuit breakers (RCCB) installed. However, the illustrated parts catalog incorrectly references a 35-amp RCCB, which is specific to Dowty propellers. A 35-amp RCCB on a 50-amp electrical circuit, if not corrected, could result in erroneous tripping of the RCCBs (even though an overload condition does not exist), premature failure of the RCCBs, loss of power to the feather pump system, and consequent reduced controllability of the airplane.

Explanation of Relevant Service Information

Saab has issued Service Bulletin 340-61-038, dated January 30, 2002, which describes procedures for a one-time inspection to identify the part number of the RCCB located in electrical compartment 407VU and the RCCB located in electrical compartment 408VU, and replacement of RCCBs with certain part numbers with new RCCBs having different part numbers if necessary. Accomplishment of the actions specified in the service bulletin is intended to adequately address the identified unsafe condition. The LFV classified this service bulletin as mandatory and issued Swedish airworthiness directive 1-172, dated January 31, 2002, in order to assure the continued airworthiness of these airplanes in Sweden.

FAA's Conclusions

This airplane model is manufactured in Sweden and is type certificated for operation in the United States under the provisions of section 21.29 of the Federal Aviation Regulations (14 CFR 21.29) and the applicable bilateral airworthiness agreement. Pursuant to this bilateral airworthiness agreement, the LFV has kept the FAA informed of the situation described above. The FAA has examined the findings of the LFV, reviewed all available information, and determined that AD action is necessary for products of this type design that are certificated for operation in the United States.

Explanation of Requirements of Proposed Rule

Since an unsafe condition has been identified that is likely to exist or develop on other airplanes of the same type design registered in the United States, the proposed AD would require accomplishment of the actions specified in the service bulletin described previously.

Cost Impact

The FAA estimates that 251 airplanes of U.S. registry would be affected by this proposed AD, that it would take approximately 1 work hour per airplane to accomplish the proposed inspection, and that the average labor rate is \$60 per work hour. Based on these figures, the cost impact of the proposed inspection on U.S. operators is estimated to be \$15,060 or \$60 per airplane.

The cost impact figure discussed above is based on assumptions that no operator has yet accomplished any of the proposed requirements of this AD action, and that no operator would accomplish those actions in the future if this proposed AD were not adopted. The cost impact figures discussed in AD rulemaking actions represent only the time necessary to perform the specific actions actually required by the AD. These figures typically do not include incidental costs, such as the time required to gain access and close up, planning time, or time necessitated by other administrative actions.

Regulatory Impact

The regulations proposed herein would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, it is determined that this proposal would not have federalism implications under Executive Order 13132.

For the reasons discussed above, I certify that this proposed regulation (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) if promulgated, will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A copy of the draft regulatory evaluation prepared for this action is contained in the Rules Docket. A copy of it may be obtained by contacting the Rules Docket at the location provided under the caption

ADDRESSES.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

The Proposed Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration proposes to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:

SAAB Aircraft AB: Docket 2002-NM-200-AD.

Applicability: Model SAAB 340B series airplanes equipped with Hamilton Sundstrand propellers, certificated in any category.

Note 1: This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (c) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Required as indicated, unless accomplished previously.

To prevent erroneous tripping of the remote controlled circuit breakers (RCCB) (even though an overload condition does not exist), premature failure of the RCCBs, loss of power to the feather pump system, and consequent reduced controllability of the airplane, accomplish the following:

Inspection and Replacement, If Necessary

(a) Within 6 months after the effective date of this AD, perform a one-time inspection of RCCB 29KFC located in electrical compartment 407VU, and RCCB 30KFC located in electrical compartment 408VU, to identify the part number, per the Accomplishment Instructions of Saab Service Bulletin 340-61-038, dated January 30, 2002.

(1) If both RCCBs are identified as part number (P/N) M83383-01-09, no further action is required by this paragraph.

(2) If any RCCB is identified as P/N M83383-02-07, prior to further flight, replace the RCCB with an RCCB having P/N M83383-01-09, per the service bulletin.

Part Installation

(b) As of the effective date of this AD, no person shall install an RCCB, P/N M83383-02-07, in electrical compartment 407VU or electrical compartment 408VU, on any airplane.

Alternative Methods of Compliance

(c) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, International Branch, ANM-116, Transport Airplane Directorate, FAA. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, International Branch, ANM-116.

Note 2: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the International Branch, ANM-116.

Special Flight Permits

(d) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

Note 3: The subject of this AD is addressed in Swedish airworthiness directive 1-172, dated January 31, 2002.

Issued in Renton, Washington, on November 8, 2002.

Vi L. Lipski,

*Manager, Transport Airplane Directorate,
Aircraft Certification Service.*

[FR Doc. 02-29116 Filed 11-15-02; 8:45 am]

BILLING CODE 4910-13-P

DEPARTMENT OF TRANSPORTATION**Federal Aviation Administration****14 CFR Part 39**

[Docket No. 2001-NM-295-AD]

RIN 2120-AA64

Airworthiness Directives; Boeing Model 777-200 and 777-300 Series Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: This document proposes the adoption of a new airworthiness directive (AD) that is applicable to certain Boeing Model 777-200 and 777-300 series airplanes. This proposal would require application of high-temperature sealant to the strut aft dry bay. This action is necessary to prevent leakage of hydraulic fluid into the strut aft dry bay, where high temperatures associated with the adjacent primary

exhaust nozzle may ignite the fluid, resulting in an uncontrolled fire in the strut aft dry bay. This action is intended to address the identified unsafe condition.

DATES: Comments must be received by January 2, 2003.

ADDRESSES: Submit comments in triplicate to the Federal Aviation Administration (FAA), Transport Airplane Directorate, ANM-114, Attention: Rules Docket No. 2001-NM-295-AD, 1601 Lind Avenue, SW., Renton, Washington 98055-4056. Comments may be inspected at this location between 9:00 a.m. and 3:00 p.m., Monday through Friday, except Federal holidays. Comments may be submitted via fax to (425) 227-1232. Comments may also be sent via the Internet using the following address: *9-anm-nprmcomment@faa.gov*. Comments sent via fax or the Internet must contain "Docket No. 2001-NM-295-AD" in the subject line and need not be submitted in triplicate. Comments sent via the Internet as attached electronic files must be formatted in Microsoft Word 97 for Windows or ASCII text.

The service information referenced in the proposed rule may be obtained from Boeing Commercial Airplane Group, P.O. Box 3707, Seattle, Washington 98124-2207. This information may be examined at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington. **FOR FURTHER INFORMATION CONTACT:** John Vann, Aerospace Engineer, Propulsion Branch, ANM-140S, FAA, Seattle Aircraft Certification Office, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 227-1024; fax (425) 227-1181.

SUPPLEMENTARY INFORMATION:**Comments Invited**

Interested persons are invited to participate in the making of the proposed rule by submitting such written data, views, or arguments as they may desire. Communications shall identify the Rules Docket number and be submitted in triplicate to the address specified above. All communications received on or before the closing date for comments, specified above, will be considered before taking action on the proposed rule. The proposals contained in this action may be changed in light of the comments received.

Submit comments using the following format:

- Organize comments issue-by-issue. For example, discuss a request to change the compliance time and a request to change the service bulletin reference as two separate issues.

- For each issue, state what specific change to the proposed AD is being requested.

- Include justification (e.g., reasons or data) for each request.

Comments are specifically invited on the overall regulatory, economic, environmental, and energy aspects of the proposed rule. All comments submitted will be available, both before and after the closing date for comments, in the Rules Docket for examination by interested persons. A report summarizing each FAA-public contact concerned with the substance of this proposal will be filed in the Rules Docket.

Commenters wishing the FAA to acknowledge receipt of their comments submitted in response to this action must submit a self-addressed, stamped postcard on which the following statement is made: "Comments to Docket Number 2001-NM-295-AD." The postcard will be date stamped and returned to the commenter.

Availability of NPRMs

Any person may obtain a copy of this NPRM by submitting a request to the FAA, Transport Airplane Directorate, ANM-114, Attention: Rules Docket No. 2001-NM-295-AD, 1601 Lind Avenue, SW., Renton, Washington 98055-4056.

Discussion

The FAA has received a report indicating that one operator had found coked hydraulic fluid in the strut aft dry bay, which is located directly above the primary exhaust nozzle. Investigation revealed that hydraulic fluid had leaked from the strut aft fairing through an unsealed gap between the strut aft bulkhead, the diagonal brace fitting, and the cowl fairing. This unsealed gap, if not corrected, permits leakage of hydraulic fluid into the strut aft dry bay, where high temperatures associated with the adjacent primary exhaust nozzle may ignite the fluid. The result would be an uncontrolled fire in the strut aft dry bay, which lacks fire detection or fire extinguishing equipment.

Explanation of Relevant Service Information

The FAA has reviewed and approved Boeing Alert Service Bulletin 777-54A0016, dated January 25, 2001, which describes procedures for application of high temperature sealant to fill the gap between the strut aft bulkhead, the diagonal brace fitting, and the aft cowl fairing. Accomplishment of the actions specified in the service bulletin is intended to adequately address the identified unsafe condition.